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Construction Engineering
Research Laboratory

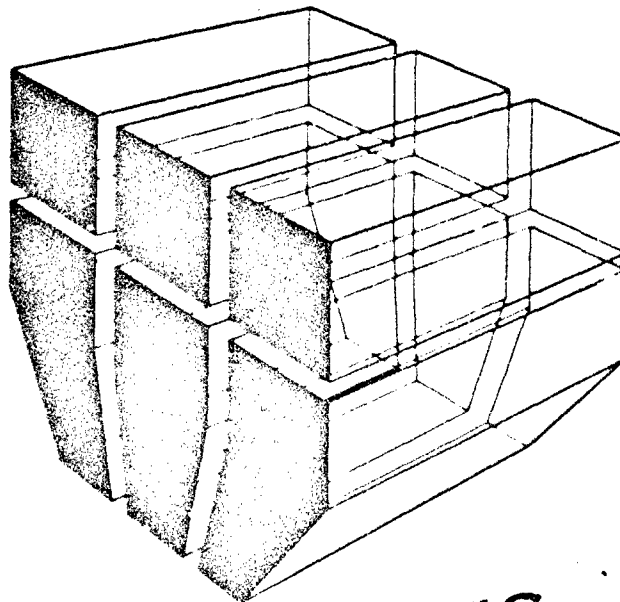


TECHNICAL REPORT M-338
December 1983

12

MUNITIONS STORAGE CONCEPTS
FOR USE IN FLAT TERRAIN
VOLUME II. MUNITIONS STORAGE LAYOUTS

AD A139564



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19. KEY WORDS (Continue on reverse side if necessary and identify by block number) weapons storage magazines (ordnance)		
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) The objective of this project was to develop feasible concepts for weapon storage facilities located in flat terrain with a high water table. The result of the study is preliminary design of six feasible storage complexes. Both aboveground and underground structures are proposed. The special construction practices necessitated by the high water table are addressed. Each design is capable of containing the hazardous effects of an internal explosion within the bay of occurrence. The facilities are		

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,also designed to survive such external threats as detonation of 300,000 pounds of high explosive 100 meters away from the structure, a direct hit by a 500 pound bomb, a B747 aircraft impact on the roof, and a sophisticated terrorist attack. Furthermore, under chemical warfare conditions, the facilities can continue the weapon loadout process while maintaining a clean environment within the structure. Twenty-five year life cycle cost estimates are included for each of the six designs.

Volume I contains a narrative description of the evolutionary design process that led to the six proposed designs, a technical discussion of the design constraints imposed by the various threat scenarios, and a detailed description of the six facility designs. Volume II documents the engineering calculations supporting the structural designs and the life cycle cost calculations. Plan and section drawings are provided for each of the six facilities.

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FOREWORD

This research was conducted for the Defense Nuclear Agency (DNA) by the Engineering and Materials Division (EM) of the U.S. Army Construction Engineering Research Laboratory (CERL) under DNA RDT&E funds for FY 1982, Task Code A99QAXFC. The study was prepared for CERL by the Southwest Research Institute, San Antonio, TX, under DACA88-82-C-0013. Dr. Anthony Kao was the CERL Contract Project Officer. MAJ L. T. Messenger was the DNA Technical Monitor.

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Dr. R. Quattrone is Chief of CERL-EM. COL Paul Theuer is Commander and Director of CERL and Dr. L. R. Shaffer is Technical Director.

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APPENDIX 1

MUNITIONS STORAGE LAYOUTS

PRELIMINARY LAYOUTS AND EVALUATIONS SUMMARIES

SHEET NO.
OF

SUBJECT: LAYOUT

SPONSOR: CERL

BY: meb

DATE: 10/10/19 19

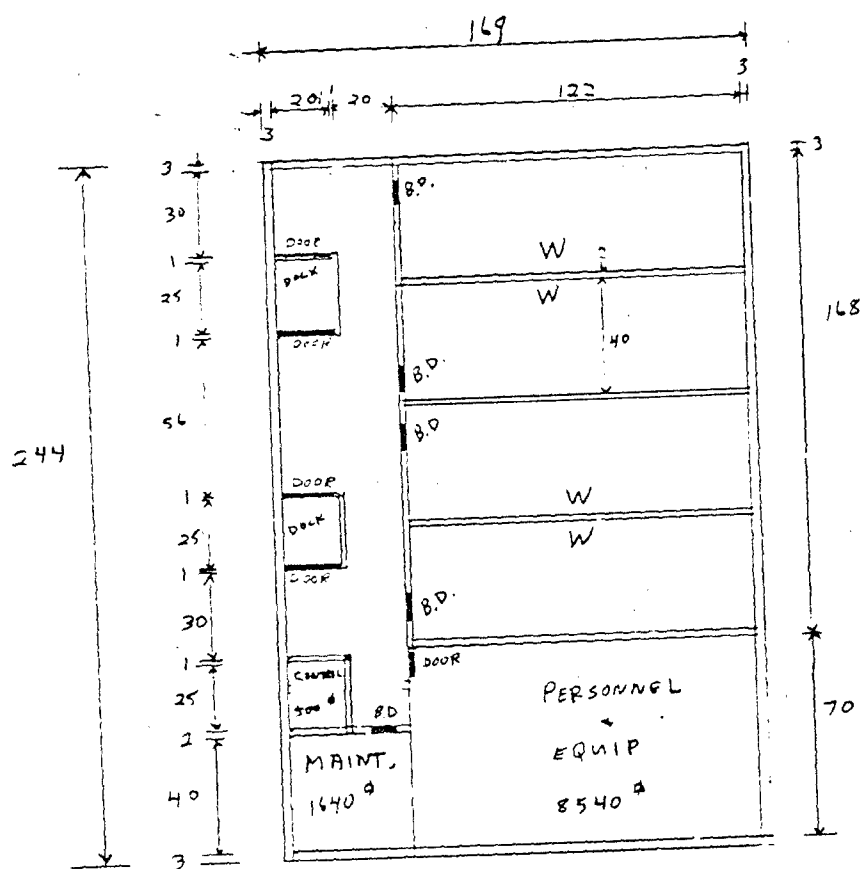
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DATE CHECKED: _____ 19____

- | | | | | | |
|--------|-----|---|-------|---|--------|
| 43 381 | 30 | 3 | HEB'S | 3 | SQUARE |
| 43 382 | 100 | 3 | HEB'S | 3 | SQUARE |
| 43 380 | 200 | 3 | HEB'S | 3 | SQUARE |

<u>MULTIPLIERS:</u>	<u>FLOOR AREA RANKING:</u>	<u>EXTERIOR PERIM.</u>	<u>INT. AREA</u>
H = 3 (HIGH)	30 - 32.5K = 10	8160 - 8480 = 10	5000 - 5500
M = 2 (MEDIUM)	32.5 - 35 = 9	8640 - 9200 = 9	5500 - 6000
L = 1 (LOW)	35 - 37.5 = 8	9200 - 9720 = 8	6000 - 6500
PERCENT	37.5 - 40 = 7	9720 - 10240 = 7	6500 - 7000
	40 - 42.5 = 6	10240 - 10760 = 6	7000 - 7500
CEILING HT.	42.5 - 45 = 5	10760 - 11304 = 5	7500 - 8000
Ave Wall = 10'	45 - 47.5 = 4	11304 - 11828 = 4	8000 - 8500
PAC-MAN = 15'	47.5 - 50 = 3	11828 - 12352 = 3	8500 - 9000
PITS = 15'	50 - 52.5 = 2	12352 - 12900 = 2	9000 - 9500
	52.5 - 55 = 1	12900 - 13400 = 1	9500 - 10000
	55 → 00 = 0	13400 → 0 = 0	10000 →

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: LAYOUT
BY: B. MOKRIS DATE: 8 JUL 19 82 CHECKED BY: mtb DATE CHECKED: 19



$$W = 4800 \text{ ft}^2/\text{ea}$$

$$S = 8540 \text{ ft}^3$$

$$M = 1640 \text{ ft}^3$$

$$L = 500 \text{ ft}^2/\text{ea}$$

$$C = 500 \text{ ft}^2$$

Total Area = 41.236 ft^2

EXTERIOR AREA = 8260 ft²

INTERIOR AREA = 7210 ft²

BLAST Door No. = 5

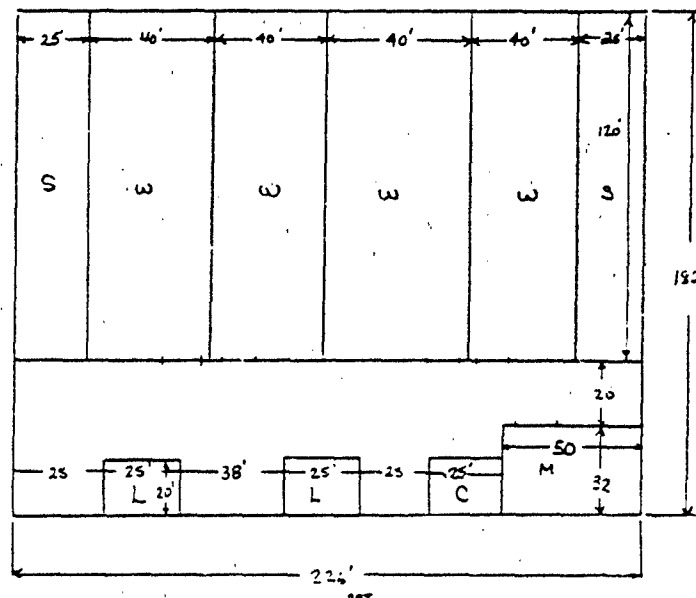
DESIGN 1-1

120x40 BAT

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BY: LAV DATE: 19 CHECKED BY: not DATE CHECKED: 19



$W = 4900 \text{ ft}^2/\text{ea}$
 $S = 6000 \text{ ft}^2$
 $M = 1600 \text{ ft}^2$
 $L = 500 \text{ ft}^2/\text{ea}$
 $C = 500 \text{ ft}^2/\text{ea}$
 TOTAL AREA = 41132 ft^2
 EXTERIOR WALL = 8160 ft^2
 INTERIOR WALL = 8420 ft^2
 BLAST DOOR NO = 5

1-4

DESIGN 1-2
120x40 BAY



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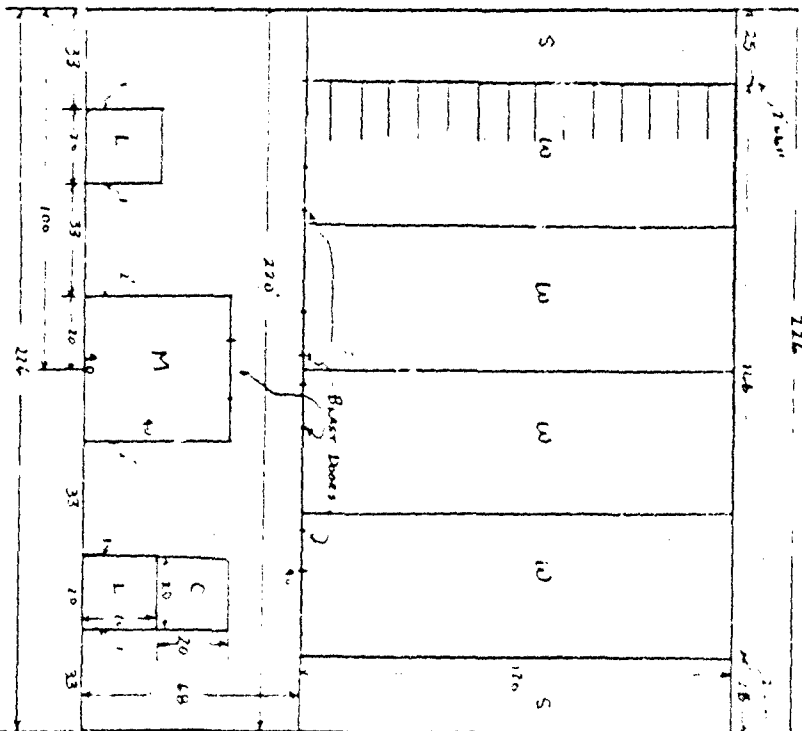
SUBJECT AYBUT

BY L. C. J. DATE 19

CHECKED BY met

DATE CHECKED 19

Positive Glass - 1/2 in Concrete
Each Bay is 7' 0" wide

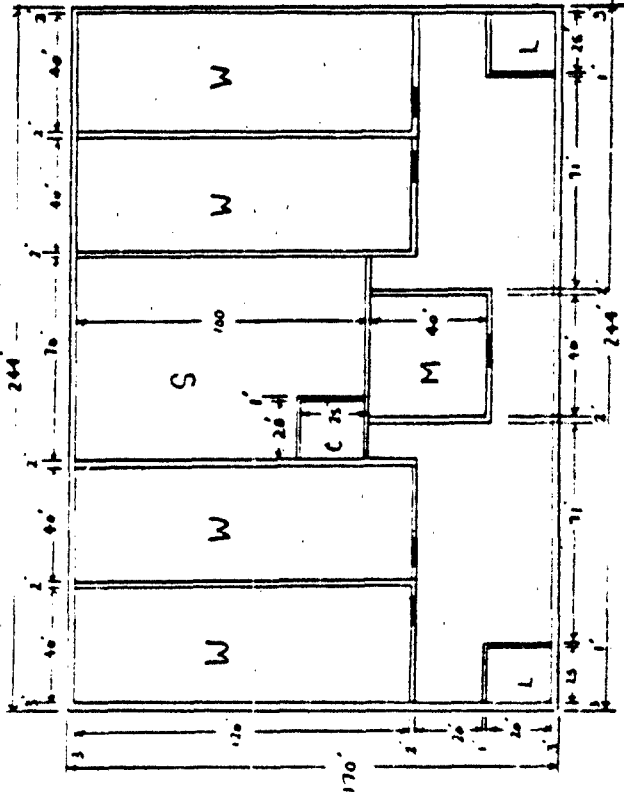


Design - 1-3

5' Footprint
(Footprint)
L - 4800 sq ft
M - 4800
C - 4800
S - 4800

Total Area Between
48296 sq ft
Entered above - 8180
Entered above - 4800 sq ft
Blower Door No. = 7
120 X 40 Bay

1-4



C = 500 ft²
 L = 500 ft²/ea
 S = 6454 ft²
 M = 1600 ft²
 W = 1600 ft²/ea
 Total Area = 41,980 ft²
 Exterior Wall = 8280 ft
 Interior Wall = 8000 ft
 No. Doors = 5
 No. Windows = 12

Design No 1-4
 Rooms 283

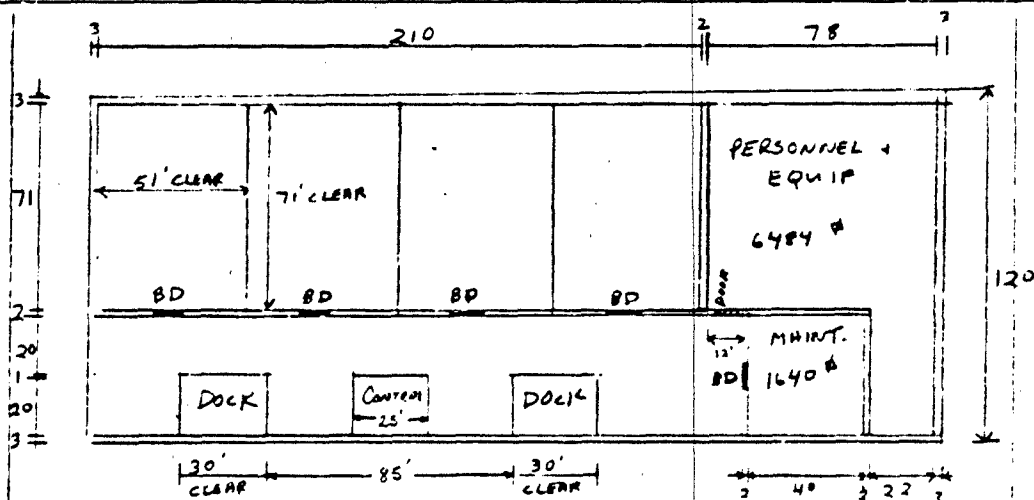
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OF _____

Floor plan of a house with overall dimensions 172' (width) by 219' (depth). The plan includes a front porch (6' wide), a living room (17' x 25'), a dining room (12' x 25'), a kitchen (10' x 25'), a bathroom (5' x 6'), a bedroom (12' x 12'), and a bedroom (12' x 12'). The front porch is 6' wide. The living room is 17' wide and 25' deep. The dining room is 12' wide and 25' deep. The kitchen is 10' wide and 25' deep. The bathroom is 5' wide and 6' deep. The bedroom is 12' wide and 12' deep. The bedroom is 12' wide and 12' deep. The front porch is 6' wide. The living room is 17' wide and 25' deep. The dining room is 12' wide and 25' deep. The kitchen is 10' wide and 25' deep. The bathroom is 5' wide and 6' deep. The bedroom is 12' wide and 12' deep. The bedroom is 12' wide and 12' deep.

DESIGN 1-5

SHEET NO
OF

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: LAYOUT - 51 X 71 INTERIOR BAY
BY: B. MORRIS DATE: 12 JUL 19 83 CHECKED BY: mtf DATE CHECKED: 19



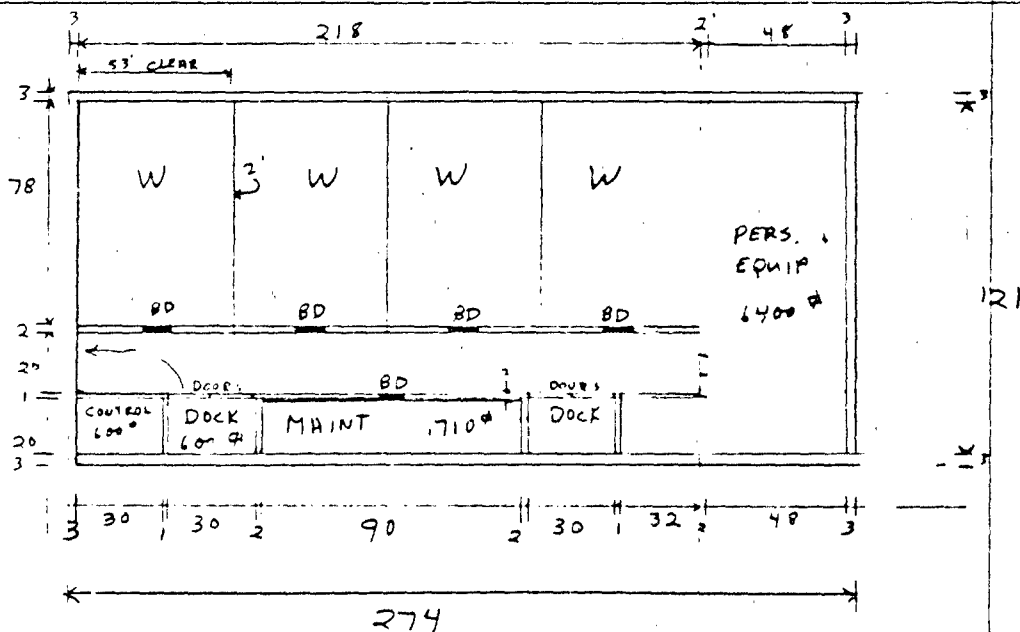
1-8

DESIGN 1-6

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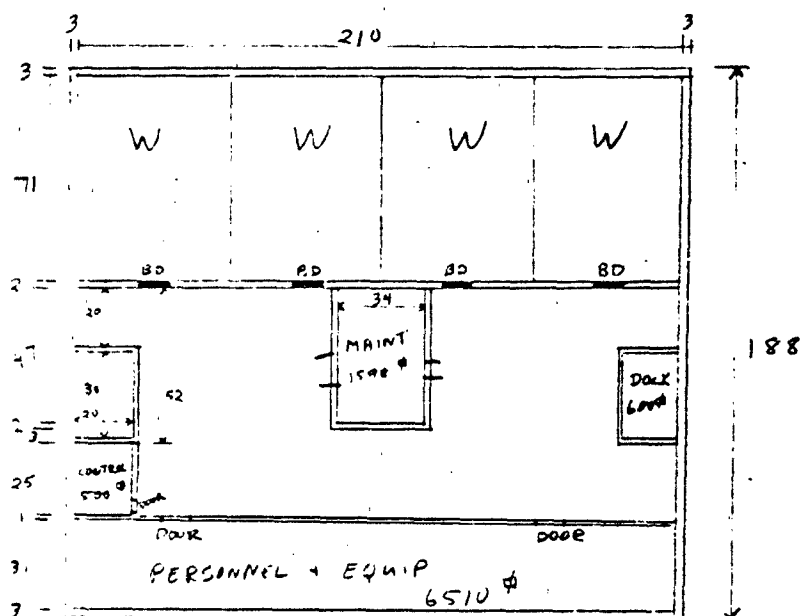
PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: LAYOUT USING 78 X 53' BAYS
BY: E. MURRIS DATE: 12 JUN 92 CHECKED BY: *[signature]* DATE CHECKED: 19



$W = 4134 \text{ ft}^2/\text{ca}$
 $S = 6400 \text{ ft}^2$
 $M = 1710 \text{ ft}^2$
 $L = 600 \text{ ft}^2/\text{ca}$
 $C = 600 \text{ ft}^2$
 TOTAL AREA = 34,798 ft^2
 EXTERIOR WALL = 12030 ft^2
 INTERIOR WALL = 9910 ft^2
 BLAST DOOR NO = 5
 DOCKING WALL BAY

SHEET NO. _____ OF _____

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: LAYOUT - 5' x 7' INTERIOR BAYS
BY: E. MORRIS DATE: 13 JUL 49 #2 CHECKED BY: muf DATE CHECKED: 19



DIVIDING WALL BAY



PIT BAY

DESIGN NO. 1-9
RANDED 228

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OF

PROJECT NO.: 02-7092

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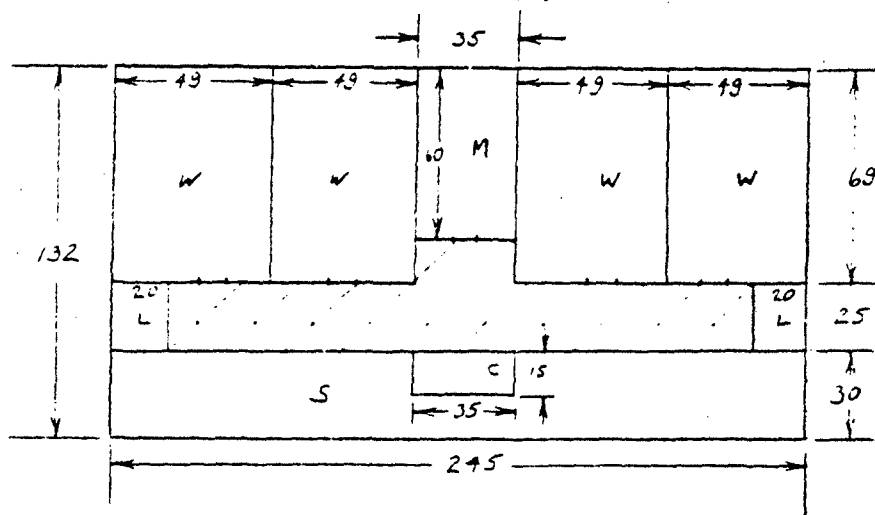
SUBJECT: LAYOUT

BY: MW

DATE: 19

CHECKED BY: *mw*

DATE CHECKED: 19



$$W = 3970 \text{ ft}^2/\text{ea}$$

$$S = 6925 \text{ ft}^2$$

$$M = 2100 \text{ ft}^2$$

$$L = 500 \text{ ft}^2/\text{ea}$$

$$C = 525 \text{ ft}^2$$

$$\text{TOTAL AREA} = 32,300 \text{ ft}^2$$

$$\text{EXTERIOR WALL} = 11,310 \text{ ft}^2 \quad (15' \text{ Tall})$$

$$\text{INTERIOR WALL} = 7605 \text{ ft}^2$$

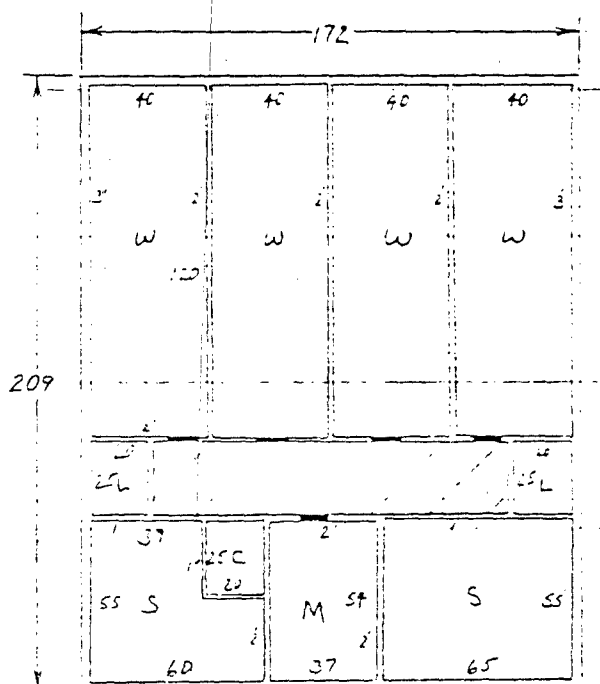
$$\text{BLAST DOOR NO.} = 5$$

PIT DAY

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SUBJECT LAYOUT
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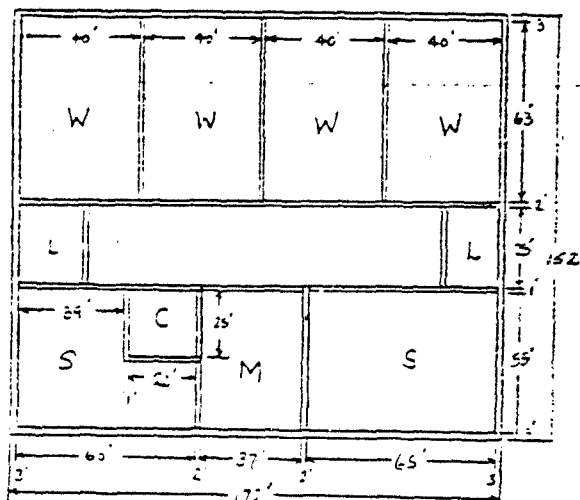
Below Ground.

$W = 4800 \text{ ft}^2/\text{ea}$
 $S = 6259 \text{ ft}^2$
 $M = 1945 \text{ ft}^2$
 $L = 500 \text{ ft}^2/\text{ea}$
 $C = 500 \text{ ft}^2$
Total Area = 35000 ft^2
Perimeter Wall = 7620 ft^2
Interior Wall = 6250 ft^2
Blow Deck No = 5

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SUBJECT: AYOUT
BY: MW DATE: 19 CHECKED BY: nwb DATE CHECKED: 19



$W = 2520 \text{ ft}^2/\text{ea}$

$S = 6329 \text{ ft}^2$

$M = 2035 \text{ ft}^2$

$L = 500 \text{ ft}^2/\text{ea}$

$C = 500 \text{ ft}^2$

TOTAL AREA = 26144 ft^2

INTERIOR WALL = 6705 (15' HIGH)

EXTERIOR WALL = 9720 (15' HIGH)

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SUBJECT: LAYOUT

BY: mw

DATE: 19

CHECKED BY: MW

DATE CHECKED: 19

	1-1	1-2	1-3	1-4
1 • TOTAL FLOOR AREA ^(H)	6 18	6 18	5 15	6 18
2 • OPER. EFFICIENCY	5 5	5 5	10 10	7 7
2a - Bay - Main - Bay ^(H)				
2b - Loadout ^(H)	10 30	10 30	10 30	10 30
3 • EXPANDABILITY ^(H)	5 5	10 10	10 10	10 10
4 • LOAD DOCK SEPARATION ^(H)	10 30	10 30	10 30	10 30
5 • PERIMETER WALL AREA/Vol. ^(H)	10 20	10 20	10 20	10 20
6 • BAY INTERIOR WALL AREA/Vol. ^(H)	6 12	4 8	3 6	5 10
7 • SQUAREDNESS OF S.M. 1 & C ^(H)	10 20	2 4	2 4	10 20
8 • QUANTITY OF BAY DOORS ^(H)	10 10	10 10	10 10	10 10
9 • WEAPONS BAY LOCATION ^(H)	8 8	8 8	8 8	8 8
10 • QUANTITY OF MAINT. BAY ^(H)	10 30	10 30	10 30	10 30
11 • EQUIPMENT NEEDS ^(H)				
- PASS WAYS ^(H)	10 30	10 30	10 30	10 30
- PROHIBITION ^(H)	10 30	10 30	10 30	10 30
- P.V. ^(H)	15 30	10 30	10 30	10 30

278

263

263

283

MULTIPLIERS:

H = 3
M = 2
L = 1

CEILING HT.
AVE WALL = 10'
PER-MAN = 15'
PITS = 15'

FLOOR AREA RANKING:

35 - 32.5K - 10
32.5 - 35 = 9
35 - 37.5 = 8
37.5 - 40 = 7
40 - 42.5 = 6
42.5 - 45 = 5
45 - 47.5 = 4
47.5 - 50 = 3
50 - 52.5 = 2
52.5 - 55 = 1
55 - 60 = 0

1-15

EXTENS. PERP.

8160 - 8470 = 10
8470 - 9200 = 9
9200 - 9732 = 8
9732 - 10260 = 7
10260 - 10780 = 6
10780 - 11300 = 5
11300 - 11828 = 4
11828 - 12352 = 3
12352 - 12970 = 2
12970 - 13400 = 1
13400 - 13800 = 0

INT. AREA

5000 - 5500
5500 - 6000
6000 - 6500
6500 - 7000
7000 - 7500
7500 - 8000
8000 - 8500
8500 - 9000
9000 - 9500
9500 - 10000
10000 - 10500

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OF _____

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: LAYOUT
BY: MUT DATE: 19 CHECKED BY: MW DATE CHECKED: 19

	1-5	1-6	1-7	1-8
1 • TOTAL FLOOR AREA ^(H)	7 21	8 24	9 27	7 21
2 • OPER. EFFICIENCY	7 7	6 6	10 10	10 10
2a - Bay - Main - Bay ^(H)				
2b - Loadout ^(H)	10 30	7 21	7 21	7 21
3 • EXPANDABILITY ^(H)	10 10	5 5	5 5	10 10
4 • LOAD DOCK SEPARATION ^(H)	10 30	10 30	10 30	10 30
5 • PERIMETER WALL AREA/Vol. ^(H)	10 20	2 4	3 6	3 6
6 • BAY INTERIOR WALL AREA/Vol. ^(H)	8 16	2 4	1 2	4 8
7 • "SQUARENESS" OF S.M.L. <C ^(H)	4 8	5 16	4 8	8 16
8 • QUANTITY OF BAY DOORS ^(H)	10 10	10 10	10 10	9 9
9 • WEEDING BAY LOCATION ^(H) (SU-K-10)	8 8	8 8	8 8	8 8
10 • QUANTITY OF MAINT. BAYS ^(H) (1 Bay = 10, 2 Bays = 5)	10 30	10 30	10 30	10 30
11 • EQUIPMENT NEEDS ^(H)				
- BASE NEEDS ^(H) (25,000 - 10,000)	10 30	7 21	7 21	7 21
- PERIMETER FLOOR AREA ^(H)	10 30	0 0	0 0	0 0
- P.W. PLOTS ^(H) (10,000 - 10,000)	10 30	7 21	7 21	7 21
	280	200	199	211

MULTIPLIERS:	FLOOR AREA RANKING:	EXTERIOR AREA	INT. AREA
H = 3	30 - 32.5K = 10	8160 - 8680 = 10	5000 - 5500
M = 2	32.5 - 35 = 9	8680 - 9200 = 9	5500 - 6000
L = 1	35 - 37.5 = 8	9200 - 9720 = 8	6000 - 6500
	37.5 - 40 = 7	9720 - 10240 = 7	6500 - 7000
	40 - 42.5 = 6	10240 - 10760 = 6	7000 - 7500
CEILING HT.	42.5 - 45 = 5	10760 - 11280 = 5	7500 - 8000
AVE WALL = 10'	45 - 47.5 = 4	11280 - 11800 = 4	8000 - 8500
PAC-MAN = 15'	47.5 - 50 = 3	11800 - 12320 = 3	8500 - 9000
PITS = 15'	50 - 52.5 = 2	12320 - 12840 = 2	9000 - 9500
	52.5 - 55 = 1	12840 - 13360 = 1	9500 - 10000
	55 - 60 = 0	13360 - 13880 = 0	10000 - 10500

1-16

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PROJECT NO.: 2-7092 SPONSOR: CEIR
SUBJECT: LAYOUT
BY: mtb DATE: 19 CHECKED BY: MTW DATE CHECKED: 19

	1-9	1-10	1-11	1-12
1 • Total Floor Area [Ⓢ]	10 30	10 30	8 24	10 30
2 • OPER. EFFICIENCY	10 10	6 6	9 10	10 10
2a - Bay - Main - Bay [Ⓢ]				
2b - Loadout [Ⓢ] <small>FRONT - 10</small>	7 21	7 21	10 30	10 30
3 • EXPANDABILITY [Ⓢ]	3 9	10 10	9 9	10 10
4 • Load DOCK SEPARATION [Ⓢ]	10 30	10 30	10 30	10 30
5 • PERIMETER WALL AREA/Vol [Ⓢ]	5 10	4 8	10 20	5 10
6 • Bay INTERIOR WALL AREA/Vol [Ⓢ]	4 8	5 10	7 14	7 14
7 • "SQUAREDNESS" OF S.M. I [Ⓢ] <small>S.D. = 10; L.D. = 9</small>	10 20	3 6	10 20	10 20
8 • QUANTITY OF BLAST DOORS [Ⓢ] <small>Ext = 8, Int = 10</small>	10 10	10 10	10 10	10 10
9 • WEAPON BAY LOCATION [Ⓢ] <small>(Ext/Int)</small>	8 8	8 8	8 8	8 8
10 • QUANTITY OF MAINT. SP. [Ⓢ] <small>1 Bay = 10, 2 Bays = 5</small>	10 30	10 30	10 30	10 30
11 • EQUIPMENT NEEDS [Ⓢ]				
- Base Needs <small>2 Front 10, 2 Rear 10</small>	7 21	7 21	10 30	10 30
- Perimeter Floor Area	0 0	0 0	10 20	10 20
- Per Doors Floor Area	7 21	7 21	10 30	10 30

228

211

294

258

MULTIPLIERS:

H = 3
M = 2
L = 1

CEILING HT.

Ave Wall = 10'

PAC-MAN = 15'

PITS = 15'

FLOOR AREA RANKING:

30 - 32.5 = 10
32.5 - 35 = 9
35 - 37.5 = 8
37.5 - 40 = 7
40 - 42.5 = 6
42.5 - 45 = 5
45 - 47.5 = 4
47.5 - 50 = 3
50 - 52.5 = 2
52.5 - 55 = 1
55 - ∞ = 0

1-17

EXTENSIVE AREA

8140 - 8630 = 10
8630 - 9200 = 9
9200 - 9732 = 8
9732 - 10260 = 7
10260 - 10780 = 6
10780 - 11304 = 5
11304 - 11828 = 4
11828 - 12352 = 3
12352 - 12900 = 2
12900 - 13400 = 1
13400 - ∞ = 0

INT. AREA

5000 - 5500
5500 - 6000
6000 - 6500
6500 - 7000
7000 - 7500
7500 - 8000
8000 - 8500
8500 - 9000
9000 - 9500
9500 - 10000
10000 - ∞



SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF BALLISTICS AND EXPLOSIVES SCIENCES
COMPUTATION SHEET

SHEET NO
OF

PROJECT NO 62-7072-001

SPONSOR CERL

SUBJECT LAYOUT

BY L.M.V.

DATE

19

CHECKED BY

mtb

DATE CHECKED

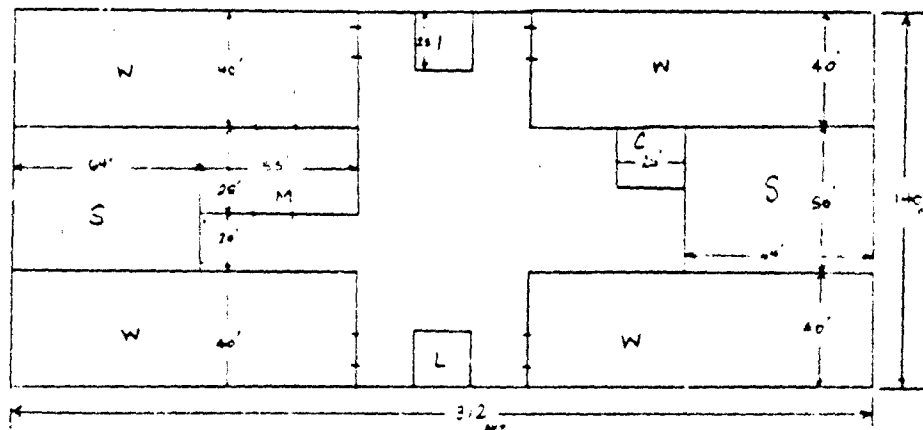
19

W = 4800 ft²/ea
S = 3400 ft²/ea
M = 1500 ft²
L = 500 ft²/ea
C = 500 ft²

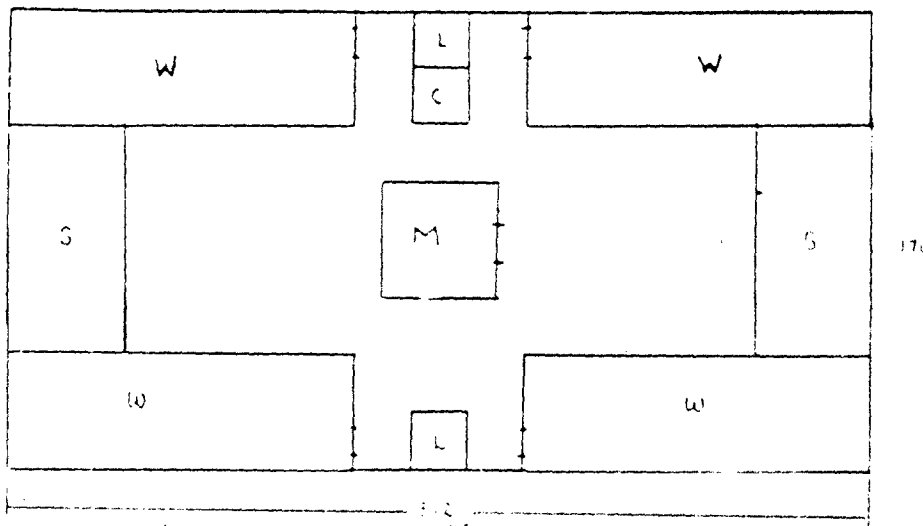
Access
Total Area = 43680 ft²
Ramp Doors = 5-6
Exterior Wall = 9040 ft
Interior Wall = 7230 ft

DESIGN
2-2

DESIGN-10A



DESIGN-10



W = 4800 ft²/ea
S = 3400 ft²/ea
M = 1500 ft²
L = 500 ft²/ea
C = 500 ft²

Access
Total Area = 53760 ft²
Ramp Doors = 5
Exterior Wall = 9040 ft
Interior Wall = 7230 ft

DESIGN
2-3

53760 ft²

SHEET NO. _____ OF _____

SPONSOR: LERL

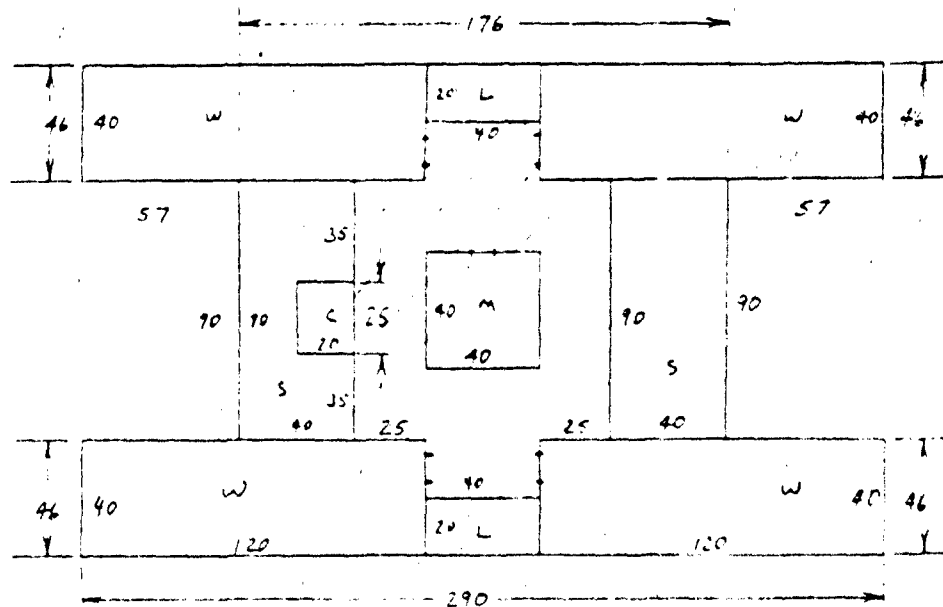
DATE CHECKED



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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO. _____
OF _____

PROJECT NO. 62-7092 SPONSOR CERL
SUBJECT: LAYOUT
BY: MW DATE 19 CHECKED BY: mtb DATE CHECKED: 19



W = 4800 ft²/ea
S = 6700 ft²
M = 1600 ft²
L = 500 ft²/ea
C = 500 ft²
TOTAL AREA = 42,500 ft²
EXTERIOR WALL = 9400 ft²
INTERIOR WALL = 2580 ft²
PLANT DOOR NO. = 5
20 x 20 ft

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COMPUTATION SHEET

SHEET NO.
OF

PROJECT NO.: 22-7092

SPONSOR: CERL

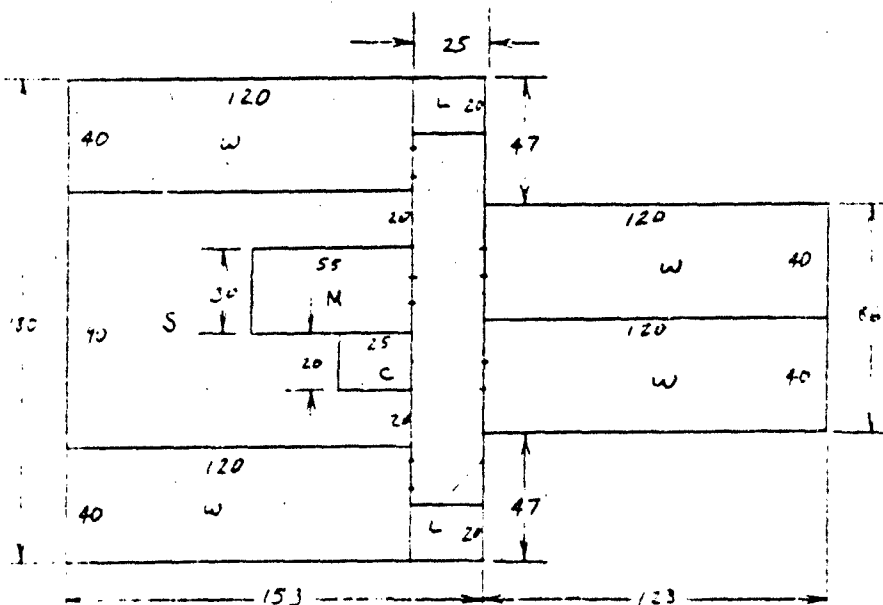
SUBJECT: LAYOUT

BY: MW

DATE: 18

CHECKED BY: *mtb*

DATE CHECKED: 18



$$W = 4800 \text{ ft}^2/\text{ea}$$

$$\text{BLAST DOOR NO} = 5$$

$$S = 6400 \text{ ft}^2$$

$$M = 1650 \text{ ft}^2$$

$$L = 500 \text{ ft}^2/\text{ea}$$

$$C = 500 \text{ ft}^2$$

$$\text{TOTAL AREA} = 38,100 \text{ ft}^2$$

$$\text{EXTERIOR WALL} = 9,120 \text{ ft}^2$$

1-22

$$\text{INTERIOR WALL} = 10,350 \text{ ft}^2$$

$$120 \times 40 \text{ BAY}$$

DESIGN 2-6

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COMPUTATION SHEET

SHEET NO.
OF

PROJECT NO. 02-7092

SPONSOR

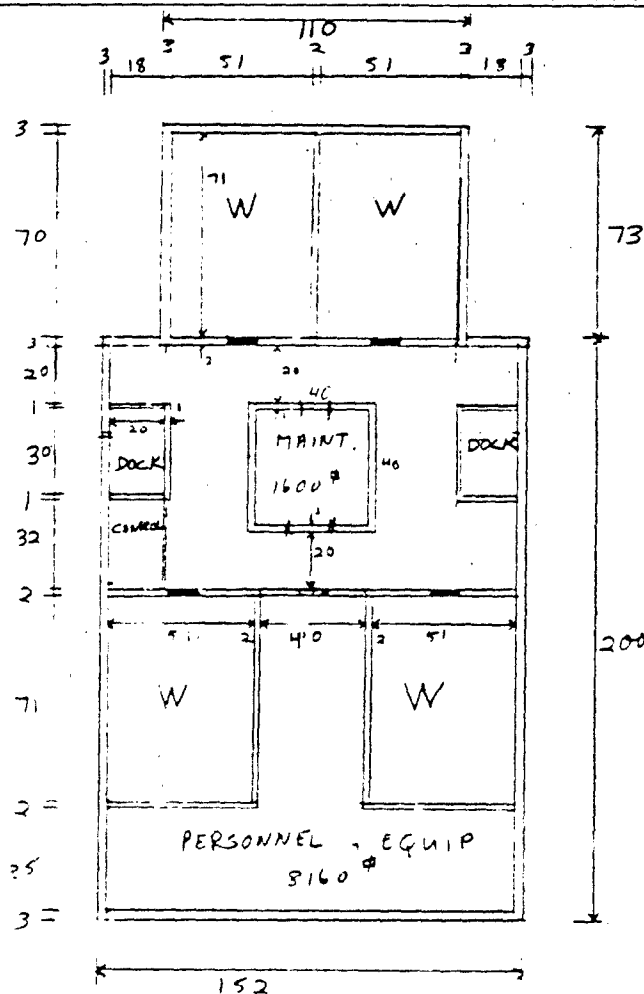
CERL

SUBJECT LAYOUT - 51' X 71' INTERIOR BAY

BY: E. MORRIS DATE 19

CHECKED BY

DATE CHECKED 19



$$W = 3621 \text{ H}^2/\text{m}$$

$$S = 8160 \text{ ft}^2$$

$$M = 1600 \text{ ft}^2$$

$$L = 400 \text{ ft}^2/\text{m}$$

$$C = 640 \text{ ft}^2$$

$$\text{TOTAL Area} = 38430 \text{ ft}^2$$

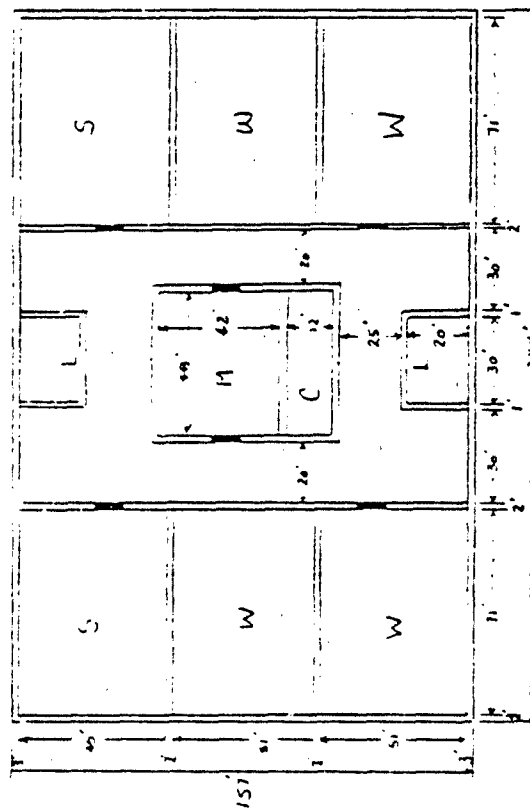
$$L \text{ Area} = 400 \text{ ft}^2/\text{m}$$

$$\text{L Area} = 400 \times 10,125 = 4,050,000 \text{ ft}^2$$

$$\text{L Area} = 400 \times 10,185 = 4,074,000 \text{ ft}^2$$

DIVIDING WALL BAY

DESIGN 2-7



Beams Center 30'

IMC MNO DESIGN

C. 1001 R.

L = 600 ft²/ea

S = 3195 ft²/ea

M = 2016 ft²

W = 3381 ft²/ea

Total Area = 78,308

Exterior Wall = 12030 (15' Wall)

Interior Wall = 10020 (15' Wall)

Blower Door No. = 6

Design Plan 2-B
Revision 112

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COMPUTATION SHEET

SHEET NO.
OF

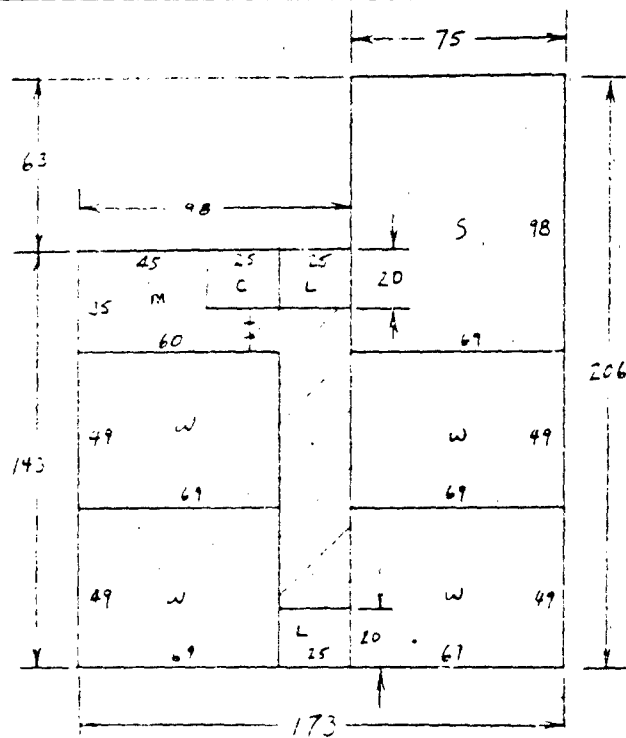
PROJECT NO.: 62-7072

SPONSOR: CERL

SUBJECT: LAYOUT

BY: M.Y. DATE: 19

CHECKED BY: mob DATE CHECKED: 19



$$W = 3331 \text{ ft}^2/\text{ea.}$$

$$\text{Exterior Wall} = 11,370 \text{ ft}^2$$

$$S = 6762 \text{ ft}^2$$

$$\text{Interior Wall} = 7305 \text{ ft}^2$$

$$M = 1800 \text{ ft}^2$$

$$L = 500 \text{ ft}^2/\text{ea.}$$

$$C = 500 \text{ ft}^2$$

$$\text{Total Area} = 29,500 \text{ ft}^2 \quad 1-25$$

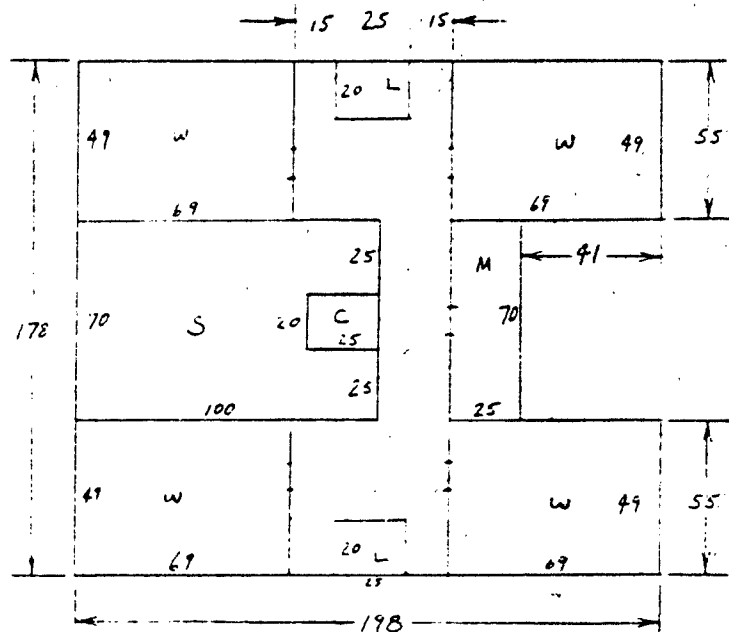
PIT BAY

DESIGN 2-9

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
OF

PROJECT NO.: 52-7092 SPONSOR: CERL
SUBJECT: AVSUT
BY: MW DATE: 19 CHECKED BY: mt DATE CHECKED: 19



W = 3391 ft²/ea

Blast Door No = 5

S = 6500 ft² Total

M = 1750 ft²

L = 500 ft²/ea

C = 500 ft²

TOTAL AREA = 32,400 ft²

EXTERIOR WALL = 12510 ft²

INTERIOR WALL = 6910 ft²

DIT BAY

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
OF

PROJECT NO.: 02-7092

SPONSOR: CERL

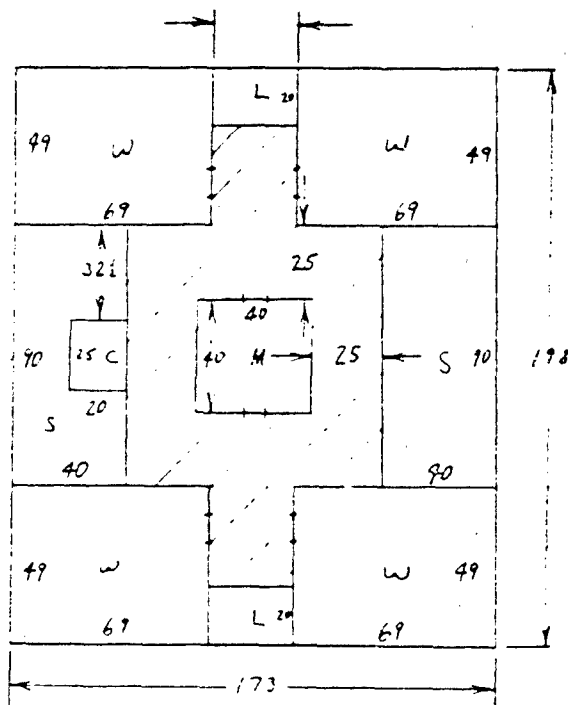
SUBJECT: LAYOUT

BY: MW

DATE: 19

CHECKED BY: *mt*

DATE CHECKED: 19



$$W = 3381 \text{ ft}^2/\text{ce}$$

$$S = 6700 \text{ ft}^2$$

$$M = 1600 \text{ ft}^2$$

$$L = 500 \text{ ft}^2/\text{ce}$$

$$C = 500 \text{ ft}^2$$

$$\text{Total Area} = 34300 \text{ ft}^2$$

$$\text{External Wall} = 11130 \text{ ft}^2$$

$$\text{Internal Wall} = 9480 \text{ ft}^2$$

$$\text{Blast Dose } U_0 = 6$$

$$\text{PIT BAY } 1-27$$

DESIGN 2-11

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO. _____
OF _____

PROJECT NO.: 22-7092 SPONSOR: CERL
SUBJECT: AIRCUT
BY: MW DATE: 19 CHECKED BY: MW DATE CHECKED: 19

	2-1	2-2	2-3	2-4
1 • TOTAL FLOOR AREA [Ⓢ]	7 21	5 15	1 3	8 24
2 • OPER. EFFICIENCY	10 10	7 7	10 10	6 6
2a - Bay - Main - Bay [Ⓢ]				
2b - Loadout [Ⓢ] <small>Furnish - 10 Crews - 7</small>	10 30	10 30	10 30	10 30
3 • EXPANSIBILITY [Ⓢ]	10 10	10 10	10 10	10 10
4 • LOAD DOCK SEPARATION [Ⓢ]	10 30	10 30	10 30	10 30
5 • PERIMETER WALL AREA/Vol. [Ⓢ]	9 18	9 18	8 16	9 18
6 • BAY INTERIOR WALL AREA/Vol. [Ⓢ]	4 8	5 10	4 8	7 14
7 • "SQUAREDNESS" OF S.M.I. [Ⓢ] <small>5 Doors = 10; 6 Doors = 9</small>	2 4	10 20	10 20	8 16
8 • QUANTITY OF BLAST DOORS [Ⓢ] <small>Ext = 8, Int = 10</small>	9 9	10 10	9 9	10 10
9 • WEAPONS BAY LOCATION [Ⓢ] <small>(Ext-Int)</small>	9 9	8 8	8 8	8 8
10 • QUANTITY OF MAINT. BAYS <small>Ext = 10, 2 Bays = 5</small>	10 30	10 30	10 30	10 30
11 • EQUIPMENT NEEDS [Ⓢ] <small>2 Floors = 10 - 2nd Floor Needs 250000 - 1st Floor Needs 100000 - 2nd Floor Needs 100000</small>	10 30 10 30 10 30	10 30 10 30 10 30	10 30 10 30 10 30	10 30 10 30 10 30
	269	278	261	286

MULTIPLIERS:

H = 3
M = 2
L = 1

CEILING HT.

Avg Wall = 10'

PAC-MAN = 15'

PITS = 10'

FLOOR AREA RANGING:

30 - 32.5K = 10
32.5 - 35 = 9
35 - 37.5 = 8
37.5 - 40 = 7
40 - 42.5 = 6
42.5 - 45 = 5
45 - 47.5 = 4
47.5 - 50 = 3
50 - 52.5 = 2
52.5 - 55 = 1
55 - 60 = 0

1-28

EXTERIOR AREA

8160 - 8680 = 10
8680 - 9200 = 9
9200 - 9720 = 8
9720 - 10240 = 7
10240 - 10760 = 6
10760 - 11280 = 5
11280 - 11800 = 4
11800 - 12320 = 3
12320 - 12840 = 2
12840 - 13360 = 1
13360 - 13880 = 0

INT. AREA

5000 - 5500
5500 - 6000
6000 - 6500
6500 - 7000
7000 - 7500
7500 - 8000
8000 - 8500
8500 - 9000
9000 - 9500
9500 - 10000
10000 - 10500
10500 - 11000

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO. _____
OF _____

PROJECT NO.: 22-7092 SPONSOR: CERL
SUBJECT: ALBOUT
BY: met DATE: 19 CHECKED BY: S. V. DATE CHECKED: 19

	2-5	2-6	2-7	2-8
1 • TOTAL FLOOR AREA ^(M)	5 15	7 21	7 21	8 24
2 • OPER. EFFICIENCY	10 10	6 6	10 10	10 10
2a - Bay - Main - Bay ^(M)				
2b - Loadout ^(M)	10 30	10 30	7 21	7 21
3 • EXPANSIBILITY ^(M)	10 10	10 10	10 10	10 10
4 • LOOSE DOCK SEPARATION ^(M)	10 30	10 30	10 30	10 30
5 • PERIMETER WALL AREA/Vol. ^(M)	8 16	9 18	2 4	4 8
6 • BAY INTERIOR WALL AREA/Vol. ^(M)	9 18	0 0	0 0	0 0
7 • "SQUAREDNESS" OF S.M.I. ^(M) <small>SQUARED = 10; CORNER = 9</small>	10 20	9 18	5 10	10 20
8 • QUANTITY OF BAY DOORS ^(M) <small>Ext = 8; Int = 10</small>	9 9	10 10	9 9	9 9
9 • WEAPONS BAY LOCATION ^(M) (T.W.K.)	8 8	8 8	8 8	8 8
10 • QUANTITY OF MAINT. BAYS ^(M) <small>1 Bay = 10; 2 Bays = 5</small>	10 30	10 30	10 30	10 30
11 • EQUIPMENT NEEDS ^(M) <small>2 Fault = 10; 1 Fault = 7</small>	10 30	10 30	7 21	7 21
- BASE NEEDS				
- PROTECTIVE FLOORING	10 30	10 30	0 0	0 0
- FIRE FIGHTING	10 30	10 30	7 21	7 21
	236	271	195	212

MULTIPLIERS:

FLOOR AREA RANGING:

EXTERIOR AREA

INT. AREA

H = 3

M = 2

L = 1

CEILING HT.

Ave Wall = 10'

PAC-MAN = 15'

PITS = 15'

30 - 32.5 = 10

32.5 - 35 = 9

35 - 37.5 = 8

37.5 - 40 = 7

40 - 42.5 = 6

42.5 - 45 = 5

45 - 47.5 = 4

47.5 - 50 = 3

50 - 52.5 = 2

52.5 - 55 = 1

55 - 60 = 0

1-29

8160 - 8470 = 10

8480 - 9200 = 9

9200 - 9732 = 8

9732 - 10260 = 7

10260 - 10780 = 6

10780 - 11304 = 5

11304 - 11828 = 4

11828 - 12352 = 3

12352 - 12900 = 2

12900 - 13400 = 1

13400 - 13900 = 0

5000 - 5500

5500 - 6000

6000 - 6500

6500 - 7000

7000 - 7500

7500 - 8000

8000 - 8500

8500 - 9000

9000 - 9500

9500 - 10000

10000 - 10500

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO. _____
OF _____

PROJECT NO.: 2-7092 SPONSOR: CERL
SUBJECT: LAYOUT
BY: WWT DATE: 19 CHECKED BY: WV DATE CHECKED: 19

	2-9	2-10	2-11				
1 • TOTAL FLOOR AREA ^(H)	10	30	9	27	9	27	
2 • OPER. EFFICIENCY	5	5	6	6	10	10	
2a - Bay - Main - Bay ^(H)							
2b - LOADOUT ^(H)	7	21	7	21	7	21	
3 • EXPANDABILITY ^(L)	10	10	10	10	10	10	
4 • LOAD DOCK SEPARATION ^(H)	10	30	10	30	10	30	
5 • PERIMETER WALL AREA/Vol. ^(H)	4	8	2	4	5	10	
6 • BAY INTERIOR WALL AREA/Vol. ^(H)	6	12	7	14	3	6	
7 • "SQUAREDNESS" OF S.M.I. ^(H) 5 Dimes = 10; 4 Dimes = 9	10	20	8	16	10	20	
8 • QUANTITY OF BLAST DOORS ^(L) Ext = 8; Int = 10	10	10	10	10	9	9	
9 • WEEDING BAY LOCATION ^(L) (Ext-Ext)	5	8	8	8	8	8	
10 • QUANTITY OF MAINT. BAYS ^(H) 1 Bay = 10; 2 Bays = 5	10	30	10	30	10	30	
11 • EQUIPMENT NEEDS ^(H) 2 Floors = 10 - BASE NEEDS ^{PER ALPH 7}	7	21	7	21	7	21	
- PERIMETER ^{PER ALPH 7}	0	0	0	0	0	0	
- PK PROTS ^{PER ALPH 7}	7	21	7	21	7	21	
	226		218		223		

MULTIPLIERS:

H = 3
M = 2
L = 1

CEILING HT.

AVE WALL = 10'

PAC-MAN = 15'

PITS = 15'

FLOOR AREA RANKINGS:

30 - 32.5K = 10
32.5 - 35 = 9
35 - 37.5 = 8
37.5 - 40 = 7
40 - 42.5 = 6
42.5 - 45 = 5
45 - 47.5 = 4
47.5 - 50 = 3
50 - 52.5 = 2
52.5 - 55 = 1
55 - 60 = 0

1-30

EXTERNAL AREA

8160 - 8680 = 10
8680 - 9200 = 9
9200 - 9720 = 8
9720 - 10240 = 7
10240 - 10760 = 6
10760 - 11280 = 5
11280 - 11800 = 4
11800 - 12320 = 3
12320 - 12840 = 2
12840 - 13360 = 1
13360 - 13880 = 0

INT. AREA

5000 - 5500
5500 - 6000
6000 - 6500
6500 - 7000
7000 - 7500
7500 - 8000
8000 - 8500
8500 - 9000
9000 - 9500
9500 - 10000
10000 - 10500

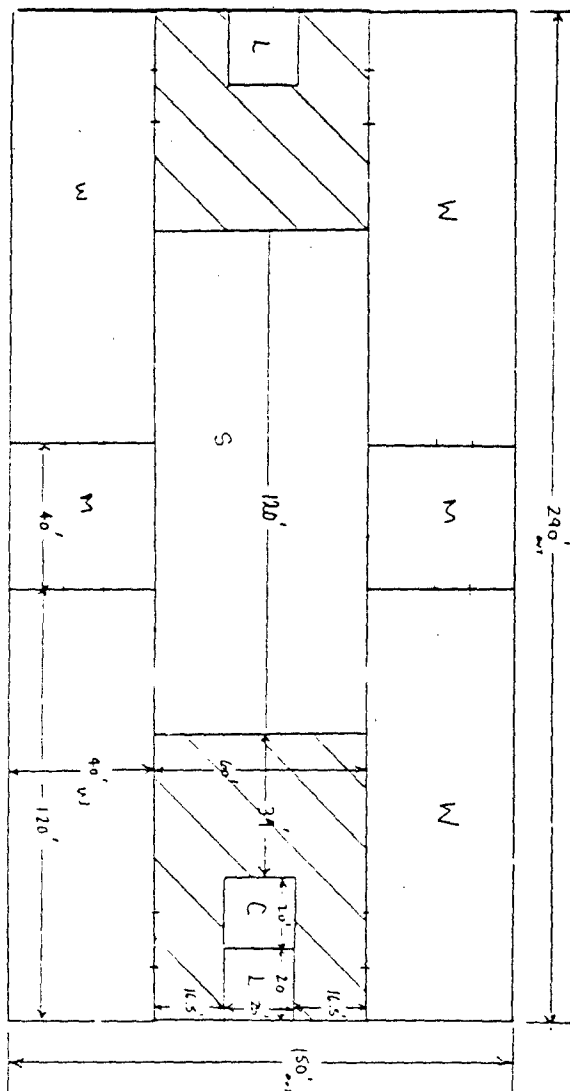


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DEPARTMENT OF BALLISTICS AND EXPLOSIVES SCIENCES
COMPUTATION SHEET

SHEET NO. _____

OF _____

PROJECT NO. 62-7092 SPONSOR CERL
SUBJECT LAYOUT
BY LMV DATE 19 CHECKED BY mtb DATE CHECKED 19



Space Footnotes

(M) = 4500 sq ft Area = 9200 ft²
S = 9400 Total Area = 13500 ft²
M = 1600 sq ft Entrance Area = 8900 ft²
L = 500 sq ft Interior Area = 7200 ft²
C = 500 Blast Door No. = 8

120x40 BA

Design 3-1



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COMPUTATION SHEET

SHEET NO. _____
OF _____

PROJECT NO.: 02-7012

SPONSOR: CERL

SUBJECT: LAYOUT

BY: LMV

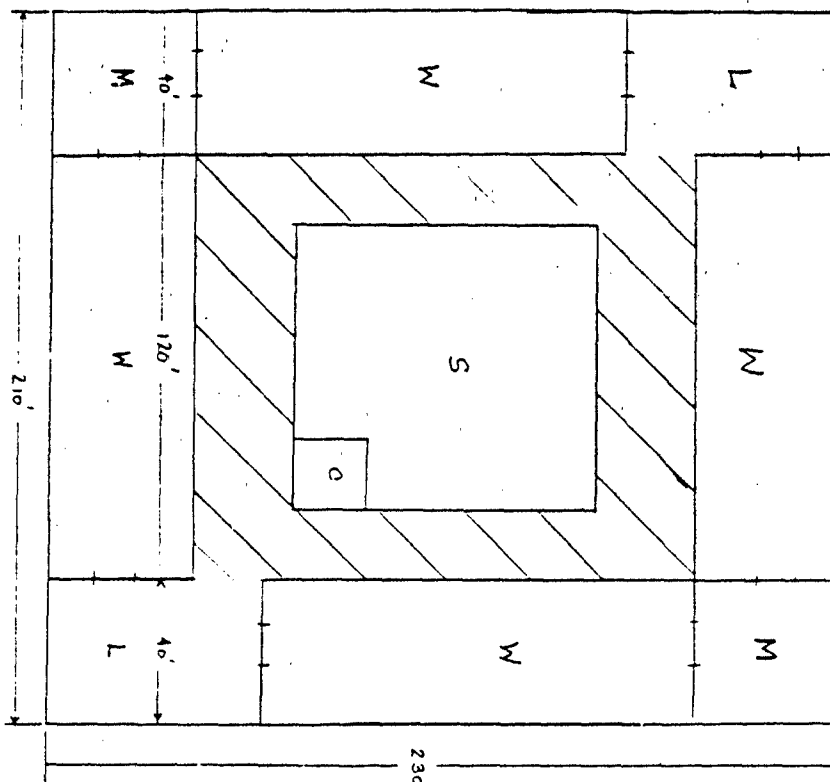
DATE: _____

19

CHECKED BY: nmf

DATE CHECKED: _____

19



Design 3-2

Quant Footage
U = 4800 ea. (4)
S = 6000
M = 1600 ea. (2)
L = 2400 ea. (2)
C = 400
Access = 8000 ft²
Total Area = 48,500 ft²
Entrance Area = 8500 ft²
Interior Area = 3800 ft²
Blast Door = 8
120 x 40 EAV

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO
OF

PROJECT NO.: 22-7092

SPONSOR: CERL

SUBJECT: LAYOUT

BY: *msb*

DATE: 19

CHECKED BY: *msb*

DATE CHECKED: 19

	3-1		3-2							
1 • Total Floor Area (A)	5	15	3	15						
2 • OPER. EFFICIENCY	10	10	10	10						
2a - Bay - Main - Bay (A)										
2b - Loadout (A)	10	30	10	30						
3 • EXPANSIBILITY (A)	10	10	10	10						
4 • Load Dock Separation (A)	10	30	10	30						
5 • PERIMETER WALL AREA/Vol (A)	5	18	5	18						
6 • Bay Interior Wall Area/Vol (A)	10	20	5	20						
7 • Squaredness of Bay (A)	5	20	5	20						
8 • QUANTITY OF Bay Doors (A)	5	5	5	5						
9 • WEAPONS Bay Location (A)	8	8	8	8						
10 • QUANTITY OF Maint. Bay (A)	5	15	5	15						
11 • EQUIPMENT NEEDS (A)	10	30	10	30						
- Base (A)										
- Equipment (A)		30		30						
- Bay (A)	10	30	10	30						
	271		271							

MULTIPLIERS:

H = 3
M = 2
L = 1

Floor Area Ranges:

30 - 32.5 = 10
32.5 - 35 = 9
35 - 37.5 = 8
37.5 - 40 = 7
40 - 42.5 = 6
42.5 - 45 = 5
45 - 47.5 = 4
47.5 - 50 = 3
50 - 52.5 = 2
52.5 - 55 = 1
55 - 60 = 0

Equipment Base

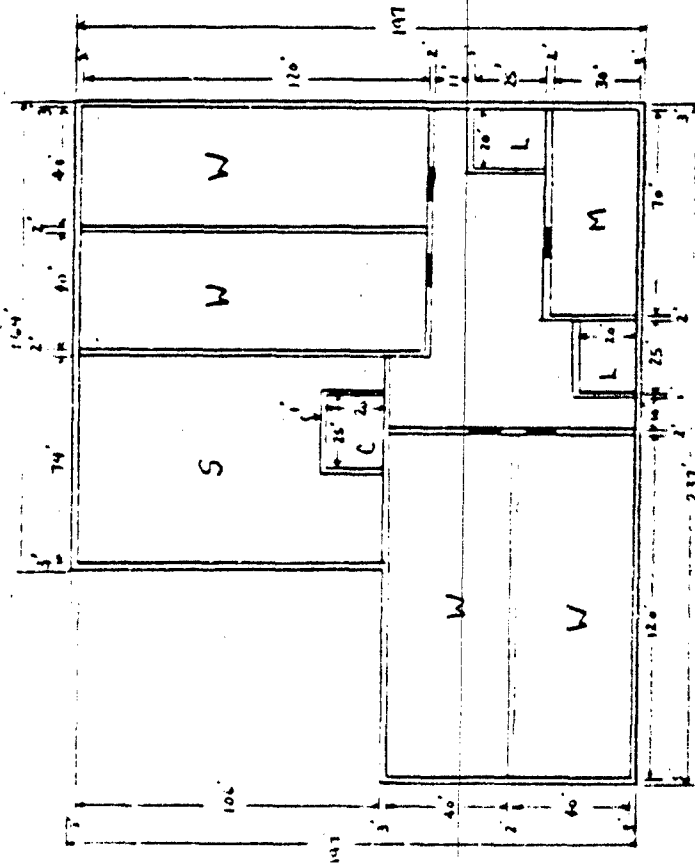
500 - 540 = 0
540 - 580 = 1
580 - 620 = 2
620 - 660 = 3
660 - 700 = 4
700 - 740 = 5
740 - 780 = 6
780 - 820 = 7
820 - 860 = 8
860 - 900 = 9
900 - 940 = 10
940 - 980 = 11
980 - 1000 = 12
1000 - 1040 = 13
1040 - 1080 = 14
1080 - 1120 = 15
1120 - 1160 = 16
1160 - 1200 = 17
1200 - 1240 = 18
1240 - 1280 = 19
1280 - 1320 = 20
1320 - 1360 = 21
1360 - 1400 = 22
1400 - 1440 = 23
1440 - 1480 = 24
1480 - 1520 = 25
1520 - 1560 = 26
1560 - 1600 = 27
1600 - 1640 = 28
1640 - 1680 = 29
1680 - 1720 = 30
1720 - 1760 = 31
1760 - 1800 = 32
1800 - 1840 = 33
1840 - 1880 = 34
1880 - 1920 = 35
1920 - 1960 = 36
1960 - 2000 = 37
2000 - 2040 = 38
2040 - 2080 = 39
2080 - 2120 = 40
2120 - 2160 = 41
2160 - 2200 = 42
2200 - 2240 = 43
2240 - 2280 = 44
2280 - 2320 = 45
2320 - 2360 = 46
2360 - 2400 = 47
2400 - 2440 = 48
2440 - 2480 = 49
2480 - 2520 = 50
2520 - 2560 = 51
2560 - 2600 = 52
2600 - 2640 = 53
2640 - 2680 = 54
2680 - 2720 = 55
2720 - 2760 = 56
2760 - 2800 = 57
2800 - 2840 = 58
2840 - 2880 = 59
2880 - 2920 = 60
2920 - 2960 = 61
2960 - 3000 = 62
3000 - 3040 = 63
3040 - 3080 = 64
3080 - 3120 = 65
3120 - 3160 = 66
3160 - 3200 = 67
3200 - 3240 = 68
3240 - 3280 = 69
3280 - 3320 = 70
3320 - 3360 = 71
3360 - 3400 = 72
3400 - 3440 = 73
3440 - 3480 = 74
3480 - 3520 = 75
3520 - 3560 = 76
3560 - 3600 = 77
3600 - 3640 = 78
3640 - 3680 = 79
3680 - 3720 = 80
3720 - 3760 = 81
3760 - 3800 = 82
3800 - 3840 = 83
3840 - 3880 = 84
3880 - 3920 = 85
3920 - 3960 = 86
3960 - 4000 = 87
4000 - 4040 = 88
4040 - 4080 = 89
4080 - 4120 = 90
4120 - 4160 = 91
4160 - 4200 = 92
4200 - 4240 = 93
4240 - 4280 = 94
4280 - 4320 = 95
4320 - 4360 = 96
4360 - 4400 = 97
4400 - 4440 = 98
4440 - 4480 = 99
4480 - 4520 = 100

Floor Area

500 - 540 = 0
540 - 580 = 1
580 - 620 = 2
620 - 660 = 3
660 - 700 = 4
700 - 740 = 5
740 - 780 = 6
780 - 820 = 7
820 - 860 = 8
860 - 900 = 9
900 - 940 = 10
940 - 980 = 11
980 - 1000 = 12
1000 - 1040 = 13
1040 - 1080 = 14
1080 - 1120 = 15
1120 - 1160 = 16
1160 - 1200 = 17
1200 - 1240 = 18
1240 - 1280 = 19
1280 - 1320 = 20
1320 - 1360 = 21
1360 - 1400 = 22
1400 - 1440 = 23
1440 - 1480 = 24
1480 - 1520 = 25
1520 - 1560 = 26
1560 - 1600 = 27
1600 - 1640 = 28
1640 - 1680 = 29
1680 - 1720 = 30
1720 - 1760 = 31
1760 - 1800 = 32
1800 - 1840 = 33
1840 - 1880 = 34
1880 - 1920 = 35
1920 - 1960 = 36
1960 - 2000 = 37
2000 - 2040 = 38
2040 - 2080 = 39
2080 - 2120 = 40
2120 - 2160 = 41
2160 - 2200 = 42
2200 - 2240 = 43
2240 - 2280 = 44
2280 - 2320 = 45
2320 - 2360 = 46
2360 - 2400 = 47
2400 - 2440 = 48
2440 - 2480 = 49
2480 - 2520 = 50
2520 - 2560 = 51
2560 - 2600 = 52
2600 - 2640 = 53
2640 - 2680 = 54
2680 - 2720 = 55
2720 - 2760 = 56
2760 - 2800 = 57
2800 - 2840 = 58
2840 - 2880 = 59
2880 - 2920 = 60
2920 - 2960 = 61
2960 - 3000 = 62
3000 - 3040 = 63
3040 - 3080 = 64
3080 - 3120 = 65
3120 - 3160 = 66
3160 - 3200 = 67
3200 - 3240 = 68
3240 - 3280 = 69
3280 - 3320 = 70
3320 - 3360 = 71
3360 - 3400 = 72
3400 - 3440 = 73
3440 - 3480 = 74
3480 - 3520 = 75
3520 - 3560 = 76
3560 - 3600 = 77
3600 - 3640 = 78
3640 - 3680 = 79
3680 - 3720 = 80
3720 - 3760 = 81
3760 - 3800 = 82
3800 - 3840 = 83
3840 - 3880 = 84
3880 - 3920 = 85
3920 - 3960 = 86
3960 - 4000 = 87
4000 - 4040 = 88
4040 - 4080 = 89
4080 - 4120 = 90
4120 - 4160 = 91
4160 - 4200 = 92
4200 - 4240 = 93
4240 - 4280 = 94
4280 - 4320 = 95
4320 - 4360 = 96
4360 - 4400 = 97
4400 - 4440 = 98
4440 - 4480 = 99
4480 - 4520 = 100

Ceiling Ht.

Ave. Ht. = 10'
Pac-Man = 15'
Pile = 10'



C = 500 sq.
 L = 500 sq./ea.
 S = 7844 sq.
 M = 2100 sq.
 W = 4800 sq.

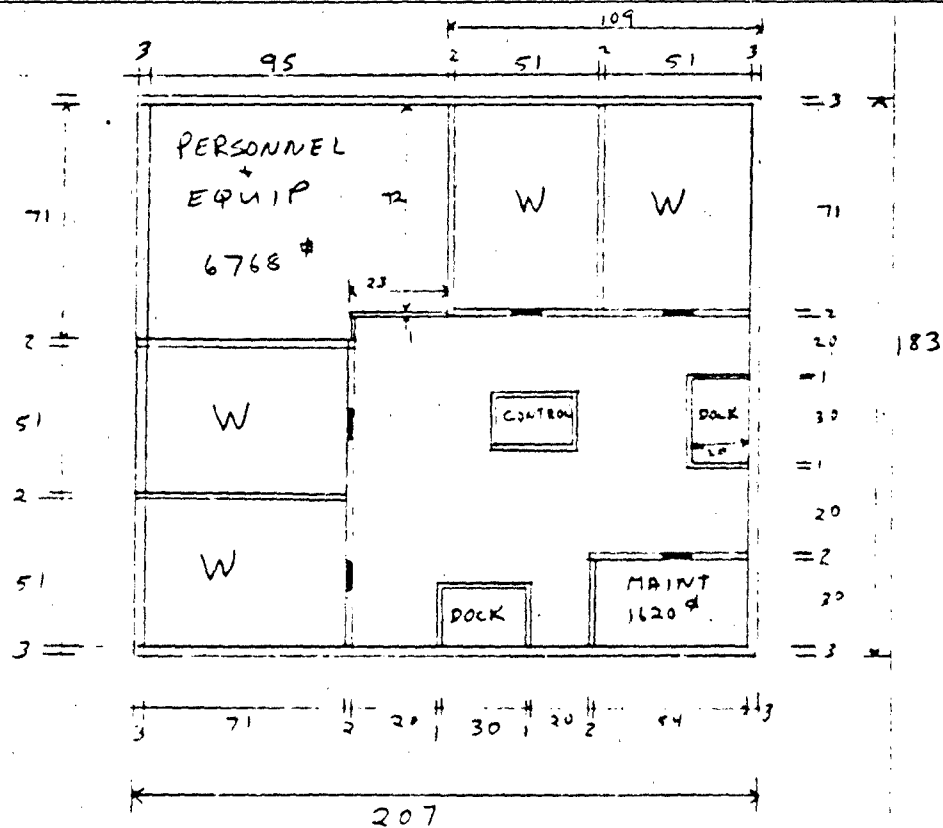
Total Area = 38732 sq.
 Exterior Wall = 8630 sq.
 Interior Wall = 6780 sq.
 No. Glass Doors = 5
 120 x 40 Bay

Design No. 4-1
 Revision 296

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO. _____
OF _____

PROJECT NO. 02-7092 SPONSOR CERL
SUBJECT LAYOUT - 51 771 INTERIOR BAY
BY W.D. DATE 19 CHECKED BY W.D. DATE CHECKED 19



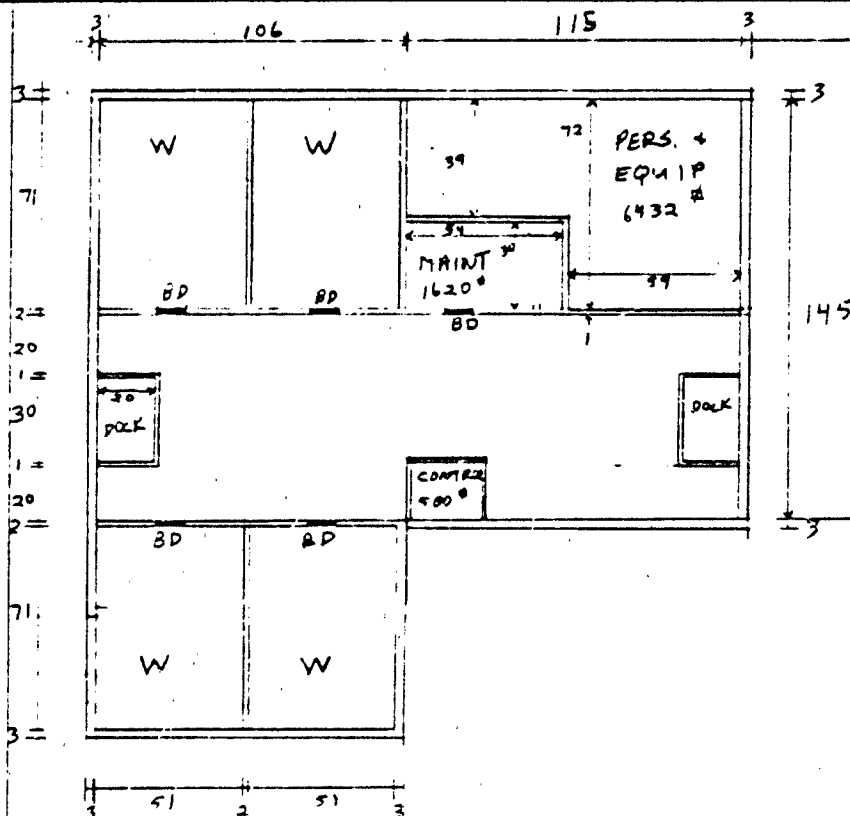
W = 3621 ft²
S = 6768 ft²
M = 1620 ft²
L = 600 ft²
C = 600 ft²
TOTAL AREA = 37,881 ft²
EXTERIOR WALL = 11,700 ft² (15' Tall)
INTERIOR WALL = 8580 ft²
BLAST DOOR NO. = 5
DIVIDING WALL BAY

DESIGN 4-2

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
OF

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: LAYOUT - 51' x 31' INTERIOR BAYS
BY: B. MORRIS DATE: 13 JUL 19 92 CHECKED BY: not DATE CHECKED: 19



W = 1581 ft²
S = 6432 ft²
M = 1620 ft²
L = 600 ft²/m
C = 500 ft²
Total Area = 42,307
Exterior Wall = 13,380 ft² (115' Tall)
Interior Wall = 7515 ft²
Blast Door No. = 5
DIVIDING WALL BAY

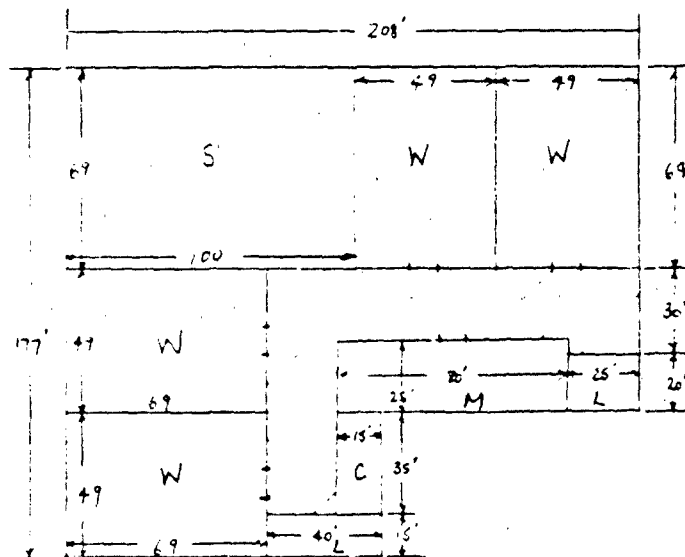
1-36

Design 4-3

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO. _____
OF _____

PROJECT NO.: 62-7092 SPONSOR: CERL
SUBJECT: LAYOUT
BY: MMW DATE: 19 CHECKED BY: mtb DATE CHECKED: 19



W = 3391 ft²/ea
S = 6900 ft²
M = 2000 ft²
L = 500 ft²/ea
C = 525 ft²
TOTAL AREA = 31,800 ft²
EXTERIOR WALL = 11,550 ft²
INTERIOR WALL = 9105 ft²
BLAST DOOR NO. = 5

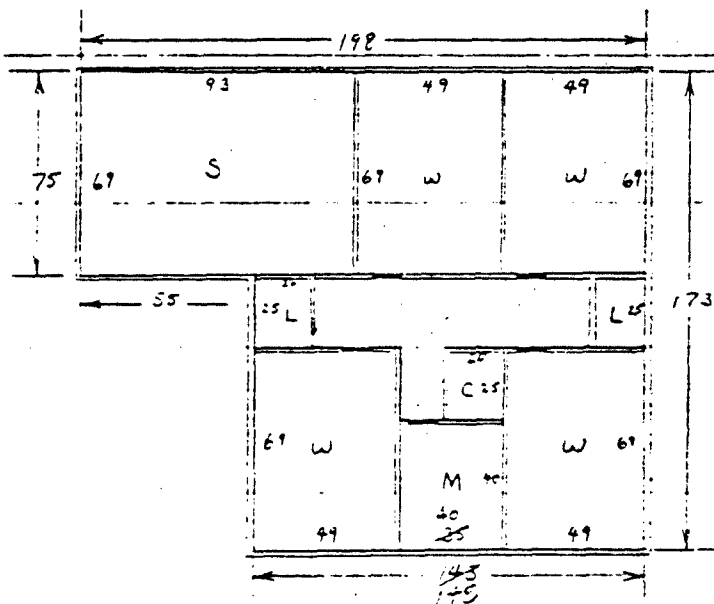
PIT BAY 1-37

DESIGN 4-4

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO. _____
OF _____

PROJECT NO.: 22-7092 SPONSOR: CERL
SUBJECT: LAYOUT
BY: M. W. DATE: 19 CHECKED BY: mtb DATE CHECKED: 19



ABOVE GROUND

W = 3381 sq/ft (F.T. BARS)
S = 6417 sq/ft
M = 1200 sq/ft
L = 500 sq/ft
C = 500 sq/ft
TOTAL AREA = 28864 sq/ft
EXTERIOR WALL = 11,130 sq/ft
INTERIOR WALL = 7605 sq/ft
BLAST DOOR NO = 5

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO. _____
OF _____

PROJECT NO.: 62-7092

SPONSOR: CERL

SUBJECT: - A-5 OUT

BY: mmh

DATE: 19

CHECKED BY: MW

DATE CHECKED: 19

	61	62	63	64	A-5
1 • TOTAL FLOOR AREA ^(H)	8 24	7 21	6 18	10 30	10 30
2 • OPER. EFFICIENCY	10 10	5 5	10 10	5 5	10 10
2a - Bay - Main - Bay ^(H)					
2b - London ^(H)	10 30	7 21	7 21	7 21	7 21
3 • EXPANDABILITY ^(L)	10 10	10 10	10 10	10 10	10 10
4 • LEAD DOOR SEPARATION ^(H)	3 30	10 30	10 30	10 30	10 30
5 • PERIMETER WALL AREA/Vol. ^(H)	10 20	4 8	1 2	4 8	5 10
6 • BAY INTERIOR WALL AREA/Vol. ^(H)	7 14	3 6	5 10	2 4	5 10
7 • "SQUAREDNESS" OF S.M. ^(H)	10 20	0 20	10 20	7 14	10 20
8 • QUANTITY OF BAY DOORS ^(L)	10 10	10 10	10 10	10 10	10 10
9 • WEATHING BAY LOCATION ^(L)	3 8	3 8	6 8	5 8	6 8
10 • QUANTITY OF MAINT. BAYS ^(H)	10 30	10 30	10 30	10 30	10 30
11 • EQUIPMENT NEEDS ^(H)	10 30	7 21	7 21	7 21	7 21
- Bay Needs ^(H)					
- Perimeter Floor Area ^(H)	10 30	0 0	0 0	0 0	0 0
- Bay Floor Area ^(H)	10 30	7 21	7 21	7 21	7 21
	296	211	211	212	231

MULTIPLIERS:

H = 3
M = 2
L = 1

CEILING HT.

Avg Wall = 10'
PAC-MAN = 15'
PITS = 10'

FLOOR AREA RANGING:

30 - 32.5 = 10
32.5 - 35 = 9
35 - 37.5 = 8
37.5 - 40 = 7
40 - 42.5 = 6
42.5 - 45 = 5
45 - 47.5 = 4
47.5 - 50 = 3
50 - 52.5 = 2
52.5 - 55 = 1
55 - 60 = 0

1-39

EXTENSIVE AREA

8100 - 8400 = 10 : 5000 - 5500
8400 - 9200 = 9 : 5500 - 6000
9200 - 9732 = 8 : 6000 - 6500
9732 - 10260 = 7 : 6500 - 7000
10260 - 10780 = 6 : 7000 - 7500
10780 - 11300 = 5 : 7500 - 8000
11300 - 11820 = 4 : 8000 - 8500
11820 - 12352 = 3 : 8500 - 9000
12352 - 12900 = 2 : 9000 - 9500
12900 - 13400 = 1 : 9500 - 10000
13400 - = 0 : 10000 -



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DEPARTMENT OF BALLISTICS AND EXPLOSIVES SCIENCES
COMPUTATION SHEET

SHEET NO. _____
OF _____

PROJECT NO. 22-7092

SPONSOR CERL

SUBJECT LAYOUT

BY MV

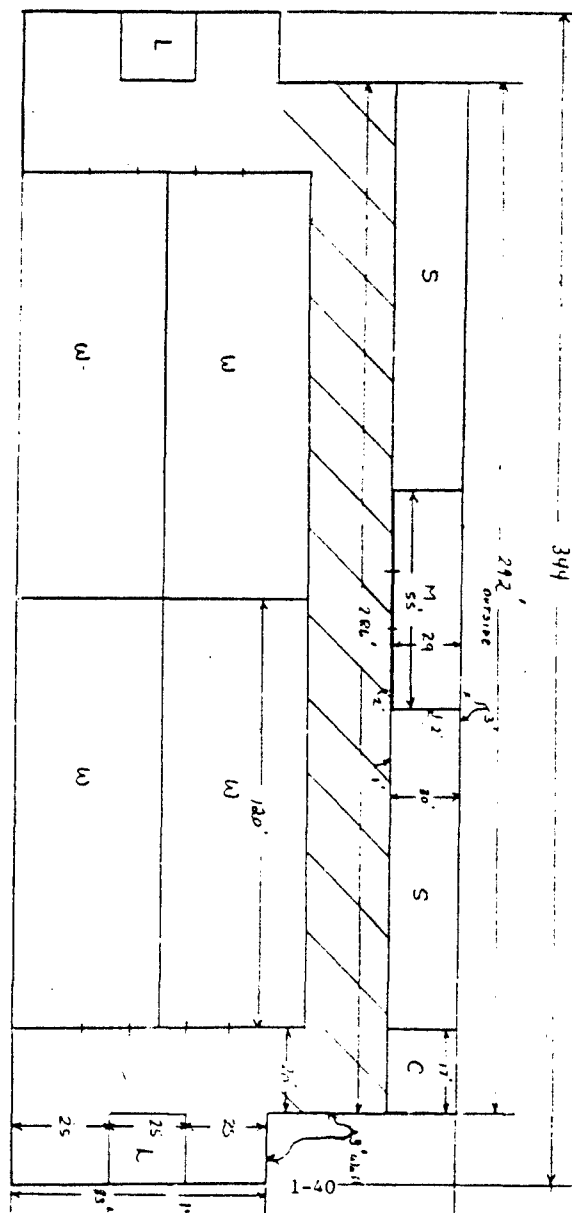
DATE _____

19 _____

CHECKED BY nwb

DATE CHECKED _____

19 _____



(1) : 4800 ft²/ca
S : 6270 ft²
M : 1595 ft²
L : 500 ft²/ca
C : 510 ft²

Total Area : 45448 ft²
Exterior Area : 9700 ft²
Interior Area : 3330 ft²
Blasr Dose = 5

120140 BAY

DESIGN 5-1



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COMPUTATION SHEET

SHEET NO.
1 OF 1

PROJECT NO. 02-7092

SPONSOR CERL

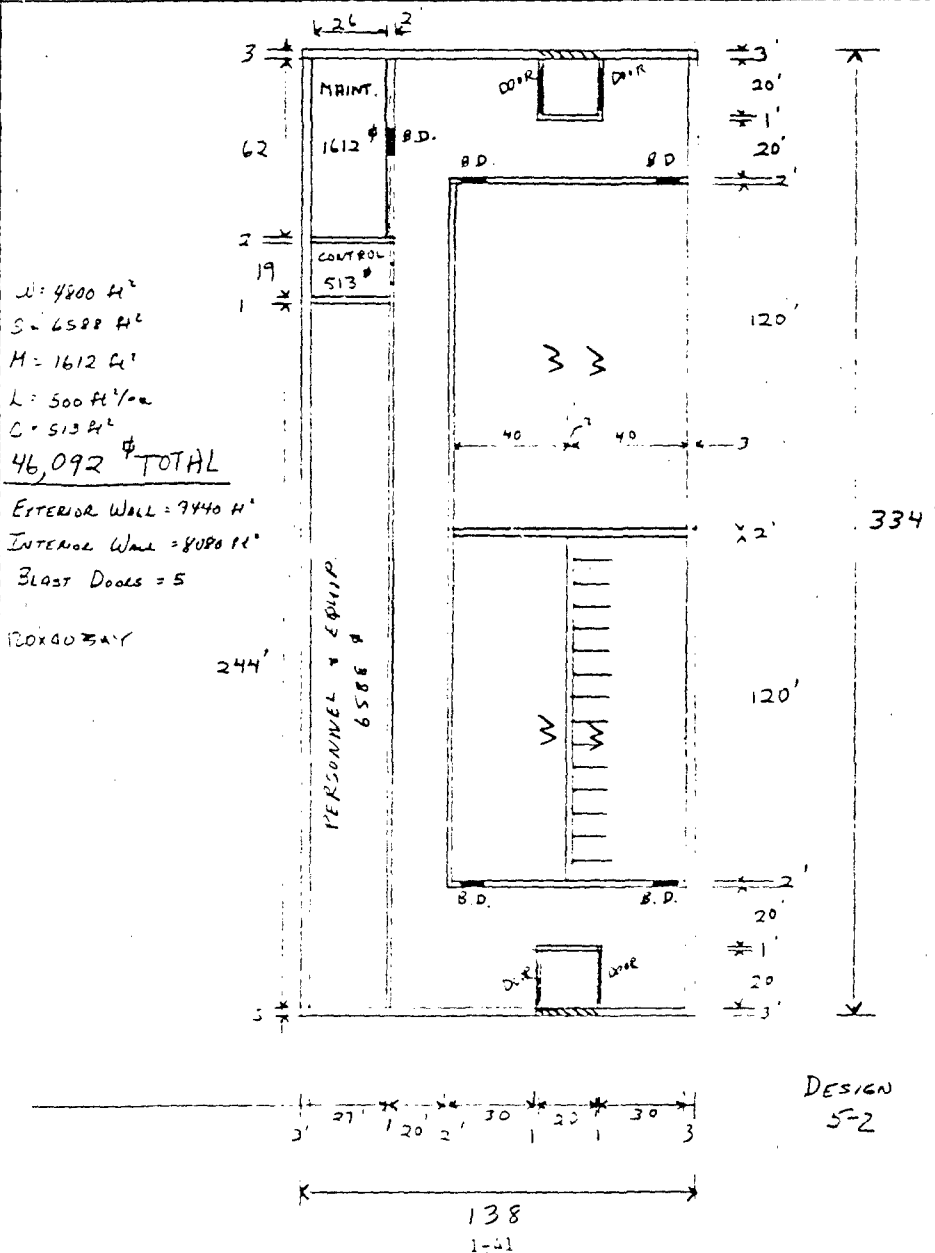
SUBJECT LAYOUT

BY E. MORRIS

DATE 7 JUL 19 82

CHECKED BY *WMB*

DATE CHECKED. 19



SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

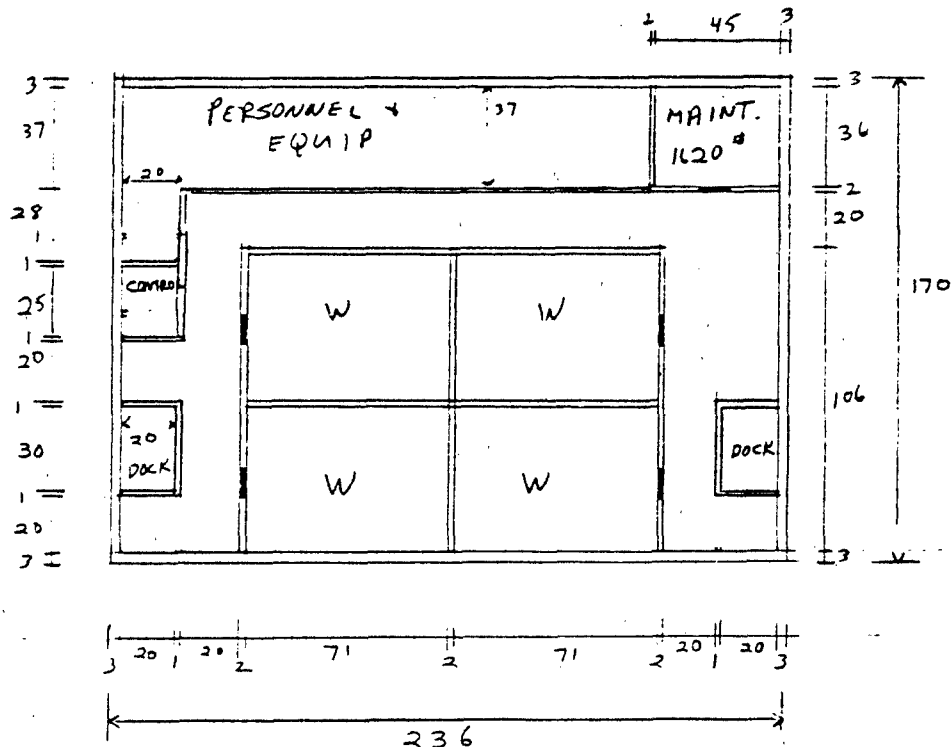
SHEET NO.
OF

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: LAYOUT - 51' x 71' INTERIOR BAY
BY: B. MORRIS DATE: 13 JUL 19 82 CHECKED BY: meb DATE CHECKED: 19

interior wall - $671 \times 15 = 10,065$
exterior wall - $812 \text{ ft length} \times 15' \text{ ht} = 12,180$

S.B.D.

40 120'



W: 3621 ft²/ea
S: 7231 ft²
M: 1620 ft²
L: 600 ft²/ea
C: 500 ft²

TOTAL AREA = 40120 ft²
EXTERIOR AREA = 12180 ft²
INTERIOR AREA = 10065 ft²
BLAST DOOR NO. = 5
DIVIDING WALL BAY
1-42

DESIGN 5-3

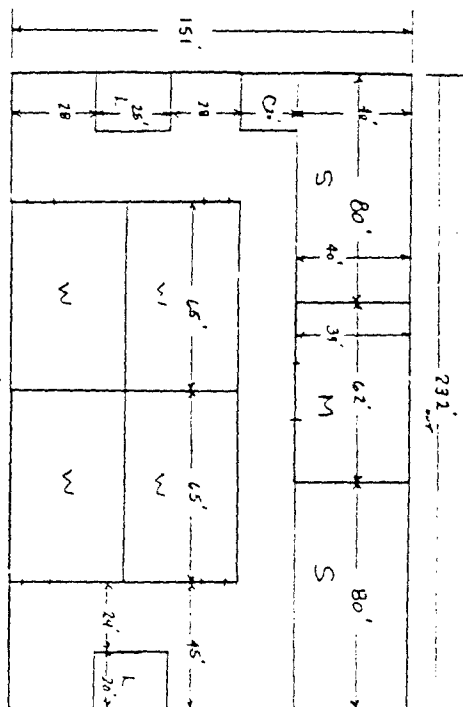


SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF BALLISTICS AND EXPLOSIVES SCIENCES
COMPUTATION SHEET

SHEET NO. _____
OF _____

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: LAYOUT
BY: LMV DATE: 19 CHECKED BY: mtb DATE CHECKED: 19

STACKED MUNITIONS



W = 2400 ft²/ea
S = 6400 ft²
M = 2418 ft²
L = 500 ft²/ea
C = 400 ft²

Total Area = 35032 ft²
Exterior Area = 13788 ft²
Interior Area 10819 ft²
Door Doors = 6
Design
S-4
PIT HALL

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO. _____
OF _____

PROJECT NO.: 62-7092 SPONSOR: CERL
SUBJECT: LAYOUT
BY: WUT DATE: 19 CHECKED BY: WV DATE CHECKED: 19

		5-1		5-2		5-3		5-4	
1 •	TOTAL FLOOR AREA ^(H)	3	9	4	12	6	18	8	24
2 •	OPER. EFFICIENCY	2	2	1	1	2	2	3	3
2a	- Bay - Main - Bay ^(H)								
2b	- Loadout ^(H)	10	30	10	30	7	21	7	21
3 •	EXPANDABILITY ^(H)	10	10	10	10	10	10	10	10
4 •	LOAD DOCK SEPARATION ^(H)	10	30	10	30	10	30	10	30
5 •	PERIMETER WALL AREA/Vol. ^(H)	8	16	8	16	3	6	0	0
6 •	BAY INTERIOR WALL AREA/Vol. ^(H)	4	8	4	8	0	0	0	0
7 •	"SQUAREDNESS" OF S.M.I. ^(H) 5 DOORS = 10; CORNERS = 9	0	0	0	0	2	4	10	20
8 •	QUANTITY OF BAY DOORS ^(H) Ext = 8; Int = 10	10	10	10	10	10	10	10	10
9 •	WEAPONS BAY LOCATION ^(H) (SW/K...)	8	8	8	8	8	8	8	8
10 •	QUANTITY OF MAINT. BAYS ^(H) 1 Bay = 10, 2 Bays = 5	10	30	10	30	10	30	10	30
11 •	EQUIPMENT NEEDS ^(H)								
	- BASE NEEDS ^(H) 2 Floors = 10	10	30	10	30	7	21	7	21
	- PERIMETER FLOOR: 100	10	30	10	30	0	0	0	0
	- PINE ROADS ^(H) 100 = 100	10	30	10	30	7	21	7	21
		243		245		181		198	

MULTIPLIERS:

H = 3
M = 2
L = 1

CEILING HT.

AVE WALL = 10'
PRE-MAN = 15'
PITS = 15'

FLOOR AREA RANKINGS:

30 - 32.5K = 10
32.5 - 35 = 9
35 - 37.5 = 8
37.5 - 40 = 7
40 - 42.5 = 6
42.5 - 45 = 5
45 - 47.5 = 4
47.5 - 50 = 3
50 - 52.5 = 2
52.5 - 55 = 1
55 - 60 = 0

1-44

EXTERIOR AREA

5100 - 8480 = 10 = 5000 - 5500
8480 - 9200 = 9 = 5500 - 6000
9200 - 9732 = 8 = 6000 - 6500
9732 - 10250 = 7 = 6500 - 7000
10250 - 10780 = 6 = 7000 - 7500
10780 - 11304 = 5 = 7500 - 8000
11304 - 11828 = 4 = 8000 - 8500
11828 - 12352 = 3 = 8500 - 9000
12352 - 12900 = 2 = 9000 - 9500
12900 - 13400 = 1 = 9500 - 10000
13400 - = 0 = 10000 -

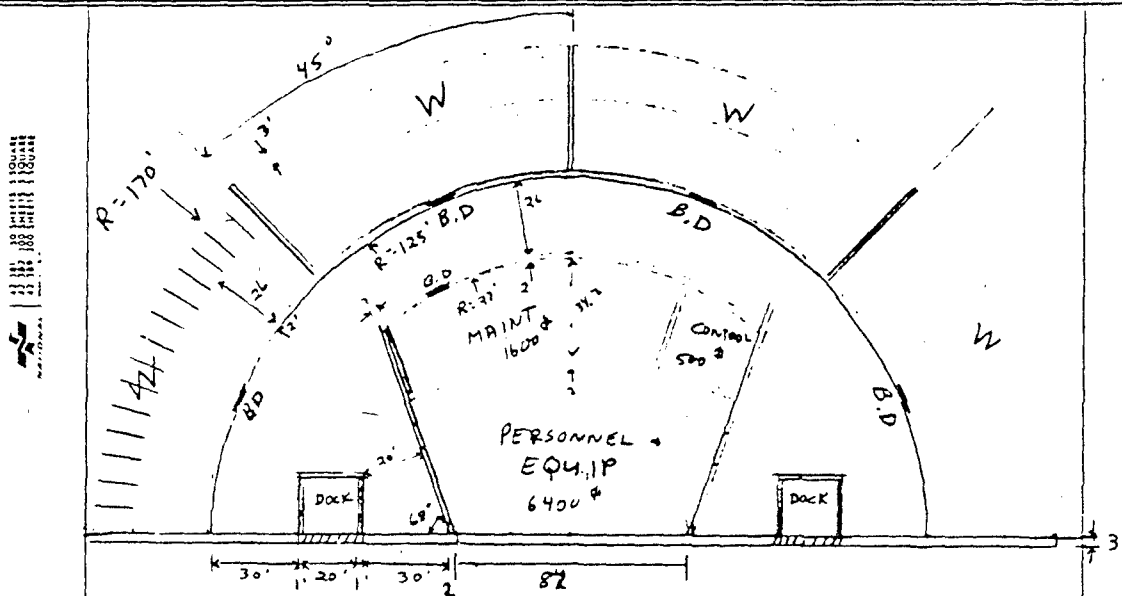
INT. AREA



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COMPUTATION SHEET

SHEET NO.
1 OF 2

PROJECT NO. 02-7092 SPONSOR CERL
SUBJECT CIRCULAR LAYOUT
BY E. MORRIS DATE 7 JUL 19 82 CHECKED BY mob DATE CHECKED 19



$$W = 4600 \text{ ft}^2/\text{ea (avg)}$$

$$S = 6400 \text{ ft}^2$$

$$M = 1600 \text{ ft}^2$$

$$L = 400 \text{ ft}^2/\text{ea}$$

$$C = 500 \text{ ft}^2$$

$$\text{TOTAL AREA} = 46,416 \text{ ft}^2$$

$$\text{EXTERIOR WALL} = 9740 \text{ ft}^2$$

$$\text{INTERIOR WALL} = 4920 \text{ ft}^2$$

$$\text{BLAST DOOR NO} = 6$$

$$100 \times 40 \text{ ft}$$

DESIGN 6-1



SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF BALLISTICS AND EXPLOSIVES SCIENCES
COMPUTATION SHEET

SHEET NO.
2 OF 2

PROJECT NO. 02-7092

SPONSOR: CERL

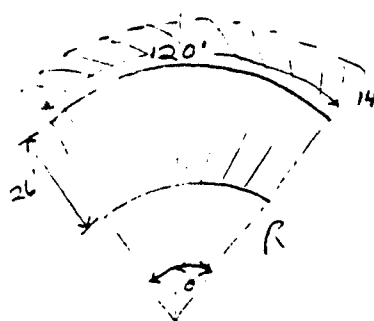
SUBJECT: LAYOUT (CIRCULAR)

BY: E. MORRIS

DATE: 7 JUL 79

CHECKED BY: *mtb*

DATE CHECKED: 19



$$\text{Arc Length} = 2\pi R \frac{\theta}{360} = 120'$$

$$R = \frac{120 \times 360}{2\pi\theta}$$

$$R = 6875.5 \frac{1}{\theta}$$

θ	R	$\Delta \text{ area}$
30	229	13,729
45	153	9,193
60	115	6,925
75	92	5,540
36	191	11,461
20	344	

$$\text{Area in } 26' \text{ section} = \frac{\theta}{360} [\pi (R^2 - (R-26)^2)]$$

θ	Area (26' section)	Area (14')	Total
30	2941	1730	4671
36	2908	1742	4650
45	2859	1759	4618
60	2777	1789	4566
75	2689	1814	4503

straight

$$120 \times 40 = 4800$$

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
1 OF 2

PROJECT NO. 02-7092

SPONSOR:

CERL

SUBJECT: CIRCULAR LAYOUT - DOUBLE STACKED

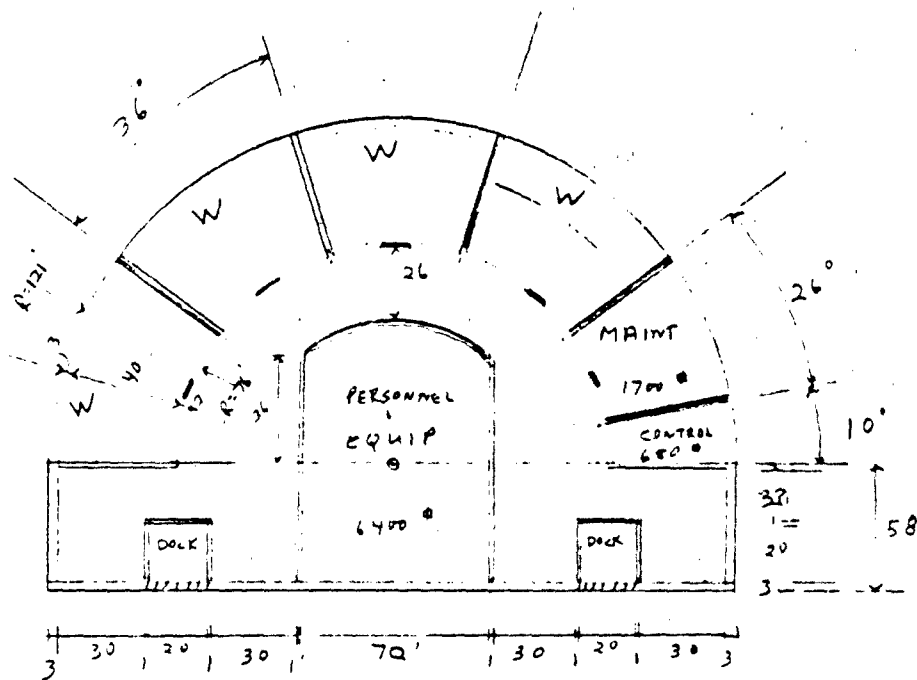
BY: B. MURRIS

DATE: JUL 19 82

CHECKED BY:

DATE CHECKED:

19



$$W = 2425 \text{ ft}^2/\text{ea (Avg)}$$

$$S = 6400 \text{ ft}^2$$

$$M = 1700 \text{ ft}^2$$

$$L = 400 \text{ ft}^2/\text{ea}$$

$$C = 680 \text{ ft}^2$$

$$\text{TOTAL AREA} = 37,034 \text{ ft}^2$$

$$\text{EXTERIOR WALL} = 13,284 \text{ ft}^2$$

$$\text{INTERIOR WALL} = 9370 \text{ ft}^2$$

$$\text{BLAST DOOR NO} = 5$$

1-47

DOUBLE STACKED BAY

DESIGN 6-2



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COMPUTATION SHEET

SHEET NO.
2 of 2

PROJECT NO. 02-7092

SPONSOR

CERL

SUBJECT CIRCULAR LAYOUT - DOUBLE STACKED

BY E. MORRIS

DATE F. 344 19 82

CHECKED BY

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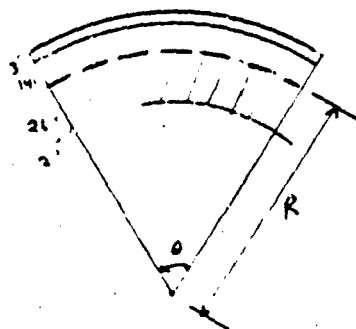
DATE CHECKED

19

$$\text{total arc length} = 7.8 + 9.1 + 2 = 18.9'$$

$$\text{arc length} = 2\pi R \frac{\theta}{360} = 65'$$

$$R = 3724 \frac{1}{\theta}$$



θ	$R(M)$	total area
30°	124	2471
36°	104	2463
45°	83	2419
60°	62	2346

$$\text{total area} = \frac{\theta}{360} \pi [(R+14)^2 - (R-26)^2]$$

$$\text{for max area, } \frac{\theta}{360} \pi [118' - 78'] = 1600 \Rightarrow \theta = 23.3$$

$$\text{for min area, } \frac{\theta}{360} \pi [118' - 78'] = 500 \Rightarrow \theta = \frac{7.3'}{30.6'} = 0.2$$

$$\text{for } R = \frac{3724}{\theta}, \quad \theta = \frac{3724}{R}$$

R	θ
78	38
100	37.2

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
OF

PROJECT NO. 02-7092

SPONSOR: CERL

SUBJECT LA 500T

BY: ST

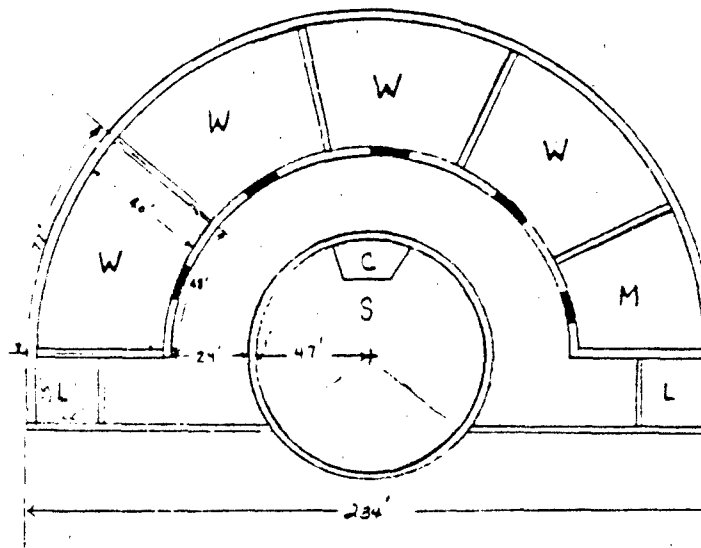
DATE

19

CHECKED BY WBT

DATE CHECKED

19



$$W = 2425 \text{ ft}^3/\text{cu} \text{ (avg)}$$

$$S = 6440 \text{ ft}^3$$

$$M = 1700 \text{ ft}^3$$

$$L = 500 \text{ ft}^3/\text{cu}$$

$$C = 500 \text{ ft}^3$$

$$\text{Total Area} = 27668 \text{ ft}^2$$

$$\text{External Wall} = 10251 \text{ ft}^2$$

$$\text{Internal Wall} = 8352 \text{ ft}^2$$

$$\text{Blast Door No} = 5$$

DOUBLE STACED BAY

DESIGN 6-3



SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF BALLISTICS AND EXPLOSIVES SCIENCES
COMPUTATION SHEET

SHEET NO.
OF

PROJECT NO. 02-7092

SPONSOR

CERL

SUBJECT CIRCULAR LAYOUT

BY E. MORRIS

DATE

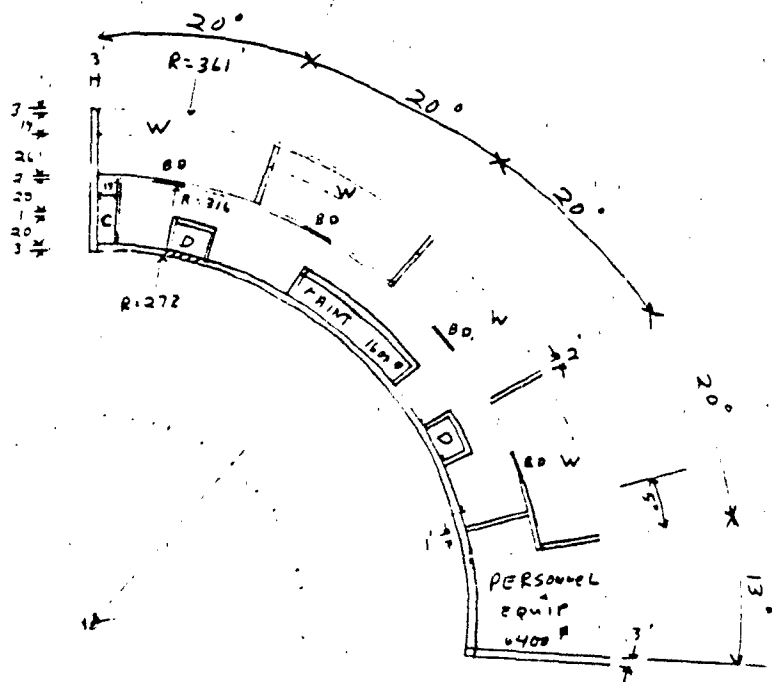
19

CHECKED BY

nmf

DATE CHECKED

19



$$\text{Total Area} = \frac{20}{360} \pi (361^2 - 272^2) + 2(3)(M) = 46,256 \text{ ft}^2$$

W = 4900 ft² (again)

S = 6400 ft²

M = 1600 ft²

L = 500 ft²/ea

C = 588 ft²

TOTAL AREA = 46,256 ft²

EXTERNAL WALL = 11930 ft

INTERNAL WALL = 7215 ft

BUILT DOOR NO = 5

DOUBLE STACKED BAY

1-50

DESIGN 6-K

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO. _____
OF _____

PROJECT NO.: 02-7092

SPONSOR: SEEL

SUBJECT: AYDUT

BY: TRUTH

DATE: 19

CHECKED BY: MW

DATE CHECKED: 19

	6-1	6-2	6-3	6-4
1 • TOTAL FLOOR AREA ^(H)	4 12	8 24	8 24	4 12
2 • OPER. EFFICIENCY	3 3	3 3	3 3	3 3
2a - Bay - Main - Bay ^(H)				
2b - Loadout ^(H)	10 30	7 21	7 21	10 30
3 • EXPANSABILITY ^(H)	10 10	10 10	5 5	5 5
4 • LOAD DOCK SEPARATION ^(H)	10 30	10 30	10 30	10 30
5 • PERIMETER WALL AREA/Vol ^(H)	9 18	1 2	5 10	3 6
6 • BAY INTERIOR WALL AREA/Vol ^(H)	10 20	4 8	4 8	6 12
7 • "SQUAREDNESS" OF S.M.I. ^(H)	10 20	10 20	10 20	2 4
5 Doors = 10; 4 Doors = 9				
8 • QUANTITY OF BAY DOORS ^(H)	10 10	10 10	10 10	10 10
Eq. = 8, Int. = 10				
9 • WEAPONS BAY LOCATION ^(H) (Fur. K.)	8 8	8 8	8 8	8 8
10 • QUANTITY OF MAINT. BAYS ^(H)	10 30	10 30	10 30	10 30
11 • EQUIPMENT NEEDS ^(H)				
- BAY VOLUME ^(H)	10 30	7 21	7 21	10 30
- PERIMETER FLOOR AREA	10 30	0 0	0 0	10 30
- P.W. FLOOR AREA ^(H)	10 30	7 21	7 21	10 30

281

208

211

240

MULTIPLIERS:

H = 3
M = 2
L = 1

FLOOR AREA RANGES:

30 - 32.5 = 10
32.5 - 35 = 9
35 - 37.5 = 8
37.5 - 40 = 7
40 - 42.5 = 6
42.5 - 45 = 5
45 - 47.5 = 4
47.5 - 50 = 3
50 - 52.5 = 2
52.5 - 55 = 1
55 - 60 = 0

EXTENDED AREA

8160 - 8420 = 0
8420 - 9200 = 9
9200 - 9732 = 8
9732 - 10260 = 7
10260 - 10780 = 6
10780 - 11304 = 5
11304 - 11828 = 4
11828 - 12352 = 3
12352 - 12900 = 2
12900 - 13400 = 1
13400 - 14000 = 0

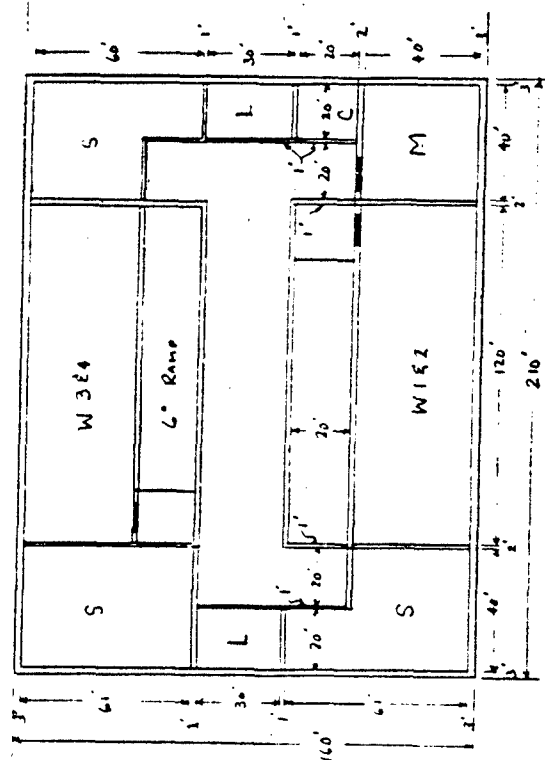
INT. AREA

5000 - 5500
5500 - 6000
6000 - 6500
6500 - 7000
7000 - 7500
7500 - 8000
8000 - 8500
8500 - 9000
9000 - 9500
9500 - 10000
10000 - 10500

CEILING HT.

AVE WALL = 10'
PAC-MAN = 15'
PITS = 15'

1-51



Areas

C = 400 ft²
 L = 600 ft²
 S = 6440 ft²
 M = 1600 ft²
 W = 4800 ft²
 Total Area = 33,600 ft²
 Exterior Wall = 14800 ft²
 Interior Wall = 8400 ft²
 Blower Door 110 = S
 120 x 40 15A/15

Design No. 7-1
 Printed 276

SHEET NO. _____ OF _____

PROJECT NO: 02-7092 SPONSOR: CERL
SUBJECT: LAYOUT
BY: mtb DATE: 19 CHECKED BY: SW DATE CHECKED: 19

[illegible]

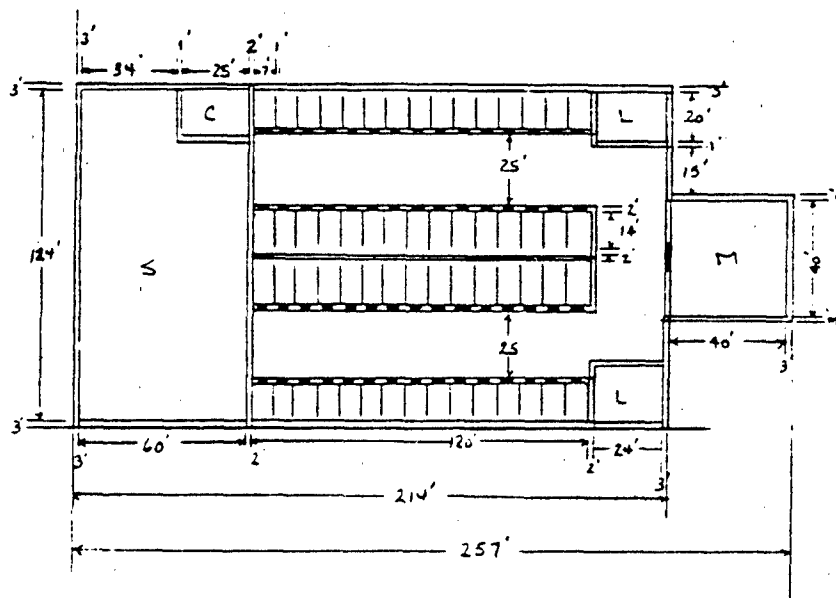
276

<u>MULTIPLIERS:</u>	<u>FLOOR AREA RANKING:</u>	<u>EXPOSED AREA</u>	<u>INT. AREA</u>
H = 3	30 - 32.5K = 10	5120 - 2680 = 10	5000 - 5500
M = 2	32.5 - 35 = 9	8680 - 9200 = 9	5500 - 6000
L = 1	35 - 37.5 = 8	9200 - 9720 = 8	6000 - 6500
	37.5 - 40 = 7	9720 - 10260 = 7	6500 - 7000
	40 - 42.5 = 6	10260 - 10780 = 6	7000 - 7500
CEILING HT.	42.5 - 45 = 5	10780 - 11300 = 5	7500 - 8000
Ave Wall = 10'	45 - 47.5 = 4	11300 - 11820 = 4	8000 - 8500
PRO-MAN = 15'	47.5 - 50 = 3	11820 - 12350 = 3	8500 - 9000
PITS = 16'	50 - 52.5 = 2	12350 - 12900 = 2	9000 - 9500
	52.5 - 55 = 1	12900 - 13400 = 1	9500 - 10000
	55 → ∞ = 0	13400 → ∞ = 0	10000 → ∞

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
OF

PROJECT NO.: 02-7012 SPONSOR: CERL
SUBJECT: A YOV
BY: XW DATE: 19 CHECKED BY: *mtb* DATE CHECKED: 11



W = 1920 ft²
S = 6894 ft²
M = 1600 ft²
L = 480 ft²
C = 500 ft²
TOTAL AREA = 29798 ft²
INTERIOR WALL = 6780
EXTERIOR WALL = 7620 ft²
BLAST DOOR NO = 61

INDIVIDUAL STORAGE BAY

SHEET NO.
OF

SPONSOR: CERL

BY: mw

DATE: _____ 19 ____

CHECKED BY: MMW

DATE CHECKED: _____ 19____

3-1

- | | | | | | | | | | |
|----|--|----|----|--|--|--|--|--|--|
| 1 | TOTAL FLOOR AREA ^(H) | 10 | 30 | | | | | | |
| 2 | OPER. EFFICIENCY | 8 | 8 | | | | | | |
| 2a | - Bay - Main - Bay ^(H) | | | | | | | | |
| 2b | - Loadou ^(H) FL=10
CR=7 | 10 | 30 | | | | | | |
| 3 | EXPANSABILITY ^(L) | 5 | 5 | | | | | | |
| 4 | LOAD DOCK SEPARATION ^(H) 2150/10
2150/10
CR=10 | 10 | 30 | | | | | | |
| 5 | PERIMETER WALL AREA/Vol ^(H) | 10 | 30 | | | | | | |
| 6 | BAY INTERIOR WALL AREA/Vol ^(H) | 7 | 14 | | | | | | |
| 7 | "SQUAREDNESS" OF S.M. I. ^(M)
S.Dimes = 10; L.Dimes = 9 etc | 9 | 18 | | | | | | |
| 8 | QUANTITY OF BAY DOORS ^(H)
E=10, I=10 | 0 | 0 | | | | | | |
| 9 | DESIGNING BAY LOCATION ^(H) (E=10, I=10) | 9 | 9 | | | | | | |
| 10 | QUANTITY OF MAINT. BAY ^(H)
2 Bays = 10, 2 Bays = 5 | 10 | 30 | | | | | | |
| 11 | EQUIPMENT NEEDS ^(H) | | | | | | | | |
| | - Base Needs FL=10
2150/10 CR=7 | 10 | 30 | | | | | | |
| | - Production Needs FL=10
2150/10 CR=7 | 10 | 30 | | | | | | |
| | - 2nd Floor FL=10
2150/10 CR=7 | 10 | 30 | | | | | | |

284

MULTIPLIERS:

Flood Area Ranking:

Extreme Area

INT. AREA

4.3

30-32.5K -10

8160-8680 * 10 , 5000-5500

$$M = 2$$

32: 25 29

$$8680 - 9200 = 9 : 5500 - 6000$$
$$L = 1$$

25 - 27.5 - 8

0200-5732 18: 6500-6500

275-40 = 7

9722 - 16260 : 7 - 4500 - 7000

$$40 - 42 i = 4$$

103102-10380 26.1 7000 - 7500

475 - 45 = 5

10780-11304 5 7500 - 8000

45 - 47.8 - 4

11304-11328 141 1000-1500

475-50

11879-11880 - 11880 - 11880 - 11880

60. 53 5 5 2

17352-17430 : 10 : 600 : 2500

525-66-1

128

$$L \rightarrow \infty : 0$$

13000 → 10000 →

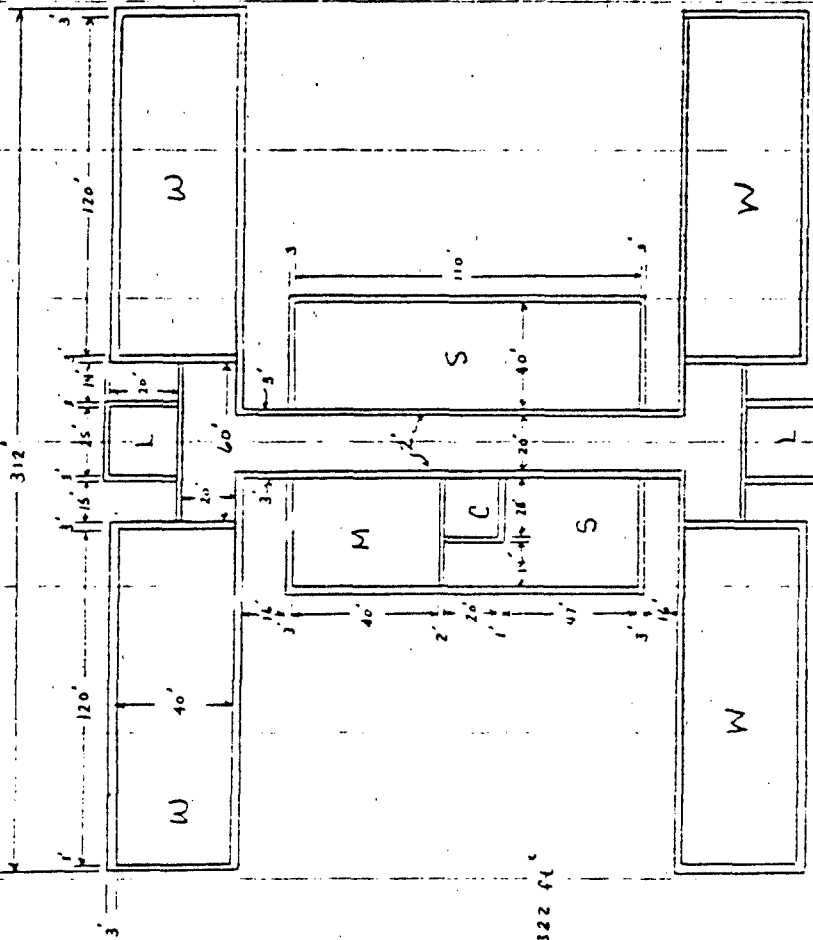
CEILING HT.

Five walls = 10'

PAC-MAN - 15

$P_{175} = 15'$

Design 9-1



$W = 4800 \text{ ft}^2$
 $L = 500 \text{ ft}^2$
 $C = 500 \text{ ft}^2$
 $S = 6560 \text{ ft}^2$
 $M = 1600 \text{ ft}^2$
 Total Area = 41322 ft^2
 120 x 40 BAY

SHEET NO. _____ OF _____

[illegible]

MULTIPLIERS:	FLOOR AREA RANKING:	EXT. AREA	INT. AREA
H = 3	30 - 22.5 = 10	8160 - 8680 = 10	5000 - 5500
M = 2	32.5 - 35 = 9	8680 - 9200 = 9	5500 - 6000
L = 1	35 - 37.5 = 8	9200 - 9732 = 8	6000 - 6500
	37.5 - 40 = 7	9732 - 10260 = 7	6500 - 7000
	40 - 42.5 = 6	10260 - 10780 = 6	7000 - 7500
CEILING 4T.	42.5 - 45 = 5	10780 - 11304 = 5	7500 - 8000
Avg Wall 10'	45 - 47.5 = 4	11304 - 11828 = 4	8000 - 8500
PAC-MAN = 16'	47.5 - 50 = 3	11828 - 12352 = 3	8500 - 9000
PITS = 15'	50 - 52.5 = 2	12352 - 12980 = 2	9000 - 9500
	52.5 - 55 = 1	12980 - 13400 = 1	9500 - 10000
	55 → ∞ = 0	13400 → ∞ = 0	10000 →

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COMPUTATION SHEET

SHEET NO.
OF

PROJECT NO.: 02-7092

SPONSOR: CERL

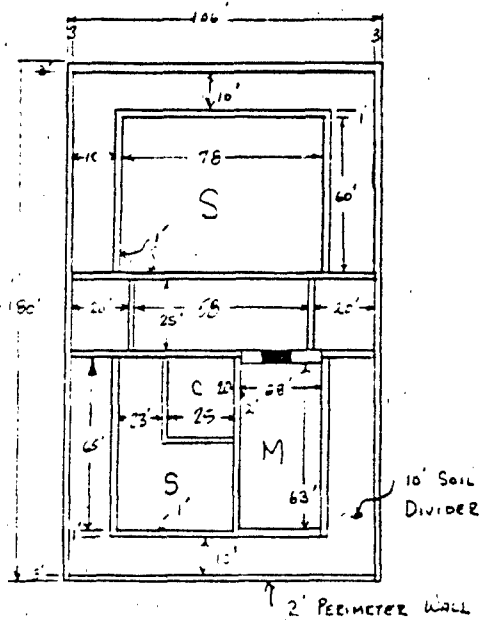
SUBJECT: LAYOUT

BY: MW

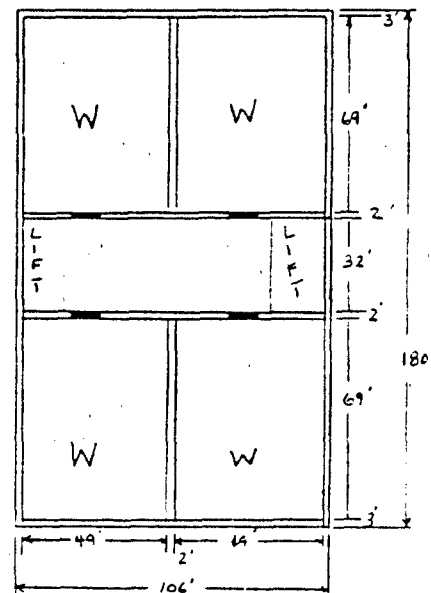
DATE: 19

CHECKED BY: *mw*

DATE CHECKED: 19



ABOVE GROUND



UNDER GROUND

$$U = 3351 \text{ ft}^2/\text{ea} \text{ (PIT BAYS)}$$

$$S = 6890 \text{ ft}^2$$

$$M = 1624 \text{ ft}^2$$

$$L = 500 \text{ ft}^2/\text{ea}$$

$$C = 900 \text{ ft}^2$$

$$\text{TOTAL AREA} = 19060 \text{ ft}^2 \text{ (LOWER FLOOR)} + 13022 \text{ ft}^2 \text{ (UPPER FLOOR)} = 32,102$$

$$\text{INTERIOR WALL} = 6890 \text{ ft}^2$$

$$\text{EXTERIOR WALL} = 15450$$

$$\text{BLAST DOOR NO.} = 5$$

PIT BAYS

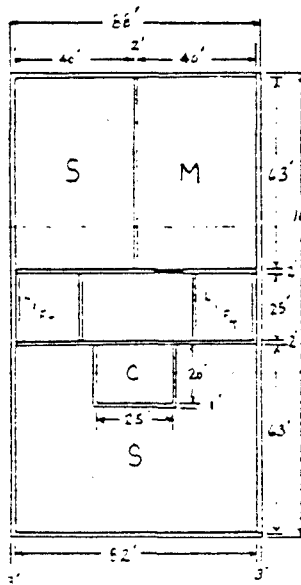
1-58

10-2

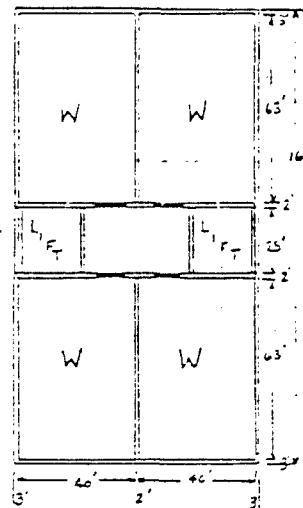
SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO
OF

PROJECT NO.: C2-7092 SPONSOR: CERL
SUBJECT: LAYOUT
BY: Y.W. DATE: 19 CHECKED BY: mtb DATE CHECKED: 19



ABOVE GROUND



BELOW GROUND

W = 2520 ft²/ea
S = 7119 ft²
M = 2520 ft²
L = 500 ft²/ea
C = 300 ft²
TOTAL AREA = 14168 ft²/floor
INTERIOR WALL = 5400 ft²
EXTERIOR WALL = 12450 ft²
BLAST DOOR No = 5

DISCREPANCY
INTERIOR WALL = 5400 (1.5) = 8100
EXTERIOR WALL = 12450 (1.5) = 18675

DISCREPANCY
INTERIOR = 105 (10) = 1050
EXTERIOR = 495 (10) = 4950

DESIGN 10-1
STACKED MUNITIONS

SHEET NO. _____ OF _____

SPONSOR: CERL

CHECKED BY: MM

DATE: _____ 19____

CHECKED BY: MM

DATE CHECKED: _____ 19____

282 224

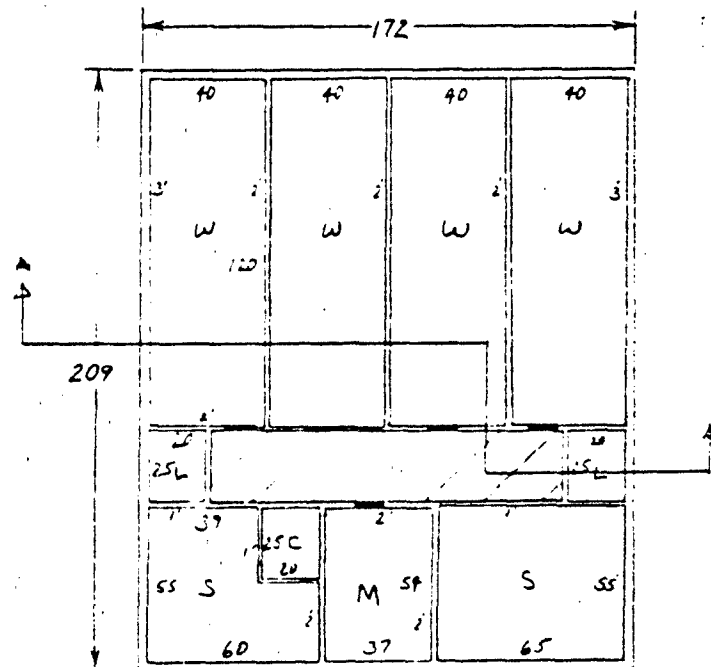
<u>MULTIPLIERS:</u>	<u>FLOOR AREA RANKING:</u>	<u>EXTENDED PERM</u>	<u>INT. AREA</u>
H = 3	30 - 32.5K = 10	8160 - 8680 = 10	5000 - 5500
M = 2	32.5 - 35 = 9	8680 - 9200 = 9	5500 - 6000
L = 1	35 - 37.5 = 8	9200 - 9720 = 8	6000 - 6500
	37.5 - 40 = 7	9720 - 10240 = 7	6500 - 7000
	40 - 42.5 = 6	10240 - 10760 = 6	7000 - 7500
CEILING HT.	42.5 - 45 = 5	10760 - 11280 = 5	7500 - 8000
Ave Wall = 10'	45 - 47.5 = 4	11280 - 11800 = 4	8000 - 8500
PAC-MAN = 15'	47.5 - 50 = 3	11800 - 12320 = 3	8500 - 9000
PITS = 15'	50 - 52.5 = 2	12320 - 12840 = 2	9000 - 9500
	52.5 - 55 = 1	12840 - 13360 = 1	9500 - 10000
	55 → ∞ = 0	13360 → ∞ = 0	10000 → ∞

FINAL SIX LAYOUTS SELECTED FOR CONCEPT EVALUATION

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COMPUTATION SHEET

SHEET NO.
OF

PROJECT NO.: 02-7092 SPONSOR: CERI
SUBJECT: FINAL LAYOUT
BY: MYV DATE: 19 CHECKED BY: WOT DATE CHECKED: 19



Below Ground.

W = 4800 ft²/ea
S = 6329 ft²
M = 1999 ft²
L = 500 ft²/ea
C = 500 ft²
Total Area = 35928 ft²
Perimeter Wall = 7620 ft²
Interior Wall = 6050 ft²
Block Dice No = 5

120 x 40 DAYS

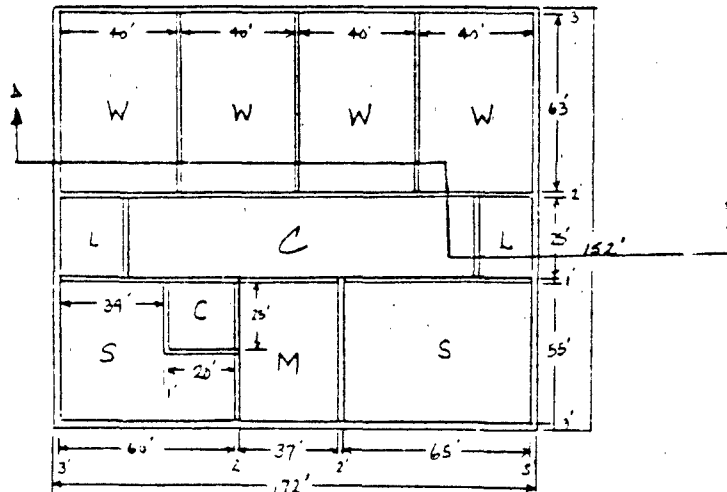
1-62

Design 1-11

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
OF

PROJECT NO.: CD-7092 SPONSOR: CERL
SUBJECT: FINAL LAYOUT
BY: MW DATE: 19 CHECKED BY: mtb DATE CHECKED: 19



W = 2520 ft²/ea

S = 6329 ft²

M = 2035 ft²

L = 500 ft²/ea

C = 500 ft²

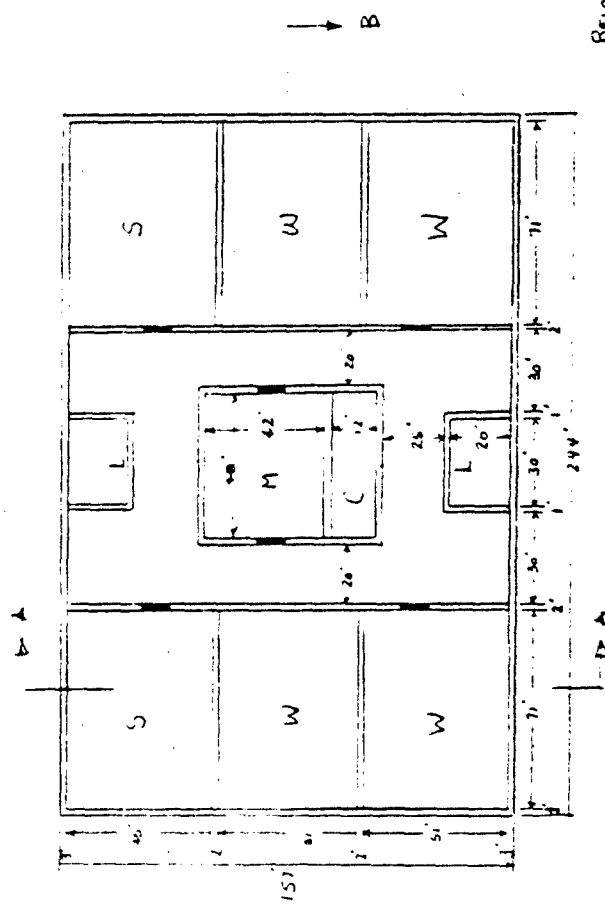
TOTAL AREA = 26144 ft²

INTERIOR WALL = 6705 (15' High)

EXTERIOR WALL = 9720 (15' High)

DESIGN 1-12

STACKED MUNITIONS

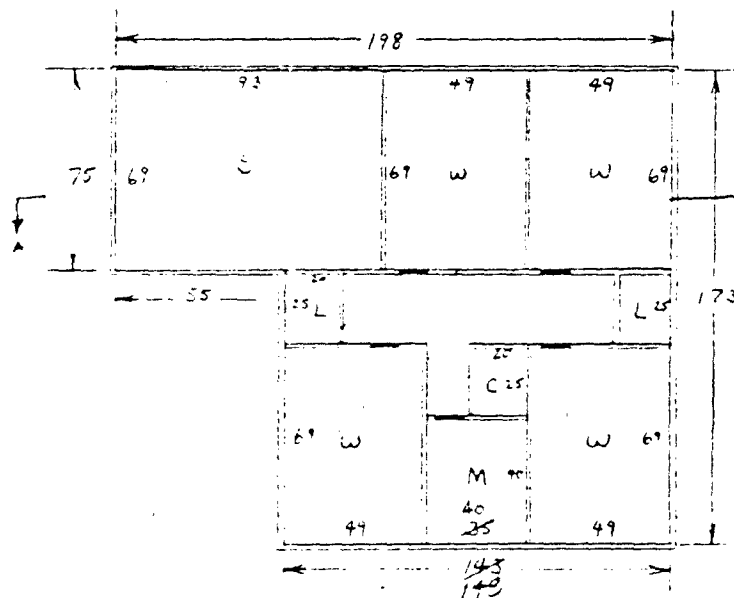


DESIGN No. 2-8
 RANKING 212

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
OF

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: FINAL LAYOUT
BY: WV DATE: 18 CHECKED BY: mt DATE CHECKED: 19



ABOVE GROUND

$W = 3381 \text{ ft}^2$ (FIT BARS)

$S = 6417 \text{ ft}^2$

$M = 1400 \text{ ft}^2$

$L = 500 \text{ ft}^2/\text{ea}$

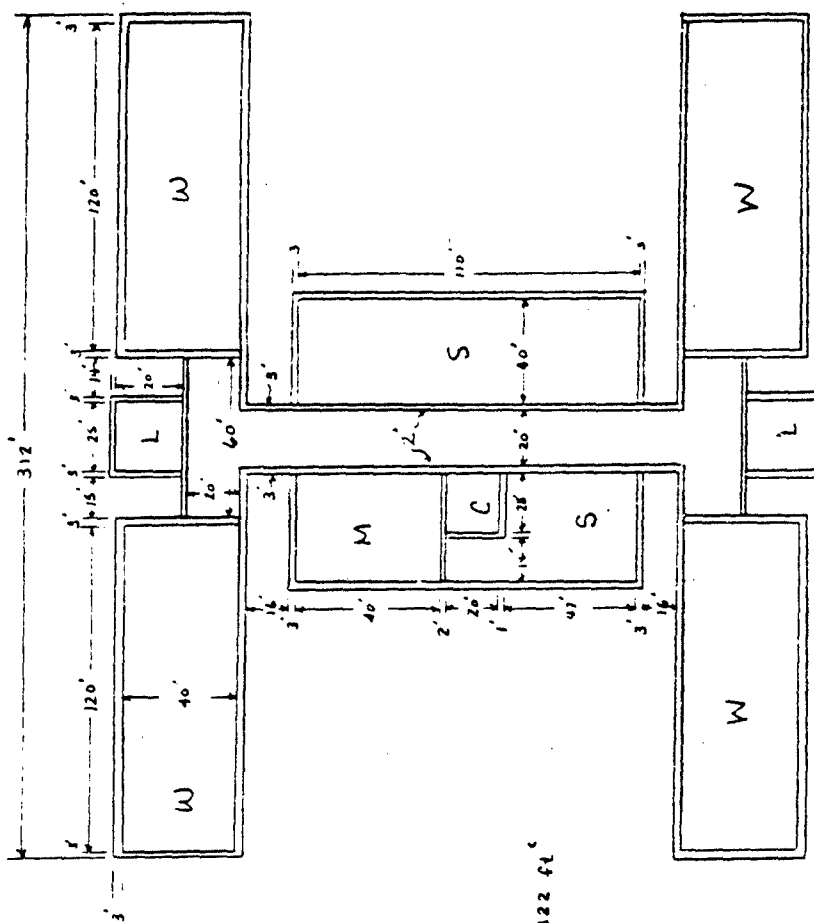
$C = 500 \text{ ft}^2$

TOTAL AREA = 29,864 ft^2

EXTERIOR WALL = 11,130 ft^2

INTERIOR WALL = 7605 ft^2

BLAST DOOR NOT S



$W = 4800 \text{ ft}^2$
 $L = 500 \text{ ft}^2$
 $C = 500 \text{ ft}^2$
 $S = 6560 \text{ ft}^2$
 $M = 1600 \text{ ft}^2$
 Total Area = 41322 ft²
 120 x 40 BAY

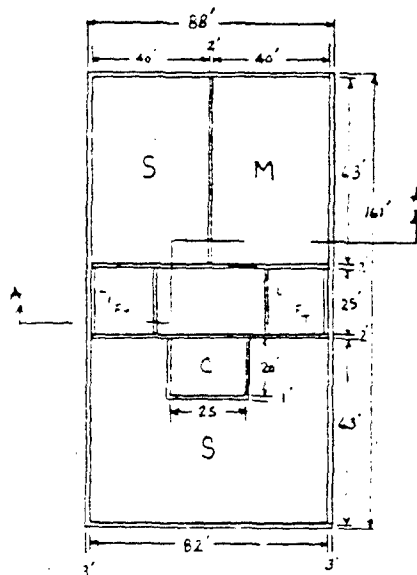
1-66

CONCEPT # 6
 (NO SECTION PROVIDED)
 Design 9-1

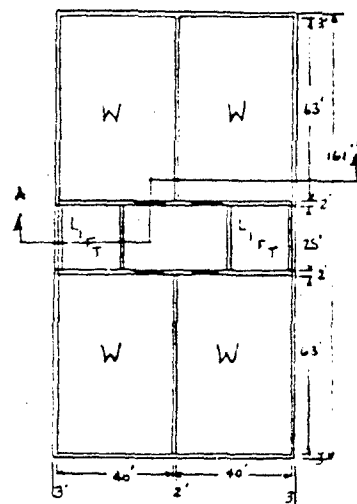
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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
OF

PROJECT NO.: C2-7092 SPONSOR: CERL
SUBJECT: FINAL LAYOUT
BY: MW DATE: 19 CHECKED BY: mtb DATE CHECKED: 19



ABOVE GROUND



BELOW GROUND

W = 2520 ft²/ea
S = 7114 ft²
M = 2520 ft²
L = 500 ft²/ea
C = 500 ft²
TOTAL AREA = 14168 ft²/floor
INTERIOR WALL = 5400 ft²
EXTERIOR WALL = 12450 ft²
BLIND DOOR No = 5

QUOTIENT
INTERIOR WALL = 292 (14) = 4088
EXTERIOR WALL = 499 (15) = 7485
ABOVE GROUND
INTERIOR = 105 (10) = 1050
EXTERIOR = 499 (10) = 4990

DESIGN 10-1
STACKED MUNITIONS

APPENDIX 2

CHEMICAL THREAT

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
OF

PROJECT NO. CA-7092 SPONSOR: CERL
SUBJECT: INDEX - APPENDIX 2
BY: SWI DATE 9-27 19 82 CHECKED BY: _____ DATE CHECKED: _____ 19 _____

APPENDIX 2
CHEMICAL THREAT

	PAGES
1. AGENT DESCRIPTIONS	2-2 to 2-6
2. AIR SUPPLY ARRANGEMENT	2-7 to 2-9
3. CHEMICAL DEFENSE ENSEMBLE	2-10
4. DON/DOFF FACILITY LAYOUT	2-11
5. DON/DOFF PROCEDURES	2-12 to 2-13
6. WIND DYNAMIC PRESSURE	2-14 to 2-15
7. PURGING AIR REQUIREMENTS	2-16 to 2-17
8. PRESSURIZATION REQUIREMENTS	2-18 to 2-20

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
1 OF 19

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: CHEMICAL AGENT THREAT
BY: *gurb* DATE: 8-18-79 82 CHECKED BY: DATE CHECKED: 19

BLISTER AGENTS

Blister agents are primarily chlorinated organic compounds often containing sulphur and nitrogen. Some can be classified as chlorinated amines. Blister agents and nerve agents fall into the same classes as many pesticides and herbicides. Blister agents irritate and/or poison cells in the eyes, lungs and skin. Effects are usually somewhat delayed (several hours). There is very little immediate pain upon contact. The agent causes severe nose and throat irritation, eye damage, and skin blistering and swelling. When death occurs, it is usually the result of infection. Lethal doses are shown in Table 1. Blister agents include mustard (5-6 types), phosgene oxime (CX), and Lewisite (L). Mustard was used during World War I. Many of the later blister agents are odorless. The duration of effectiveness varies from agent to agent.

The concentration at which minor physiological effects begin is generally taken to be 4 ug/cm^2 for mustard agent HD (Reference 1).

1. MIL-STD-282, "Filter Units, Protective Clothing, Gas-Mask Components and Related Products: Performance Test Methods," Department of Defense, 28 May 1956 with change 2 dated 3 December 1974.

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
2 OF 19

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: CHEMICAL AGENT THREAT
BY: mtf DATE: 8-18-79 82 CHECKED BY: DATE CHECKED: 19

NERVE AGENTS

Nerve agents can be referred to as organophosphorus compounds. They are extremely toxic and fast acting, as shown in Table 1. A few tens of milligrams of liquid agent can cause death in less than 15 minutes. Very minute concentrations around the eye will lead to marked miosis. The nerve agents enter the body through the respiratory system, the eyes, or exposed skin. Both liquid and vapor forms ^{will} rapidly penetrate the skin. Nerve agents act by upsetting the balance between the sympathetic and parasympathetic nervous systems, resulting in convulsions, coma, and death. Some examples of nerve agents are tabun (GA), sarin (GB), soman (GD) and agent VX (or VR55). Nerve agents have a relatively long duration of effectiveness (days as opposed to hours). Agents like VX are absorbed by vegetation and can have long-term effects.

The concentration at which minor physiological effects begin is generally taken to be $4 \text{ } \mu\text{g}/\text{cm}^2$ for agent VX (2). For years, the maximum acceptable concentration for GB was set at $1.25 \text{ } \mu\text{g}/\text{cm}^2$ (1), and a similar value was assumed for thickened GD (2). However, tests conducted in 1977 indicate that concentrations of GB as high as 4 to $5 \text{ } \mu\text{g}/\text{cm}^2$ preclude physiological effects, and $10 \text{ } \mu\text{g}/\text{cm}^2$ can be used as the break point for GB if a slight cholinesterase despression can be tolerated (3).

- (2) McGrath, Anna S., et al., "Evaluation of Candidate Agent Protective Materials for Use by United States Air Force Personnel (u)," U.S. Army Chemical Systems Laboratory, ARCSL-SP-79006, ADCO176854, April 1979, Confidential
- (3) Jaynes, Capt. Edgar N., "Report of Active Duty Trainings," DF, DRDAR-CLW-P, through Chief, Individual Protection Branch, and Chief, Physical Protection Division to Director of Personnel and Force Development, U.S. Army Chemical Systems Laboratory, 17 June 1977.

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
3 OF 19

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: CHEMICAL AGENT THREAT
BY: TWW/ DATE: 8-18-82 CHECKED BY: DATE CHECKED: 19

BLOOD AGENTS

Blood agents prevent transfer of oxygen from the blood to the body tissue. Death occurs as a result of interference with the liver, kidneys and lungs. Blood agents include hydrogen cyanide (AC), cyanogen chloride (CK), and arsine (SA). Blood agents must be disseminated as a gas. They are absorbed into the body primarily by breathing, and have a short effectiveness duration. Hydrogen cyanide may cause death within 15 minutes. Arsine, on the other hand, may take anywhere from several hours to several weeks to develop maximum effects. Lethal doses are summarized in Table 1.

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
4 OF 19

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: CHEMICAL AGENT THREAT
BY: [signature] DATE: 8-16-82 CHECKED BY: DATE CHECKED: 10

CHOKING AGENTS

Choking agents can be characterized in general terms as halogenated compounds of a type that readily hydrolyze in the presence of moisture in the respiratory system to form hydrochloric acid, thus irritating and inflaming the nose and throat membranes causing swelling. The lungs fill with fluid. Death occurs from lack of oxygen. Lethal doses are given in Table 1. Examples of such agents are phosgene (CG) and diphosgene (DP). Phosgene was used extensively in World War I. At least 80 percent of the chemical agent fatalities in this war were caused by phosgene. It has a low persistence rating. A delay of three or four hours will typically occur between exposure and the onset of disabling symptoms for the concentrations expected on the battlefield. Most deaths will occur within 24 hours. The choking agents must be disseminated as gases to be effective since they enter the body primarily through the respiratory system.

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO
5 OF 17

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: CHEMICAL AGENT THREAT
BY: WWT DATE: 8-18-82 CHECKED BY: DATE CHECKED: 19

TABLE 1. CHEMICAL AGENT LETHAL DOSES

Agent	Exposure Conditions	Median Lethal Dosage (mg-min/m ³)
<u>Blister Agents</u>		
• HD distilled mustard	Inhalation	1,500
	Skin absorption	10,000
• HN-3 nitrogen mustard	Inhalation	1,500
	Skin absorption	10,000 (est)
• L lewisite	Inhalation	1,200-1,500
	Skin absorption	100,000
<u>Nerve Agents</u>		
• GA tabun	Inhalation (resting)	400
• GB sarin	Inhalation (resting)	100
	(mild activity)	70
• GD soman	Inhalation (resting)	GB-GA range
<u>Blood Agents</u>		
• AC hydrogen cyanide	Inhalation	Wide variation with concentration γ
		2,000 for $\gamma = 200$ mg/m ³
		4,500 for $\gamma = 150$ mg/m ³
• CY cyanogen chloride	Inhalation	11,000
• SA arsine	Inhalation	5,000
<u>Choking Agents</u>		
• CG phosgene	Inhalation	3,200
• DP diphosgene	Inhalation	3,200

Source: TM 3-215, "Military Chemistry and Chemical Agents,"
Departments of the Air Force and Army, Washington, D.C.,
December 1963 with change 1 dated 16 March 1965.

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COMPUTATION SHEET

SHEET NO
6 OF 19

PROJECT NO: 02-7092 SPOI: CE RL
SUBJECT: Air Supply Arrangements
BY: mtb DATE: 7-25-92 CHECKED BY: _____ DATE CHECKED: 19

Air Supply Options

1. Blow through system, no recirculation
2. Recirculation air through NC only.
3. Recirculation air through a particulate filter and A/C
4. Recirculation air through CBR filters and A/C.
5. Parallel banks with detector and auto switching.

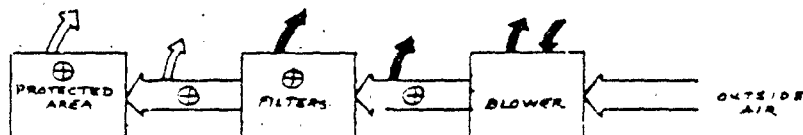
Positive Pressure Systems:

$\uparrow \downarrow$ - leaks

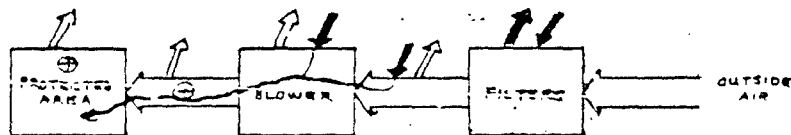
⊕ - highest pressure area

RED - contaminated air.

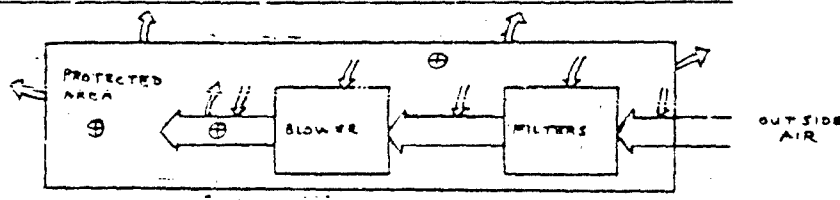
OK
MUST SUIT-UP
FOR FILTER
CHANGEOUT



WILL
NOT
PROTECT



OK
UNTIL
FILTER
CHANGEOUT
OR BLOWER
SHUTDOWN

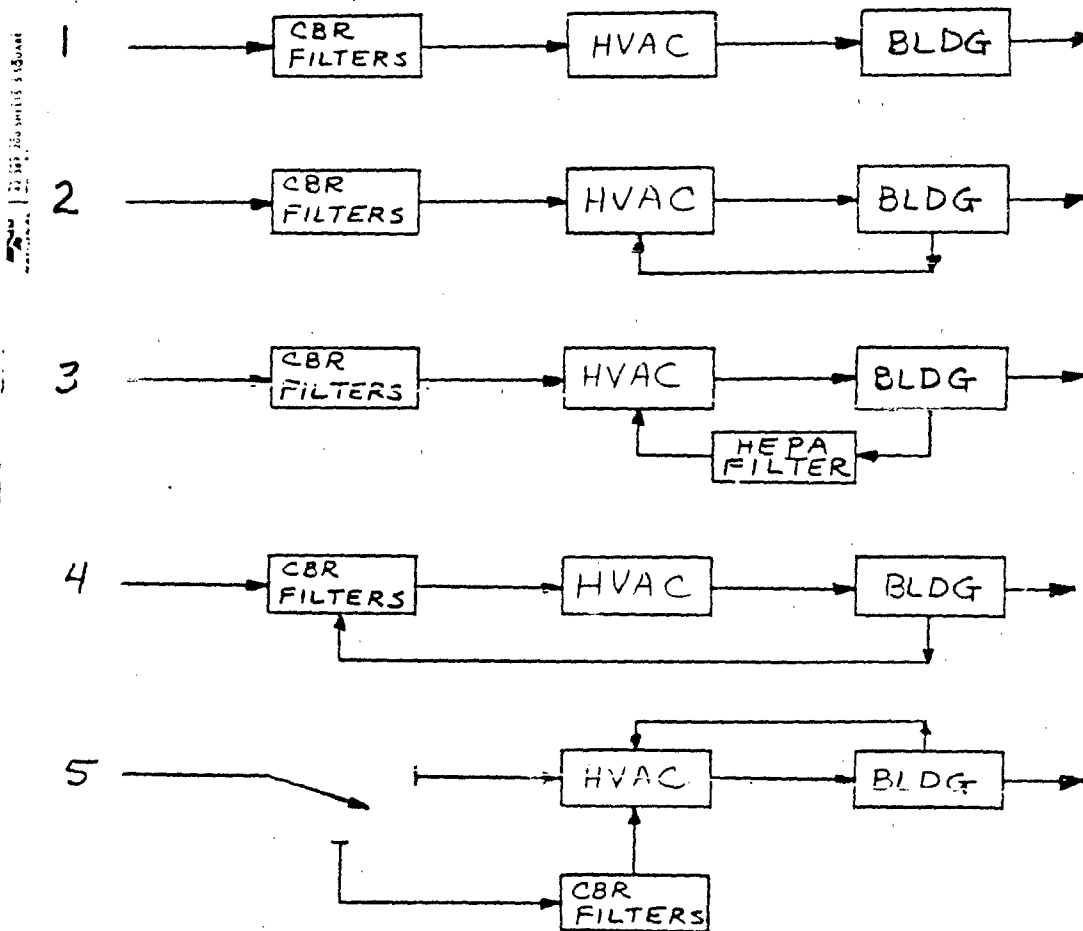


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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
7 OF 19

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: AIR SUPPLY ARRANGEMENTS
BY: ~~WWT~~ DATE: 7-23-1982 CHECKED BY: DATE CHECKED: 19

AIR SUPPLY OPTIONS

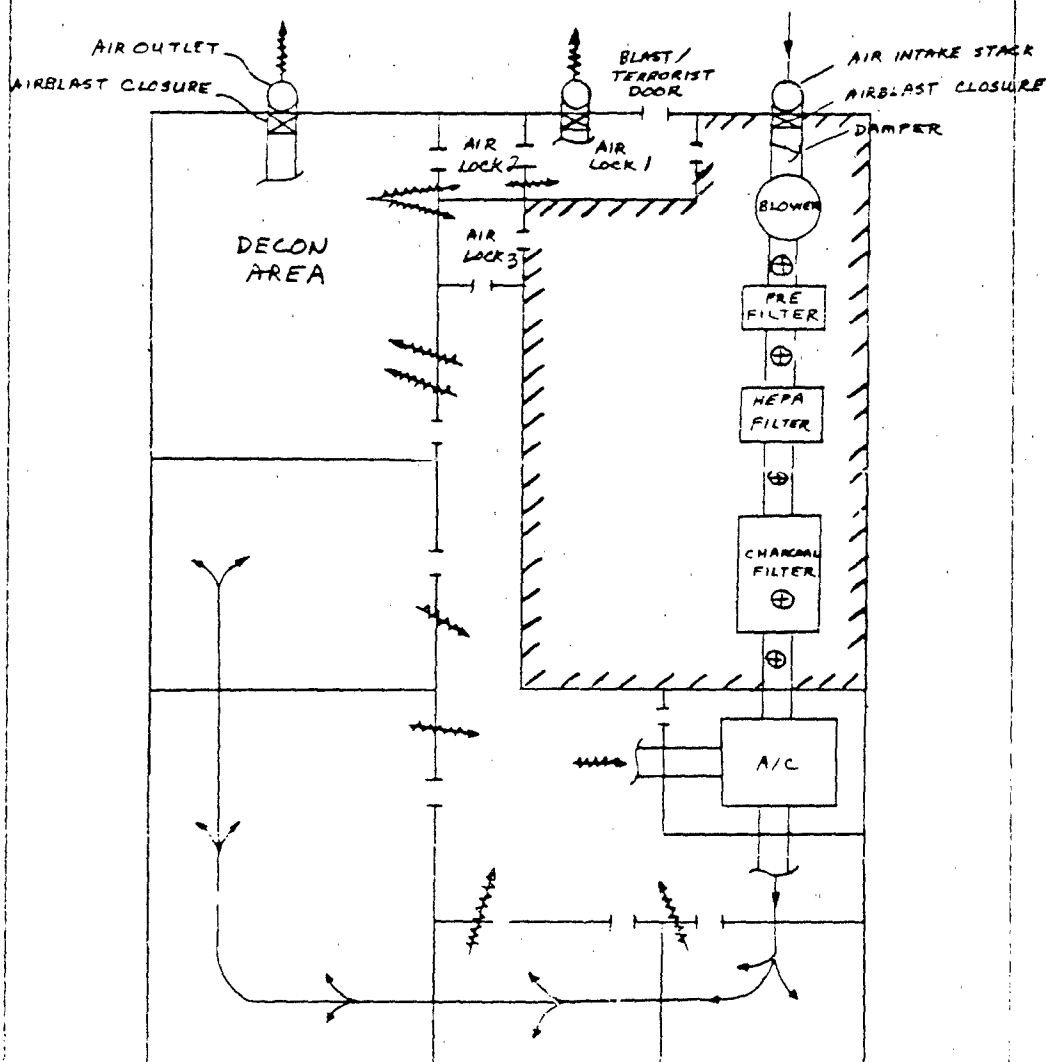


* CBR = CHEMICAL / BIOLOGICAL / RADIOLOGICAL
HVAC = HEATING, VENTILATING AIR CONDITIONING

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
8 OF 19

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: AIR SUPPLY ARRANGEMENTS
BY: mlt DATE: 7-26 1982 CHECKED BY: DATE CHECKED: 19



CLEAN CONDITIONED AIR →
RETURN AIR ~~~~

Not to scale

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
9 OF 19

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: CHEMICAL DEFENSE ENSEMBLE
BY: nwt DATE: 8-17 1982 CHECKED BY: DATE CHECKED: 19

U.S. Army Personal Protection Equipment

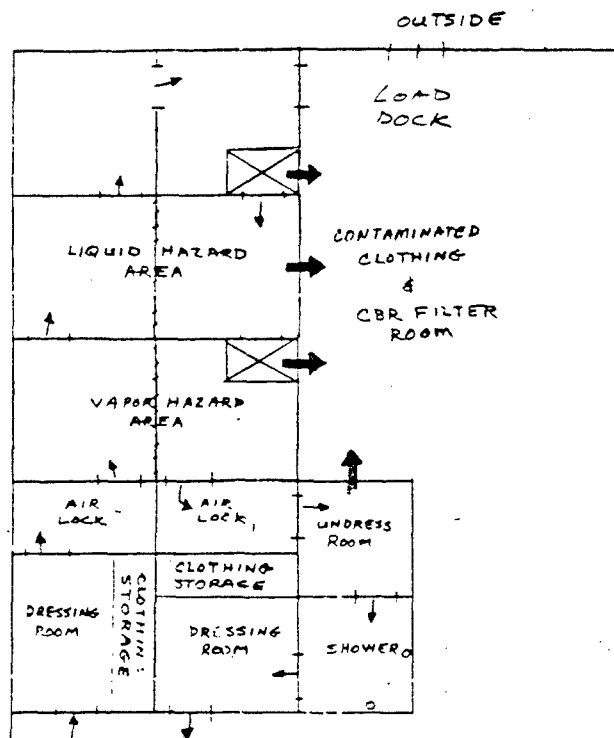
1. Mask, chemical - biological: Field, M17A1 FSN 4240-00-926-4201
Designed to protect face, eyes, & respiratory tract
Self contained drinking system
Voice mitter
2. Filter element, M13A2 FSN 4240-00-165-5226
Charcoal canister
3. Hood, CB Mask: Field, NBC-M16A2 FSN 4240-00-999-0420
Designed to protect head and neck
Butyl coated nylon
4. Suit, chemical protective NSN 8415-00-177-5008
Charcoal impregnated material
5. Footwear covers, chemical protective NSN 8430-01-021-5978
Butyl rubber overshoes
Worn over standard leather combat boot
6. Glove set, chemical protective NSN 8415-00-033-3518
Butyl rubber gloves
25 mil thick
Cotton inserts.

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO. 10 OF 19

PROJECT NO.: 02-292 SPONSOR: CERL
SUBJECT: CONTAMINATION CONTROL AREA
BY: mvb DATE: 8-4-1982 CHECKED BY: _____ DATE CHECKED: _____ 19

DON/DOFF FACILITY



→ ← Scale 2'

$$\begin{aligned} 46' \times 20' &= 920 \\ 16' \times 8' &= 128 \\ \hline &1048 \text{ ft}^2 \end{aligned}$$

$$\begin{aligned} \text{Protected } 24' \times 20' &= 720 \\ 16' \times 8' &= 128 \\ \hline &848 \text{ ft}^2 \end{aligned}$$

↓ PROCESS FLOW
X MOVABLE PARTITIONS

☒ SHUFFLE BOX

→ CLOTHING CHUTE

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
11 of 19

PROJECT NO.: C2-7092 SPONSOR: CERL
SUBJECT: Don/DoFF Procedures
BY: WAB DATE: 8-11 1982 CHECKED BY: _____ DATE CHECKED: 19

TYPICAL DOFF PROCEDURES

DOORWAY TO LIQUID HAZARD AREA:

1. Remove overboots
2. Shuffle feet and hands in Fullers Earth
3. Proceed into Liquid Hazard Area.

INSIDE LIQUID HAZARD AREA:

4. Dust mask with Fullers Earth
5. Remove combat boots and pull on a clean plastic tube sock before touching floor.
6. Remove helmet
7. Remove hood
8. Remove outer ensemble pants and shirt with help from another person. Do not remove gloves or gas mask.
9. Dust hands, feet, mask with Fullers Earth
10. Proceed to doorway between Liquid and Vapor Hazard Areas.

DOORWAY TO VAPOR HAZARD AREA:

11. Have LHA attendant peel off ^{rubber} gloves, and pull a clean tube sock over each hand.
12. Hold breath, close eyes while attendant removes mask and replaces with a clean one.
13. Clear and seal the mask.
14. Proceed to Vapor Hazard Area.

INSIDE VAPOR HAZARD AREA:

15. Replace tube socks with clean ones (hands and feet)
16. Remove charcoal undergarment.
17. Proceed to Airlock entry

DOORWAY TO AIRLOCK:

18. Pass hand and foot plastic tube socks back into VHA as you step into the airlock.
19. Remove cotton glove inserts and pass them back into VHA.
20. Close Airlock door.

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
12 OF 19

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: Don/Doff Procedures
BY: mtb DATE: 8-11 19 82 CHECKED BY: _____ DATE CHECKED: _____ 19 _____

TYPICAL DOFF PROCEDURES (CON'T)

INSIDE AIRLOCK:

21. Set purging air timer for 2 minutes.
22. When timer rings open door to undress room (not the door to the VHA).
23. Enter the Undress Room.

INSIDE THE UNDRESS ROOM:

24. Remove socks
25. Remove underwear
26. Set purging air timer for 2 minutes.
27. When timer rings, remove mask
28. Exit into shower room

INSIDE SHOWER ROOM:

29. Take hot shower with liberal use of soap.
30. Wash hair
31. Dry and step into dressing room.

INSIDE DRESSING ROOM

32. Full body scan with chemical agent detector/alarm.
33. Don clean work clothes.
34. Enter protected shelter area.

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
13 OF 19

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: wind pressures
BY: gwb DATE: 8-2-1982 CHECKED BY: DATE CHECKED: 19

$$\text{Dynamic wind pressure} = p_w = \frac{1}{2} \rho v^2$$

$$\rho_{\text{air}} = \frac{0.001293 H}{76(1+0.00367 t)} \text{ gm/ml} \quad \text{per CRC Hdbk of Physics \& Chemistry 1966-69, pg F-8}$$

H = atmospheric pressure (cm of Hg)
t = temperature (°C)

$$1 \text{ gm/ml} = 62.42621 \text{ lbm/ft}^3 \\ = 0.03612628 \text{ lbm/in}^3 \quad \text{where } 1 \text{ lbm} = \frac{1 \text{ lbf}}{g}$$

$$1 \text{ mile/hr} = 1.466667 \text{ ft/sec} \\ = 17.6 \text{ in/sec}$$

$$g = \text{acceleration due to gravity} = 32.1725 \text{ ft/sec}^2 \quad \text{@ sea level, 45°N lat.} \\ = 386.070 \text{ in/sec}^2$$

$$1 \text{ lbf/in}^2 = 27.68068 \text{ inH}_2\text{O} \quad \text{at } 4^\circ\text{C}$$

25 mph wind @ 4°C, 76.0 cm Hg

$$\rho_{\text{air}} = \frac{(0.001293)(76)}{(76)[1+(0.00367)(4)]} = 0.00127429 \text{ gm/ml} = .00004604 \text{ lbm/in}^3$$

$$g = 386.070 \text{ in/sec}^2$$

$$\rho_{\text{air}} = (.00004604) / (386.070) \quad \left(\frac{\text{lbm}}{\text{in}^3} \right) / \left(\frac{\text{in}}{\text{sec}^2} \right) \\ = 1.1925 \times 10^{-7} \text{ lbf-sec}^2/\text{in}^4$$

$$v = 25 \text{ miles/hr} = 440 \text{ in/sec}$$

$$p_w = \left(\frac{1}{2} \right) (1.1925 \times 10^{-7}) (440)^2 \quad \frac{\text{lbf-sec}^2}{\text{in}^4} \cdot \frac{\text{in}^2}{\text{sec}^2} \\ = 0.011543 \text{ lbf/in}^2 \\ = 0.32 \text{ inH}_2\text{O}$$

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
14 OF 19

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: Wind pressure
BY: *gwb* DATE: 8-2-79 *82* CHECKED BY: DATE CHECKED: 19

25 mph wind @ 21.1°C (70°F), 76.0 cm Hg

$$\rho_{\text{air}} = \frac{(0.001293)(76)}{(76)[1 + (0.00367)(21.1)]} = 0.00128007 \text{ gm/ml} = 4.335 \times 10^{-5} \text{ lbm/in}^3$$

$$g = 386.070 \text{ in/sec}^2$$

$$p_{\text{air}} = (4.335 \times 10^{-5})(386.070) = 1.1229 \times 10^{-7} \text{ lbf-sec}^2/\text{in}^4$$

$$v = 25 \text{ miles/hr} = 440 \text{ in/sec}$$

$$p_w = \left(\frac{1}{2}\right)(1.1229 \times 10^{-7})(440)^2$$

$$= 0.010869 \text{ lbf/in}^2$$

$$\approx 0.30 \text{ in H}_2\text{O}$$

50 mph wind @ 21.1°C (70°F), 76.0 cm Hg

$$p_{\text{air}} = 1.1229 \times 10^{-7} \text{ lbf-sec}^2/\text{in}^4$$

$$v = 50 \text{ miles/hr} = 880 \text{ in/sec}$$

$$p_w = \frac{1}{2}(1.1229 \times 10^{-7})(880)^2$$

$$= 0.043479 \text{ lbf/in}^2$$

$$\approx 1.20 \text{ in H}_2\text{O}$$

Mears, Merton D., "Handbook on Collective Protection," ARCSL-SP-79003, US Army Chemical Systems Laboratory, August 1979

Note: Reference states on pg 12 that

"... it has been arbitrarily established that no area of a stationary protected system that is adjacent to a possibly contaminated area should be below 0.3 in. H₂O overpressure. This is equivalent to the impact pressure of a 25 mph wind on a flat vertical wall normal to the wind direction. The effectiveness of an agent attack diminishes rapidly as the wind approaches 25 mph."

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
15 OF 17

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: PURGING AIR REQUIREMENTS
BY: mwb DATE: 8-6-82 CHECKED BY: DATE CHECKED: 19

DOFF

Reference * calls for a 2 minute stay in the airlock, and a 2 minute stay in the undressing area.

Assume a 1000:1 reduction in concentration is required during the T=2 minute period.

Airlock Volume = 5' x 10' x 10' = 500 ft³ = V

$$\frac{C}{C_0} = e^{-\frac{QT}{V}}$$

$$\frac{1}{1000} = e^{-\frac{Q(2)}{500}}$$

$$.001 = e^{-0.004Q}$$

$$Q = -\frac{\ln .001}{0.004}$$

$$Q = 1727 \text{ cfm.}$$

Undressing Area Volume = 8' x 8' x 10' = 640 ft³ = V

$$\frac{C}{C_0} = e^{-\frac{QT}{V}}$$

$$\frac{1}{1000} = e^{-\frac{Q(2)}{640}}$$

$$.001 = e^{-0.003125Q}$$

$$Q = -\frac{\ln .001}{0.003125}$$

$$Q = 2210 \text{ cfm}$$

* Sears, Col. William J., "An Evaluation of USAF and RAF Aircrew Chemical Defense Shelter Processing Procedures," SAM-TR-81-5 USAF School of Aerospace Medicine, February 1981, F040
2-17

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
16 OF 19

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: AIRFLOW RATE ROOMS
BY: msb DATE: 8-25-79 CHECKED BY: _____ DATE CHECKED: _____ 19__

Reference *, pg 136 indicates
20 room air changes per hour
is the accepted standard for conventional clean rooms

$$\begin{aligned} \therefore \text{Flow rate} &= 20 V \text{ ft}^3/\text{hr} \\ &= \frac{20V}{60} \text{ ft}^3/\text{min} \\ &= \frac{V}{3} \text{ ft}^3/\text{min} \quad \text{where } V \text{ is the room volume.} \end{aligned}$$

If this criteria were applied to the 848 ft² floor space of the contamination control area, the resulting flow rate required (assuming a 10 ft ceiling height) is found as follows:

$$V = (848)(10) = 8480 \text{ ft}^3$$

$$Q = \frac{8480}{3} = 2827 \text{ ft}^3/\text{min}$$

* Austin, Philip R., Design and Operation of Clean Rooms, Business News Publishing Co., Detroit, MI, 1970.

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
17 OF 19

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: SHELTER PRESSURIZATION
BY: mlt DATE: 8-25-82 CHECKED BY: DATE CHECKED: 19

Assume 10' ceiling height all parts of contamination control area.
Assumed pressurization requirements:

	inches H ₂ O	Volume (ft ³)
• liquid hazard area	0.3	10 X 20 X 10 = 2000
• vapor hazard area	0.4	10 X 20 X 10 = 2000
• 2 airlocks, undress room, shower	0.5	5 X 20 X 10 + 16 X 8 X 10 = 2280
• 2 dressing rooms, 2 clothing storage areas	0.6	11 X 20 X 10 = 2200

Compute required "static" mass of air needed in each area to produce the desired pressures.

Methodology:

Equation of state for P-V-T behavior of gases at low density

$$PV = mRT$$

$$m = \frac{PV}{RT} \quad \text{where } m = \text{total mass of the enclosed air (lbm)}$$

P = pressure in enclosed area (lb_f/ft²)

V = total enclosed volume (ft³)

T = air temperature (°R)

R = Universal gas constant in mass units
 $R_u = 53.34 \text{ (ft}^3 \cdot \text{lb}_f \text{) / (lbm} \cdot \text{°R)}$

Useful conversion factors

$$g = 32.174 \text{ ft/sec}^2$$

$$\text{lb}_f = (\text{lbm})(g) = 32.174 \text{ lbm} \cdot \text{ft/sec}^2$$

$$\text{°R} = \text{°F} + 459.67$$

$$1 \text{ in H}_2\text{O (39.2°F)} = 5.20218 \text{ lb}_f/\text{ft}^2 = 0.03612628 \text{ lb}_f/\text{in}^2$$

$$1 \text{ in H}_2\text{O (60°F)} = 5.1982676 \text{ lb}_f/\text{ft}^2 = 0.03609908 \text{ lb}_f/\text{in}^2$$

$$1 \text{ in H}_2\text{O (78°F)} = 5.1948819 \text{ lb}_f/\text{ft}^2 \text{ (linear extrapolation)}$$

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
15 OF 19

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: SHELTER PRESSURIZATION
BY: nwt DATE: 8-25-79 82 CHECKED BY: DATE CHECKED: 19

Liquid Hazard Area

$$\begin{aligned} P &= 0.3 \text{ in H}_2\text{O} \\ &= (0.3)(5.1948819) \text{ lb}_f/\text{ft}^2 \text{ at } 78^\circ\text{F} \\ &= 1.5584646 \text{ lb}_f/\text{ft}^2 \text{ at } 78^\circ\text{F} \end{aligned}$$

$$V = 2000 \text{ ft}^3$$

$$R = 53.34 \text{ (ft-lbf/lbm-}^\circ\text{R)}$$

$$\begin{aligned} T &= 78^\circ\text{F} \\ &= 78 + 459.67 \text{ }^\circ\text{R} \\ &= 537.67^\circ\text{R} \end{aligned}$$

$$m = \frac{PV}{RT} = \frac{(1.5584646)(2000)}{(53.34)(537.67)} = 0.1087 \text{ lbm}$$

Vapor Hazard Area

$$\begin{aligned} P &= 0.4 \text{ in H}_2\text{O} \\ &= (0.4)(5.1948819) \text{ lb}_f/\text{ft}^2 \text{ at } 78^\circ\text{F} \\ &= 2.0779528 \text{ lb}_f/\text{ft}^2 \text{ at } 78^\circ\text{F} \end{aligned}$$

$$V = 2000 \text{ ft}^3$$

$$R = 53.34 \text{ (ft-lbf/lbm-}^\circ\text{R)}$$

$$\begin{aligned} T &= 78^\circ\text{F} \\ &= 537.67^\circ\text{R} \end{aligned}$$

$$m = \frac{PV}{RT} = \frac{(2.0779528)(2000)}{(53.34)(537.67)} = 0.1449 \text{ lbm}$$

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
17 OF 17

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: SHELTER PRESSURIZATION
BY: mtc DATE: 8-25-79 CHECKED BY: _____ DATE CHECKED: 19

2 airlocks + undress room + shower

$$\begin{aligned} P &= 0.5 \text{ in H}_2\text{O} \\ &= (0.5)(5.1948819) \text{ lbf/ft}^2 \text{ at } 78^\circ\text{F} \\ &= 2.597441 \text{ lbf/ft}^2 \text{ at } 78^\circ\text{F} \end{aligned}$$

$$V = 2280 \text{ ft}^3$$

$$R = 53.34 \text{ (ft} \cdot \text{lbf / lbm} \cdot \text{R}^\circ)$$

$$\begin{aligned} T &= 78^\circ\text{F} \\ &= 537.67^\circ\text{R} \end{aligned}$$

$$m = \frac{PV}{RT} = \frac{(2.597441)(2280)}{(53.34)(537.67)} = 0.2065 \text{ lbm}$$

APPENDIX 3

ARCHITECT ENGINEER NARRATIVE

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SOUTHWEST RESEARCH INSTITUTE
SAN ANTONIO, TEXAS

MUNITIONS STORAGE CONCEPTS FOR USE IN FLAT TERRAIN
FACILITY CONCEPTS

INTRODUCTION

This facility concepts study has been prepared under a subcontract agreement with Southwest Research Institute in conjunction with their project to develop munitions storage concepts for use in flat terrain for the Department of the Army, Construction Engineering Research Laboratory at Champaign, Illinois.

The diagrams, descriptive narratives and other information contained in this report present six conceptual solutions for a weapons storage facility in flat terrain based on criteria and guidelines provided by Southwest Research Institute, including blast wall design data and chemical hazard control measures. Close coordination between the Southwest Research Institute and Bernard Johnson Incorporated project teams has been maintained to evaluate all conceptual solutions and ideas for feasibility of construction as well as for functional and security requirements.

This report presents the six conceptual facility plans, as coordinated with the Southwest Research project team. It includes, in addition to conceptual facility diagrams, narrative descriptions of proposed building systems, budgetary estimates of probable construction costs and a comparative analysis of the six concepts.

It is intended that this conceptual study be preliminary to and establish a basis for a definitive architectural/engineering design program.

The facility study focuses primarily on requirements established by Southwest Research Institute during initial investigation of the criteria and project requirements. Basic provisions of all concepts respond to requirements for storage or warehousing of munitions, handling of the munitions within the storage facility, security of the storage facility from accidental or intended damage or destruction and personnel safety.

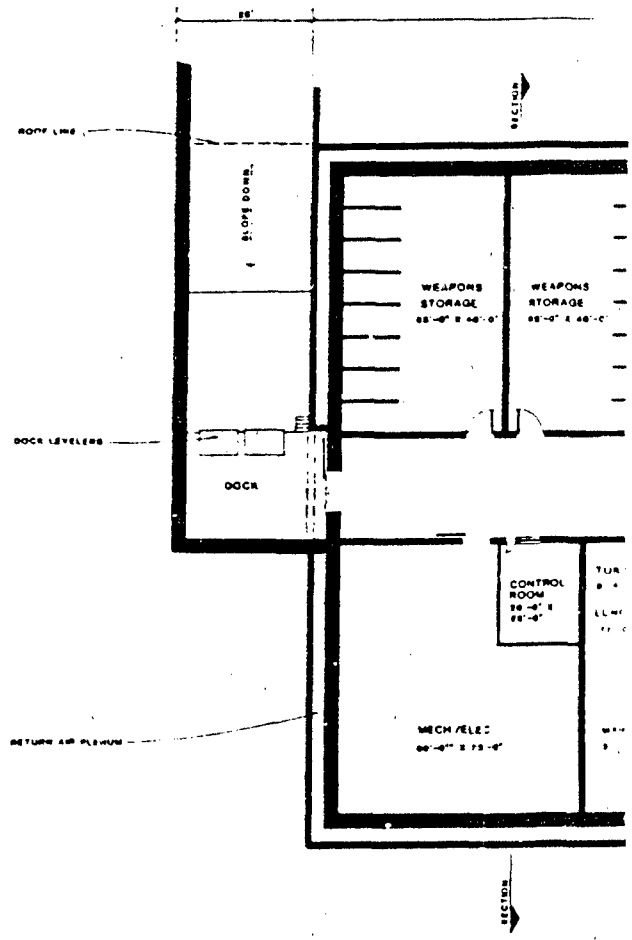
Each concept was developed around four munitions storage bays or rooms, a munitions maintenance/service room, control room, load out docks and contamination control area. Appropriate provisions were then made for circulation, support and utilities. Backup utility systems are included in all concepts to permit continuation of operations should the primary system be lost or fail.

There are no site related considerations included in this study since a definitive site was not identified. Building site preparation and foundation provisions were based on readily available information and previous experiences in designing foundations for similar facilities in the Houston-Galveston, Texas vicinity. It should be recognized that subsurface soil conditions may vary widely within a given area.

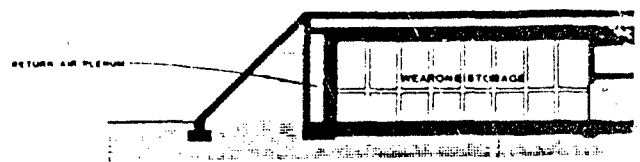
Estimates of probable construction costs are based on readily available unit costing factors for the type of facility described and adjusted with assumed contingency factors based on current experience in Houston, Texas. Escalation is not included in estimates since the schedule for starting construction is unknown.

Both objectives and subjective factors were utilized in a comparative analysis of six concepts to determine the most appropriate concepts for further consideration.

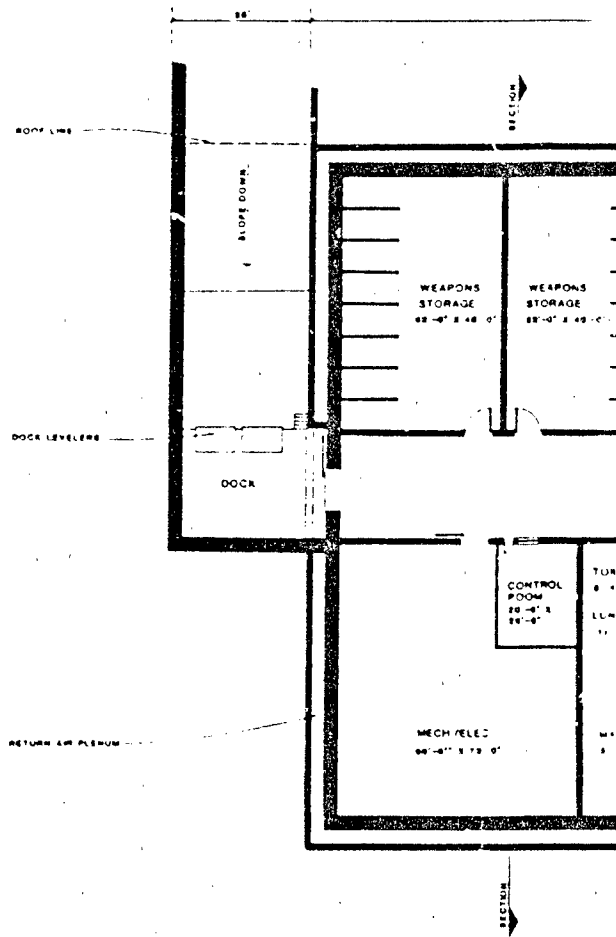
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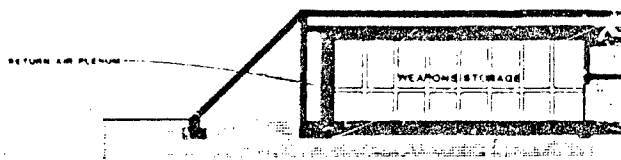
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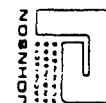
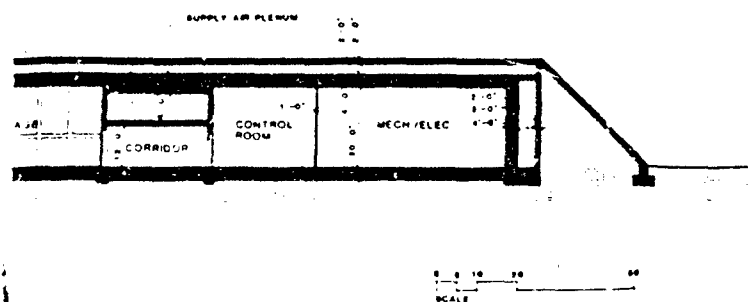
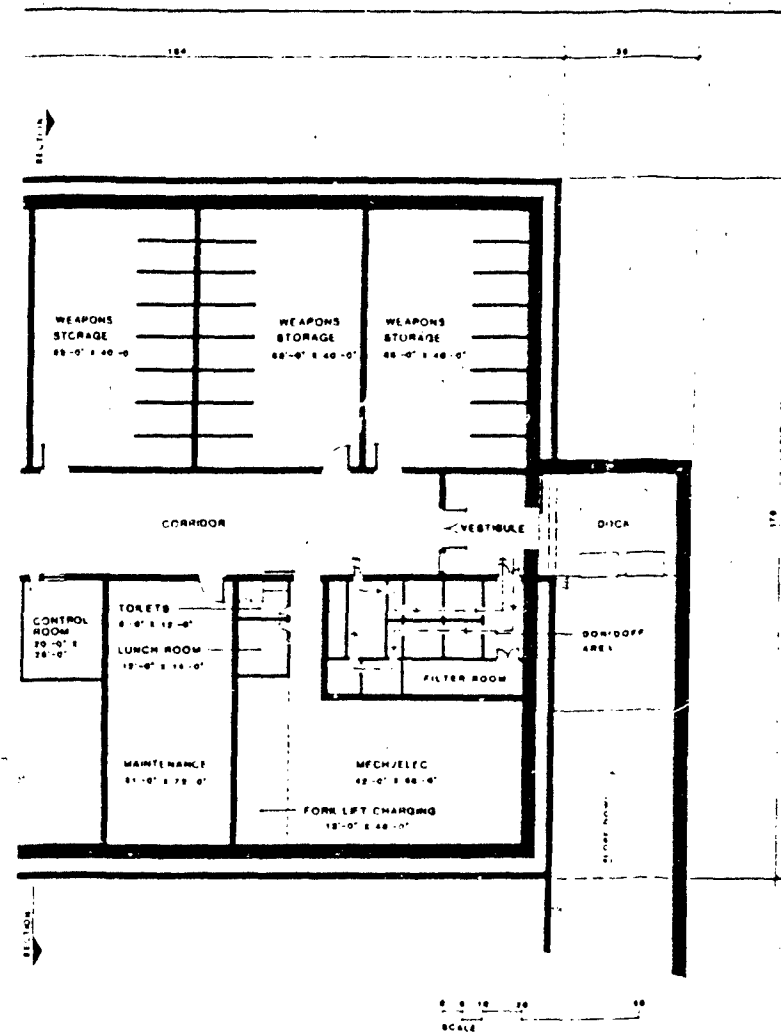
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PLAN
CONCEPT NO. 1



SECTION
CONCEPT NO. 1



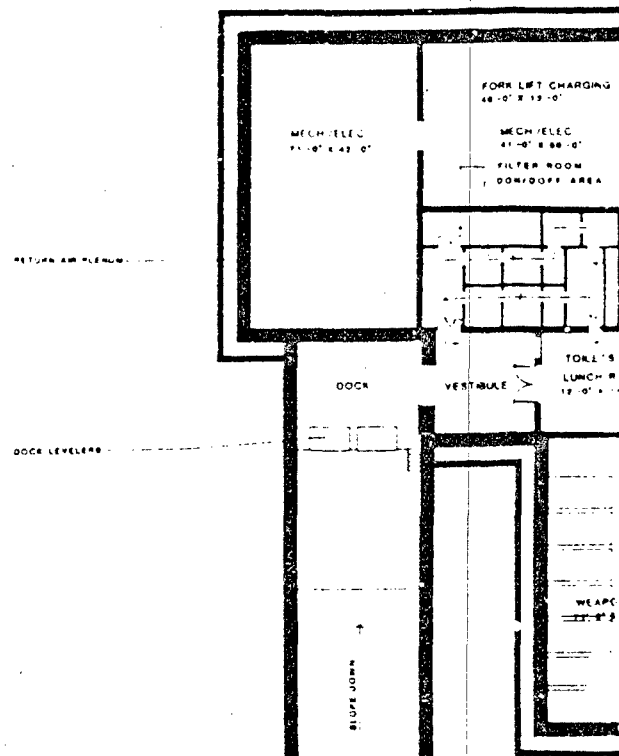
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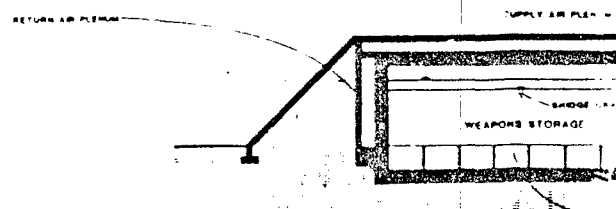
AMMUNITION STORAGE FACILITY
CONCEPT NUMBER 1
PLAN AND SECTION

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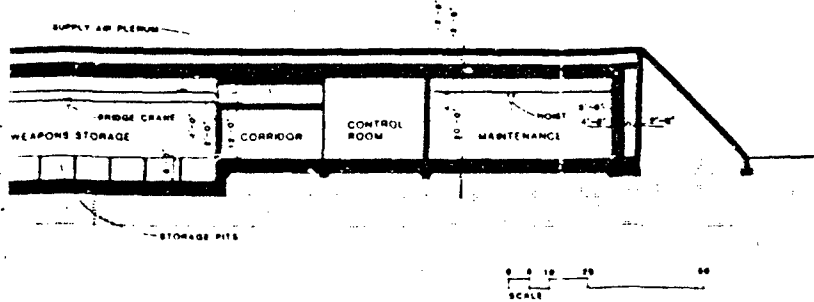
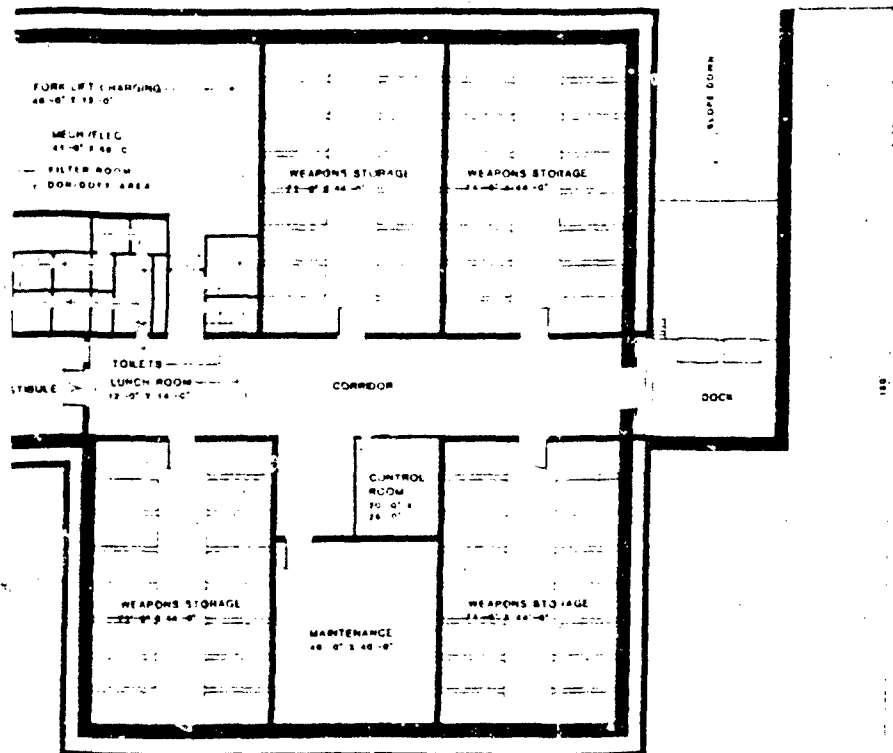
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PLAN ABOVE GRADE
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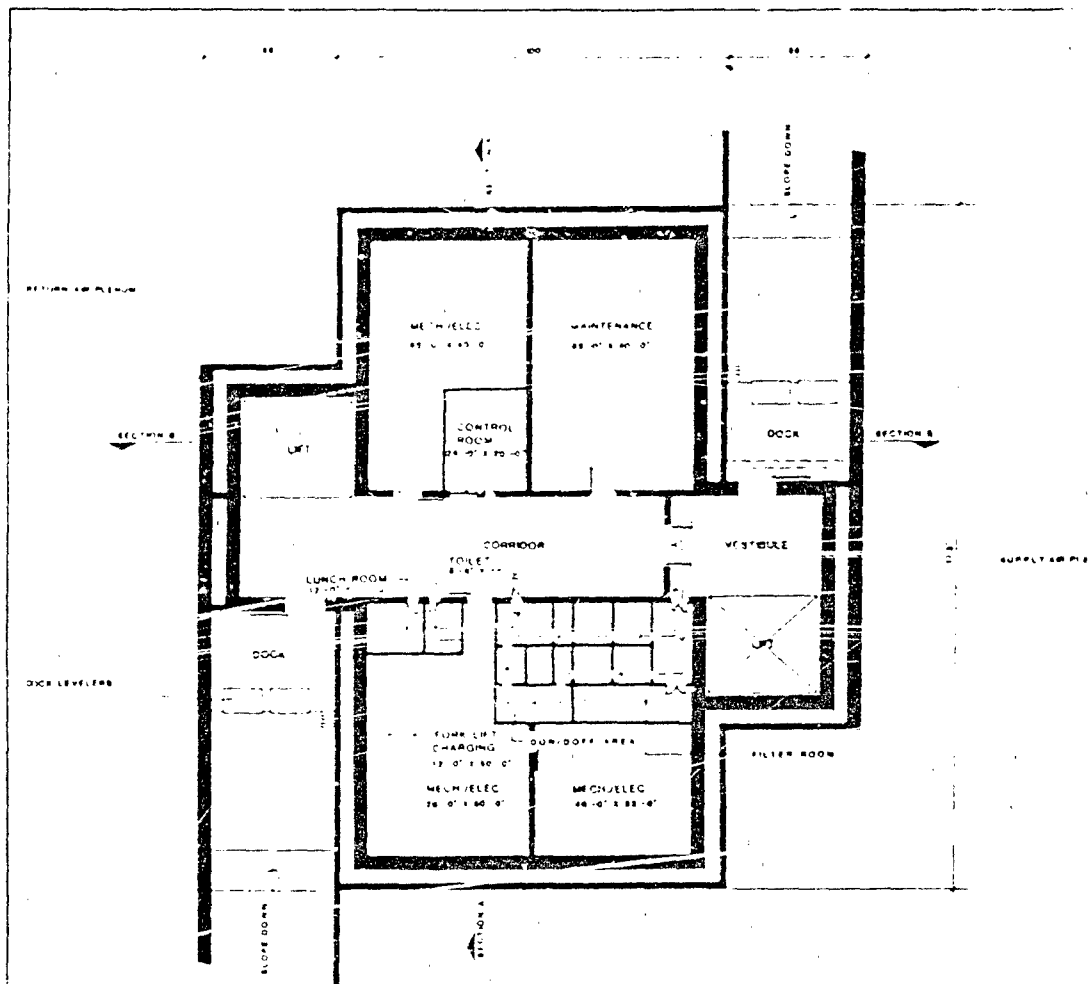


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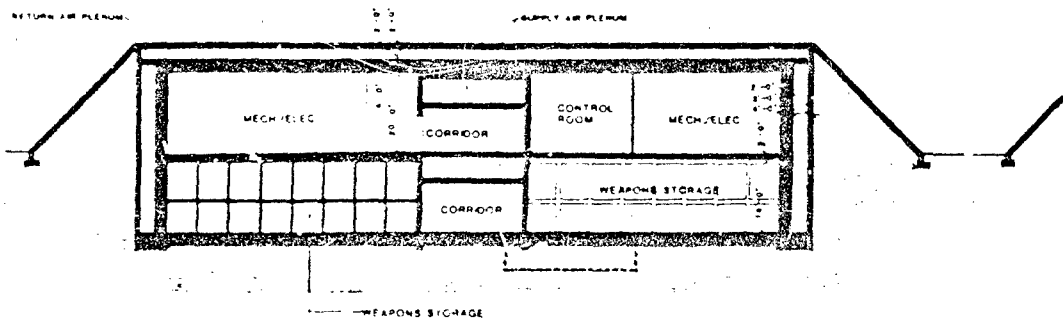
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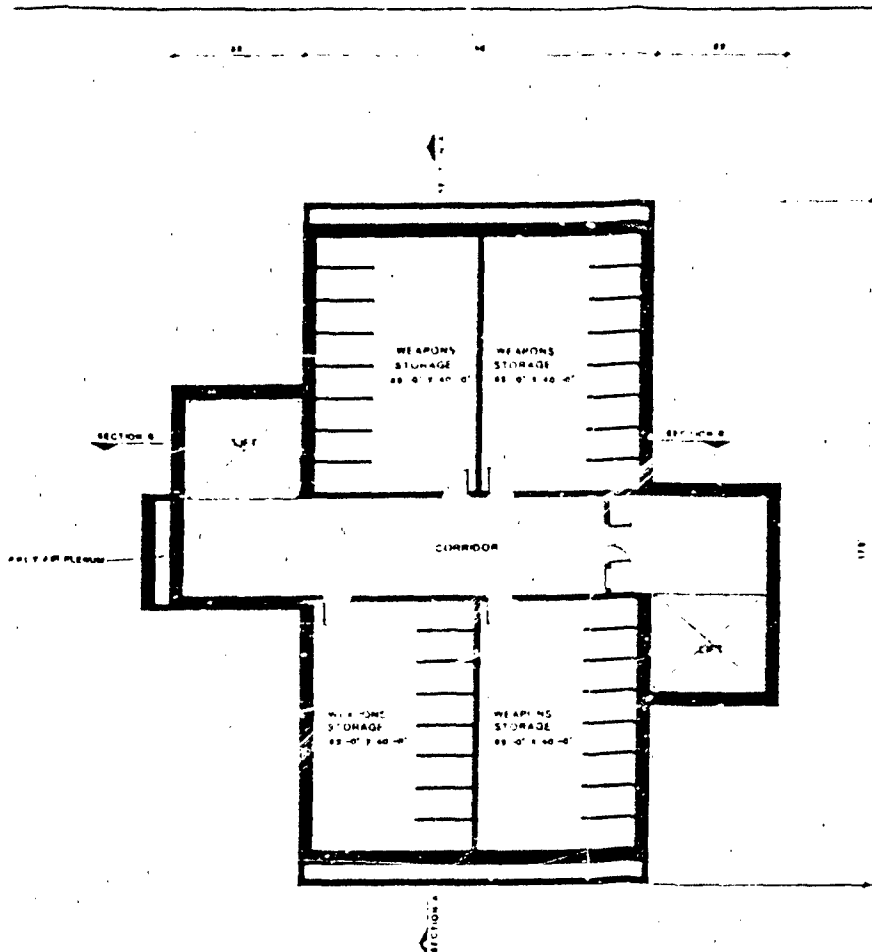
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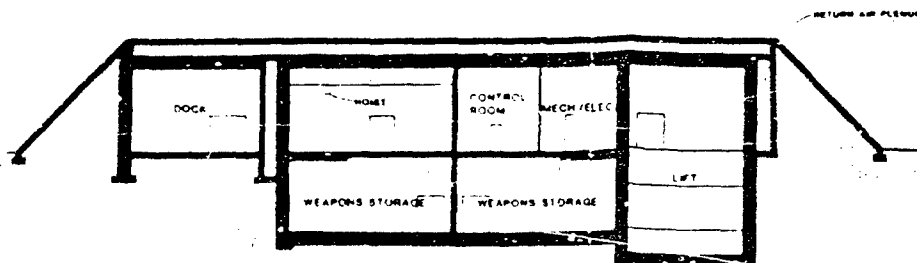
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PLAN BELOW GRADE
CONCEPT NO. 3

0 10 20 30
SCALE



SECTION B
CONCEPT NO. 3

0 10 20 30
SCALE



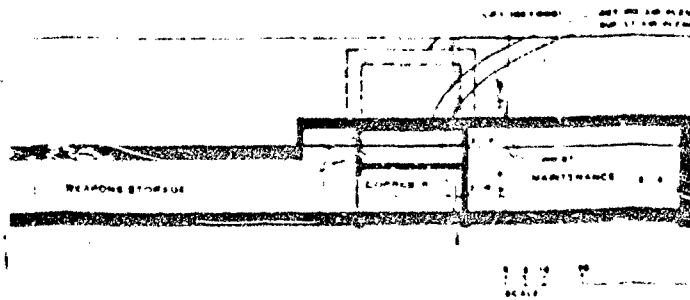
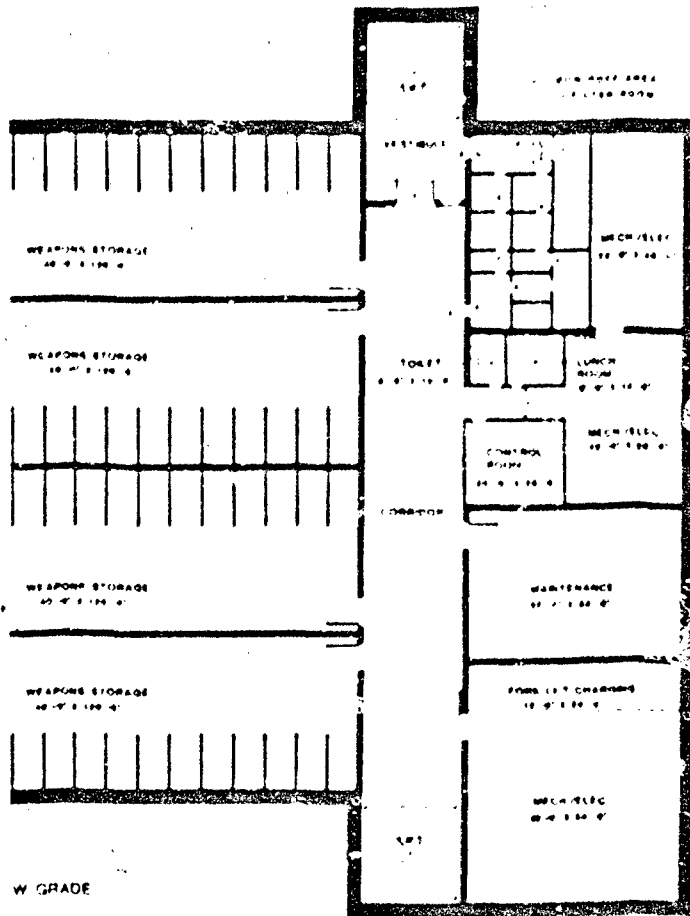
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AMMUNITION STORAGE FACILITY
CONCEPT NUMBER 3

PLAN AND SECTION

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A3



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AMMUNITION STORAGE FACILITY
CONCEPT NUMBER 4

PLAN AND SECTION

A4

DOCK LEVELING

DOCK #1

DOCK #2

CORRIDOR

MAINTENANCE

CONTROL ROOM

CORRIDOR

DOCK #3

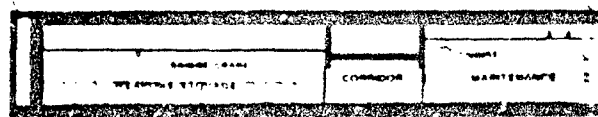
RETURN AIR PLUMB

DOCK #4

DOCK #5

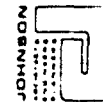
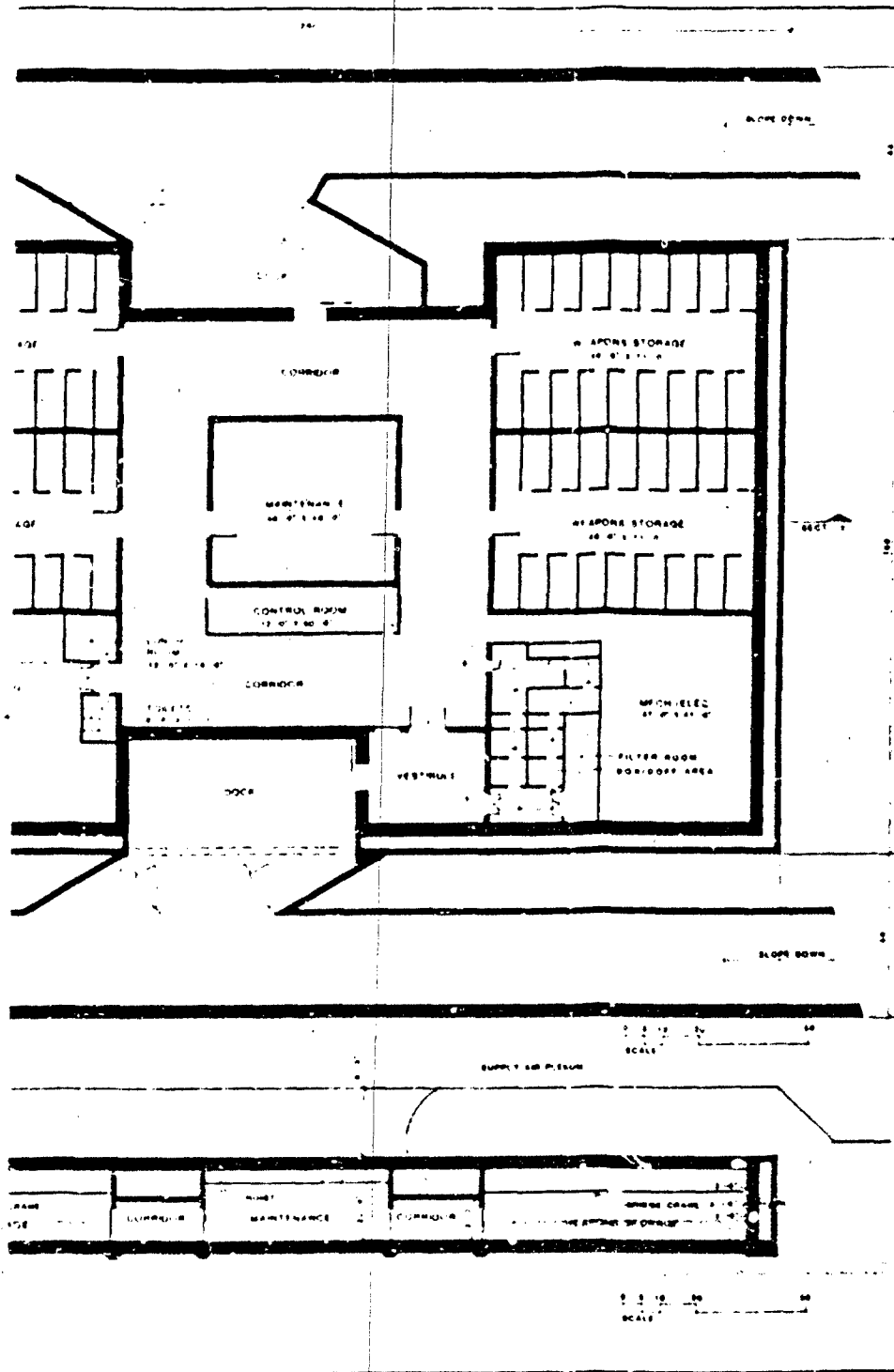
PLAN BELOW GRADE
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RETURN AIR PLUMB



SECTION
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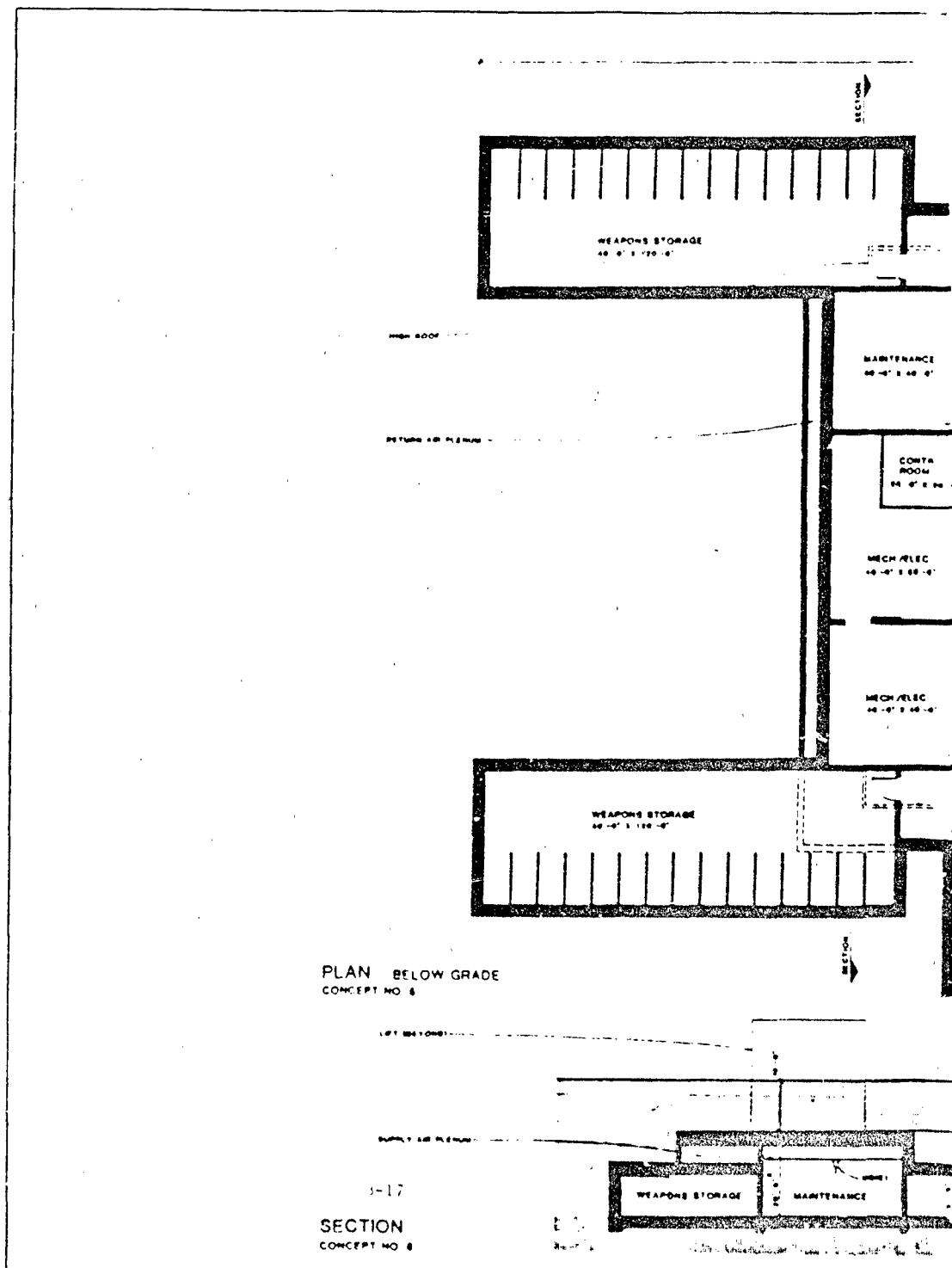
1-1

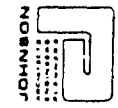
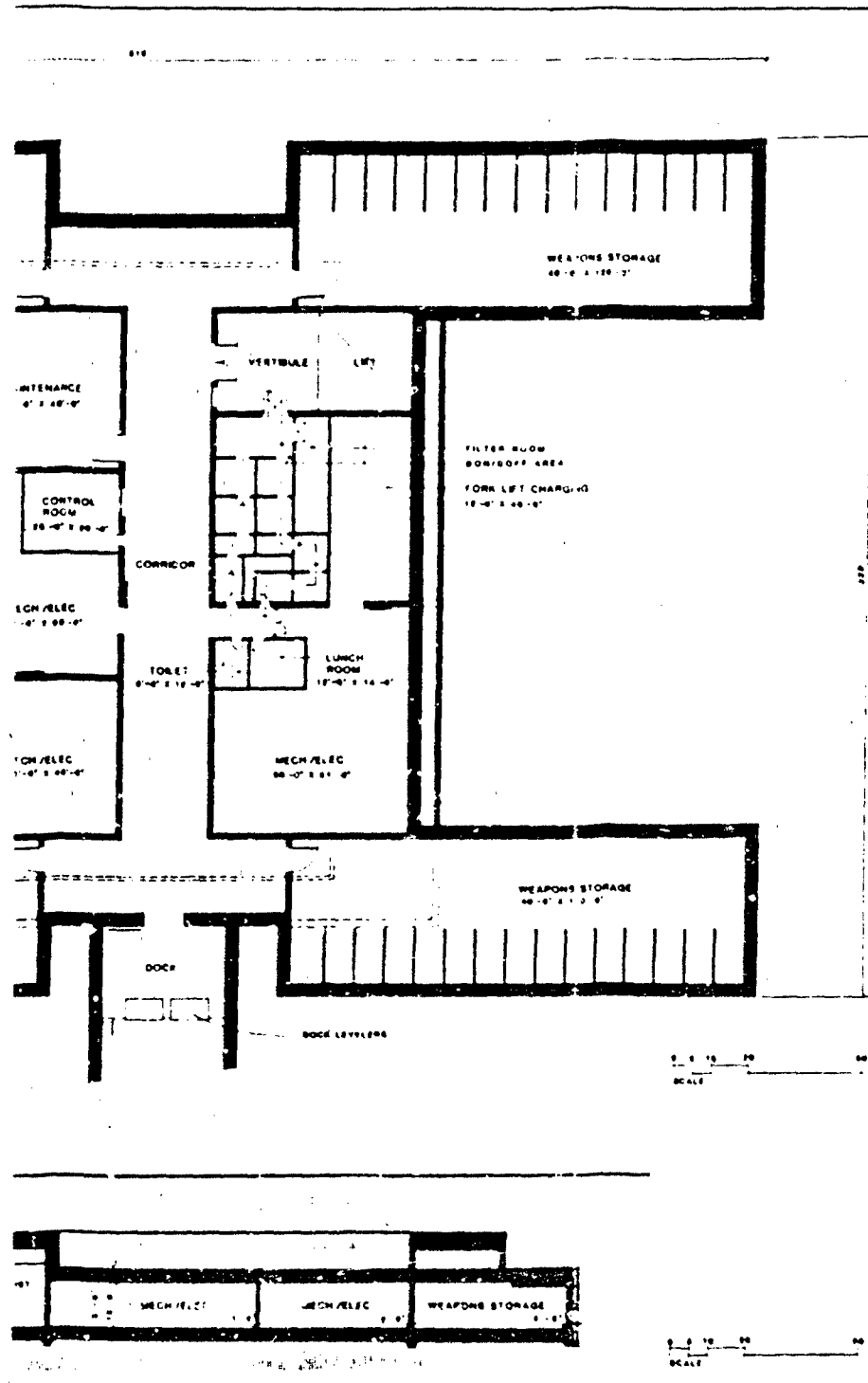


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JOHNSON
ARCHITECTS
ENGINEERS
PLANNERS
1000 WESTINGHOUSE BLVD., SUITE 1100, TEXAS 77060

BERNARD JOHNSON INCORPORATED
ARCHITECTS ENGINEERS PLANNERS

DATE	DESCRIPTION

AMMUNITION STORAGE FACILITY
CONCEPT NUMBER 6
PLAN AND SECTION

DATE	
BY	
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A6

SECTION II

ARCHITECTURAL NARRATIVE

PART 1 - GENERAL

The six architectural concepts presented in this description and in the drawings in Section One were developed from criteria supplied by Southwest Research Institute and respond to requirements identified for function, security and safety. Although a preliminary analysis of applicable building, life safety and fire codes was conducted, compliance with these code requirements has been waived by direction from the principal investigator for Southwest Research Institute in favor of more rigid requirements for security and functional capability.

PART 2 - CONCEPTS

2.1 BASIC REQUIREMENTS

- A. Each of the six (6) concepts developed included the basic requirements for weapons storage bays or rooms, a weapons maintenance room, a control room, mechanical/electrical room(s), loading dock(s), and air-lock or vestibule, a contamination control (change) area and circulation corridor.
- B. Basic material handling requirements for all concepts include a two-ton capacity overhead bridge crane in the weapons maintenance room and a battery powered fork lift truck for general handling and transport of weapons units within the facility. Dock levellers are required at each loading dock position.

2.2 CONCEPT NO. 1

- A. Space Requirements: Concept No. 1 consists of a total gross area of 36,357 SF, with a net area of 18,050 SF. Functional areas within the facility are: Weapons Storage Rooms (four at 2,520 SF each); Control Room (500 SF); Toilets (102 SF); Lunch Room (168 SF); Maintenance Area (2,108 SF); Contamination Control Area (1,500 SF); Fork-lift Charging Area (624 SF); Mechanical/Electrical Areas (two at a total of 5,768 SF); Vestibule (625 SF); Corridor (3,525 SF); and two Covered Docking Areas (3,069 SF total).
- B. Materials Systems
 - 1. Walls: The structure consists of specially reinforced waterproofed concrete walls, designed to withstand and contain an accidental explosion with no structural damage to the facility. Explosions occurring in

the Weapons Storage Rooms or Maintenance Area will be contained within that respective space.

2. Roof: The roof will be reinforced concrete with an elastic waterproofing membrane and protection board.
3. Floors: Floors in all areas will be hardened, reinforced concrete.
4. Ceilings: Ceilings in all areas will be reinforced concrete.
5. Doors and Windows: A wire-glass glazed window in a hollow-metal frame will be located between the Control Room and Corridor. All Weapons Storage Rooms and the Maintenance Area will have 7' X 9' blast doors. Personnel doors will be 3' X 7' hollow metal doors. Areas requiring forklift access will have sliding metal doors, except at the Contamination Control Area, where swinging metal doors will be used.

Exterior doors will be sliding metal doors, resistant to blast and forcible entry.

6. Finishes: All interior concrete walls and ceilings shall be painted with alkyd enamel paint and primer.

All interior concrete floors shall be exposed hardened concrete.

Metal doors and frames shall be painted with enamel paint over factory primer.

Industrial machinery, equipment and piping shall be painted with linseed alkyd-resin paint and primer.

2.3 CONCEPT NO. 2

- A. Space Requirements: Concept No. 2 consists of a total gross area of 39,200 SF, with a net area of 20,095 SF. Functional areas within the facility are: Weapons Storage Rooms (four at 3,124 SF each); Control Room (500 SF); Toilets (102 SF); Lunch Room (168 SF); Maintenance Area (1,800 SF); Contamination Control Area (1,500 SF); Forklift Charging Area (624 SF); Mechanical/Electrical Rooms (2 at a total of 5,513 SF); Vestibule (624 SF); Corridor (4,038 SF); and two covered Docking Areas (3,007 SF total).

B. Materials Systems

1. Walls: The structure consists of specially reinforced, waterproofed concrete walls, designed to withstand and contain an accidental explosion with no structural damage to the facility. Explosions occurring in the Weapons Storage Rooms or Maintenance Area will be contained within the respective space.
2. Roof: The roof will be reinforced concrete with an elastic waterproofing membrane and protection board.
3. Floors: Floors in all areas will be hardened, reinforced concrete.
4. Ceilings: Ceilings in all areas will be reinforced concrete.
5. Doors and Windows: A wire-glass glazed window in a hollow-metal frame will be located between the Control Room and Corridor. All weapons Storage Rooms and the Maintenance Area will have 7' X 9' blast doors. Personnel doors shall be 3' X 7' hollow metal doors. Areas requiring forklift access shall have sliding metal doors, except at the Contamination Control Area, where swinging metal doors will be used.

Exterior doors will be sliding metal doors, resistant to blast and forcible entry.

6. Finishes: All interior concrete walls and ceilings shall be painted with alkyd enamel paint and primer.

All interior concrete floors shall be exposed, hardened concrete.

Metal doors and frames shall be painted with enamel paint over factory primer.

Industrial machinery, equipment and piping shall be painted with linseed alkyd-resin paint and primer.

7. Handling Equipment: Two-ton bridge cranes are also required in each of weapons storage rooms.

2.4 CONCEPT NO. 3

- A. Space Requirements: Concept No. 3 consists of a total gross area of 43,775 SF, with a net area of 17,538 SF. Functional areas within the facility are: Weapons Storage Rooms (four at 2,520 SF each); Control Room (500 SF); Toilets (102 SF); Lunch Room (168 SF); Maintenance Area

(2,520 SF); Contamination Control Area (1,500 SF); Forklift Charging Area (600 SF); Mechanical/Electrical Areas (three at a total of 4,710 SF); two Vestibules (2,000 SF total); Corridor (5,500 SF); Lifts (two at 725 SF each); and two Covered Docking Areas (2,170 SF total).

B. Materials Systems

1. Walls: The structure consists of specially reinforced, waterproofed concrete walls, designed to withstand and contain an accidental explosion with no structural damage to the facility. Explosions occurring in the Weapons Storage Rooms or Maintenance Area will be contained within that respective space.
2. Roof: The roof will be reinforced concrete with an elastic waterproofing membrane and protection board.
3. Floors: Floors in all areas will be hardened, reinforced concrete.
4. Ceilings: Ceilings in all areas will be reinforced concrete.
5. Doors and Windows: A wire-glass glazed window in a hollow-metal frame will be located between the Control Room and Corridor. All Weapons Storage Rooms and the Maintenance Area will have 7' X 9' blast doors. Personnel doors shall be 3' X 7' hollow metal doors. Areas requiring forklift access shall have sliding metal doors, except at the Contamination Control Area, where swinging metal doors will be used.

Exterior doors will be sliding metal doors, resistant to blast and forcible entry.

6. Finishes: All interior concrete walls and ceilings shall be painted with alkyd enamel paint and primer.

All interior concrete floors shall be exposed, hardened concrete.

Metal doors and frames shall be painted with enamel paint over factory primer.

Industrial machinery, equipment and piping shall be painted with linseed alkyd-resin paint and primer.

2.5 CONCEPT NO. 4

- A. Space Requirements: Concept No. 4 consists of a total gross area of 48,100 SF, with a net area of 27,436 SF.

Functional areas within the facility are: Weapons Storage Rooms (four at 3,266 SF each); Control Room (600 SF); Toilets (102 SF); Lunch Room (168 SF); Maintenance Area (2,208 SF); Contamination Control Area (1,500 SF); Forklift Charging Area (728 SF); Mechanical/Electrical Areas (two at a total of 4,661 SF); Vestibule (750 SF); Corridor (8,150 SF); and two Covered Docking Areas (9,170 SF total). Bridge cranes will be required in the four Weapons Storage Rooms and a 2-ton hoist shall be required in the Maintenance Area.

B. Materials Systems

1. Walls: The structure consists of specially reinforced, waterproofed concrete walls, designed to withstand and contain an accidental explosion with no structural damage to the facility. Explosions occurring in the Weapons Storage Rooms or Maintenance Area will be contained within that respective space.
2. Roof: The roof will be reinforced concrete with an elastic waterproofing membrane and protection board.
3. Floors: Floors in all areas will be hardened, reinforced concrete.
4. Ceilings: Ceilings in all areas will be reinforced concrete.
5. Doors and Windows: Wire-glass glazed windows in a hollow metal frame will be located between the Control Room and Corridor. All Weapons Storage Rooms and the Maintenance Area will have 7' X 9' blast doors. Personnel doors shall be 3' X 7' hollow metal doors. Areas requiring forklift access shall be sliding metal doors, except at the Contamination Control Area, where swinging metal doors shall be used.

Exterior doors will be sliding metal doors, resistant to blast and forcible entry.

6. Finishes: All interior concrete walls and ceilings shall be painted with alkyd enamel paint and primer.

All interior concrete floors shall be exposed, hardened concrete.

Metal doors and frames shall be painted with enamel paint over factory primer.

Industrial machinery, equipment and piping shall be painted with linseed alkyd-resin paint and primer.

2.6 CONCEPT NO. 5

A. Space Requirements: Concept No. 5 consists of a total gross area of 44,020 SF, with a net area of 25,814 SF. Functional areas within the facility are: Weapons Storage Rooms (four at 4,800 SF each); Control Room (500 SF); Toilets (102 SF); Lunch Room (168 SF); Maintenance Area (1,998 SF); Contamination Control Area (1,500 SF); Forklift Charging Area (648 SF); Mechanical/Electrical Areas (three at a total of 5,060 SF); Vestibule (500 SF); Corridor (3,978 SF); two Covered Docking Areas (1,800 SF total); and two lifts (625 SF each).

B. Materials Systems

1. Walls: The structure consists of specially reinforced, waterproofed concrete walls, designed to withstand and contain an accidental explosion with no structural damage to the facility. Explosions occurring in the Weapons Storage Rooms or Maintenance Area will be contained within that respective space.
2. Roof: The roof will be reinforced concrete with an elastic waterproofing membrane and protection board.
2. Floors: Floors in all areas will be hardened, reinforced concrete.
4. Ceilings: Ceilings in all areas will be reinforced concrete.
5. Doors and Windows: A wire-glass glazed window in a hollow-metal frame will be located between the Control Room and Corridor. All Weapons Storage Rooms and the Maintenance Area will have 7' X 9' blast doors. Personnel doors shall be 3' X 7' hollow metal doors. Areas requiring forklift access shall have sliding metal doors, except at the Contamination Control Area, where swinging metal doors shall be used.

Exterior doors shall be sliding metal doors, resistant to blast and forcible entry.

6. Finishes: All interior concrete walls and ceilings shall be painted with alkyd enamel paint and primer.

All interior concrete floors shall be exposed, hardened concrete.

Metal doors and frames shall be painted with enamel paint over factory primer.

Industrial machinery, equipment and piping shall be painted with linseed alkyd-resin paint and primer.

2.7 CONCEPT NO. 6

A. Space Requirements: Concept No. 6 consists of a total gross area of 49,137 SF, with a net area of 25,404 SF. Functional areas within the facility are: Weapons Storage Rooms (four at 4,800 SF each); Control Room (500 SF); Toilets (102 SF); Lunch Room (168 SF); Maintenance Area (1,600 SF); Contamination Control Area (1,500 SF); Forklift Charging Area (576 SF); Mechanical/Electrical Areas (three at a total of 3,763 SF); Vestibule (650 SF); Corridor (6,221 SF); Lift (645 SF); and two Covered Docking Areas (1,860 SF total).

B. Materials Systems

1. Walls: The structure consists of specially reinforced, waterproofed concrete walls, designed to withstand and contain an accidental explosion with no structural damage to the facility. Explosions occurring in the Weapons Storage Rooms or Maintenance Area will be contained within that respective space.
2. Roof: The roof will be reinforced concrete with an elastic waterproofing membrane and protection board.
3. Floors: Floors in all areas will be hardened, reinforced concrete.
4. Ceilings: Ceilings in all areas will be reinforced concrete.
5. Doors and Windows: A wire-glass glazed window in a hollow-metal frame will be located between the Control Room and Corridor. All Weapons Storage Rooms and the Maintenance Area will have 7' X 9' blast doors. Personnel doors shall be 3' X 7' hollow metal doors. Areas requiring forklift access shall have sliding metal doors, except at the Contamination Control Area, where swing metal doors shall be used.

Exterior doors shall be sliding metal doors, resistant to blast and forcible entry.

6. Finishes: All interior concrete walls and ceilings shall be painted with alkyd enamel paint and primer.

All interior concrete floors shall be exposed, hardened concrete.

Metal doors and frames shall be painted with enamel paint over factory primer.

Industrial machinery, equipment and piping shall be painted with linseed alkyd-resin paint and primer.

STRUCTURAL NARRATIVE

PART 1 - GENERAL

1.1 BASIC CHARACTERISTICS

Of the six concepts presented, there are four variations in location of the structures relative to the ground surface, as follows:

- A. Concepts No. 1 and No. 2 are single story buildings with the finish floor located approximately at grade level. The perimeter walls are surrounded by an earthen berm covered with concrete. A shallow concrete covered earth fill covers the building roof.
- B. Concept No. 3 is a two story building with the upper floor level approximately at grade and the basement level fully below grade (partially buried structure). Perimeter walls and roof are covered in the same manner as concepts No. 1 and No. 2.
- C. Concepts No. 4 and No. 5 are single story buildings with the roof level approximately at grade and covered by a deep earthen berm.
- D. Concept No. 6 is a single story building with the roof level located beneath deep earth fill.

1.2 SIMILAR STRUCTURAL SYSTEMS

- A. The above concepts have construction similarities to other structures not uncommon in the Houston-Galveston area. Below grade sanitary structures, below grade parking garages, basement levels of multi-story buildings and, of course, the abandoned Coast Artillery gun emplacement at Fort Crockett in Galveston are examples of structures with similar concepts.

PART 2 - SUPERSTRUCTURE

2.1 BASIC REQUIREMENTS

- A. Because the buildings are designed to prevent forced entry, resist shock loading either from external sources or internal accidents and are to be either earth covered or buried, the construction will be of reinforced concrete.

2.2 ROOFS

- A. Roofs will be of heavy reinforced concrete construction designed to support the earth cover and to resist the

shock loading. It is anticipated that the roof thickness will range from 3 ft. - 6 in. to 6 ft. depending on the particular span; the roof slab will be reinforced with two layers of steel bars in each direction, and in addition, be reinforced with vertical stirrups for the shear induced by the shock loading.

2.3 WALLS

- A. Perimeter wall will be comprised of an interior wall approximately 4 ft. thick, a 3 ft. plenum space and a 2 ft. exterior wall. The exception is concept No. 4 in which both the supply and return plenum space are located above the ceiling of the corridor and part of the weapons storage bays.
- B. The portions of interior perimeter wall at the weapons storage bays and maintenance bays will be of laced reinforced concrete construction; these walls could be subjected to the overpressures from an H. E. accident. The remaining portions of the perimeter wall will not require lacing.
- C. The exterior perimeter wall and the dock and ramp walls will be of conventional (not laced) reinforced concrete construction.
- D. The interior load bearing walls at the weapons storage and maintenance bays will be of laced reinforced concrete construction; other interior walls will be conventional (not laced) reinforced concrete construction.

2.4 FLOORS

- A. Concept No. 3, which is two story, will have a laced reinforced concrete floor at the upper level; this floor is the ceiling of the weapons storage bays.

PART 3 - FOUNDATIONS

3.1 GEOTECHNICAL ASSUMPTIONS

- A. Specific recommendations regarding foundation design would come as a result of a geotechnical investigation based on sampling and testing of soil materials at the site or sites selected.
- B. Surface and subsurface materials in the Houston-Galveston area are predominately clay although the material in the immediate vicinity of the coast becomes predominately sandy. The strength of these materials varies throughout the area; for example, bearing capacities for footings founded at a ten foot depth can range from 2,500 to 5,000 pounds per square foot or more. If the surface

materials are very poor, they can be stabilized with hydrated lime in order to provide a better subgrade for slabs at grade and pavements. Subsurface materials can be improved with lime injection.

3.2 CONCEPT NO. 1 AND NO. 2 (AT GRADE STRUCTURES)

- A. Floor Slab: Since the perimeter walls and many of the interior walls are subject to shock loading, a heavy mat type reinforced concrete floor slab will be used to transfer the loads and stresses from these walls. The mat will be approximately equal in thickness to the perimeter wall and will be reinforced with two layers of reinforcing steel in each direction.
- B. Mat type floor slabs in the coastal vicinities may require piling if the sandy materials do not provide adequate support for a slab on grade.
- C. The slabs at the loading dock and ramps will be 6 to 8 in. thick slab on grade.
- D. Both perimeter walls and interior walls are subject to extremely high axial loading imposed by the roof and the weight of the wall itself. These loads will be resisted by piling, either augered or driven, depending on the specific geotechnical recommendations.

3.3 CONCEPT NO. 3 THROUGH 5 (PARTIALLY OR FULLY BURIED STRUCTURES)

- A. Floor Slab (basement floor only for Concept No. 3): In addition to transferring shock loading (and backfill loading at the perimeter) from the walls, the mat slab also resists hydrostatic uplift pressures. These uplift forces are counteracted by the dead weight of the structure plus the fill.
- B. The vertical forces acting through walls will be distributed to the rigid mat slab through bending action at sites where the subsurface material has adequate bearing capacity. At sites where the subsurface material is weak, piling will be used beneath the walls as required to aid the mat in resisting the vertical loads. At perimeter walls, the skin friction of the adjacent soil will also aid in resisting the vertical loads.

3.4 BUCYANCY

- A. The concepts presented have adequate weight to resist the hydrostatic uplift forces with an appropriate safety factor. A final buoyancy check will be based on information from a geotechnical investigation and the final design.

3.5 WATERPROOFING

- A. The structures will have exterior waterproofing of base slabs, walls and roofs to prevent moisture from permeating the concrete.

HEATING, VENTILATING AND
AIR CONDITIONING NARRATIVE

PART 1 - GENERAL

1.1 BASIC CRITERIA

- A. ASHRAE Manuals: "Equipment", "Systems", "Fundamentals" and "Applications".
- B. NFPA 90A.
- C. Standard Building Code.

1.2 DESIGN CONDITIONS

A. Outdoor Temperatures:

Summer: 95°FDB, 80°FWB
Winter: 28°FDB

B. Indoor Temperatures:

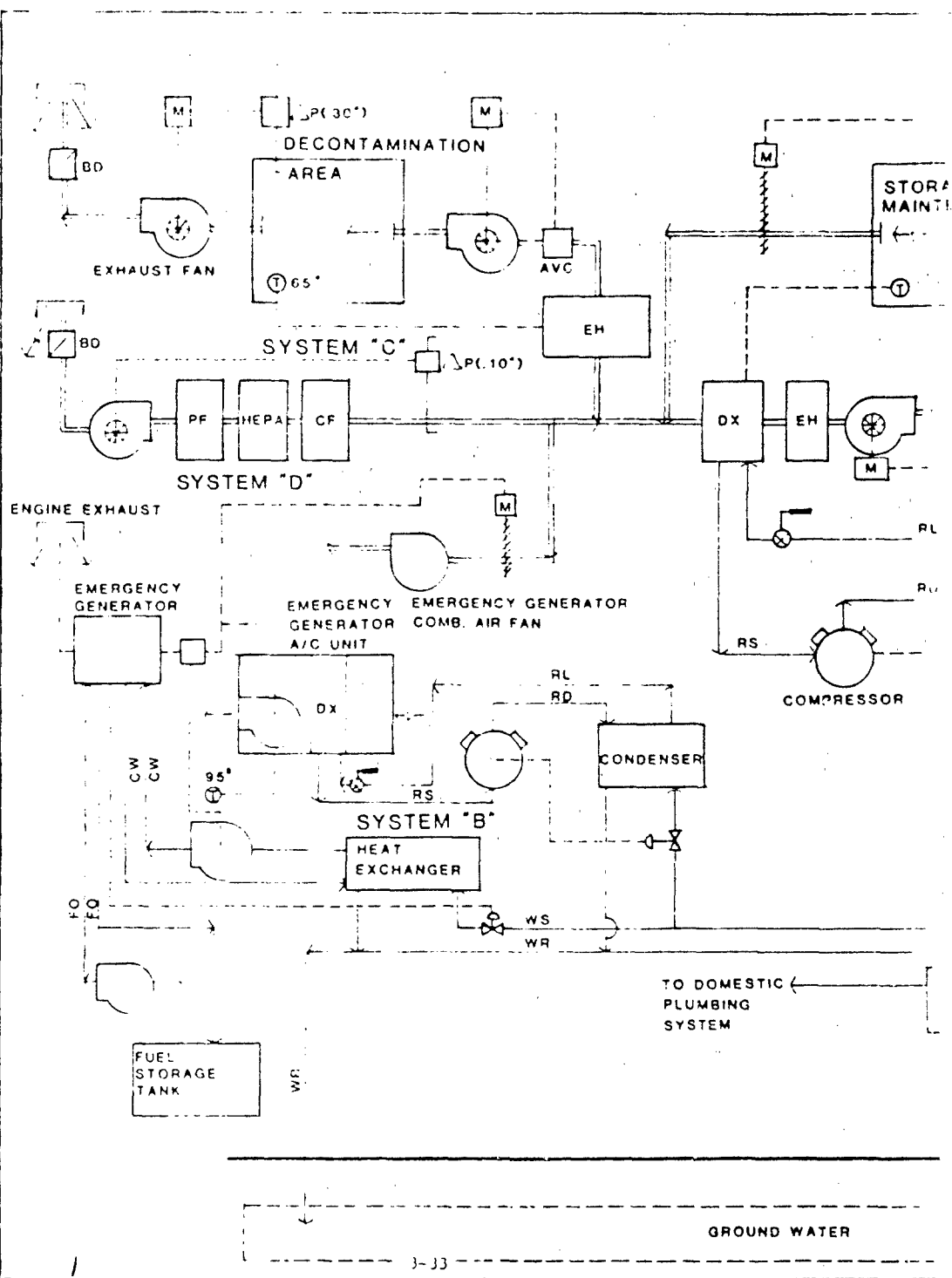
	<u>Cooling</u>	<u>Heating</u>
Weapons Storage	80°FDB, 60% R.H.(max.)	65°FDB
Corridor, Control Room, Maintenance, Toilets, Lunch Room	75°FDB, 60% R.H.(max.)	72°FDB
Mechanical and Electrical Equipment Rooms	85°FDB	65°
Contamination Control	Ventilation	65°FDB

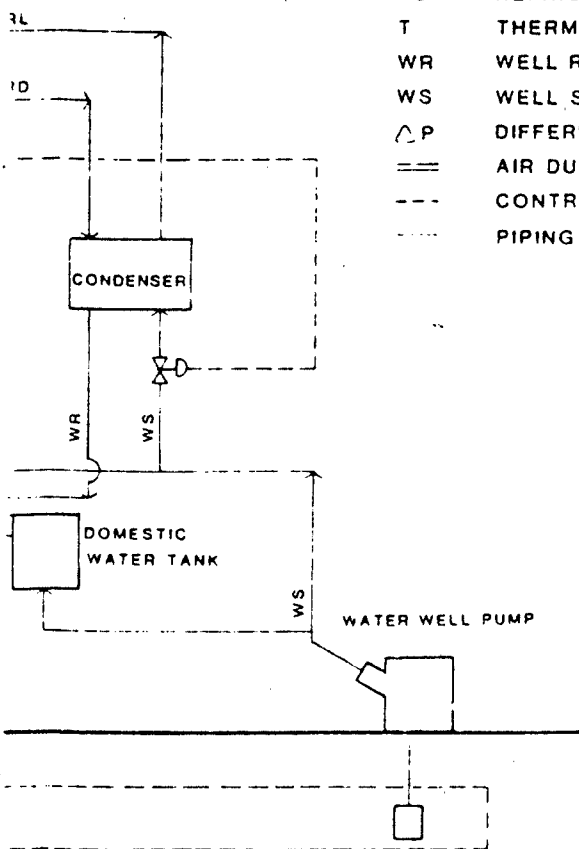
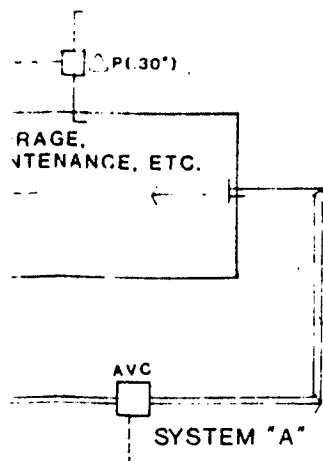
C. Pressure Relationships (Related to Ambient Pressure)

Corridor +.30" w.g.
Dress Room +.30" w.g.
Shower +.25" w.g.
Undress Room +.25" w.g.
Air Locks +.20" w.g.
Vapor Hazard +.15" w.g.
Liquid Hazard +.10" w.g.

1.3 HVAC SYSTEM SCHEMATIC

- A. The following page shows a schematic diagram of a heating ventilating and air conditioning system for the facility.





LEGEND

AVC	AIR VOLUME CONTROL
BD	BLAST DAMPER
CF	CHARCOAL FILTER
CW	GENERATOR COOLING WATER
DX	REFRIGERATION COOLING COIL
EH	ELECTRIC DUCT HEATER
FO	FUEL OIL
HEPA	ABSOLUTE FILTER
M	DAMPER MOTOR
PF	PRE-FILTER
RD	REFRIGERATION HOT GAS
RL	REFRIGERATION LIQUID
RS	REFRIGERATION SUCTION GAS
T	THERMOSTAT
WR	WELL RETURN WATER
WS	WELL SUPPLY WATER
ΔP	DIFFERENTIAL PRESSURE CONTROL
==	AIR DUCTS
---	CONTROL CIRCUIT
...	PIPING

AMMUNITION STORAGE FACILITY CONCEPT NUMBERS 1-6 HVAC SYSTEM SCHEMATIC	DWG NO	JOB NO	DATE	SCALE
	DWN BY:	CHKD BY	APPROVED:	
	JOHNSON AMMUNITION STORAGE FACILITY CONCEPT NUMBERS 1-6 HVAC SYSTEM SCHEMATIC			

- B. The diagram shows general system arrangement but not necessarily the final design solution.

PART 2 - AIR CONDITIONING AND VENTILATION SYSTEMS

2.1 SYSTEM A (AIR CONDITIONING)

- A. System shall serve weapons bays, corridor, maintenance, control room, toilets, lunch room and mechanical and electrical equipment rooms. Air handling and refrigeration equipment shall be located in the mechanical equipment room.
- B. System shall be recirculating air type with sufficient outside air to make-up exhaust from toilet and to provide the required positive pressure inside the building.
- C. The plenum above the corridor is enclosed on all four sides by reinforced concrete construction. This area shall be used as supply air duct to convey air to the four weapons bays. A reinforced concrete plenum around the outside building perimeter shall be used to return air from the weapons bays.
- D. Air shall be supplied to each bay through a HEPA filter mounted securely in the wall and open to the ceiling supply air plenum. Air shall be returned from each bay through a HEPA filter mounted securely in the wall adjacent to the return air plenum. The purpose of the HEPA filters is to prevent contaminated air from backing up from one weapons bay to the others in the event of an explosion.
- E. Necessary dampers and controls shall be provided to maintain the indoor building pressure at the required positive pressure in relation to the outdoor pressure.
- F. Condenser cooling water for the refrigeration equipment shall be supplied from the injection water well system located in the mechanical equipment room.

2.2 SYSTEM B (AIR CONDITIONING)

- A. System shall be designed to remove the heat rejected to the mechanical room by the emergency generator. Air handling and refrigeration equipment shall be located in the mechanical room with the emergency generator.
- B. System shall be the 100% recirculating air type with no outside air connections.
- C. System shall be interlocked with the emergency generator so that it operates only when the emergency generator is in operation.

D. A combustion air fan shall be provided and shall be interlocked to run when the generator is in operation. Fan shall supply prefiltered and unconditioned outside air to the mechanical room.

D. Condenser cooling water for the refrigeration equipment shall be supplied from the injection water well system.

2.3 SYSTEM C (CONTAMINATION CONTROL AREA VENTILATION)

A. A ventilation system shall be provided for the contamination control area.

B. System shall ventilate the area with 100% prefiltered outside air.

C. System shall be equipped with supply fan exhaust fan, dampers and automatic controls required to maintain the required space pressure relationships in the various spaces.

D. Heating shall be included to maintain the required space temperature.

E. Blast dampers shall be provided in the exhaust air duct which discharges air from the building. Discharge opening shall be designed to prevent entrance by intruders and to prevent explosives from being placed into the opening.

2.4 SYSTEM D (OUTSIDE AIR FILTRATION)

A. Provide an outside air filtration system to supply prefiltered outside air for the contamination control area ventilation system, main building air conditioning system and emergency generator combustion air fan.

B. The intake duct extending out of the building shall be equipped with a blast damper. Intake opening shall be designed to prevent entrance by intruders and to prevent explosives from being placed into the opening.

C. System shall include an outside air fan which intakes outside air and discharges the air through a roughing filter, HEPA filter and finally through a charcoal filter.

D. System shall be equipped with necessary controls to automatically control the volume of air delivered by the supply fan to compensate for increased fan static due to dust loading of the filters.

PART 3 - DESIGN CRITERIA RATIONAL

2.1 AIR CONDITIONING AND VENTILATION SYSTEMS

A. Comfort air conditioning is specified for the control

room, maintenance, corridor, lunch room and toilets to provide a comfortable working environment for the operating personnel. Weapons bays and mechanical/electrical rooms are also air conditioned.

- B. Consideration was given to ventilating the weapons storage bays and mechanical/electrical rooms with outside air as opposed to air conditioning. A ventilation system would require relatively large volumes of air be taken in and exhausted from the building. This would require much larger intake and exhaust openings and also a much larger outside air filtration system. Large intake and exhaust openings would be more difficult to secure against intrusion and blast. Air conditioning of these areas was selected as the most feasible alternative.

2.2 EMERGENCY GENERATOR AND REFRIGERATION EQUIPMENT HEAT REJECTION SYSTEM

- A. Consideration was given to the use of a remote air cooled radiator for the emergency generator and remote air cooled condensers for the refrigeration equipment. The remote units would be located outdoors and subject to damage by intruders. This was considered unacceptable.
- B. Consideration was given to the same type system except the emergency generator radiator and air cooled condensers would be located within the blast resistant structure. The air cooled radiator and air cooled condensers require large volumes of air to carry away heat rejected from the equipment. This would necessitate large openings for intake air and exhaust air. Large openings would be difficult to secure against intrusion and blast. This arrangement was rejected as unacceptable.
- C. The use of a cooling tower was also considered. Pumps would circulate cooling tower water between the tower and the emergency generator and refrigeration equipment heat. This type arrangement was considered unacceptable for the same reasons as the air cooled heat exchanger concept was rejected.
- D. An injection type water well system was selected as the most desirable means of removing heat rejected from the emergency generator and refrigeration equipment. A well pump would circulate water through the equipment heat exchangers and then return the underground water system. The well would also supply the domestic water plumbing system. The well equipment could be located within the blast resistant structure and inaccessible to intruders.

PLUMBING AND FIRE PROTECTION NARRATIVE

PART 1 - GENERAL

1.1 BASIC CRITERIA

- A. Standard Building Code.
- B. NFPA 10
- C. NFPA 13

1.2 BASIS OF DESIGN

- A. Sanitary waste and vent system shall be designed on the "Per Fixture Unit" basis.
- B. Domestic cold water system shall be designed on the "Per Fixture Unit" basis.
- C. Domestic hot water system shall be designed on the "Per Fixture Unit" basis.
- D. Portable fire extinguishers shall comply with the requirements of NFPA-10.
- E. Fire sprinkler systems shall be designed in accordance with NFPA-13.

1.3 OUTSIDE UTILITIES

- A. Fire water and sanitary sewer lines shall terminate five feet outside of the building line.
- B. Extending the utilities beyond the five foot line is not included in the scope of this report.

PART 2 - SYSTEMS DESCRIPTION

2.1 DOMESTIC WATER SYSTEMS

- A. Domestic cold water shall be piped to all plumbing fixtures including lavatories, water closets, urinals, service sinks and hose outlets.
- B. Hot water shall be supplied to lavatories and service sinks. Hot water shall be supplied by electric water heaters.

2.2 WATER WELL SYSTEM

- A. An injection type water well shall be provided with sufficient capacity to supply the domestic water demand and cooling water requirements of the emergency generator and air conditioning system refrigeration equipment.

- B. Cooling water for the emergency generator and refrigeration compressors shall be pumped through heat exchangers on the equipment and then returned to the ground.

2.3 FIRE SPRINKLER SYSTEM

- A. A wet pipe fire sprinkler system shall be provided for all areas in the facility. The hazard classification for the various occupancies will be determined during the design phase. Water supply shall be from the main base water system.
- B. The need for high volume deluge system in the weapons storage bays may be considered in the design phase.

2.4 FIRE EXTINGUISHERS

- A. Hand held portable fire extinguishers shall be provided in the main service corridor and in mechanical and electrical equipment rooms.
- B. Extinguishers shall be surface mounted on wall brackets.

2.5 SANITARY WASTE SYSTEM

- A. A complete sanitary waste and vent system shall be extended to all plumbing fixtures, equipment and flow drains and connected to the sanitary sewer main.
- B. Provide a duplex sewerage ejector with sufficient lift to pump sewerage into the gravity sewer.
- C. All vent stacks shall be designed to be bomb proof.

2.6 COMPRESSED AIR SYSTEM

- A. A compressed air piping system shall be provided to serve the facility including pneumatic door operators, pneumatic tool air outlets and air operated hoists.
- B. A duplex air compressor shall be provided to supply compressed air to the system.

ELECTRICAL NARRATIVE

PART 1 - BASIC REQUIREMENTS

1.1 GENERAL

- A. Electrical work shall conform with the requirements of the latest edition of the National Electrical Code, Life Safety Code, and local codes and ordinances. In case of conflict, local codes and ordinances shall govern.
- B. Materials and equipment shall conform to and be in accordance with the latest applicable standards of National Electrical Manufacturers Association (NEMA), American National Standards Institute (ANSI), and Insulated Cable Engineers Association (ICEA) where standards have been established for the specific items of materials and equipment. Where inspection categories have been established by Underwriters Laboratories, materials and equipment shall bear their label.
- C. An EMR clean environment shall be provided in accordance with DARCOM 385-100, 6-14.

PART 2 - SERVICE AND DISTRIBUTION

2.1 MUNITIONS STORAGE

- A. A new substation will be provided in the building. Equipment shall consist of a power center 300 KVA dry type transformer and bolted type air circuit breaker substation for secondary distribution. System voltage shall be 480 volts, 3 phase, 4 wire secondary.
- B. The transformer will be supplied from a load break switch located in the substation.
- C. The primary service feed from the utility to the substation will be installed in concrete encased conduit. Cable will be fire proofed whenever exposed.
- D. The transformer will be sized to provide 100% spare capacity.
- E. The substation will supply power to the parking lot lighting.
- F. The estimated total connected load for the substation will be 157.5 KVA.

2.2 SECONDARY DISTRIBUTION

- A. Secondary distribution within buildings will be at 480

volts. Generally, HVAC systems will be served at 480 volts and the fluorescent lighting system at 120 or 277 volts.

- B. Dry type transformers will be installed to provide 208/120 volt service for lighting convenience receptacles and miscellaneous equipment.
- C. Secondary distribution, 480/277 and 208/120 volt systems within buildings, will be wire in conduit and circuit breaker panelboards.
- D. Equipment used to distribute power to loads shall be 480 volt, 3 phase, 3 wire motor control centers, 480 volt, 3 phase, 4 wire power panels and 208/120 volt, 3 phase, 4 wire lighting and receptacle panels. All motor control centers power panels, dry type transformers and lighting panels are to be located in separate rooms for electrical equipment.
- E. Ammeters, voltmeters and KWH-demand meters are to be provided in addition to utility metering.

PART 3 - LIGHTING SYSTEMS

3.1 GENERAL

- A. Lighting systems will conform to engineering practice using standards from the Illuminating Engineers Society handbook as guidelines.
- B. Systems will utilize energy efficient light sources. Lamp life and ease of maintenance will be given effective consideration.
- C. Roads, parking, and walks will be illuminated to approximately 0.5 footcandle on roads and 2.0 footcandle on parking and walk areas using high pressure sodium as a light source.
- D. Control room will be illuminated to approximately 75 maximum footcandles using 2' x 4' fluorescent fixtures. Fixtures will be switched locally providing a multiple level of lighting. Generally, three lamp fixtures will be used with center lamp and two outside lamps switched separately.
- E. Maintenance will be illuminated to approximately 90 footcandles using fluorescent fixtures, switched locally. Multi-level lighting will be provided.
- F. Warehouse areas will be illuminated to approximately 30 footcandles using Class I Group D, Class II Group G, Division 2 light source fixtures.

- G. General and miscellaneous areas will be considered on an individual basis. Special attention and treatment will be given to all areas.
- H. Emergency lighting will be provided to insure safe egress in case of power failure or under other emergency conditions.

PART 4 - MAINTENANCE AND CONTROL

4.1 GENERAL

- A. Electrical services and outlets will be provided to service maintenance equipment as required.
- B. An emergency generator will be provided to service selected areas and control equipment in the event of power failure or other emergency. Equip with diesel fuel day tank, underground storage tank, heat exchanger and engine exhaust systems.
- C. All systems and equipment will be provided with grounding conductors.

PART 5 - AUXILIARY SYSTEMS

5.1 GENERAL

- A. Auxiliary systems will be provided complete for the following:
 - 1. Fire alarm system.
 - 2. Closed circuit T. V. surveillance system at control room.
 - 3. A low level distributed sound system, fully equalized will be provided for the building.
 - 4. Computer terminal system will be provided.
 - 5. Security monitoring will be provided through door monitors, sound detectors, motion detectors as required.
 - 6. Individual systems for sensing environmentally hazardous chemicals.
 - 7. Safety monitoring of storage rooms will be provided with pressure monitors set for 5 psi.
- B. An empty conduit and pull box system only will be provided for telephone system.
- C. Auxiliary systems will be connected to the emergency generator system through a uninterruptable power supply.

PART 6 - LIGHTING PROTECTION AND STATIC GROUNDING

6.1 GENERAL

- A. Lighting protection system in accordance with DARCOM 385-100 and NFPA 78 will be provided.
- B. Static grounding system in accordance with DARCOM 385-100 and NFPA 77 will be provided.

SECTION III

3-45

ESTIMATES OF PR

Item	Description	Unit	Concept 1		Concept 2		Q
			Quantity	Cost	Quantity	Cost	
1	General Conditions			\$ 651,600		\$ 738,500	
2	Building Earthwork						
A.	Excavation	\$5.00/ CY	18,900 CY	105,900	29,160 CY	163,300	31
B.	Fill & Subgrade Prep.	2.37/ CY	10,500 CY	59,100	20,600 CY	59,100	32
3	Concrete Work						
A.	Concrete	61.63/ CY	22,630 CY	1,440,000	27,460 CY	1,747,300	28
B.	Forms	1.75/ SF	286,124 SF	50,000	346,514 SF	606,400	354
C.	Reinforcing Steel	1,162/ Ton	2,554 Tons	3,223,100	3,097 Tons	3,908,700	
D.	Pilings	1,000/ Ea.	400/ Ea.	400,300	483 Ea.	482,700	4
4	Masonry						
A.	Brickwork	1,250/ Ton	4 Tons	5,000	2.9 Tons	9,900	
B.	Railing, Etc.	10/ LF	50 LF	500	50 LF	500	
5	Wood & Plastics						
A.	Thermal & Moisture Prot.	2.10/ SF	36,400 SF	295,000	39,200 SF	317,500	3
6	Doors & Windows						
A.	Interior Doors	119,600/ Ea.	5 Ea.	493,000	5 Ea.	693,000	
B.	Exterior Doors	8,000/ Ea.	2 Ea.	16,000	2 Ea.	16,000	
C.	Special Int. Doors	2,005/ Ea.	19 Ea.	38,100	20 Ea.	40,100	
D.	Hollow Metal Doors	400/ Ea.	3 Ea.	1,200	3 Ea.	1,200	
E.	Windows	1/ 12/ SF	33 SF	400	33 SF	400	
F.	Hardware	5,000/ Lot	1 Lot	5,000	1 Lot	5,000	
G.	Finishes	3.45/ SF	36,400 SF	140,000	39,200 SF	151,300	2
7	Specialties	12,000/ Lot	1 Lot	12,000	1 Lot	12,000	
8	Equipment						
A.	Case-work	175/ LF	126 LF	22,100	126 LF	22,100	
B.	Book Lockers	3,700/ Ea.	4 Ea.	14,800	4 Ea.	14,800	
9	Furnishings						
10	Special Construction						
11	Conveying Systems						
A.	Hoists, Pneumatic	4,500/ Ea.	1 Ea.	4,500	1 Ea.	4,500	
B.	Lifts, Hydraulic	40,000/ Ea.			1 Ea.	5,000	
C.	Bridge Crane	5,000/ Ea.					
12	Mechanical						
A.	Heat, Vent. & A. C. Sys.						
1.	Direct Expansion Sys.	1,000/ Ton	16.8 Tons	110,400	17.8 Tons	112,400	
2.	Free Blow A. C. Sys.	400/ Ton	13.8 Tons	11,000	14.1 Tons	11,300	
3.	Constant, Cent. Vent. Sys.	3.91/ CFM	7,400 CFM	28,900	7,520 CFM	29,400	
4.	Air Filtration Sys.	11.15/ CFM	6,900 CFM	76,900	7,050 CFM	78,600	
5.	Unorg. Gen. Vent. & Cool. Sys.	6,500/ Sys.	1 Sys.	6,500	1 Sys.	6,500	
6.	Planning & Start-up	Lump Sum		11,400		11,400	
B.	Pumping Systems						
1.	Air Compressor Sys.	6,000/ Sys.	1 Sys.	6,000	1 Sys.	8,000	
2.	Domestic Piping & Fixt.	36.63/ LF	1,000 LF	37,000	1,030 LF	37,900	
3.	Perimeter Drain Sys.	23/ LF	920 LF	21,200	940 LF	21,600	
4.	Water Well System	205,000/ Ea.	1 Ea.	205,000	1 Ea.	205,000	
D.	Fire Protection System	1,347/ SF	24,000 SF	32,200	24,100 SF	32,700	
13	Electrical						
A.	Basic Plant Service	34,474/ SF	36,400 SF	1,414,900	39,200 SF	1,523,700	4
B.	Lighting	275/ SF	36,400 SF	27,300	39,200 SF	29,400	1
C.	Emergency Generator Sys.	2,037/ SF	36,400 SF	73,900	39,200 SF	79,600	4
D.	LPS	2,114/ SF	36,400 SF	77,900	39,200 SF	83,900	
E.	Fire Alarm & Monitor Sys.	771/ SF	36,400 SF	25,800	39,200 SF	27,800	4
Subtotal				\$ 9,193,800		\$11,194,600	
Contractor's Markup				783,500		903,700	
Subtotal				9,977,300		12,100,300	
Design Contingency				1,583,700		1,830,000	
Total Estimated Cost				\$12,140,800		\$14,930,300	
Total Square Footage				36,357		39,200	
Cost per Square Foot				\$ 334		\$ 38	
35 Year Life Cycle Cost							
Building				\$14,226,000		\$16,192,000	
Equipment				206,000		221,000	
Utilities				902,000		930,000	
Total				\$15,334,000		\$17,343,000	

Assume completion in 1984

TABLES OF PROBABLE CONSTRUCTION COST

Concept 3		Concept 4		Concept 5		Concept 6	
Cost	Quantity	Cost	Quantity	Cost	Quantity	Cost	Quantity
1738,560		\$ 422,200		\$ 443,400		\$ 431,400	
							\$ 868,900
183,200	31,500 CY	129,600	19,500 CY	274,400	49,000 CY	274,400	82,357 CY
59,100	32,020 CY	16,100	33,500 CY	88,100	30,700 CY	88,100	51,846 CY
247,360	24,100 CY	17,757,600	39,240 CY	1,800,000	30,525 CY	1,942,300	30,950 CY
807,400	154,100 SF	820,300	392,750 SF	869,800	385,965 SF	873,930	390,460 SF
109,700	3,118 Tons	3,848,500	3,421 Tons	4,317,500	3,442 Tons	4,342,800	3,490 Tons
192,700	404 Ea.	133,600	532 Ea.	132,100	535 Ea.	535,000	543 Ea.
4,300	4 Tons	5,000	4 Tons	5,000	7.9 Tons	8,900	34 Tons
500	30 LF	500	30 LF	500	50 LF	500	30 LF
107,500	39,500 SF	120,000	34,440 SF	441,000	49,827 SF	403,600	55,555 LF
693,000	5 Ea.	693,000	5 Ea.	693,000	5 Ea.	431,600	5 Ea.
16,000	2 Ea.	16,000	2 Ea.	16,000	2 Ea.	16,000	1 Ea.
40,100	28 Ea.	42,100	22 Ea.	44,100	15 Ea.	48,100	27 Ea.
1,200	3 Ea.	1,200	3 Ea.	1,200	3 Ea.	1,200	3 Ea.
400	10 SF	400	10 SF	400	30 SF	400	30 SF
5,000	1 Lot	5,000	1 Lot	5,000	1 Lot	5,000	1 Lot
151,000	38,900 SF	160,000	37,000 SF	144,300	34,212 SF	132,100	33,766 SF
12,000	1 Lot	12,000	1 Lot	12,000	1 Lot	12,000	1 Lot
22,100	18 Ea.	22,100	18 Ea.	22,100	18 Ea.	22,100	40 LF
14,800	4 Ea.	14,800	4 Ea.	14,800	4 Ea.	14,800	2 Ea.
4,500	1 Ea.	4,500	1 Ea.	4,500	1 Ea.	4,500	1 Ea.
5,000	2 Ea.	5,000	2 Ea.	5,000	2 Ea.	5,000	1 Ea.
112,800	40 Tons	112,800	40 Tons	112,800	40 Tons	112,800	42 Tons
11,300	15 Tons	11,300	15 Tons	12,600	15 Tons	12,600	15.8 Tons
29,400	8,300 CFM	29,400	8,300 CFM	31,200	8,000 CFM	31,200	8,400 CFM
78,600	7,500 CFM	78,600	7,500 CFM	83,600	7,000 CFM	83,600	7,875 CFM
6,500	1 Sys.	6,500	1 Sys.	6,500	1 Sys.	6,500	1 Sys.
11,400	2,000	11,400	2,000	11,400	2,000	11,400	2,000
4,000	1 Sys.	4,000	1 Sys.	4,000	1 Sys.	4,000	1 Sys.
37,900	1,100 LF	40,300	1,100 LF	42,300	1,100 LF	40,300	1,100 LF
11,600	1,000 LF	11,600	1,000 LF	11,600	1,000 LF	11,600	1,050 LF
205,000	1 Ea.	205,000	1 Ea.	205,000	1 Ea.	205,000	1 Ea.
32,700	26,000 SF	32,700	26,000 SF	34,900	26,000 SF	34,900	27,300 SF
23,700	43,775 SF	23,700	43,775 SF	23,700	43,775 SF	23,700	43,775 SF
29,400	43,775 SF	29,400	43,775 SF	29,400	43,775 SF	29,400	43,775 SF
79,600	43,775 SF	79,600	43,775 SF	79,600	43,775 SF	79,600	43,775 SF
83,900	43,775 SF	83,900	43,775 SF	83,900	43,775 SF	83,900	43,775 SF
27,800	43,775 SF	27,800	43,775 SF	27,800	43,775 SF	27,800	43,775 SF
109,500		109,500		109,500		109,500	
603,000		603,000		603,000		603,000	
100,000		100,000		100,000		100,000	
910,000		910,000		910,000		910,000	
930,750		930,750		930,750		930,750	
19,200		19,200		19,200		19,200	
350		350		350		350	
102,000		\$17,567,000		\$14,019,000		\$14,029,000	
121,000		211,000		211,000		211,000	
930,000		930,000		930,000		930,000	
930,000		\$14,019,000		\$14,019,000		\$14,019,000	

SECTION IV

COMPARATIVE ANALYSIS

The following table presents a comparative rating of the six facility concepts using six separate comparison factors. Three of the factors: efficiency, life cycle cost and total cost are strictly objective, while the other three factors are subjective, requiring judgement and knowledge of project objectives to apply ratings. In order to minimize any tendencies for bias ratings, the BJI project team applied the ratings independently and developed averages to arrive at a single rating for each subjective factor. Explanations of rating factors follow the table.

Based on total ratings for all factors and the rating scale (lowest total equal best rating), Concept No. 1 is the most favorable solution, Concept No. 3 is second and Concept No. 4 is third.

COMPARATIVE RATING TABLE

RATING FACTORS	CONCEPTS					
	No. 1	No. 2	No. 3	No. 4	No. 5	No. 6
Efficiency (Net-to-Gross Area Ratio)	5	4	6	2	1	3
Life Cycle Cost	1	4	2	3	5	6
Total Cost	1	2	3	5	4	6
Function	1	5	2	3	6	4
Security	3	4	2	1	5	6
Safety	1	2	4	6	3	5
TOTALS	12	21	19	20	24	30

Rating Scale: Highest = 1; Moderate = 3-4; Poorest = 6

1. Efficiency is defined as the "net" assignable area in relation to the "gross" space and is calculated by dividing the total net square feet by the total gross square feet.

$$\text{Efficiency} = \frac{\text{net square feet}}{\text{gross square feet}}$$

Net square feet (area) is the area assigned for facility operation and includes circulation within a functional area, but not outside of it. Therefore, corridor and vestibule areas are excluded. Also, mechanical/electrical space is excluded. Gross square feet (area) is the total area within the building including structure, partitions, hallways, building support areas (such as mechanical/electrical areas, plenums, etc.) and toilets, also including 50 percent of the covered dock area.

2. Life cycle cost includes the 25 year cost of maintaining and operating the building and equipment.
3. Total cost includes the price for the facility construction including all architectural, mechanical, electrical, plumbing and structural.
4. Function is based on how well the building plan meets the "program" criteria. Involved in this analysis are considerations concerning building configuration (Does the "plan" lead to spaces larger than program requirements?), and the need for special equipment or conveying systems (Does the type of weapons storage or building configuration require additional cranes or hoists in order to perform the facility's function?).
5. Security is determined by facility accessibility and internal control.
6. Safety is rated both for personnel egress and internal safety in case of a hazard.

APPENDIX 4

DESIGN DRAWINGS

Plan and section drawings for the six baseline munition storage facilities are contained in the envelope attached to this volume of the report.

APPENDIX 4

DESIGN DRAWINGS

APPENDIX 5

CONCEPT DESIGN

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
1 OF 13

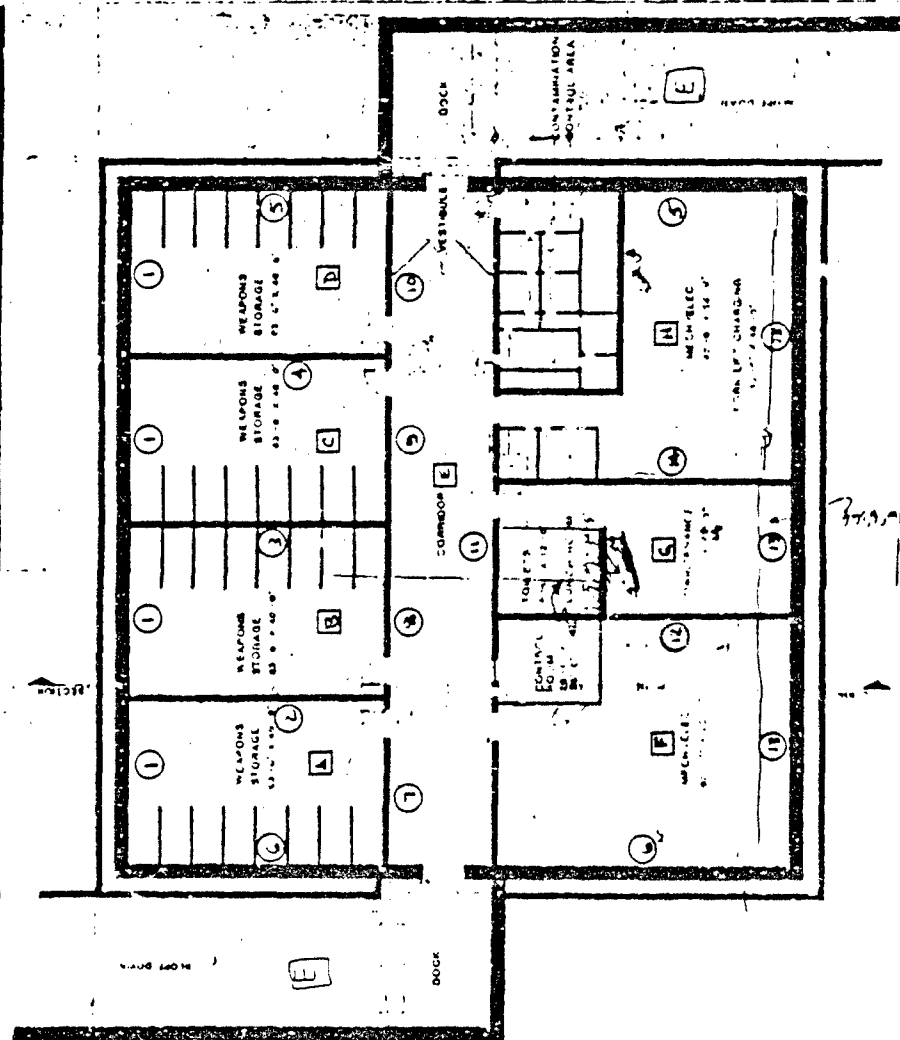
PROJECT NO.: 02-7092 SPONSOR: CORP
SUBJECT: DESIGN RECONSTRUCTION SUMMARY CONCEPT 1
BY: D. K. K. DATE: 29 Sept 82 CHECKED BY: DATE CHECKED: 19

CONCEPT No. 1					
WALL/ROOF	TRIM	BAR (FLS)	SHEATHING		SPAN (IN) OR (FT x IN)
1	48	FLOOR	#1012	#1012	240"
2	24	F	#9012	#1012	216"
3	24	F	#7012	#1012	240"
4	24	F	#9012	#1012	216"
5	48	F	#2012	#2012	240"
6	48	F	#2012	#2012	240"
7,8,9,10	24	F	#3012	#2012	36"
11,12,13	24	F	(2) #3012	#1012	240"
13	48	F	#1012	#1012	240"
A	48	F	#1012	#9012	63' x 40'
B	48	F	#1012	#2012	63' x 40'
C	48	F	#1012	#9012	63' x 40'
D	48	F	#1012	#3012	63' x 40'
E	72	F	(3) #10012	#2012	30' 1-way
F	48	F	#1012	#11012	60' x 73'
G	48	F	#11012	#11012	31' x 48'
H	48	F	#11012	#11012	71' x 42'

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
2 OF 13

PROJECT NO.: 02-7092 SPONSOR: CDR
SUBJECT: CONCEPT 1 DOUBLE STACKED
BY: D. KETCHUM DATE: 27 SEP 19 82 CHECKED BY: DATE CHECKED: 19



SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO
3 OF 13

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: DESIGN REQUIREMENT SUMMARY CONCEPT 2
BY: XXXXXXXX DATE: 28 OCT 19 92 CHECKED BY: _____ DATE CHECKED: _____ 19__

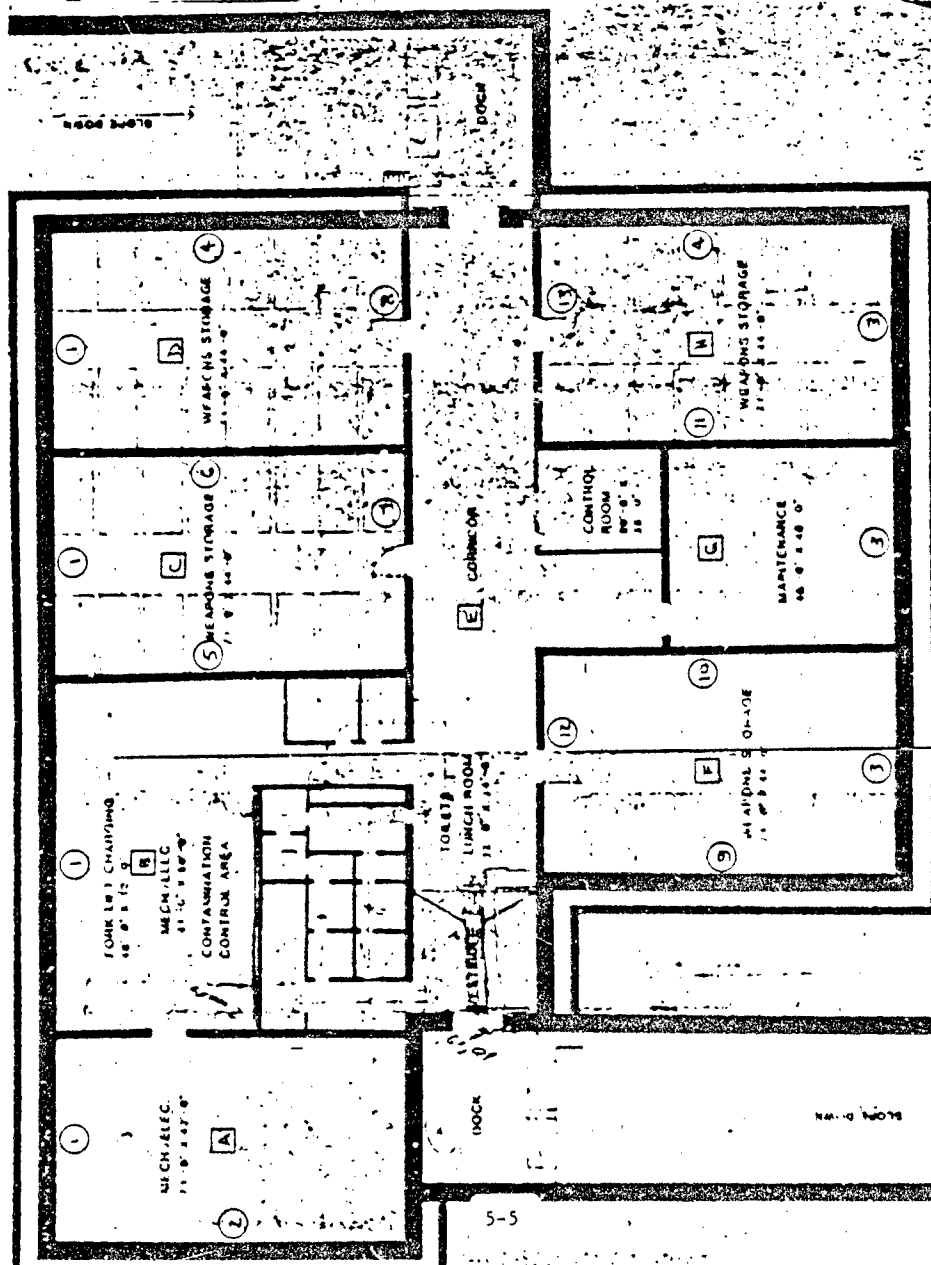
CONCEPT No. 2

NOTE: LINES 1 - 1000
NUMBERS 1 - 1000

WAVE/F	T (in)	Bar (F.L.S)	SIZE / SPACING		SIZE (in) or (ft x ft)
			SHORT SAW	LONG SAW	
1	48	F	#10/12	#10/12	240"
		L	#5		
2	48	F	#10/12	#11/12	240"
		L	#5		
3	48	F	#10/12	#11/12	240"
		L	#5		
4	48	F	#10/12	#11/12	240"
		L	#5		
5,6	24	F	#10/12	#6/12	240"
		L	#5 mm x 2		
7,8,12	24	F	#10/12	#6/12	240"
		L	#5 mm x 2		
9	48	F	#10/12	#10/12	
		L			
10,11	24	F	(2) #10/12	#6/12	240"
		L	#7/12		
A	48	F	#11/12	#10/12	71 x 42
		S	#5		
B	48	F	#10/12	#10/12	71 x 42
		S	#5		
C	48	F	#10/12	#10/12	71 x 44
		S	#5		
D	48	F	#11/12	#10/12	71 x 44
		S	#5		
E	72	F	3 #10/12	#9/12	30' 1" x 44'
		S	#5		
F	48	F	#10/12	#10/12	71 x 44
		S	#5		
G	48	F	#11/12	#11/12	45 x 40
		S	#5		
H	48	F	#10/12	#10/12	71 x 44
		S	#5		

SHEET NO.
4 OF 13

PROJECT NO.: 02-7092 SPONSOR: CORL
SUBJECT: CONCEPT 2 - PITS
BY: _____ DATE: 22 MAR 19 CHECKED BY: _____ DATE CHECKED: _____ 19 _____



SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
5 OF 13

PROJECT NO: 02-7092

SPONSOR: CDR

SUBJECT: DESIGN REQUIREMENT SUMMARY CONCEPT 3

BY: [Signature]

DATE: 28 Jan 1982

CHECKED BY:

DATE CHECKED: 19

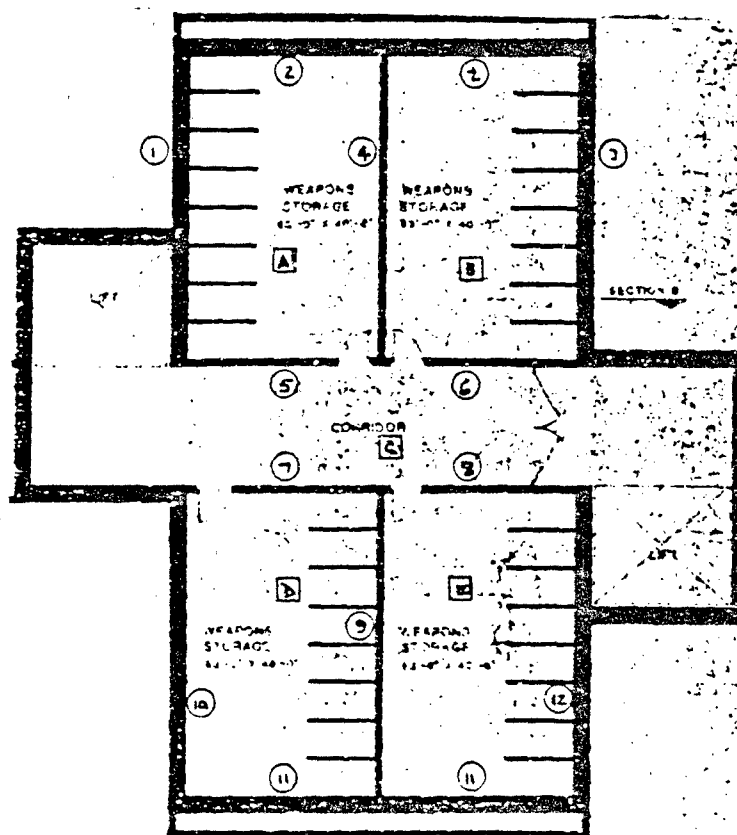
Notes: ALL DIMENSIONS IN INCHES
UNLESS OTHERWISE SPECIFIED

WALL/ROOF	CONCEPT NO. 3		SIZE / SPACING		SIZE (IN) OR (FT x IN)
	T (IN)	BAR (F.L.S)	SHORT SPAN	LONG SPAN	
1	48	F	#11 @ 12	#6 @ 12	492"
		L			
2, 11	48	F	#9 @ 12	#6 @ 12	492"
		L			
3, 13	48	F	#11 @ 12	#6 @ 12	492"
		L			
4	24	F	#9 @ 12	#6 @ 12	216"
		L	#5 #2		
5, 6, 7	24	F	#9 @ 12	#6 @ 12	96"
		L	#5 #2		
10, 12	48	F	#11 @ 12	#6 @ 12	492"
		L			
A, B, D, E	48	F	#11 @ 12	#9 @ 12	63' x 40'
		S	#5		
C	72	F	(3) #10 @ 12	#9 @ 12	30' 1-1/4" x 4
		S	#5		
9	24	F	#7 @ 12	#6 @ 12	240"
		L	#5 #2		
CRANE / FLOOR	36	F	(2) #9 @ 12	#7 @ 12	80' x 156'
		L	#5 #2		

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
6 OF 13

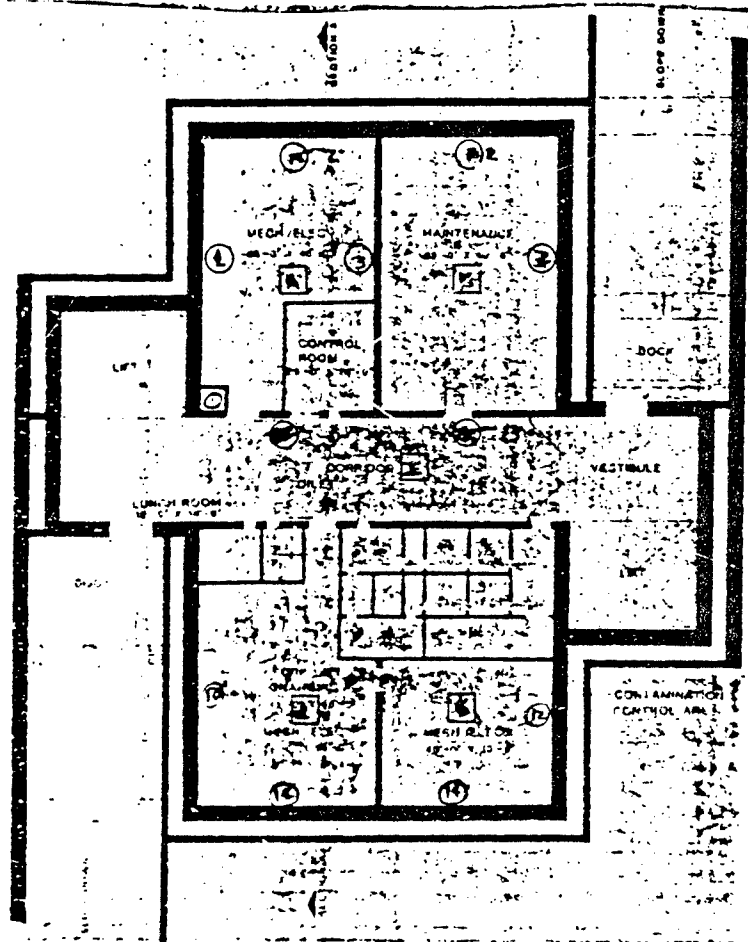
PROJECT NO.: D2-7092 SPONSOR: CITEL
SUBJECT: 2 FLOW STAGE
BY: _____ DATE: 19 CHECKED BY: _____ DATE CHECKED: 19



SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
7 OF 13

PROJECT NO.: 02-7092 SPONSOR: CRL
SUBJECT: COMSTT #3 - DENSE SMOKE - ABOVE GRADE
BY: D. K. M. DATE: 14 Feb 19 52 CHECKED BY: DATE CHECKED: 19



SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
9 OF 17

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: DESIGN REQUIREMENT SUMMARY
BY: DKT/SHW DATE: 20 SEP 19 82 CHECKED BY: _____ DATE CHECKED: _____ 19 ____

CONCRETE 4

NUMBERS # WALLS
NOTE - LOTIONS = RADES

WALL/ROOF	T _c (in)	BAR(E,WS)	SIZE/SPACING		SIZE (IN) OR (FT x FT)
			SHORT SPAN	LONG SPAN	
1,3	24	F L	#7@12 #5@12	#6@12	156"
2	24	F L	#7@12 #5@12	#6@12	144"
4,5,6,7	24	F L	#9@12 #5@12	#6@12	144"
8,9,10	24	F L	(2) #9@12 #7@12	#6@12	240"
11	36	F L	#10@12	#10@12	240"
12	36	F L	#10@12	#10@12	240"
13	36	F L	#9@12	#9@12	156"
14	36	F L	#11@12	#9@12	156"
15	36	F L	#9@12	#9@12	156"
16	36	F L	#10@12	#10@12	240"
A	36	F S	#11@12 #5	#9@12	26' 1-WAY
B	36	F S	#11@12 #5	#9@12	26' 1-WAY
C	36	F S	#11@12 #5	#9@12	26' 1-WAY
D	36	F S	#11@12 #5	#9@12	26' 1-WAY
E	48	F S	#11@12 #5	#9@12	25' 1-WAY
F	36	F S	#10@12 #5	#10@12	52' x 48'
G	36	F S	#10@12 #5	#9@12	52' x 48'
H	36	F S	#9@12	#9@12	37' x 54'
I	36	F S	#10@12 #5	#10@12	62' x 54'

5-9

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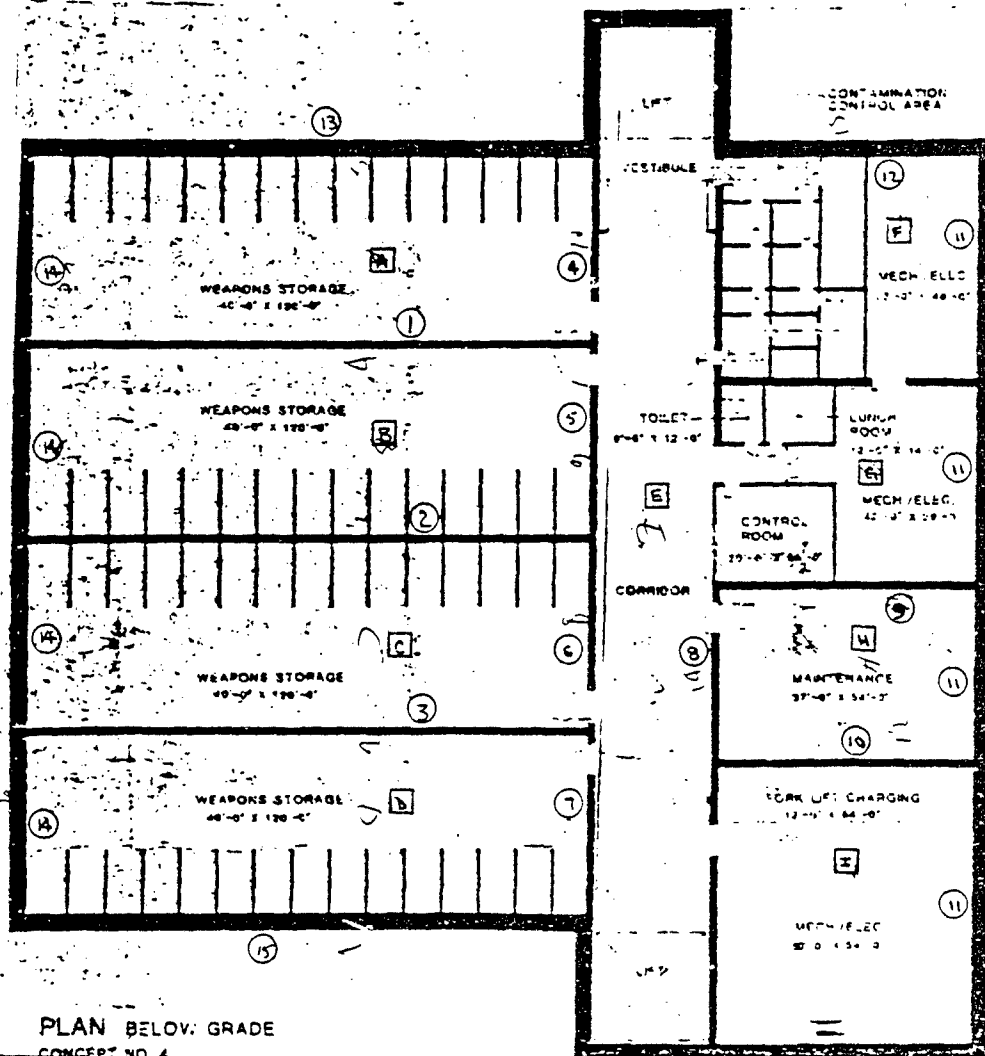
PROJECT NO.: 02-7592

SPONSOR: CERL

SUBJECT: CONCEPT 4 LONG BAY

BY: DATE: 2-5-79

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PLAN BELOW GRADE
CONCEPT NO. 4

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SHEET NO.
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PROJECT NO.: 92-7092 SPONSOR: CERL
SUBJECT: Design and Construction Summary Sheet 5
BY: [Signature] DATE: 2/19/92 CHECKED BY: DATE CHECKED: 19

CONCRETE No. 5

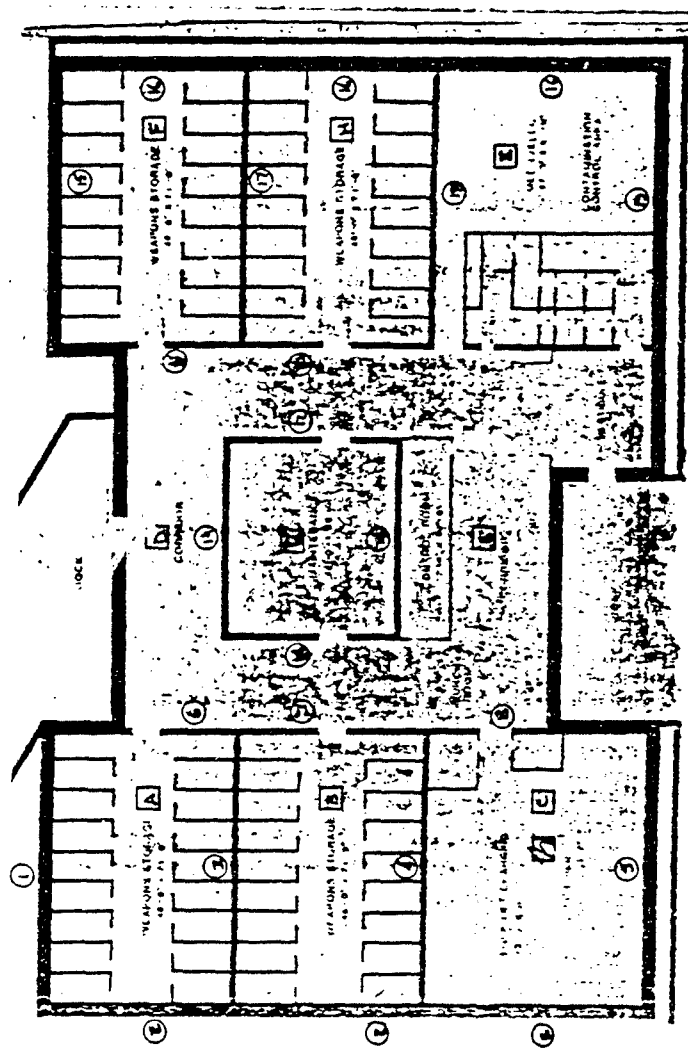
LETTERING RANGES
NOTE: NUMBERS & LETTERS

WALL/DOOR	Tc (in)	BAR (F, S)	SIZE/SPACING		SIZE (in) OR (FT x FT)
			SHORT SPAN	LONG SPAN	
1	36	F L	#11 @ 12	#6 @ 12	240"
2	36	F L	(2) #9 @ 12	#6 @ 12	240"
3, 4, 17, 18	24	F L	#7 @ 12 #5 @ 12	#6 @ 12	240"
5	36	F L	#11 @ 12	#6 @ 12	240"
6, 7, 9, 10	24	F L	#10 @ 12 #5 @ 12	#6 @ 12	240"
11, 12, 13, 14	24	F L	(2) #9 @ 12 #7 @ 12	#6 @ 12	240"
15	36	F L	#11 @ 12	#6 @ 12	240"
16	36	F L	(2) #9 @ 12	#6 @ 12	240"
19	36	F L	(2) #9 @ 12	#6 @ 12	240"
A	36	F S	#10 @ 12 #5	(2) #9 @ 12	46' x 71'
B	36	F S	#10 @ 12 #5	(2) #9 @ 12	46' x 71'
C	36	F S	#11 @ 12 #5	(2) #9 @ 12	57' x 71'
D	36	F S	(2) #9 @ 12	#6 @ 12	25' 1-WAY
E	36	F S	(2) #9 @ 12	#6 @ 12	25' 1-WAY
F	36	F S	#10 @ 12 #5	(2) #9 @ 12	46' x 71'
G	36	F S	#11 @ 12	#11 @ 12	48' x 46'
H	36	F S	#10 @ 12 #5	(2) #9 @ 12	46' x 71'
I	36	F S	#11 @ 12 #5	(2) #9 @ 12	57' x 71'

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SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: CONCEPT 5 MATE
BY: DATE: 19 CHECKED BY: DATE CHECKED: 19



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SHEET NO.
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PROJECT NO.: 02 7092 SPONSOR: CERL
SUBJECT: EXTRUSION FLOW-INDUCED STRESS ANALYSIS
BY: DATE: 2/21/92 CHECKED BY: DATE CHECKED: 19

CONCRETE No. 6

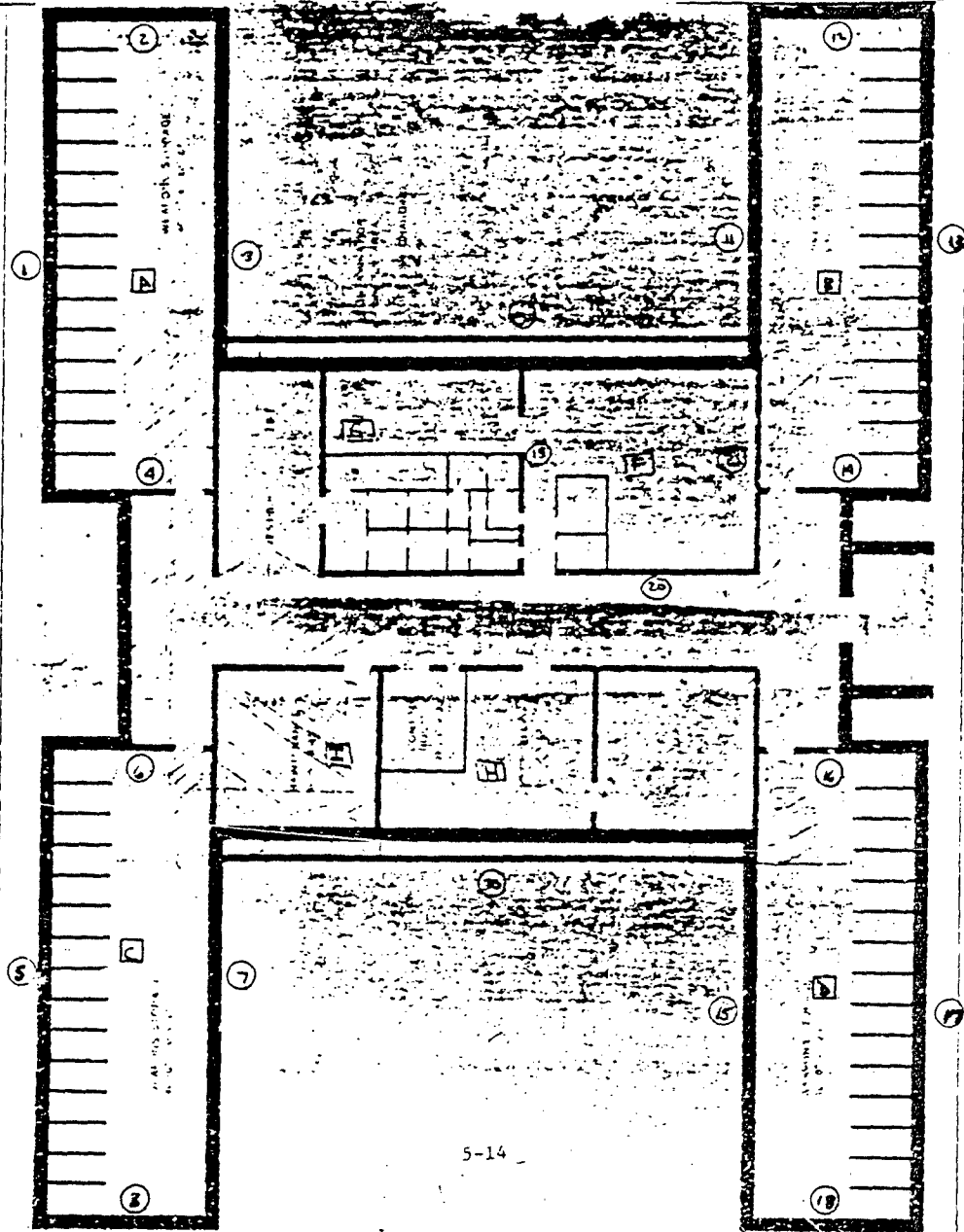
NUMBERS = WALLS
Note: LORRISSE BASE

WALL/POLE	T _c (in)	BAR (F.L.S)	SIZE/SPACING		SIZE (in) or (ft x ft)
			SUB-SOIL	LONG CHAN	
1,13	36	F	#10@12	#6@12	156"
		L			
2,12	36	F	#9@12	#6@12	156"
		L			
3,11	36	F	#10@12	#6@12	156"
		L			
4,9,14,16	24	F	#9@12	#6@12	144"
		L	#5@2		
5,17	36	F	#11@12	#6@12	156"
		L			
7,15	36	F	#10@12	#6@12	156"
		L			
8,18	36	F	#9@12	#6@12	156"
		L			
9	36	F	#10@12	#6@12	240"
		L			
10	36	F	#10@12	#6@12	240"
		L			
19,20,21	24	F	(2) #9@12	#6@12	240"
		L	#7@12		
A	36	F	#11@12	#9@12	26' 1-WAY
		S	#5		
B	36	F	#11@12	#9@12	26' 1-WAY
		S	#5		
C	36	F	#11@12	#9@12	26' 1-WAY
		S	#5		
D	36	F	#11@12	#9@12	26' 1-WAY
		S	#5		
E	36	F	#10@12	#9@12	25' 1-WAY
		S	#5		
F	36	F	#10@12	#10@12	40 x 53
		S	#5		
G	36	F	#10@12	#10@12	40 x 40
		S	#5		
H	36	F	#10@12	#10@12	40 x 53
		S	#5		
I	36	F	#8@12	#9@12	40' x 40'
		S	#5		

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PROJECT NO.: 02-7092 SPONSOR: CEPL
SUBJECT: DATE: 1988 CHECKED BY: DATE CHECKED: 19



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SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: DISCUSSION OF METHODS USED
BY M. WHITNEY DATE: SEP 19 82 CHECKED BY: DATE CHECKED: 19

DISCUSSION OF METHODS USED

THE ANALYSIS BY SWRI INCLUDED
CONSIDERATION OF THE FOLLOWING ITEMS

1. INTERNAL EXPLOSION BLAST LOADS
2. WEAPONS AND MAINTENANCE BAY STRUCTURAL DESIGN
3. EXTERIOR SKIN STRUCTURAL DESIGN FOR AIRCRAFT IMPACT

1. INTERNAL BLAST LOADS

WHEN AN EXPLOSION OCCURS IN A CONFINED OR PARTIALLY
CONFINED STRUCTURE THE PRESSURE HISTORY EXPERIENCED BY
WALL AND ROOF SURFACES CONSISTS OF TWO DISTINCT PHASES.
THE FIRST PHASE IS THE SHOCK LOADING PHASE WHICH CONSISTS
OF A VERY SHORT DURATION, HIGH PRESSURE PULSE FOLLOWED BY
SEVERAL PRESSURE PULSES OF INCREASINGLY LOWER MAGNITUDE
AS THE BLAST WAVE REVERBERATES WITHIN THE STRUCTURE.
THIS FIRST PHASE IS A FUNCTION OF CHARGE WEIGHT AND
STANDOFF. THE SECOND PHASE IS CALLED THE QUASI-STATIC
PHASE AND IS A LONGER DURATION PRESSURE PULSE WHOSE PEAK
MAGNITUDE IS DETERMINED BY CHARGE WEIGHT AND ROOM VOLUME.
THE DURATION OF THE QUASI-STATIC PHASE IS FIXED BY THE AVAILABLE
VENT AREA, PEAK QUASI-STATIC PRESSURE, AND ROOM VOLUME.

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SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: DISCUSSION OF METHODS USED
BY: M. WHITNEY DATE: 25 Sept 87 CHECKED BY: DATE CHECKED: 19

SHOCK PHASE - FIGURES 1a, FROM REF [1] WERE USED TO DEFINE INTERNAL SHOCK LOADING PARAMETERS INCLUDING IMPULSE AND PRESSURE. FIGURES 1 AND 2 WERE USED TO CALCULATE IMPULSE LOADS AS FOLLOWS. FOR THE CALCULATION OF AVERAGE PEAK SHOCK PRESSURE OVER A PLATE, FIGURES 1 AND 4 WERE USED. FIGURE 4 WAS USED FOR VALUES OF X/R WITHIN THE RANGE SHOWN, AND FIGURE 1 WAS USED FOR Z/W (SIMILAR TO THAT DONE FOR IMPULSE) AND X/R GREATER THAN THAT SHOWN IN FIG. 4.

ONCE SHOCK IMPULSE AND PRESSURE WERE OBTAINED AS DESCRIBED ABOVE, A SHOCK DURATION, t_d , WAS SOLVED FOR ASSUMING A TRIANGULAR PULSE AS DESCRIBED IN IN REF. (1), i.e. $t_d = \frac{2i}{P}$. A FACTOR OF 1.75 WAS USED, WHERE APPLICABLE, TO P AND i TO ACCOUNT FOR SHOCK REVERBERATIONS AS SUGGESTED IN REF (1), AND t_d IS THEREBY KEPT THE SAME.

QUASI-STATIC PHASE - PEAK QUASI-STATIC PRESSURE AND DURATION WERE CALCULATED USING FIGURE 5 FROM DOE/TIC-11269⁽¹⁾. FIGURE 5 GIVES PEAK QUASI-STATIC PRESSURE AS A FUNCTION OF CHARGE WEIGHT AND ROOM VOLUME.

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BY: WITNEY DATE: 22 Sept 82 CHECKED BY: DATE CHECKED: 19

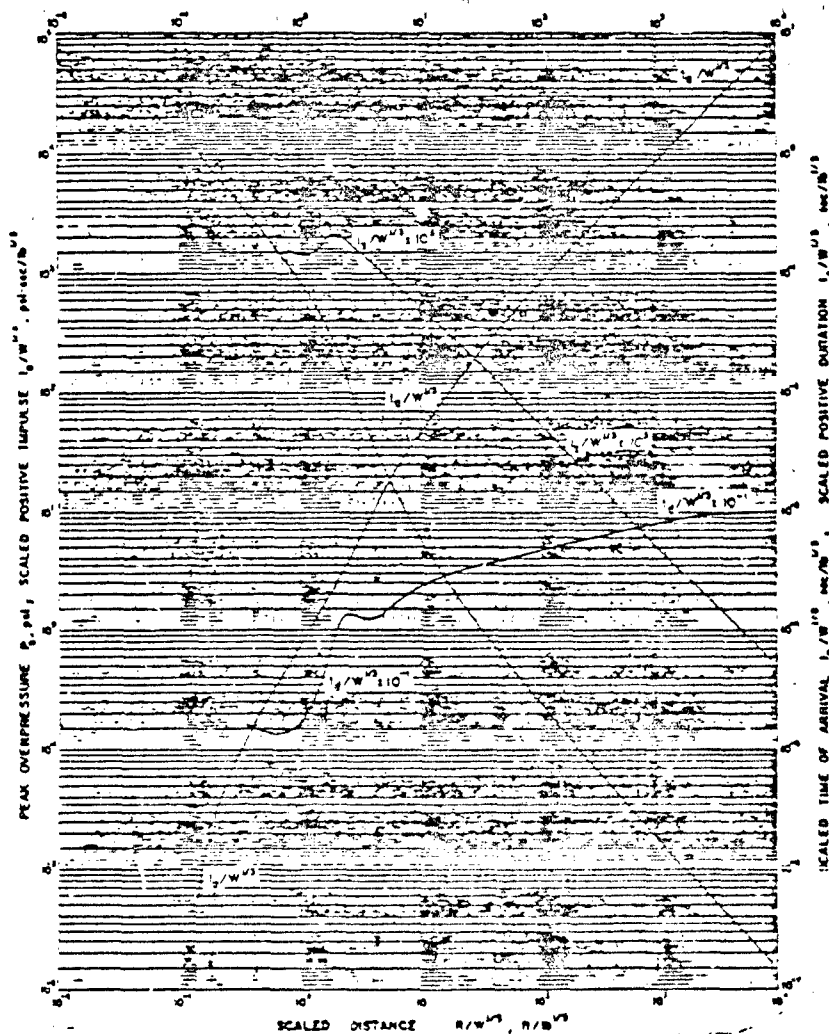


FIGURE 1 - SIDE-ON BLAST WAVE PROPERTIES FOR BARE SPHERICAL TNT
FOR AN EXPLOSION IN AIR
(FROM REF. 1, FIGURE 4.5)

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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: DISCUSSION
BY: H. W. T. DATE: 25 APR 87 CHECKED BY: DATE CHECKED: 19

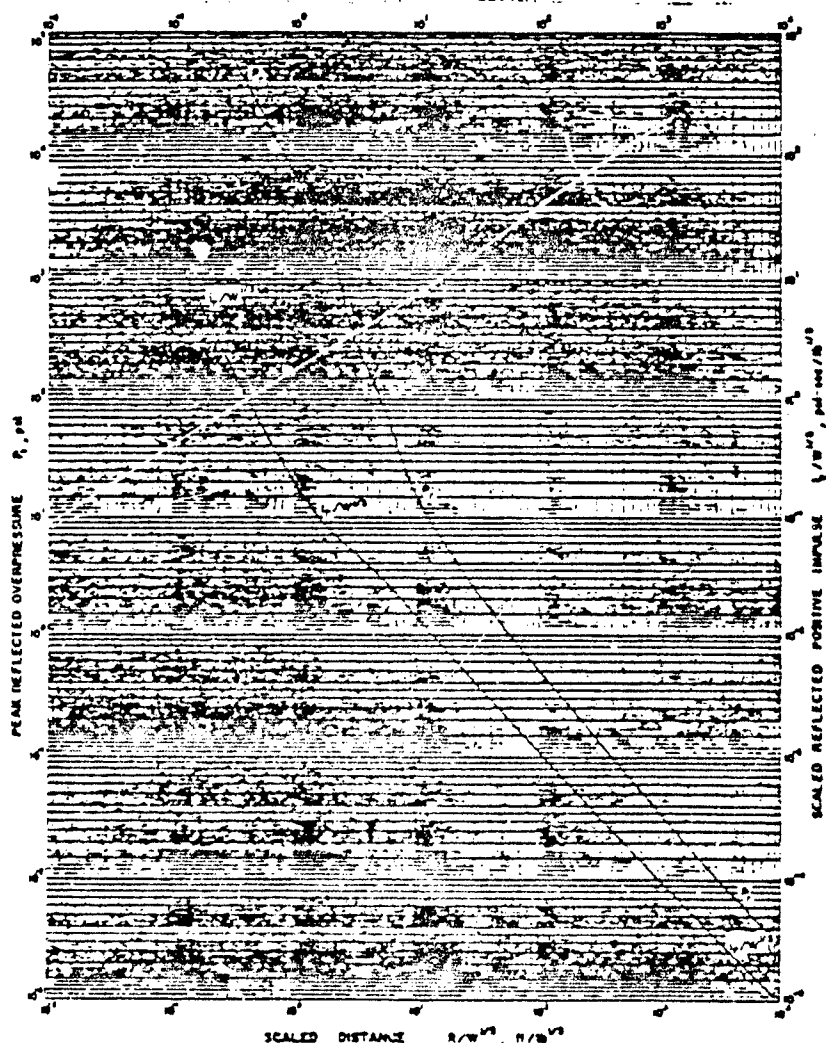


FIGURE 2. NORMALLY REFLECTED BLAST WAVE
PROPERTIES FOR BARE, SPHERICAL TNT
FOR AN EXPLOSION IN AIR
(REF. (1), FIGURE 4.6)

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PROJECT NO.: 02-7092

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DATE CHECKED: 19

$I/V^{1/3}$ VS X/R FOR VARIOUS Z VALUES

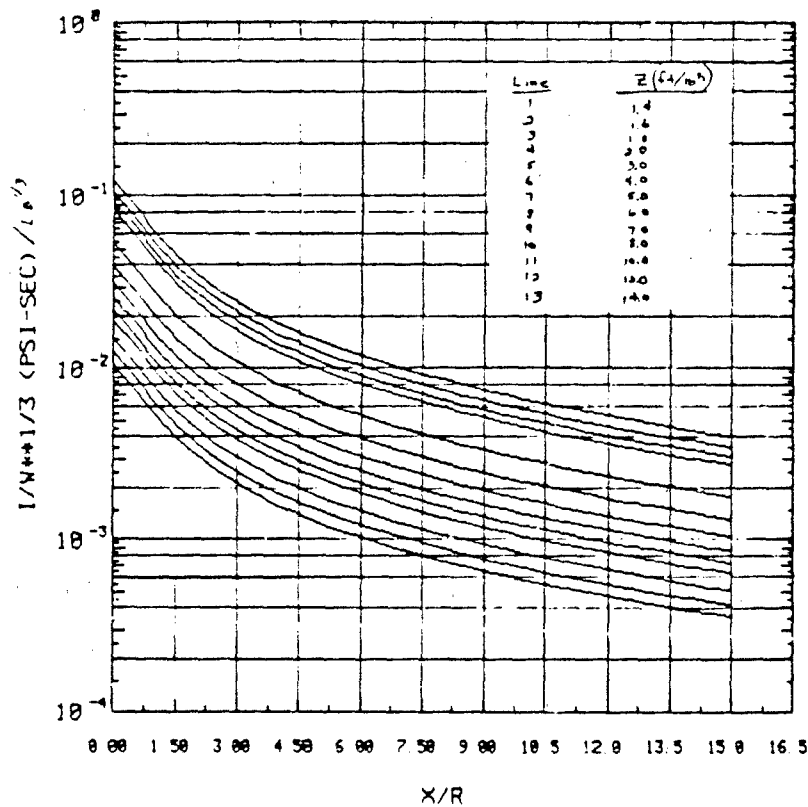


FIGURE 3a. Specific Impulse, $1.4 \leq Z \leq 14.0$

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SUBJECT: DISCUSSION
BY: M. WHITNEY DATE: 2 Sept 92 CHECKED BY: DATE CHECKED: 19

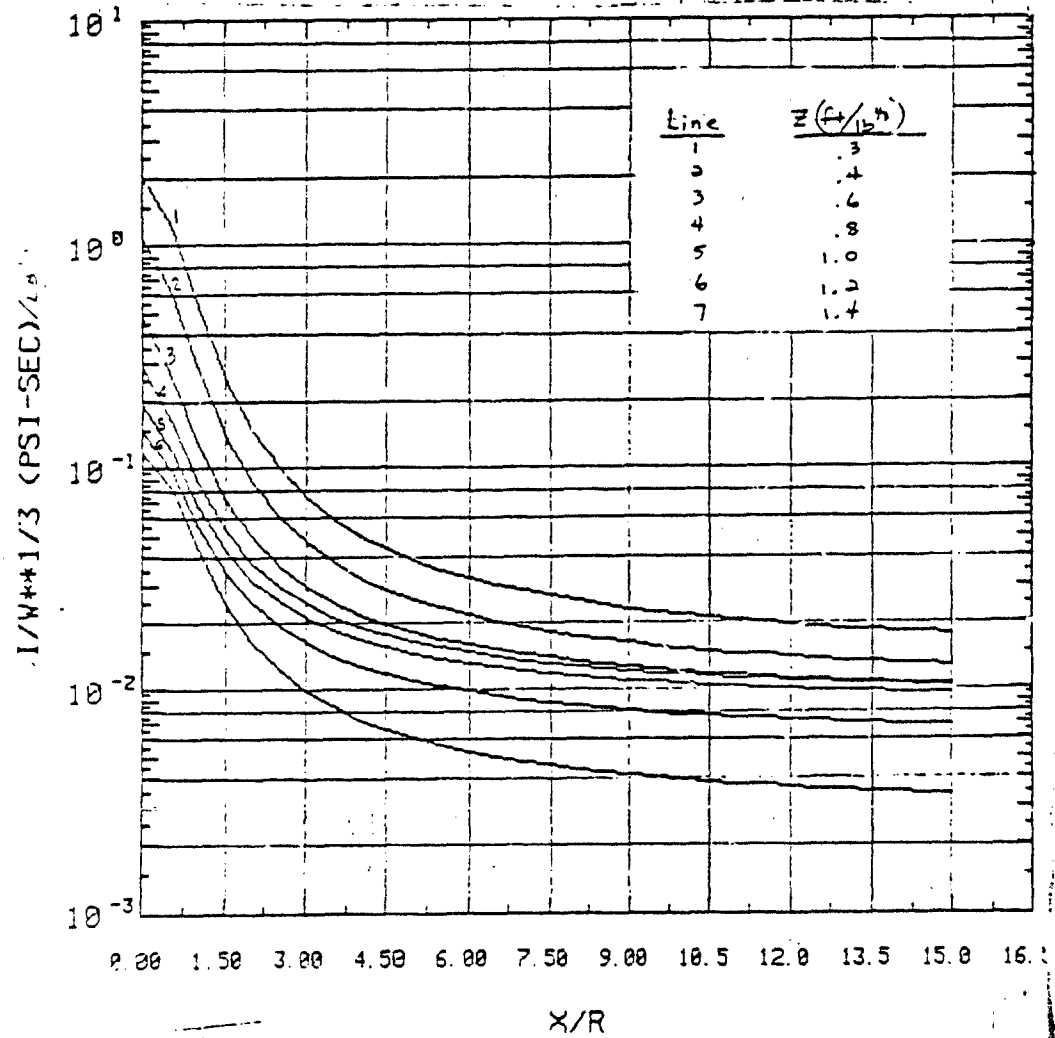


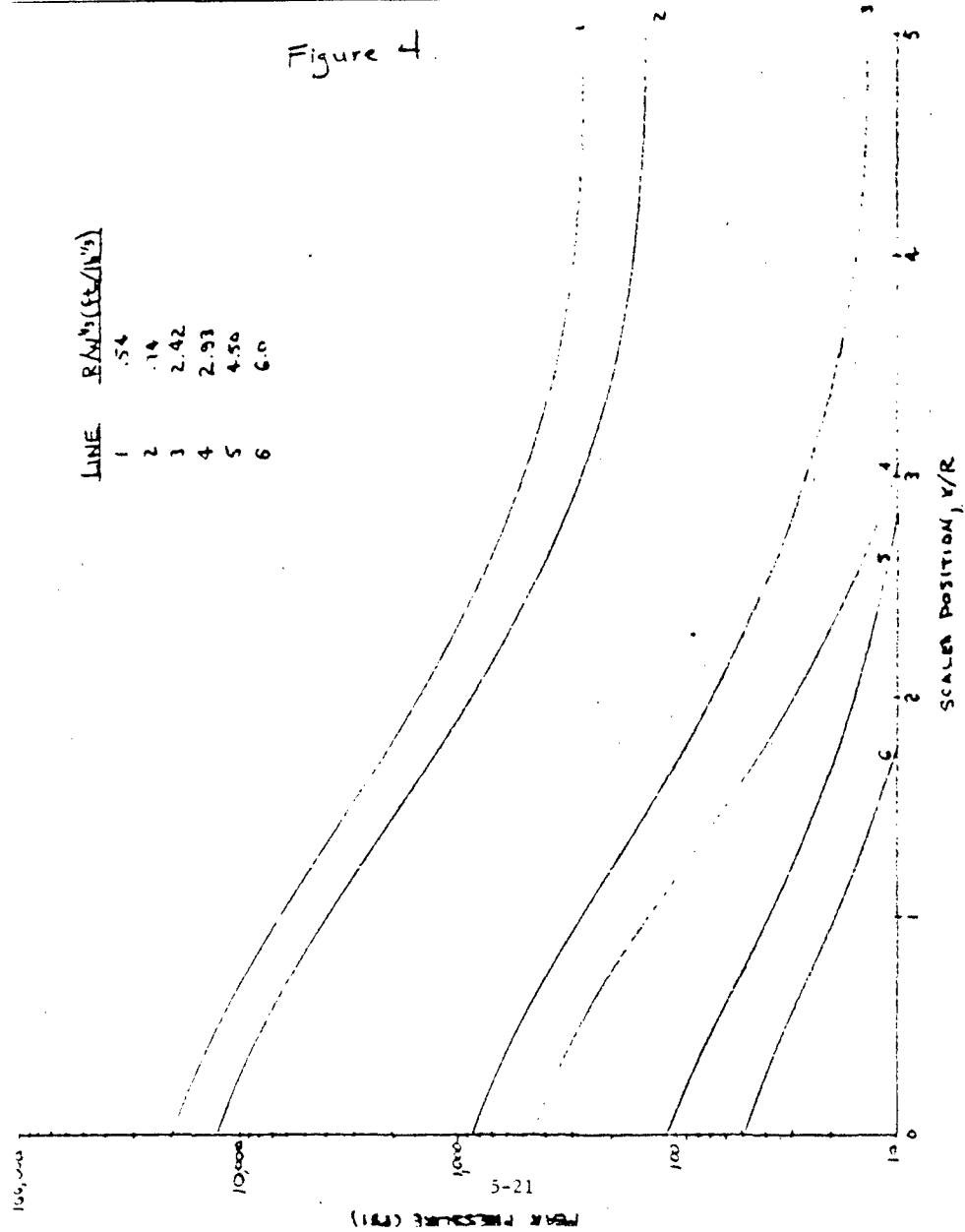
FIGURE 36. SPECIFIC IMPULSE, $0.35 \leq Z \leq 1.4$

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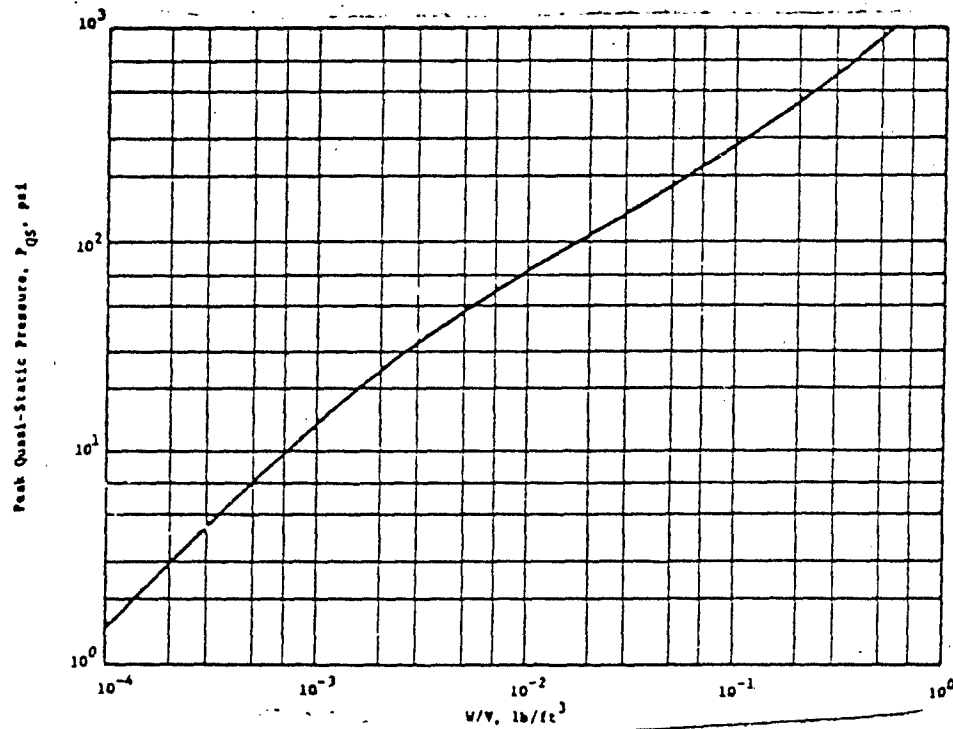
PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: PEAK PRESSURE VS. SCALED POSITION FOR DIFFERENT SCALED POSITIONS
BY: D. KETCHUM DATE: 30 SEPT 19 92 CHECKED BY: DATE CHECKED: 19

Figure 4



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Area	Length	Width	Area
1	43	141	6063
2	43	138	5934
3	43	136	5848
4	43	134	5762
5	43	132	5676
6	43	130	5590
7	43	128	5504
8	43	126	5418
9	43	124	5332
10	43	122	5246
11	43	120	5160
12	43	118	5074
13	43	116	4988
14	43	114	4902
15	43	112	4816
16	43	110	4730
17	43	108	4644
18	43	106	4558
19	43	104	4472
20	43	102	4386
21	43	100	4300
22	43	98	4214
23	43	96	4128
24	43	94	4042
25	43	92	3956
26	43	90	3870
27	43	88	3784
28	43	86	3698
29	43	84	3612
30	43	82	3526
31	43	80	3440
32	43	78	3354
33	43	76	3268
34	43	74	3182
35	43	72	3096
36	43	70	3010
37	43	68	2924
38	43	66	2838
39	43	64	2752
40	43	62	2666
41	43	60	2580
42	43	58	2494
43	43	56	2408
44	43	54	2322
45	43	52	2236
46	43	50	2150
47	43	48	2064
48	43	46	1978
49	43	44	1892
50	43	42	1806
51	43	40	1720
52	43	38	1634
53	43	36	1548
54	43	34	1462
55	43	32	1376
56	43	30	1290
57	43	28	1204
58	43	26	1118
59	43	24	1032
60	43	22	946
61	43	20	860
62	43	18	774
63	43	16	688
64	43	14	602
65	43	12	516
66	43	10	430
67	43	8	344
68	43	6	258
69	43	4	172
70	43	2	86
71	43	0	0



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PROJECT NO.: 02-7092

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SUBJECT: DISCUSSION

BY: T. WHITNEY DATE: 29 Sept 92

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BEFORE ANY BLAST CALCULATIONS ARE MADE,
THE SPECIFIED EXPLOSIVE WT IS INCREASED
BY A 1.2 FACTOR AS SUGGESTED IN
TMS-1300 [2].

2. WEAPONS AND MAINTANCE BAY STRUCTURAL DESIGN

THE WALLS WERE DESIGNED AS
FIXED - FIXED ONE WAY MEMBERS SPANNING
FROM FLOOR TO CEILING. DIVIDING WALLS
WERE TREATED EITHER AS FIXED-FIXED
OR CANTILEVER AS NECESSARY. 1° OF ROTATION
WAS CHOSEN FOR REUSABLE DESIGN FOR WALLS
AND 12° ROTATION FOR DIVIDING WALLS. THE
ROOF SLAB DESIGNS ARE SET BY THE AIRCRAFT
IMPACT EXCEPT FOR ONE CASE. THIS WAS
TREATED AS A FOUR-SIDED FIXED TWO WAY
SLAB. THE ONE WAY DESIGN WAS TREATED
AS FOLLOWS:

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PROJECT NO.: 02-7092

SPONSOR: CSTR

SUBJECT: DISCUSSION

BY: M. W. H. DATE: 29/06/52

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DATE CHECKED: 19

A COMPUTER PROGRAM, "BEAMDES", USING BIGGS' METHOD AS DESCRIBED IN TMS-1300¹ AND THE U.S. ARMY SUPPRESSIVE SHIELDS HANDBOOK², WAS WRITTEN AT SWRI TO DESIGN ONE-WAY REINFORCED CONCRETE MEMBERS. THIS METHOD CONVERTS A DYNAMICALLY LOADED STRUCTURE WITH DISTRIBUTED MASS AND LOADS, TO AN EQUIVALENT ONE DEGREE-OF-FREEDOM STRUCTURE. THE PROGRAM ACCOUNTS FOR THE COMBINED EFFECT OF THE SHOCK AND QUASI-STATE LOADS.

THE DESIGN PROCESS USED BY SWRI, EMPLOYING THE "BEAMDES" COMPUTER CODE, IS PRESENTED IN FLOWCHART FORM AS FIGURE 6. PRELIMINARY STRUCTURAL DESIGNS WERE MADE USING PRELIMINARY BLAST LOADS. THE "BEAMDES" CODE ENABLED SWRI TO DETERMINE THE REQUIRED STEEL AREAS FOR VARIOUS VALUES OF SUPPORT ROTATION, WALL THICKNESS AND SUPPORT CONDITIONS.

THE NEXT STEP WAS TO DESIGN THE LACING REINFORCEMENT AND DESIGN FOR SUPPORT REGIONS, ACCORDING TO TMS-1300¹. THESE PROCEDURES WILL NOT BE DESCRIBED IN DETAIL HERE SINCE CHAPTERS 5, 6 & 9 DISCUSS THESE METHODS FULLY USING EXAMPLE PROBLEMS.

DOCUMENTATION OF THE "BEAMDES" CODE FOLLOWS.

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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: DISCUSSION
BY: M. W. H. TROY DATE: 30 APR 82 CHECKED BY: DATE CHECKED: 19

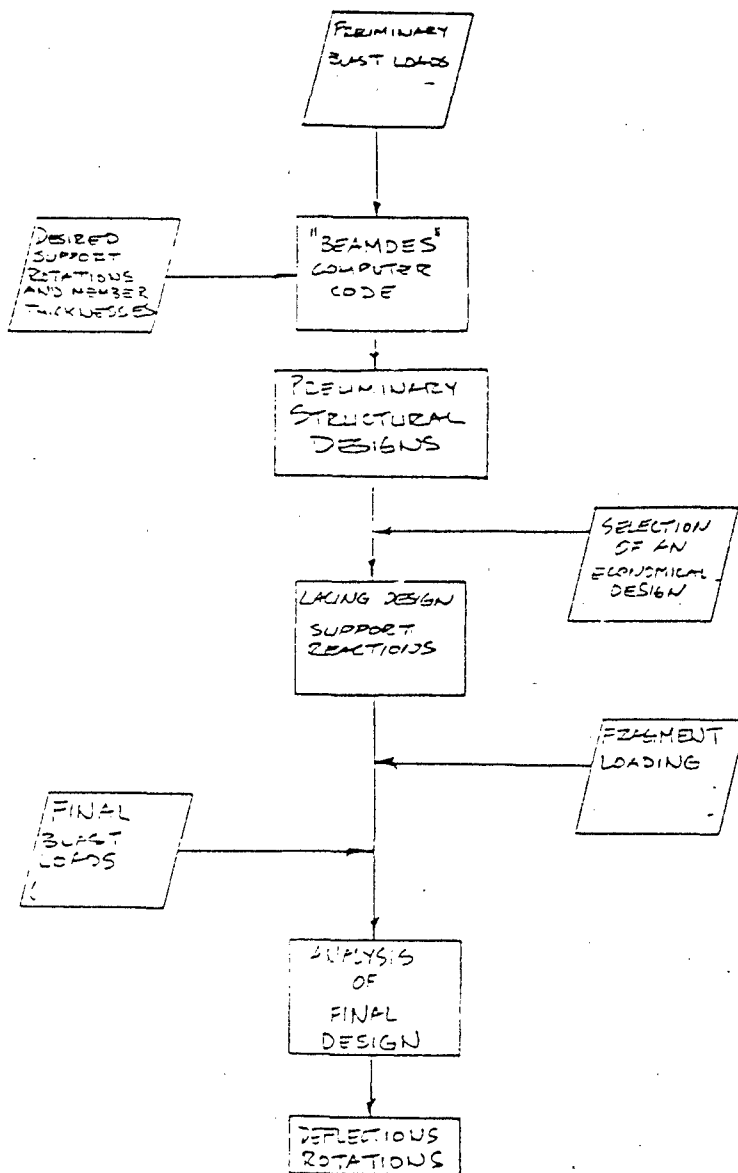


Fig 6 ONE-WAY REINFORCED CONCRETE DESIGN PROCESS

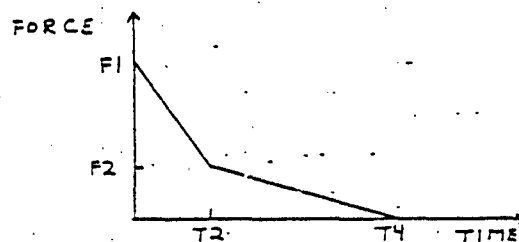
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: DISCUSSION
BY: B. B. B. DATE: 29 SEP 92 CHECKED BY: DATE CHECKED: 19

THE "BEAMDES" PROGRAM WAS WRITTEN TO PROVIDE A MEANS OF DESIGNING REINFORCED CONCRETE BEAMS AND ONE-WAY SLABS IN ACCORDANCE WITH TM 5-1300^(a). REQUIRED INPUT, COMPUTATIONAL SCHEME AND OUTPUT ARE DESCRIBED BELOW.

THE PROGRAM USES THE FORCE-TIME LOADING FUNCTION AS SHOWN. THE FORCES F_1 AND F_2 REPRESENT



FORCE-TIME FUNCTION

SENT THE PEAK SHOCK AND QUASI-STATIC PRESSURES, RESPECTIVELY, MULTIPLIED BY THE BEAM LENGTH, L , MULTIPLIED BY 1.0 INCH FOR A UNIT STRIP. END SUPPORT CONDITIONS OF FIXED-FIXED, FIXED-PINNED OR PINNED-PINNED ARE ALSO INPUT. MATERIAL PROPERTY INPUTS CONSIST OF THE CONCRETE CRUSHING STRENGTH, F_c' , AND THE STEEL DYNAMIC

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BY: B. M. M. M.

DATE: 29 SEP 19 82

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YIELD STRENGTH, F_{DY} . THE DESIRED MAXIMUM ROTATION IS INPUT, AS IS THE TOTAL THICKNESS OF THE CONCRETE SECTION AND THE THICKNESS OF COVER ABOVE OR BELOW THE TENSION AND COMPRESSION STEEL. THE MOMENT ARM OF THE REINFORCING STEEL, d_c , IS CALCULATED FROM THESE INPUTS. A DOUBLY REINFORCED CONCRETE SECTION IS ASSUMED WITH EQUAL TENSION AND COMPRESSION AREAS AND BAR STRENGTHS.

THE SOLUTION PROCEDURE IS AN ITERATIVE ONE IN WHICH THE AREA OF TENSION AND COMPRESSION STEEL IS VARIED BETWEEN 0.01 IN²/IN AND 3.0 IN²/IN OF BEAM/SLAB WIDTH. AN AVERAGE MOMENT OF INERTIA, I_A , FOR THE SECTION IS CALCULATED USING EQUATIONS 5-20, 5-21 AND 5-22 OF TMS-1300. THE ULTIMATE MOMENT CAPACITY OF THE SECTION, M_u , IS CALCULATED AS $M_u = A_s F_{DY} d_c$ PER EQN. 5-7 OF TMS-1300.⁽²⁾ THE ELASTIC AND ELASTIC-PLASTIC STIFFNESSES, K_{EL} AND K_{ELPL} , ARE THEN DETERMINED FOR THE APPROPRIATE END SUPPORT CONDITIONS USING TABLE 5-11 OF TMS-1300.⁽²⁾ THE CODE THEN CALCULATES THE

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SUBJECT: DISCUSSION

BY: B. Evans

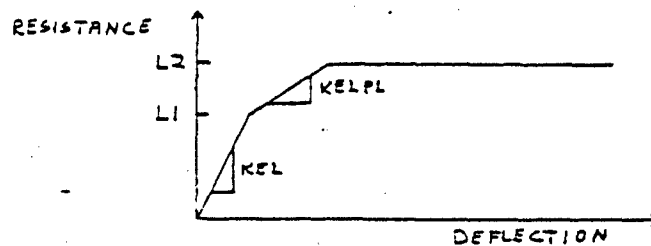
DATE: 23 SEP 87

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ELASTIC AND PLASTIC MAXIMUM RESISTANCES, L_1 AND L_2 , BASED ON TABLES FOUND IN "SUPPRESSIVE SHIELDS",³ HNDM-1110-1-2, AND BIGGS' INTRODUCTION TO STRUCTURAL DYNAMICS.⁴ THE TOTAL RESISTANCE-DEFLECTION RELATIONSHIP IS SHOWN BELOW. APPROPRIATE



RESISTANCE - DEFLECTION CURVE

LOAD-MASS FACTORS, K_{LM} , ARE ALSO DETERMINED BASED ON BIGGS⁴ AND TMS-1300.²

ONCE ALL OF THE ABOVE INPUT PARAMETERS FOR A SINGLE-DEGREE-OF-FREEDOM SIMULATION ARE ESTABLISHED, THE CODE CALCULATES THE MAXIMUM DEFLECTION OF THE BEAM/SLAB USING A NUMERICAL TECHNIQUE OUTLINED IN BIGGS. THIS CALCULATED MAXIMUM DEFLECTION IS COMPARED WITH THE DESIRED DEFLECTION, AND, IF IT DIFFERS BY MORE THAN 1%, THE AREA OF STEEL IS ADJUSTED

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BY: R. M. M. M.

DATE: 20 SEP 91

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AND THE CALCULATIONS REPEATED. ONCE THE CALCULATED AND DESIRED DEFLECTIONS DIFFER BY LESS THAN 1%, THE REQUIRED STEEL AREA AND OTHER INPUT AND SECTION PROPERTIES ARE OUTPUT BY THE CODE.

THE PROGRAM USED FOR THE TWO-WAY SLAB DESIGN IS SIMILAR TO THAT DISCUSSED ABOVE.

THE NEXT PAGE INCLUDES A SUMMARY OF CONCRETE PROPERTIES USED IN THE ANALYSIS.

THE DESIGN METHODOLOGY FOR BLAST DOOR DESIGN IS BASED ON METHODS DESCRIBED IN REF [3]. THE BLAST DOOR DESIGN CALCULATIONS ARE SELF EXPLANATORY AS ATTACHED. REFER TO THE CALCULATIONS SHEETS FOR A DISCUSSION OF METHODS USED.

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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: DISCUSSION
BY: M. WHITNEY DATE: 22 SEP 19 82 CHECKED BY: DATE CHECKED: 19

SUMMARY OF MATERIAL PROPERTIES:

CONCRETE:

$$f'_c = 4000 \text{ psi}, E_c = 3.62 \times 10^6 \text{ psi}$$

REINFORCING STEEL:

$$f_y = 60,000 \text{ psi}, f_u = 90,000 \text{ psi}, E_s = 29 \times 10^6 \text{ psi}$$

FOR SUPPORT ROTATION $0^\circ < \theta < 2^\circ$

$$\text{STATIC DESIGN STRESS}, f_s = f_y = 60,000 \text{ psi}$$

FOR SUPPORT ROTATION $2^\circ < \theta < 5^\circ$

$$\text{STATIC DESIGN STRESS}, f_s = f_y + \frac{1}{4}(f_u - f_y)$$

$$f_s = 60 + \frac{1}{4}(90 - 60)$$

$$f_s = 67,500 \text{ psi}$$

FOR SUPPORT ROTATION $\theta \geq 5^\circ$

$$\text{STATIC DESIGN STRESS}, f_s = \frac{1}{2}(f_u + f_y)$$

$$f_s = \frac{1}{2}(60 + 90) = 75,000 \text{ psi}$$

DYNAMIC INCREASE FACTOR (DIF) = 1.2

$$f_{dy} = (\text{DIF}) f_s$$

$$\text{FOR } 0^\circ < \theta < 2^\circ, f_{dy} = 1.2(60,000) = 72,000 \text{ psi}$$

$$\text{FOR } 2^\circ < \theta < 5^\circ, f_{dy} = 1.2(67,500) = 81,000 \text{ psi}$$

$$\text{FOR } \theta \geq 5^\circ, f_{dy} = 1.2(75,000) = 90,000 \text{ psi}$$

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CER
SUBJECT: EXTERIOR SHELL DESIGNS
BY: DATE 19 CHECKED BY: DATE CHECKED: 25 SEP 62

3. EXTERIOR SHELL STRUCTURAL DESIGN FOR AIRCRAFT IMPACT

THE ANALYSIS DETAILS WHICH FOLLOW WERE PRODUCED USING A NUMBER OF IN-HOUSE DEVELOPED COMPUTER CODES DESIGNED TO NUMERICALLY INTEGRATE THE EQUATIONS OF MOTION SUGGESTED IN BIGGS' INTRODUCTION TO STRUCTURAL DYNAMICS OR TO PERFORM OTHER MATHEMATICAL COMPUTATIONS. CODES USED ARE DESCRIBED BELOW:

'BEAMDOS' - ITERATIVE DESIGN OF REINFORCED CONCRETE ONE WAY SLABS OR BEAMS SUBJECTED TO TIME-VARYING PRESSURE LOADS. IT USES BIGGS' SINGLE-DEGREE-OF-FREEDOM MODEL WITH STRUCTURAL PARAMETERS AS CALCULATED BY BIGGS OR TMS-1300.

'SLABDES' - SIMILAR TO ABOVE BUT FOR TWO-WAY REINFORCED CONCRETE SLABS.

'ARCH' - CALCULATES REACTIONS FOR FIXED-END ARCHES BASED ON ARCH SPAN, THICKNESS, AND RISE. GOVERNING EQUATIONS ARE FOUND IN FORMULAS FOR STRESS AND STRAIN BY ROARK AND YOUNG.

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COMPUTATION SHEET

SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CECIL
SUBJECT: EXTERNAL SHELL DESIGNS
BY: DATE: 19 CHECKED BY: D. K. T. DATE CHECKED: 25 SEP 19 82

'ONEDF' - SINGLE-DEGREE-OF-FREEDOM MAXIMUM RESPONSE OF
AN UNDAMPED MASS-SPRING SYSTEM TO A
SPECIFIED TIME-VARYING LOAD. EQUATIONS OF
MOTION ARE FROM BIGGS.

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COMPUTATION SHEET

SHEET NO.
19 OF 19

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: EXTERIOR SHELL DESIGN
BY: DATE: 19 CHECKED BY: SKETCH DATE CHECKED: 25 OCT 82

REFERENCES

- (1) DOE/TIC-11268, "A MANUAL FOR PREDICTION OF BLAST AND FRAGMENT LOADINGS ON STRUCTURES," DEPARTMENT OF ENERGY, AMARILLO AREA OFFICE, PANTEX PLANT, NOVEMBER, 1980.
- (2) TMS-1300, "STRUCTURES TO RESIST THE EFFECTS OF ACCIDENTAL EXPLOSIONS," DEPT OF THE ARMY TECH. MANUAL TMS-1300, DEPT. OF THE NAVY PUBLICATION NAVFAC P-397, DEPARTMENT OF THE AIR FORCE MANUAL AFM 88-22, JUNE 1969.
- (3) "Suppressive Shields Structural Design and Analysis Handbook," HNDM-1110-1-2, U.S. Army Corps of Engineers, Huntsville Division, 18 Nov 1977.
- (4) Biggs, J.M. Introduction to Structural Dynamics, The McGraw-Hill Book Co., New York, New York, 1964.

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
1 OF 47

PROJECT NO.: DA-7092 SPONSOR: CERL
SUBJECT: BLAST ANALYSIS
BY: NRS DATE: 23 Sep 19 82 CHECKED BY: _____ DATE CHECKED: 19

III BLAST ANALYSIS

The following pages include interior blast load calculations for the 4 munitions storage bay designs and a typical maintenance bay design. The blast analysis was made using methods described in DOE/TIC-11268. The blast analysis is organized as follows:

	<u>Page</u>
1. Summary	5-35
2. Internal Blast Loading	5-36
2.1 Shock Loading	5-36
2.1.1 Long Bay Design	5-37
2.1.2 Pits Bay Design	5-45
2.1.3 Maze Bay Design	5-54
2.1.4 Double-stacked Bay Design	5-61
2.1.5 Maintenance Bay Design	5-69
2.2 Quasi-static Loading	5-75
2.2.1 Long Bay Design	5-76
2.2.2 Pits Bay Design	5-77
2.2.3 Maze Bay Design	5-78
2.2.4 Double-stacked Design	5-79
2.2.5 Maintenance Bay Design	5-80

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PROJECT NO.: 02-7092 SPONSOR: CEK
SUBJECT: BLAST ANALYSIS
BY: NRS DATE: 29 Sep 19 82 CHECKED BY: _____ DATE CHECKED: 19

1. Summary

Table 1 - Summary of Internal Loads

Bay Description	P (psi)	i (psi-sec)	t ₁ (sec)	Pas (psi)
<u>Long Bay Design</u>				
wall A	3673	0.68	0.00037	20
wall B	4073	0.72	0.00035	
wall C	79	0.25	0.0063	
dividing wall	4073	0.72	0.00035	
ceiling	833	0.42	0.001	
door	100	0.23	0.0046	
<u>P+3 Bay Design</u>				
wall A	1500	0.51	0.00068	17
wall B	3200	0.58	0.00036	
dividing wall	12,250	1.14	0.00019	
ceiling	125	0.25	0.0039	
door	75	0.28	0.0032	
<u>Maze Bay Design</u>				
wall A	1930	0.56	0.00061	19
wall B	2170	0.44	0.00040	
dividing wall	12,250	1.14	0.00019	
ceiling	200	0.28	0.0027	
door	470	0.35	0.0015	
<u>Corridor-Stacked Bay</u>				
wall A	2632	0.53	0.0004	28
wall B	5780	0.95	0.00033	
wall C	69	0.23	0.0063	
dividing wall	5750	0.95	0.00033	
ceiling	780	0.30	0.00077	
door	100	0.23	0.0046	
<u>Maintenance Bay</u>				
wall	2770	0.42	0.00036	37
ceiling	238	0.32	0.0027	
door	17,348	1.19	0.00014	

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SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEETSHEET NO.
3 OF 47PROJECT NO.: 92-7092 SPONSOR: CERL
SUBJECT: BLAST ANALYSIS
BY: NRS DATE: 29 Sep 82 CHECKED BY: DATE CHECKED: 192 Internal Blast Loading

The following pages include calculations of shock and quasi-static loads determined using methods described in Section II.

2.1 Shock Loading

Shock loads were calculated in each of the four storage bays for the 9.7 ft x 3.1 ft x 3.3 ft munition should there be a detonation. Shock loads were also calculated for a typical maintenance bay. By sponsor instruction, no transport scenarios were analyzed. The worst case shock pressure, impulse and duration are summarized in Table 1 of Section I. The charge weight of 78.5 lb_{TNT} is increased by a factor of 1.2 as specified in TMS-1300. When the charge is considered as located on the floor, the charge weight is doubled for use with air blast curves.

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PROJECT NO.: 02-7092-001 SPONSOR: CERL
SUBJECT: BLAST ANALYSIS
BY: WRS DATE: 21 Sep 19 82 CHECKED BY: DATE CHECKED: 19

2.1.1 Shock Loadings for Long Bay

Summary of Blast Loads - Long Bay			
Component	P (psi)	i (psi-sec)	t _a (sec)
wall A	3673	0.68	0.00037
wall B	4078	0.72	0.00035
wall C	79	0.25	0.0063
dividing wall	4078	0.72	0.00035
ceiling	833	0.42	0.001
door	100	0.23	0.0046

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SHEET NO.

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PROJECT NO: 22-7092

SPONSOR: CERL

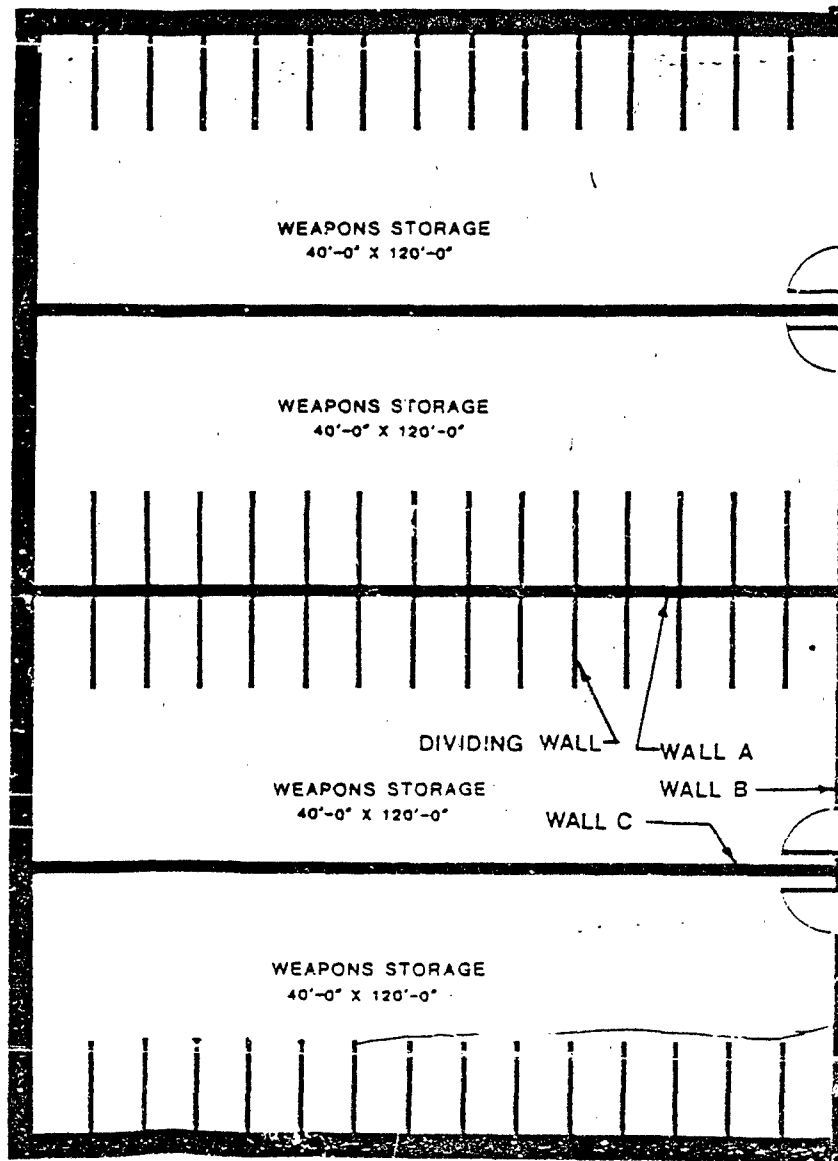
SUBJECT: Floor plan - Long Ray Concept No. 4

BY: DEL

DATE: 24 SEP 82

CHECKED BY: NRS

DATE CHECKED: 24 Sep 19 82



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COMPUTATION SHEET

SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: BLAST ANALYSIS
BY: NRS DATE: 29 82 CHECKED BY: D. ECKED: 19

WALL A - Long Bay

INPUT PARAMETERS.

CHARGE WEIGHT = 188.40 LB
PERPENDICULAR STANDOFF = 7.00 FT
PLATE LENGTH, XTOT = 7.00 FT
PLATE WIDTH, YTOT = 12.00 FT

CHARGE POSITION IN PLANE PARALLEL TO PLATE

LENGTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, XC = 3.50 FT
WIDTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, YC = 0.00 FT

OUTPUT:

AVERAGE PRESSURE = 0.2099E+04 PSI
TOTAL FORCE ON WALL = 0.2539E+08 LB
TOTAL APPLIED IMPULSE = 0.4694E+04 LB-SEC
AVERAGE SPECIFIC IMPULSE = 0.39 PSI-SEC

$$P_{AVG} = 2099 \times 1.75 = 3673 \text{ psi}$$

$$I_{AVG} = 0.39 \times 1.75 = 0.68 \text{ psi-sec}$$

$$t_1 = \frac{2L}{F} = \frac{2(65)}{3673} = 0.0037 \text{ sec}$$

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COMPUTATION SHEET

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PROJECT NO.: 02-7092

SPONSOR: CERL

SUBJECT: Blast Analysis

BY: NRS

DATE: 29 Sep 19 82

CHECKED BY:

DATE CHECKED:

19

WALL B + dividing wall - Long Bay

INPUT PARAMETERS:

CHARGE WEIGHT = 198.40 LB
PERPENDICULAR STANDOFF = 3.50 FT
PLATE LENGTH, XTOT = 12.00 FT
PLATE WIDTH, YTOT = 12.00 FT

CHARGE POSITION IN PLANE PARALLEL TO PLATE

LENGTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, XC = 6.00 FT
WIDTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, YC = 0.00 FT

OUTPUT:

AVERAGE PRESSURE = 0.2330E+04 PSI
TOTAL FORCE ON WALL = 0.4830E+08 LB
TOTAL APPLIED IMPULSE = 0.8576E+04 LB-SEC
AVERAGE SPECIFIC IMPULSE = 0.41 PSI-SEC

$$P_{avg} = 2330 \times 1.75 = 4078 \text{ psi}$$

$$I_{avg} = .41 \times 1.75 = 0.72 \text{ psi-sec}$$

$$\tau = \frac{2(0.72)}{4078} = 0.00035 \text{ sec}$$

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COMPUTATION SHEET

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PROJECT NO.: 02-7092

SPONSOR: CERL

SUBJECT: PLAST ANALYSIS

BY: NRS

DATE: 29 Sep 82

CHECKED BY:

DATE CHECKED: 19

WALL C - Long Bay

INPUT PARAMETERS:

CHARGE WEIGHT = 188.40 LB
PERPENDICULAR STANDOFF = 33.00 FT
PLATE LENGTH, XTOT = 7.00 FT
PLATE WIDTH, YTOT = 13.00 FT

CHARGE POSITION IN PLANE PARALLEL TO PLATE

LENGTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, XC = 3.50 FT
WIDTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, YC = 0.00 FT

OUTPUT:

AVERAGE PRESSURE = 0.4500E+02 PSI
TOTAL FORCE ON WALL = 0.5897E+06 LB
TOTAL APPLIED IMPULSE = 0.1795E+04 LB-SEC
AVERAGE SPECIFIC IMPULSE = 0.14 PSI-SEC

$$p_{avg} = 75 \times 1.75 = 79 \text{ psi}$$

$$i_{avg} = .14 \times 1.75 = 0.25 \text{ psi-sec}$$

$$t_d = \frac{2.0}{p} = \frac{2(.25)}{79} = 0.0063 \text{ sec}$$

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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: BLAST ANALYSIS
BY: NRS DATE: 29 Sep 82 CHECKED BY: _____ DATE CHECKED: 19

CEILING - Long Bay

INPUT PARAMETERS:

CHARGE WEIGHT = 188.40 LB
PERPENDICULAR STANDOFF = 12.00 FT
PLATE LENGTH, XTOT = 7.00 FT
PLATE WIDTH, YTOT = 40.00 FT

CHARGE POSITION IN PLANE PARALLEL TO PLATE

LENGTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, XC = 3.50 FT
WIDTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, YC = 7.00 FT

OUTPUT:

AVERAGE PRESSURE = 0.4762E+03 PSI
TOTAL FORCE ON WALL = 0.1920E+09 LB
TOTAL APPLIED IMPULSE = 0.9602E+04 LB-SEC
AVERAGE SPECIFIC IMPULSE = 0.24 PSI-SEC

$$P_{avg} = 76 \times 1.75 = 133 \text{ psi}$$

$$I_{avg} = 0.24 \times 1.75 = 0.42 \text{ psi-sec}$$

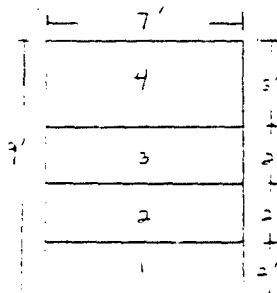
$$t_d = \frac{2.1}{r} = \frac{2.1}{533} = 0.0010 \text{ Sec}$$

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PROJECT NO.: 22-7092 SPONSOR: CERL
SUBJECT: BLAST ANALYSIS
BY: NRS DATE: 2 Aug 82 CHECKED BY: TKM DATE CHECKED: 17 Aug 82

Find the average pressure on the 7' x 9' door
of the Long Bay Design for a charge located on
the floor (assuming no dividing wall); charge is
placed 24.5 ft horizontally from center of door



$w = 78.5 \text{ lb}$
 $R = 11.5 \text{ ft}$ (worst case impulse on
door occurred for 2nd stall)

charge wt = $w \times 1.2 \times 2$ on floor

$w = 188.4 \text{ lb}$ and

$z = \frac{R}{w^{1/3}} = 2.0 \text{ ft/lb}^{1/3}$

For each wall
element shown, tabulate
 x/R , A and P_r

Element	$x \text{ (ft)}$	x/R	$A \text{ (ft}^2\text{)}$	$P_r \text{ (psi)}$	$P_r A \text{ (psi ft}^2\text{)}$
1	24.5	2.1	14	80	1120
2	24.7	2.1	14	80	1120
3	25.0	2.2	14	40	560
4	25.6	2.2	21	40	840

$\Sigma = 63 \text{ ft}^2$

$\Sigma = 3640 \text{ psi ft}^2$

$$P = \frac{3640}{63} = 58 \text{ psi} \times 1.75 = \underline{\underline{100 \text{ psi}}}$$

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PROJECT NO. 02-7092 SPONSOR: CERL
SUBJECT: BLAST ANALYSIS
BY: NRS DATE: 25 APR 82 CHECKED BY: _____ DATE CHECKED: 19

7X9 DOOR - Bay

INPUT PARAMETERS:

CHARGE WEIGHT = 188.40 LB
PERPENDICULAR STANDOFF = 11.50 FT
PLATE LENGTH, XTOT = 7.00 FT
PLATE WIDTH, YTOT = 9.00 FT

CHARGE POSITION IN PLANE PARALLEL TO PLATE

LENGTHWISE DISTANCE FROM
PLATE CORNER, XC = 28.00 FT
WIDTHWISE DISTANCE FROM
PLATE CORNER, YC = 0.00 FT

OUTPUT:

TOTAL APPLIED IMPULSE = 0.1176E+04 LB-SEC
AVERAGE SPECIFIC IMPULSE = 0.13 PSI-SEC

$$i_{avg} = .13 \times 1.75 = 0.23 \text{ psi-sec}$$

$$t_d = \frac{20}{F} = \frac{2(.23)}{100} = 0.0046 \text{ sec}$$

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PROJECT NO: 02-7092 SPONSOR: CERL
SUBJECT: BLAST ANALYSIS
BY: ERS DATE 22 SEP 19 82 CHECKED BY: _____ DATE CHECKED: 19

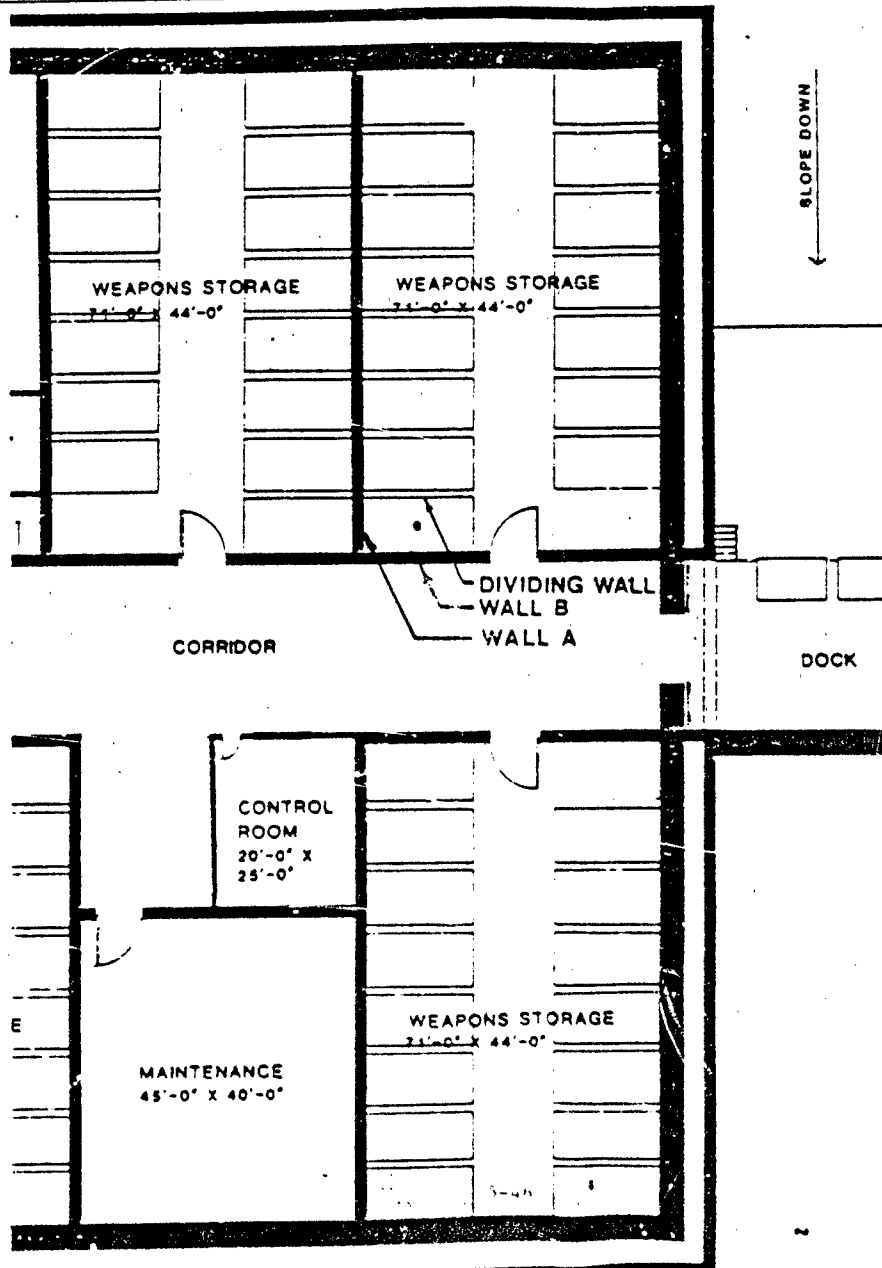
2.1.2 Shock Loadings for Pits Bay Design

Summary of Blast Loads - Pits			
Component	P_{ps}	$i(\text{psi-sec})$	$t_d(\text{sec})$
Wall A	1500	0.51	0.00068
Wall B	3200	0.58	0.00036
Ceiling	125	0.25	0.0031
loading wall	2250	1.14	0.00019
door	175	0.28	0.0032

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PROJECT NO.: 02-7091 SPONSOR: CER
SUBJECT: Floor Plan - 61+S - Concept No. 2
BY: Dey DATE: 24 SEP 82 CHECKED BY: NRS DATE CHECKED: 24 Sep 82



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COMPUTATION SHEET

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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: Blast Analysis
BY: NRS DATE: 30 Sep 19 82 CHECKED BY: _____ DATE CHECKED: _____ 19 _____

WALL A - Pits Bay

INPUT PARAMETERS:

CHARGE WEIGHT = 189.40 LB
PERPENDICULAR STANDOFF = 8.50 FT
PLATE LENGTH, XTOT = 8.00 FT
PLATE WIDTH, YTOT = 25.00 FT

CHARGE POSITION IN PLANE PARALLEL TO PLATE

LENGTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, XC = 4.00 FT
WIDTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, YC = 0.00 FT

OUTPUT

AVERAGE PRESSURE = 0.8574E+03 PSI
TOTAL FORCE ON WALL = 0.2469E+08 LB
TOTAL APPLIED IMPULSE = 0.8292E+04 LB-SEC
AVERAGE SPECIFIC IMPULSE = 0.29 PSI-SEC

PLATE MASS = 1500 LB

PLATE AREA = 200.00 SQ FT

PLATE THICKNESS = 0.005 FT

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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: BLAST ANALYSIS
BY: NRS DATE: 20 SEP 19 82 CHECKED BY: DEK DATE CHECKED: 30 SEP 19 82

WALL B - P1+3 Bay

INPUT PARAMETERS:

CHARGE WEIGHT = 198 40 LB
PERPENDICULAR STANDOFF = 4.00 FT
PLATE LENGTH, XTOT = 17.00 FT
PLATE WIDTH, YTOT = 25.00 FT

CHARGE POSITION IN PLANE PARALLEL TO PLATE

LENGTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, XC = 8.50 FT
WIDTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, YC = 12.50 FT

OUTPUT

AVERAGE PRESSURE = 0.1827E+04 PSI
TOTAL FORCE ON WALL = 0.1116E+09 LB
TOTAL APPLIED IMPULSE = 0.2046E+05 LB-SEC
AVERAGE SPECIFIC IMPULSE = 0.33 PSI-SEC

PLATE STRESS =

PLATE DEFLECTION =

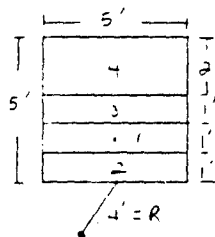
PLATE VIBRATION =

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PROJECT NO.: DD-7092 SPONSOR: CERL
SUBJECT: BLAST ANALYSIS
BY: NKS DATE: 18 Aug 19 82 CHECKED BY: TRM DATE CHECKED: 18 Aug 19 82

Find the average peak pressure on a dividing wall of Pits Bay for a charge located in the air



$$w = 78.5 \text{ lb}$$

$$R = 4.0 \text{ ft}$$

$$\text{charge wt} = w \times 1.2 \text{ in air}$$

$$W = 94.2 \text{ lb and}$$

$$Z = R/W^{1/3} = 0.88 \text{ ft/lb}^{1/3}$$

For each wall element shown tabulate x/R , A , and P_r

Element	$x(\text{ft})$	x/R	$A(\text{ft}^2)$	$P_r(\text{psi})$	$PA(\text{psi} \cdot \text{ft}^2)$
1	0	—	5	9000*	45000
2	1	0.25	5	8000	40000
3	1	0.25	5	8000	40000
4	2.5	0.63	10	5000	50000

* @ $x/R = 0$, Fig 2 was used

$$Z = 25 \text{ ft}^3$$

$$Z = 175000 \text{ psi} \cdot \text{ft}^3$$

$$P = \frac{175000}{25} = 7000 \text{ psi} \times 1.75 = \underline{\underline{12,250 \text{ psi}}}$$

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PROJECT NO.: DA-7092 SPONSOR: CERL
SUBJECT: BLAST ANALYSIS
BY: NRS DATE: 30 Sep 82 CHECKED BY: _____ DATE CHECKED: _____ 19__

DIVIDING WALL - Pits Bay

INPUT PARAMETERS:

CHARGE WEIGHT = 94.20 LB
PERPENDICULAR STANDOFF = 4.00 FT
PLATE LENGTH, XTOT = 5.00 FT
PLATE WIDTH, YTOT = 5.00 FT

CHARGE POSITION IN PLANE PARALLEL TO PLATE

LENGTHWISE DISTANCE FROM
PLATE CORNER, XC = 2.50 FT
WIDTHWISE DISTANCE FROM
PLATE CORNER, YC = 1.50 FT

OUTPUT:

TOTAL APPLIED IMPULSE = 0.2344E+04 LB-SEC
AVERAGE SPECIFIC IMPULSE = 0.65 PSI-SEC

$$C = 1.75(0.65) = 1.14 \text{ PSI-SEC}$$

$$t_A = \frac{Z_c}{P} = \frac{2(1.14)}{12250} = 0.00019 \text{ sec}$$

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COMPUTATION SHEET

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PROJECT NO.: 02-10920 SPONSOR: CERL
SUBJECT: Blast Analysis
BY: NRS DATE: 30 Sep 82 CHECKED BY: DER DATE CHECKED: 30 Sep 82

CEILING - Pits Bay

INPUT PARAMETERS:

CHARGE WEIGHT = 199.40 LB
PERPENDICULAR STANDOFF = 25.00 FT
PLATE LENGTH, XTOT = 8.00 FT
PLATE WIDTH, YTOT = 44.00 FT

CHARGE POSITION IN PLANE PARALLEL TO PLATE

LENGTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, XC = 4.00 FT
WIDTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, YC = 25.50 FT

OUTPUT:

AVERAGE PRESSURE = 0.7148E+02 PSI
TOTAL FORCE ON WALL = 0.3623E+07 LB
TOTAL APPLIED IMPULSE = 0.7254E+04 LB-SEC
AVERAGE SPECIFIC IMPULSE = 0.14 PSI-SEC

$$P_{wall} = 35 \text{ PSI}$$

$$t_{wall} = 0.245 \text{ sec}$$

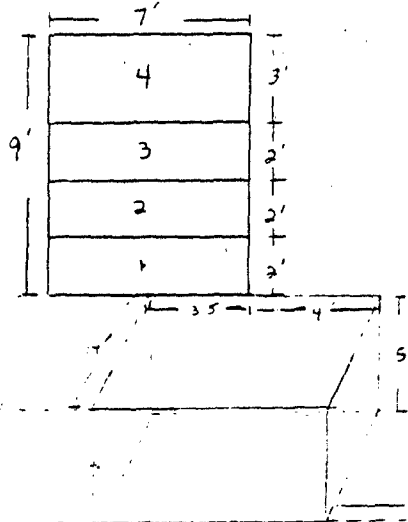
$$t_d = 3.9 \times 10^{-3} \text{ sec}$$

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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: Blast Analysis
BY: NRS DATE: 18 Aug 19 82 CHECKED BY: TKM DATE CHECKED: 19 Aug 19 82

Find the average peak pressure on the 7'x9' door of Pits Bay design for a charge located on the floor (assuming no dividing wall)



$$w = 78.5 \text{ lb}$$

$$R = 14 \text{ ft (worst case impulse on door occurred for 2nd stall)}$$

$$\text{charge wt} = w \times 1.2 = 2 \text{ on floor}$$

$$\therefore w = 188.4 \text{ lb and}$$

$$Z = R/w^{1/3} = 2.4 \text{ ft/lb}^{1/3}$$

For each wall element shown, tabulate x/R , A and P .

Element	$x \text{ (ft)}$	x/R	$A \text{ (ft}^2\text{)}$	$P \text{ (psi)}$	$P A \text{ (psi ft}^2\text{)}$
1	22.1	1.58	14	120	1680
2	22.1	1.62	14	110	1540
3	23.5	1.68	14	100	1400
4	24.7	1.76	21	80	1680

$$\Sigma = 63 \text{ ft}^2$$

$$\Sigma = 6300 \text{ psi ft}^2$$

$$P = \frac{6300}{63} = 100 \text{ psi} \times 1.75 = \underline{\underline{175 \text{ psi}}}$$

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COMPUTATION SHEET

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PROJECT NO.: 22-7092

SPONSOR: CERL

SUBJECT: BLAST ANALYSIS

BY: NRS

DATE: 30 SEP 19 82

CHECKED BY:

DATE CHECKED: 19

7X9 DOOR - Fits Bay

INPUT PARAMETERS:

CHARGE WEIGHT = 188.40 LB
PERPENDICULAR STANDOFF = 14.00 FT
PLATE LENGTH, XTOT = 7.00 FT
PLATE WIDTH, YTOT = 9.00 FT

CHARGE POSITION IN PLANE PARALLEL TO PLATE

LENGTHWISE DISTANCE FROM
PLATE CORNER, XC = 19.50 FT
WIDTHWISE DISTANCE FROM
PLATE CORNER, YC = -5.00 FT

OUTPUT:

TOTAL APPLIED IMPULSE = 0.142E+04 LB-SEC
AVERAGE SPECIFIC IMPULSE = 0.16 PSI-SEC

$$I = 1.75 (16) = 28 \text{ psi-sec}$$

$$t_d = \frac{2I}{P} = \frac{2(28)}{175} = 0.32 \text{ sec}$$

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SHEET NO.

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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: BLAST ANALYSIS
BY: NES DATE: 22 Sep 19 82 CHECKED BY: DATE CHECKED: 19

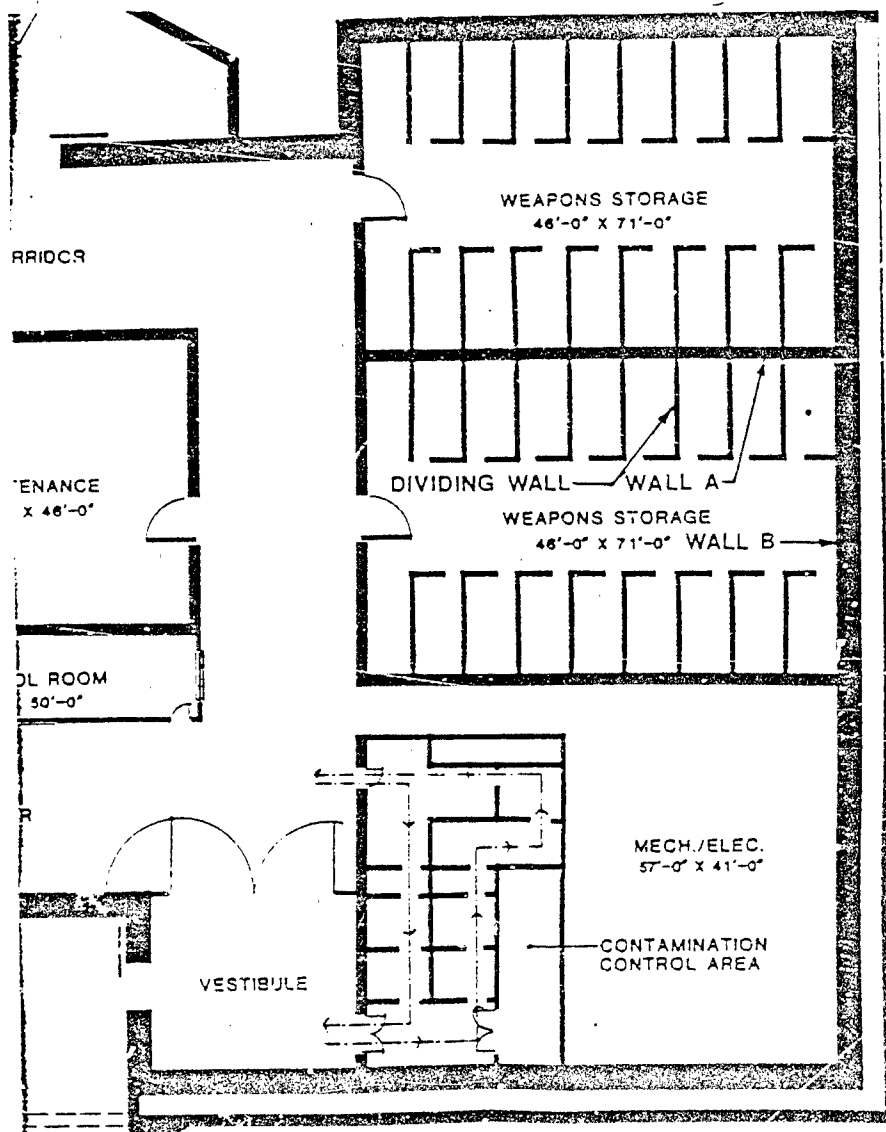
2.1.3 Shock Loading for Marge Bay Design

Summary of Blast Loads - Meze			
Component	P(psi)	\dot{c} (psi-sec)	t_d (sec)
wall A	1830	0.56	0.00061
wall B	2170	0.44	0.00040
dividing wall	12,250	1.14	0.00019
ceiling	200	0.28	0.0027
door	470	0.35	0.0015

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COMPUTATION SHEET

SHEET NO.
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PROJECT NO: 02-7092 SPONSOR: CSRL
SUBJECT: FLOOR PLAN OF DATES COMPLET - NO. 5
BY: D. [unclear] DATE: 28 SEPT 19 82 CHECKED BY: NRS DATE CHECKED: 24 SEP 19 82



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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: Blast Analysis
BY: NRS DATE: 30 Sep 19 82 CHECKED BY: DEK DATE CHECKED: 30 Sep 19 82

WALL A - maze bay

INPUT PARAMETERS:

CHARGE WEIGHT = 188.40 LB
PERPENDICULAR STANDOFF = 8.50 FT
PLATE LENGTH, XTOT = 8.00 FT
PLATE WIDTH, YTOT = 20.00 FT

CHARGE POSITION IN PLANE PARALLEL TO PLATE

LENGTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, XC = 4.00 FT
WIDTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, YC = 0.00 FT

OUTPUT:

AVERAGE PRESSURE = 0.1040E+04 PSI
TOTAL FORCE ON WALL = 0.2410E+08 LB
TOTAL APPLIED IMPULSE = 0.7431E+04 LB-SEC
AVERAGE SPECIFIC IMPULSE = 0.32 PSI-SEC

$$P_{ave} = 1830.5 \text{ PSI}$$

$$I_{ave} = 1.56 \text{ PSI-SEC}$$

$$\dot{m} = 6.1 \times 10^{-4} \text{ SEC}$$

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PROJECT NO.: 02-7092 SPONSOR: CECL
 SUBJECT: BLAST ANALYSIS
 BY: NRS DATE: 22 Sep 82 CHECKED BY: DEK DATE CHECKED: 30 Sep 82

WALL R - max 60g

INPUT PARAMETERS

CHARGE WEIGHT = 188.40 LB
 PERPENDICULAR STANDOFF = 4.00 FT
 PLATE LENGTH, XTOT = 17.00 FT
 PLATE WIDTH, YTOT = 20.00 FT

CHARGE POSITION IN PLANE PARALLEL TO PLATE

LENGTHWISE DISTANCE FROM LOWER
 LEFT PLATE CORNER, XC = 8.50 FT
 WIDTHWISE DISTANCE FROM LOWER
 LEFT PLATE CORNER, YC = 0.00 FT

OUTPUT

AVERAGE PRESSURE = 0.1237E+04 PSI
 TOTAL FORCE ON WALL = 0.6057E+06 LB
 TOTAL APPLIED IMPULSE = 0.1231E+05 LB-SEC
 AVERAGE SPECIFIC IMPULSE = 0.25 PSI-SEC

$$P_{LS} = 2.5 \text{ PSI}$$

$$I_{LS} = 1.4 \text{ PSI-SEC}$$

$$t_d = 1.0 \times 10^{-4} \text{ SEC}$$

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PROJECT NO.: 22-7092 SPONSOR: CERL
SUBJECT: BLAST ANALYSIS
BY: NKS DATE: 22 Apr 82 CHECKED BY: DATE CHECKED: 18

Dividing Wall - Marge Bay Design

The shock pressure, impulse and duration on the dividing wall of the Marge Bay are the same as that calculated for the dividing wall of the Pits Bay design (see Sec 2.1.2)

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COMPUTATION SHEET

SHEET NO.
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PROJECT NO. SR-1092 SPONSOR CERL
SUBJECT: Blast Analysis
BY SS DATE 30 Sep 19 82 CHECKED BY DK DATE CHECKED 30 Sep 19 82

CEILING - Main Bay

INPUT PARAMETERS

CHARGE WEIGHT = 188.50 LB
PERPENDICULAR STANDOFF = 20.00 FT
PLATE LENGTH, YTOT = 8.00 FT
PLATE WIDTH, YTOT = 46.00 FT

CHARGE POSITION IN PLANE PARALLEL TO PLATE

LENGTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, XC = 4.00 FT
WIDTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, YC = 37.50 FT

OUTPUT

AVERAGE PRESSURE = 0.1187E+03 PSI
TOTAL FORCE ON WALL = 0.6790E+07 LB
TOTAL APPLIED IMPULSE = 0.9691E+01 LB-SEC
AVERAGE SPECIFIC IMPULSE = 0.16 PSI-SEC

P_{max} = 0.1187E+03 PSI
 I_{max} = 0.9691E+01 LB-SEC
 t_{max} = 0.0001 SEC

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COMPUTATION SHEET

SHEET NO.
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PROJECT NO. C2-7012 SPONSOR: CERL
SUBJECT: BLAST ANALYSIS
BY: NRS DATE: 25 SEP 19 92 CHECKED BY: DEX DATE CHECKED: 30 SEP 19 92

DOOR - mize bay

INPUT PARAMETERS

CHARGE WEIGHT = 188.40 LB
PERPENDICULAR STANDOFF = 13.00 FT
PLATE LENGTH, XTOT = 7.00 FT
PLATE WIDTH, YTOT = 9.00 FT

CHARGE POSITION IN PLANE PARALLEL TO PLATE

LENGTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, XC = -10.00 FT
WIDTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, YC = 0.00 FT

OUTPUT

AVERAGE PRESSURE = 0.2677E+03 PSI
TOTAL FORCE ON WALL = 0.2427E+07 LB
TOTAL APPLIED IMPULSE = 0.1790E+04 LB-SEC
AVERAGE SPECIFIC IMPULSE = 0.20 PSI-SEC

$T_c = 4.35 \text{ sec}$

$t_{0.5} = 1.35 \text{ sec}$

$t_d = 5.45 \text{ sec}$

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COMPUTATION SHEET

SHEET NO.
28 OF 47

PROJECT NO.: C2-7092 SPONSOR: CERL
SUBJECT: BLAST ANALYSIS
BY: NRS DATE 22 Sep 19 82 CHECKED BY: DATE CHECKED: 19

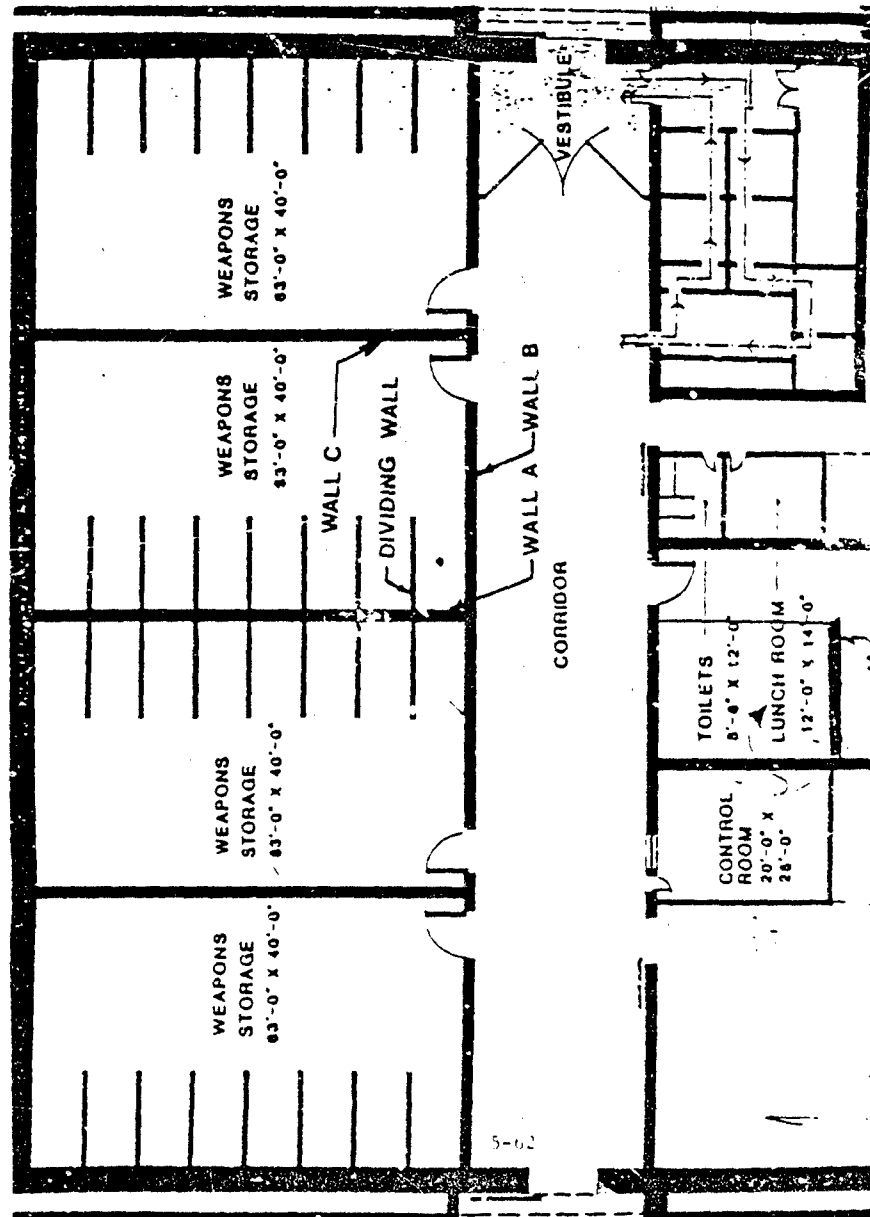
2.1.4 Double-Stacked Bay Design

Summary of Blast Loads - Double stacked			
Component	P (psi)	i (psi-sec)	T _d (sec)
wall A	2632	0.53	0.0004
wall B	5780	0.95	0.00033
wall C	68	0.23	0.0068
dividing wall	5780	0.95	0.00033
ceiling	780	0.30	0.00077
door	100	0.23	0.0046

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COMPUTATION SHEET

SHEET NO.
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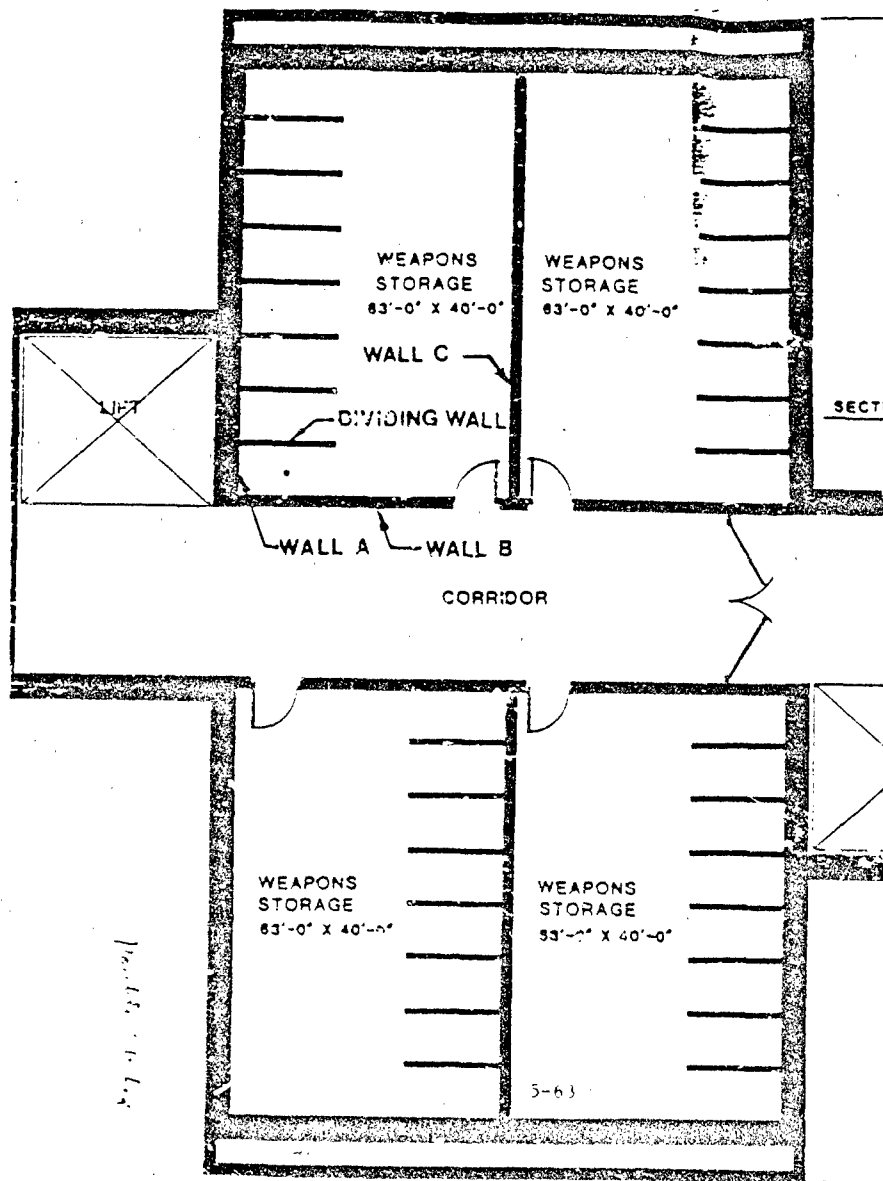
PROJECT NO.: 02-7992 SPONSOR: CERL
SUBJECT: Floor Plan - Double Stacked - Contract No. 1
BY: DEK DATE: 23 Oct 82 CHECKED BY: NRS DATE CHECKED: 27 Sep 82



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COMPUTATION SHEET

SHEET NO.
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PROJECT NO.: 02-392 SPONSOR: CERL
SUBJECT: Floor Plan - Double Stacked ~~except~~ No. 3
BY: DEK DATE: 24 SEP 19 82 CHECKED BY: LKS DATE CHECKED: 27 SEP 19 82



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PROJECT NO: 02-7092 SPONSOR: CERL
SUBJECT: Blast Analysis
BY: NRS DATE: 22 Sept 82 CHECKED BY: _____ DATE CHECKED: _____ 19

WALL A - Double-Stacked Coy

INPUT PARAMETERS:

CHARGE WEIGHT = 188.40 LB
PERPENDICULAR STANCOFF = 7.00 FT
PLATE LENGTH, XTOT = 7.00 FT
PLATE WIDTH, YTOT = 18.00 FT

CHARGE POSITION IN PLATE PARALLEL TO PLATE

LENGTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, XC = 3.30 FT
WIDTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, YC = 0.00 FT

OUTPUT:

AVERAGE PRESSURE = 0.1304E+04 PSI
TOTAL FORCE ON WALL = 0.2723E+04 LB
TOTAL APPLIED IMPULSE = 0.5475E+04 LB-SEC
AVERAGE SPECIFIC IMPULSE = 0.30 PSI-SEC

$$P_{avg} = 1504 \times 1.75 = 2632 \text{ PSI}$$

$$I_{avg} = 0.30 \times 1.75 = 0.53 \text{ PSI-SEC}$$

$$t_d = \frac{24}{P} = \frac{2(53)}{2632} = 0.0040 \text{ SEC}$$

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COMPUTATION SHEET

SHEET NO
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PROJECT NO.: C2-7092 SPONSOR: SERL
SUBJECT: BLAST ANALYSIS
BY: NRS DATE: 30 Sep 82 CHECKED BY: _____ DATE CHECKED: _____ 19____

WALL B and Dividing Wall - Double stacked

INPUT PARAMETERS

CHARGE WEIGHT = 188.40 LB
PERPENDICULAR STANDOFF = 3.50 FT
PLATE LENGTH, XTOT = 8.00 FT
PLATE WIDTH, YTOT = 8.00 FT

CHARGE POSITION IN PLANE PARALLEL TO PLATE

LENGTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, XC = 7.00 FT
WIDTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, YC = 0.00 FT

OUTPUT

AVERAGE PRESSURE = 0.3303E+04 PSI
TOTAL FORCE ON WALL = 0.3044E+09 LB
TOTAL APPLIED IMPULSE = 0.4992E+04 LB-SEC
AVERAGE SPECIFIC IMPULSE = 0.54 PSI-SEC

$$P_{avg} = 3303 \times 1.75 = 5780 \text{ psi}$$

$$I_{avg} = 0.54 \times 1.75 = 0.95 \text{ psi-sec}$$

$$t_d = \frac{2I}{P} = \frac{2(0.95)}{5780} = 0.00033 \text{ sec}$$

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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: BLAST ANALYSIS
BY: NRS DATE: 22 Sep 1982 CHECKED BY: DATE CHECKED: 19

WALL C - Double Stacked Bay

INPUT PARAMETERS:

CHARGE WEIGHT = 198.40 LB
PERPENDICULAR STANCOFF = 33.00 FT
PLATE LENGTH, XTOT = 7.00 FT
PLATE WIDTH, YTOT = 19.00 FT

CHARGE POSITION IN PLANE PARALLEL TO PLATE:

LENGTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, XC = 7.00 FT
WIDTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, YC = 0.00 FT

OUTPUT:

AVERAGE PRESSURE = 0.3913E+02 PSI
TOTAL FORCE ON WALL = 0.7100E+06 LB
TOTAL APPLIED IMPULSE = 0.2273E+04 LB-SEC
AVERAGE SPECIFIC IMPULSE = 0.13 PSI-SEC

$$p_{avg} = 39 \times 1.75 = 68 \text{ psi}$$

$$I_{avg} = 0.13 \times 1.75 = 0.23 \text{ psi-sec}$$

$$t_d = 2/p = \frac{2(23)}{68} = 0.0068 \text{ sec}$$

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COMPUTATION SHEET

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PROJECT NO.: D2-7092 SPONSOR: CERL
SUBJECT: BLAST ANALYSIS
BY: NRS DATE: 20 Sep 19 82 CHECKED BY: DEX DATE CHECKED: 30 Sep 19 82

CEILING - Double-stacked bay

INPUT PARAMETERS:

CHARGE WEIGHT = 94.20 LB
PERPENDICULAR STANDOFF = 9.00 FT
PLATE LENGTH, XTOT = 7.00 FT
PLATE WIDTH, YTOT = 40.00 FT

CHARGE POSITION IN PLANE PARALLEL TO PLATE

LENGTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, XC = 3.50 FT
WIDTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, YC = 33.00 FT

OUTPUT:

AVERAGE PRESSURE = $0.4462E+03$ PSI
TOTAL FORCE ON WALL = $0.1799E+08$ LB
TOTAL APPLIED IMPULSE = $0.6998E+04$ LB-SEC
AVERAGE SPECIFIC IMPULSE = 0.17 PSI-SEC

$P_{me} = 780.5$ PSI
 $I_{ave} = 0.30$ PSI-SEC
 $t_d = 7.7 \times 10^{-4}$ SEC

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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: BLAST ANALYSIS
BY: NRS DATE: 30 Sep 1982 CHECKED BY: _____ DATE CHECKED: _____ 19__

47	181	30 SHEETS	3 SQUARE
47	362	100 SHEETS	3 SQUARE
47	369	200 SHEETS	3 SQUARE

The shock pressure impulse and duration on the door of the Doubled stacked Bay are the same as that for the door of the Long Bay (see sec. 2.1.1)

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COMPUTATION SHEET

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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: Revised Maintenance Bay
BY: NRS DATE: 22 Sep 19 82 CHECKED BY: _____ DATE CHECKED: 19

2.1.5 Shock Loads for Maintenance Bay

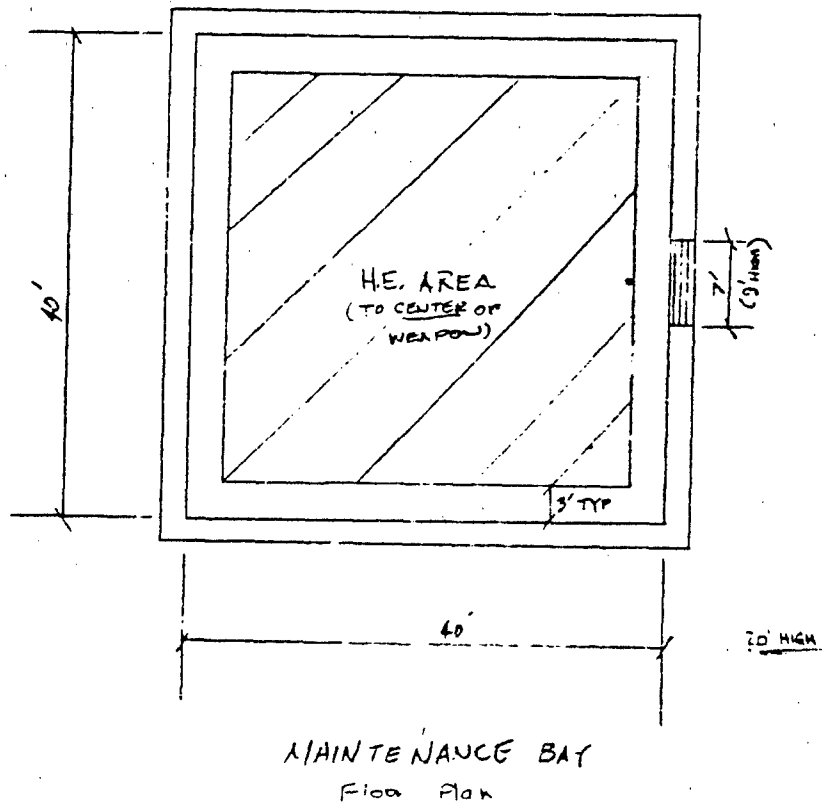
Summary of Blast Loads - Maintenance Bay			
Component	P (psi)	i (psi-sec)	t _d (sec)
wall	2770	0.42	0.00030
roof	238	0.32	0.0027
door	17248	1.19	0.00014

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COMPUTATION SHEET

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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: Blast Analysis
BY: NRS DATE: 29 Sep 19 82 CHECKED BY: DATE CHECKED: 19

11 12 13 14 15 16 17 18 19 20 21 22 23 24 25 26 27 28 29 30 31 32 33 34 35 36 37 38 39 40 41 42 43 44 45 46 47 48 49 50 51 52 53 54 55 56 57 58 59 60 61 62 63 64 65 66 67 68 69 70 71 72 73 74 75 76 77 78 79 80 81 82 83 84 85 86 87 88 89 90 91 92 93 94 95 96 97 98 99 100



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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: BLIST ANALYSIS
BY: NRS DATE: 20 SEP 82 CHECKED BY: _____ DATE CHECKED: _____ 15

WALL - Maintenance Bay

INPUT PARAMETERS

CHARGE WEIGHT = 188 40 LB
PERPENDICULAR STANDOFF = 3 00 FT
PLATE LENGTH, XTOT = 20 00 FT
PLATE WIDTH, YTOT = 20 00 FT

CHARGE POSITION IN PLANE PARALLEL TO PLATE

LENGTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, XC = 10 00 FT
WIDTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, YC = 0 00 FT

OUTPUT

AVERAGE PRESSURE = 0 1582E+04 PSI
TOTAL FORCE ON WALL = 0 5126E+08 LB
TOTAL APPLIED IMPULSE = 0 1383E+05 LB-SEC
AVERAGE SPECIFIC IMPULSE = 0 24 PSI-SEC

$$P_{avg} = 1532 \times 1.75 = 2770 \text{ psi}$$

$$I_{avg} = 0.24 \times 1.75 = 0.42 \text{ psi-sec}$$

$$t_d = 2/P = \frac{2(42)}{2770} = 0.0030 \text{ sec}$$

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COMPUTATION SHEET

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PROJECT NO. 52-7092 SPONSOR: CERL
SUBJECT: BLAST ANALYSIS
BY: NES DATE: 30 Sep 82 CHECKED BY: _____ DATE CHECKED: 18

ROOF - Maintenance Bay

INPUT PARAMETERS

CHARGE WEIGHT = 188 40 LB
PERPENDICULAR STANDOFF = 20 00 FT
PLATE LENGTH, XTOT = 20 00 FT
PLATE WIDTH, YTOT = 40 00 FT

CHARGE POSITION IN PLANE PARALLEL TO PLATE

LENGTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, XC = 10 00 FT
WIDTHWISE DISTANCE FROM LOWER
LEFT PLATE CORNER, YC = 20 00 FT

OUTPUT

AVERAGE PRESSURE = 0 1264E+03 PSI
TOTAL FORCE ON WALL = 0 1572E+08 LB
TOTAL APPLIED IMPULSE = 0 2070E+05 LB-SEC
AVERAGE SPECIFIC IMPULSE = 0 18 PSI-SEC

$$P_{avg} = 126 = 1.75 \times \underline{23^4 \text{ psi}}$$

$$I_{avg} = 0.18 \times 1.75 = \underline{0.32 \text{ psi-sec}}$$

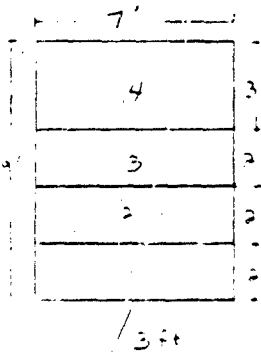
$$t_d = \frac{I_{avg}}{P} = \frac{0.32}{23^4} = \underline{0.0027 \text{ sec}}$$

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COMPUTATION SHEET

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PROJECT NO: 22-7092 SPONSOR: CERL
SUBJECT: Blast Analysis
BY: SC DATE: 3 Aug 19 52 CHECKED BY: TKM DATE CHECKED: 19 Aug 19 52

Find the average peak pressure on the
7' x 9' door of the maintenance bay for
a charge on the floor.



$$W = 78.5 \text{ lb}$$

standoff to center of wrapper,
 $R = 3 \text{ ft}$

charge wt = $w \times 2 \times 2$ on floor

$$W = 188.4 \text{ lb and}$$

$$Z = R/W^{1/3} = 0.52 \text{ ft/lb}^{1/3}$$

Using Fig. 2 for each element shown
tabulate r/R , A and P_r .

Element	r/R	r/R	$A (\text{ft}^2)$	$P (\text{psi})$	$PA (\text{psi-ft}^2)$
1	1	1	14	23500	329000
2	3	1.0	14	14000	196000
3	5	1.7	14	4000	56000
4	7.5	2.5	21	900	39700

$\Sigma P_r A = 620900 \text{ psi-ft}^2$

$$\Sigma = 63 \text{ ft}^2$$

$$\Sigma = 620900 \text{ psi-ft}^2$$

$$P = \frac{620900}{63} = 9856 \text{ psi} \times 1.75 = \underline{\underline{17248 \text{ psi}}}$$

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PROJECT NO.: 02-7092 SPONSOR: CEKL
SUBJECT: BLAST ANALYSIS
BY: NRS DATE: 30 SEP 82 CHECKED BY: _____ DATE CHECKED: _____ 19

MAINTENANCE BAY DOOR

INPUT PARAMETERS:

CHARGE WEIGHT = 188.40 LB
PERPENDICULAR STANDOFF = 3.00 FT
PLATE LENGTH, XTOT = 7.00 FT
PLATE WIDTH, YTOT = 9.00 FT

CHARGE POSITION IN PLANE PARALLEL TO PLATE

LENGTHWISE DISTANCE FROM
PLATE CORNER, XC = 3.50 FT
WIDTHWISE DISTANCE FROM
PLATE CORNER, YC = 0.00 FT

OUTPUT

TOTAL APPLIED IMPULSE = 0.6194E+04 LB-SEC
AVERAGE SPECIFIC IMPULSE = 0.68 PSI-SEC

$$I = 1.75 (.68) = 1.19 \text{ psi-sec}$$

$$\tau = \frac{20}{1} = \frac{20}{1.19} = 16.81 \text{ sec}$$

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COMPUTATION SHEET

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PROJECT NO: 02-7092 SPONSOR: CERL
SUBJECT: Blast Analysis
BY: LRS DATE: 20 Sep 19 82 CHECKED BY: DATE CHECKED: 19

2.2 Quasi-static Loading

In this section, the quasi-static loadings on each of the 4 munition storage bays and the maintenance bay are calculated. Drawings of each design are included in Section 2.1 of this appendix.

Quasi-static Load Summary	
Bay	Pos (psi)
Long Bay	20
Fits Bay	17
Roze Bay	19
Double-stacked Bay	33
Maintenance Bay	37

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PROJECT NO.: 22-7092-001 SPONSOR: CERL
SUBJECT: BLAST ANALYSIS
BY: P. RS DATE: 21 Sep 19 82 CHECKED BY: DATE CHECKED: 19

2.2.1 Quasi-static Loading for Long Bay Design

To find the quasi-static pressure corresponding to the long bay design, first find volume to be considered.

$$19 \text{ ft} \times 40 \text{ ft} \times 13 \text{ ft} = 6180 \text{ ft}^3$$

$$\text{Bay ceiling volume to subtract} = 1666 \text{ ft}^3$$

$$\text{munition volume} = 9.67 \text{ ft} \times 3.08 \text{ ft} \times 3.25 \text{ ft} = 96.8 \text{ ft}^3$$

$$96.8 \text{ ft}^3 \times 5 \text{ munitions} = 484 \text{ ft}^3$$

$$\text{Dividing wall volume} = 140 \text{ ft}^3 \times 14 \text{ walls} = 1960 \text{ ft}^3$$

$$6180 \text{ ft}^3 - 1666 \text{ ft}^3 - 484 \text{ ft}^3 - 1960 \text{ ft}^3 = 5680 \text{ ft}^3$$

$$\text{Volume to be considered} = 56,802 \text{ ft}^3$$

$$\text{weight of explosive} = 35.6 \text{ Kg} = 78.5 \text{ lb} \times 1.2 = 94.2 \text{ lb}$$

$$\frac{w}{v} = \frac{94.2 \text{ lb}}{56802 \text{ ft}^3} = 1.66 \times 10^{-3} \frac{\text{lb}}{\text{ft}^3}$$

$$\text{using Figure 5, } P_{qs} = \underline{\underline{20 \text{ psi}}}$$

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

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PROJECT NO: 02-7092 SPONSOR: CERL
SUBJECT: BLAST ANALYSIS
BY: L. S. DATE 22 Apr 82 CHECKED BY: _____ DATE CHECKED: 19

2.2.2 Quasi-static Loading for Pits Bay Design

To find the quasi-static pressure corresponding to the pits design first find the volume to be considered

$$71 \text{ ft} \times 44 \text{ ft} \times 25 \text{ ft} = 78,100 \text{ ft}^3$$

$$\begin{aligned} \text{Munition volume} &= 9.67 \text{ ft} \times 3.53 \text{ ft} \times 3.25 \text{ ft} = 96.3 \text{ ft}^3 \\ 96.3 \text{ ft}^3 \times 15 \text{ munitions} &= 1452 \text{ ft}^3 \end{aligned}$$

Dividing Wall Volume =

$$85 \text{ ft}^3 \times 4 \text{ walls (1 wall)} = 1190 \text{ ft}^3$$

$$3550 \text{ ft}^3 \times 1 \text{ wall (middle)} = 3550 \text{ ft}^3$$

$$4740 \text{ ft}^3$$

$$78,100 \text{ ft}^3 - 1452 \text{ ft}^3 - 4740 \text{ ft}^3 = 71,908 \text{ ft}^3$$

$$\text{Volume to be considered} = 71,908 \text{ ft}^3$$

$$\text{Weight of explosive} = 35.6 \text{ Kg} = 78.5 \text{ lb} \times 2 = 157 \text{ lb}$$

$$\frac{W}{V} = \frac{157 \text{ lb}}{71,908 \text{ ft}^3} = 2.18 \times 10^{-3} \frac{\text{lb}}{\text{ft}^3}$$

$$\text{using Figure 5, } P_{QS} = \underline{\underline{17 \text{ psi}}}$$

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

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PROJECT NO.: 72-7000 SPONSOR: AFOSR
SUBJECT: CUBIC ROOM: PROTECTIVE
BY: J. K. HARRIS DATE: JAN 19 82 CHECKED BY: NRS DATE CHECKED: 23 Sept 82

2.2.3 QUANTITATIVE PRESSURE IN THE MIZE ROOM

CONSIDERED VOLUME = TOTAL VOLUME - MUNITIONS VOLUME - DIVIDING WALL VOL.

$$V = V_T - V_M - V_D$$

$$V_T = 46' \times 7' \times 20' = 65320 \text{ ft}^3$$

$$V_M = 1452 \text{ ft}^3$$

$$V_D = 16 \text{ WALLS} \times 10 \text{ DEEP} \times (5' \text{ HIGH}) \times (17+8' \text{ LONG}) = 2000 \text{ ft}^3$$

$$V = 65320 - 1452 - 2000$$

$$= 61868 \text{ ft}^3$$

$$\text{CHARGE WEIGHT (W)} = 35.6 \text{ kg-W}$$

$$= 785.0 \text{ (2 SAFETY FACTOR)} = 94.2 \text{ lb.}$$

$$\frac{W}{V} = \frac{94.2}{61868} = 1.5 \times 10^{-3} \text{ lb/ft}^3$$

$$\text{USING FORMULA 5 } \underline{\underline{P_{max} = 19 \text{ PSI}}}$$

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PROJECT NO.: 22-7002 SPONSOR: CER
SUBJECT: 22-7002-19-22
BY: NRS DATE: 25 Oct 19 82 CHECKED BY: NRS DATE CHECKED: 23 Sept 82

2.2.4 Static Pressing for the Double Stacked Room

CONSIDERED VOLUME = TOTAL VOLUME - MOUNTAIN VOLUME - DIPPING WALL VOLUME

$$V = V_T - V_M - V_D$$

$$V_T = 40 \times 64 \times 2 = 51200 \text{ ft}^3$$

$$V_M = 1452 \text{ ft}^3$$

$$V_D = 8 \text{ WALLS} \times 16 \text{ LONG} \times 8 \text{ DEEP} \times 8 \text{ HIGH} \\ + 1 \text{ FLOOR} \times 16 \text{ LONG} \times 64 \text{ DEEP} \times 1 \text{ DEEP} \\ + 8 \text{ WALLS} \times 16 \text{ LONG} \times 8 \text{ HIGH} \times 1 \text{ DEEP} \\ = 2900 \text{ ft}^3$$

$$V = 51200 - 1452 - 2900 = 46848 \text{ ft}^3$$

$$W = 35.6 \text{ lb/ft}^3 = 78.5 \text{ lb}$$

$$78.5 \text{ lb} \times 12 = W = 942 \text{ lb}$$

$$\frac{W}{V} = \frac{942}{46848} = 2.25 \times 10^{-3} \text{ lb/ft}^3$$

$$US VC = 1.205, \quad \underline{\underline{P_s = 28 \text{ PSI}}}$$

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.

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PROJECT NO.: 02-7092

SPONSOR: CERL

SUBJECT:

BY: NRS

DATE: 22 Feb 1982

CHECKED BY:

DATE CHECKED:

19

2.2.5 Quasi-static Loading for Typical Maintenance Bay

Find the quasi-static pressure occurring in the maintenance bay. Estimating the room volume occupied by equipment and machinery at 20% of the bay volume we find the volume to be considered.

$$\text{Volume of bay} = 40 \text{ ft} \times 40 \text{ ft} \times 20 \text{ ft} = 32000 \text{ ft}^3$$

$$(32000 \text{ ft}^3)(2) = 6400 \text{ ft}^3 \text{ for equipment}$$

$$\text{Volume to be considered} = 32000 \text{ ft}^3 - 6400 \text{ ft}^3 = 25600 \text{ ft}^3$$

$$\text{Weight of explosive} = 35.6 \text{ Kg} = 78.5 \text{ lb} \times 1.2 = 94.2 \text{ lb}$$

$$\frac{w}{V} = \frac{94.2 \text{ lb}}{25600 \text{ ft}^3} = 3.68 \times 10^{-3} \frac{\text{lb}}{\text{ft}^3}$$

$$\text{Using Figure 5, } P_{as} = \underline{\underline{37 \text{ psi}}}$$

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COMPUTATION SHEET

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SHEET NO.
1 OF 32

PROJECT NO.: 62-7092 SPONSOR: CERL
SUBJECT: INTERIOR WALL DESIGN
BY: DRETHUM DATE: 30 SEP 19 82 CHECKED BY: DATE CHECKED: 19

INTERIOR WALL DESIGN

INTRODUCTION	5-81
SUMMARY OF RESULTS	5-82
COMPUTER RESULTS/CALCULATIONS	
LONG BAY	5-83
DOUBLE STACKED	5-91
PITS	5-99
MAINTENANCE BAY	5-104
MAZE	5-106
DOUBLE STACKED CRANK/KINK	5-111

INTRODUCTION-

THE BLAST LOADINGS FROM THE DETONATION OF EXPLOSIVES WITHIN THE STRUCTURE GOVERN THE DESIGN OF THE INTERIOR WALLS. THE BLAST LOADINGS AND THE DIAGRAMS DESIGNATING WALLS AS A, B, OR C CAN BE FOUND IN THE BLAST LOADING SECTION OF THIS APPENDIX.

THE WALLS HAVE BEEN DESIGNED FOR AN ALLOWABLE ROTATION OF 1°. THE CALCULATIONS ARE PRESENTED WITH A SUMMARY OF DESIGN REQUIREMENT RESULTS, COMPUTER OUTPUT ESTIMATING THE STEEL AREA REQUIRED IN THE WALL, AND A WORKSHEET OF BAR SIZING CALCULATIONS.

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DEPARTMENT OF ENERGETIC SYSTEMS
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2 OF 32

PROJECT NO.: 62-7092 SPONSOR: CERL
SUBJECT: INTERIOR WALL DESIGN SUMMARY
BY: DK/AM DATE: 10/11/87 CHECKED BY: AMV DATE CHECKED: 30/11/87

INTERIOR WALL DESIGN SUMMARY

COMMENT	WALL	Tc	FLUX	TEMP	LACING	A _g	EXPANDED A _{st}
LONG	A	24	7	6	5	.60	.51
BAY	B	24	9	6	5	1.0	.82
	C	24	7	6	5	.60	.51
	DIV.	12	5	4	5	.31	-
DOUBLE	A	24	7	6	5	.60	.51
STUCK	B	24	9	6	7	1.0	.84
	C	24	9	6	5	1.0	.87
	DIV.	12	5	4	5	.31	-
MAINT		24	(2)3	6	7	2.0	1.70
BAY							
PITS	A	24	10	6	5	1.27	1.11
	B	24	10	6	5	1.27	1.25
	DIV.	12	5	4	5	.31	-
MAINT	A	24	7	6	5	.60	.51
	B	24	10	6	5	1.27	1.04
	DIV.	12	5	4	5	.31	-

SPACING = 12"

LACING USES METHOD 2

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SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CORL
SUBJECT: RUMBERG BAR REQUIREMENTS
BY: DK DATE: 11-18-82 CHECKED BY: DATE CHECKED: 19

LONG BAY WALL A

SUMMARY -

$T_c = 24"$

FLEXURAL BAR #7@12

TEMPERATURE BAR #6@5

LACING BAR #5 METHOD 2

COMPUTER OUTPUT -

FIXED - FIXED SUPPORT CONDITIONS

LOAD PARAMETERS $F1 = .3100E+06$ $F2 = .1680E+04$
 $T2 = .3700E-03$ $T4 = .1000E+02$

BEAM PARAMETERS $L = 144.00$ THICK = 24.00
 $D = 3.50$ $D' = 3.50$
ROTATION = 1.00 $TNATL = .315E-02$

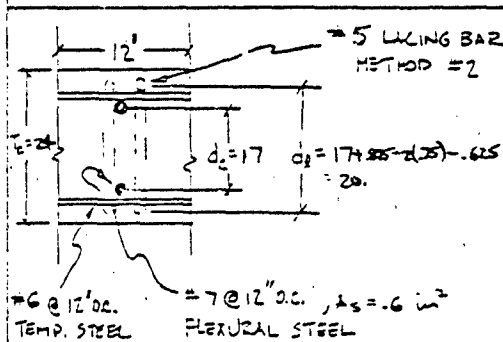
$FC' = 4000.$ $FDY = 72000.$

$IE = .1152E+04$ $IC = .5570E+02$ $IA = .5043E+03$
 $KE = .2826E+06$ $KEPL = .5653E+05$
 $LI = .2721E+04$ $L2 = .3628E+04$

REQUIRED STEEL AREA = .0257 SQ IN/IN = .3201 SQ IN/FT

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: LACING/LONG RAY/WALL A
BY: DK DATE: 24 SEP 82 CHECKED BY: DATE CHECKED: 19

LONG DESIGN



FIXED-FIXED SUPPORTS

$L = 12 \text{ FT} = 144 \text{ in}$

$M = \frac{w_u L^2}{12}$

$M = \frac{.6 (7200) 17}{12} = 6.1 \times 10^4 \frac{\text{in-lb}}{\text{in}}$

$r_u = \frac{16 M}{L^2} = \frac{16 (6.1 \times 10^4)}{(144)^2} = 4.2 \text{ psi}$

FLEX. BAR, $A_s > .0025 b d_e = .0025 (12) (17) = .51 \text{ in}^2$
TEMP. BAR, $A_s > .0015 b d_e = .0015 (12) (17) = .31 \text{ in}^2$
 $A_s = 6 \text{ BAR} = .44 \text{ in}^2 > \frac{1}{2}$

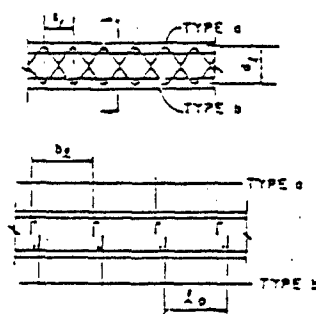
$v_u = \frac{r_u (L - d_e)}{d_e} = \frac{(47.2 \frac{\text{lb}}{\text{in}} - 17)}{17} = 153 \text{ psi}$

$p = A_s / d_e b = \frac{.60}{(17)(12)} = .0029$

$v_c = \phi (1.9 \sqrt{f'_c} + 2500 p) = .85 (1.9 \sqrt{4000} + 2500 (.0029)) = 108 \text{ psi}$

$v_{c \text{ max}} = 2.28 \phi \sqrt{f'_c} = 2.28 (.85) \sqrt{4000} = 123 \text{ psi}$

$v_c < v_{c \text{ max}}$ USE v_c $\therefore v_c = 108 = 108 \text{ psi}$



$s/d_e = \frac{12}{20} = .6$, $\frac{7(.65)}{20} = .22$

FROM FIG. 6-19 OF TMS, $\alpha = 67^\circ$

$A_T = \frac{v_u' s_e b_e}{.85 f_y (\sin \alpha + \cos \alpha)} = \frac{108 (12) (12)}{.85 (60000) (\sin 67 + \cos 67)}$

$A_T = .23 \text{ in}^2$

USE 5 BAR

METHOD #2

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: REINFORCING BAR REQUIREMENTS
BY: JK DATE: 24 SEP 19 82 CHECKED BY: DATE CHECKED: 19

LONG BAR WALL B

SUMMARY -

$T_2 = 24"$
FLEX #9 @ 12
TENT #6 @ 12
LACING #5 METHOD 2

COMPUTER OUTPUT

FIXED - FIXED SUPPORT CONDITIONS

LOAD PARAMETERS $F1 = .5872E+05$ $F2 = .2980E+04$
 $T2 = .3500E-03$ $T4 = .1000E+02$

BEAM PARAMETERS $L = 144.00$ THICK = 24.00
 $D = 3.50$ $D' = 3.50$
ROTATION = 1.00 INATL = .991E-02

$FC' = 4000.$ $FDY = 72000.$

$IE = .1152E+04$ $IC = .1202E+03$ $IA = .5351E+03$
 $KEL = .2975E+06$ $KE_L P_1 = .5950E+05$
 $L1 = .6937E+04$ $L2 = .9249E+04$

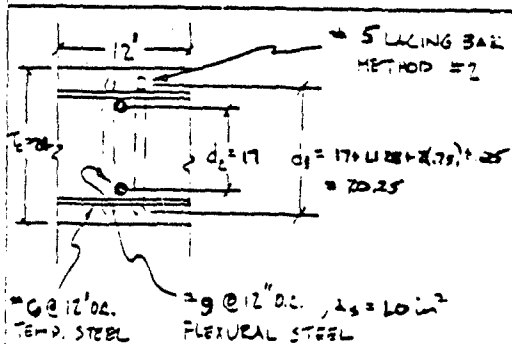
REQUIRED STEEL AREA = .0680 SQ IN/IN = .8151 SQ IN/FT

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
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PROJECT NO.: 22-7092 SPONSOR: CRRL
SUBJECT: LACING/LONG RAY/WALL B
BY: SKETCHING DATE: 23 SEP 19 82 CHECKED BY: DATE CHECKED: 19

LONG DESIGN



FIXED-FIXED SUPPORTS

$$L = 12 \text{ FT} = 144 \text{ in}$$

$$M = \frac{1.0 F_c d_c}{b}$$

$$M = \frac{1.0 (7000) 17}{12} = 10000 \text{ in-lb}$$

$$r_u = \frac{16 M}{L^2} = \frac{16 (10000)}{(144)^2} = 0.77 \text{ psi}$$

$$\text{FLEX. BAR, } A_s = 0.0029 (12) (17) = 0.59 \text{ in}^2$$

$$\text{TEMP. BAR, } A_s = 0.0015 (12) (17) = 0.31 \text{ in}^2$$

$$A_s = 6 \text{ BAR} = 0.44 \text{ in}^2$$

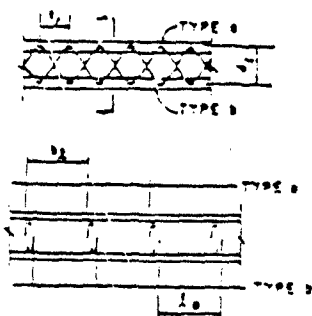
$$r_u = \frac{r_u \left(\frac{L}{d_c} - d_c \right)}{d_c} = \frac{77 \left(\frac{144}{17} - 17 \right)}{17} = 249 \text{ psi}$$

$$p = A_s / d_c b = \frac{1.0}{(17)(12)} = 0.005$$

$$r_c = \phi (1.9 \sqrt{f_c} + 2500 p) = 0.85 (1.9 \sqrt{4000} + 2500 (0.005)) = 112 \text{ psi}$$

$$r_{c \text{ max}} = 2.28 \phi \sqrt{f_c} = 2.28 (0.85) \sqrt{4000} = 123 \text{ psi}$$

$$r_c < r_u < r_{c \text{ max}} \text{ USE } r_c \quad r_c = 112 \text{ psi}$$



METHOD #2

$$x/d_c = \frac{12}{20.25} = 0.59, \quad \frac{r_u}{d_c} = \frac{249}{20.25} = 12.3$$

$$\text{FROM } 16.6-19 \text{ ON TMS, } \lambda = 66^\circ$$

$$A_s = \frac{r_u S_c b}{0.85 f_y (\sin \lambda + \cos \lambda)} = \frac{(112)(12)(12)}{0.85 (60000) (\sin 66^\circ + \cos 66^\circ)}$$

$$A_s = 0.24 \text{ in}^2$$

USE 5 BAR

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: REINFORCING BAR CORRELATION
BY: D. K. T. DATE: 2-5-82 CHECKED BY: DATE CHECKED: 19

LONG TERN WALL C

SUMMARY -

$T_c = 24"$
FLUX - 57.2
TENSILE - 46.0
LAGGING - 45.2

COMPUTER OUTPUT -

FIXED - FIXED SUPPORT CONDITIONS

LOAD PARAMETERS $F1 = .1232E+05$ $F2 = .3120E+04$
 $T2 = .5300E-02$ $T4 = .1000E+02$

BEAM PARAMETERS $L = 155.00$ THICK = 24.00
 $D = 3.50$ $D' = 3.50$
ROTATION = 1.00 INATL = .107E-01

$FC' = 4000.$ $FDY = 72000.$

$IE = .1152E+04$ $IC = .6087E+02$ $IA = .5054E+03$
 $KEL = .2231E+06$ $KELP = .4461E+05$
 $L1 = .2756E+04$ $L2 = .3475E+04$

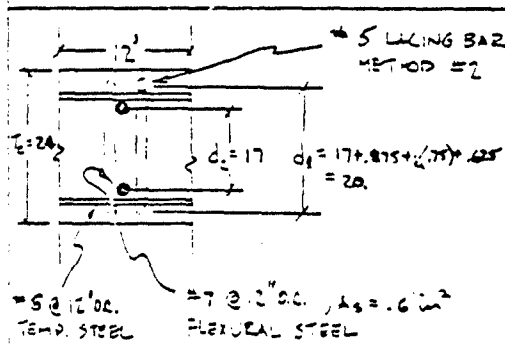
REQUIRED STEEL AREA = .0293 SQ IN/IN = .3513 SQ IN/FT

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: LACING / LONG RAY / WALKER
BY: DETRAN DATE: 2/27/78 CHECKED BY: DATE CHECKED: 19

LACING DESIGN -



FIXED-FIXED SUPPORTS

$$L = 13 \text{ FT} = 156 \text{ in}$$

$$M = \frac{A_s F_y d_c}{6}$$

$$M = \frac{.6 (72000) 17}{12} = 6.1 \times 10^4 \frac{\text{in} \cdot \text{lb}}{\text{in}}$$

$$r_u = \frac{16 M}{L^2} = \frac{16 (6.1 \times 10^4)}{(156)^2} = 40 \text{ psi}$$

$$\text{FLEX. BAR, } A_s > .0025 b d_c \left(\frac{12}{17} \right) (17) = .51 \text{ in}^2$$

$$\text{TEMP. BAR, } A_s > .0015 b d_c \left(\frac{12}{17} \right) (17) = .37 \text{ in}^2$$

$$A_s = 6 \text{ BAR} = .44 \text{ in}^2 > \frac{1}{2}$$

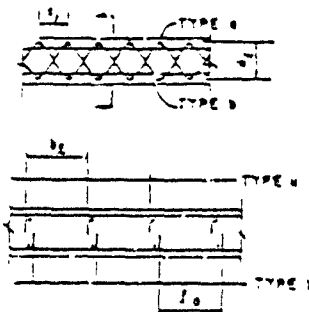
$$r_u = \frac{r_u \left(\frac{L}{d_c} - 2 \right)}{d_c} = \frac{40 \left(\frac{156}{17} - 2 \right)}{17} = 143.5 \text{ psi}$$

$$p = A_s / d_c b = \frac{.6}{(12)(17)} = .0029$$

$$r_c = \phi (1.9 \sqrt{f_c} + 2500 p) = .85 (1.9 \sqrt{4000} + 2500 (.0029)) = 108 \text{ psi}$$

$$r_{c, \text{max}} = 2.28 \phi \sqrt{f_c} = 2.28 (.85) \sqrt{4000} = 123 \text{ psi}$$

$$r_c < 108 \text{ psi} \text{ USE } r_c = 108 \text{ psi}$$



METHOD #2

$$s_1/s_2 = \frac{12}{20} = .60, \quad \frac{7D_0}{d_c} = \frac{7(.625)}{20} = .22$$

$$\text{FROM FIG. 6-19 OF TMS, } \alpha = 67^\circ$$

$$A_T = \frac{r_u' S_1 b_1}{.85 F_y (\sin \alpha + \cos \alpha)} = \frac{(108)(12)(12)}{.85(60000)(\sin 67^\circ + \cos 67^\circ)}$$

$$A_T = .23 \text{ in}^2$$

$$\text{USE } 5 \text{ BAR}$$

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COMPUTATION SHEET

SHEET NO.
9 OF 12

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: REINFORCING BAR CALCULATIONS
BY: J.K. DATE: 9/24/92 CHECKED BY: DATE CHECKED: 19

LONG BAY DIVIDING WALL

SUMMARY -

MAX DEFLECTION = 8.09"

TL = 12"

BLEX = 5 1/2"

TEMP = 40 1/2"

COMP. OUTPUT - LAGS = 5 mm 2

STRUCTURAL ELEMENT : DIVIDING WALL

LOADS ON 1" SECTION :

PEAK SHOCK FORCE = .5800E+05 LB

PEAK QUASI-STATIC FORCE = ***** LB

SHOCK FORCE DURATION = .3500E-03 SEC

QUASI-STATIC FORCE DURATION = .1000E+01 SEC

MATERIAL STRENGTHS :

FC' = 4000. PSI

FDY = 90000. PSI

CROSS-SECTION :

TOTAL CONCRETE THICKNESS = 12.0 INCHES

COVER (TOP AND BOTTOM) = 2.00 INCHES

EFFECTIVE STEEL MOMENT ARM = 7.00 INCHES

STEEL SIZE AND SPACING = # 5 AT 12 INCHES

STEEL AREA (EACH FACE) = .31 SQ IN/FT

SUPPORT CONDITIONS : FIXED - FIXED

RESPONSE

MAXIMUM DEFLECTION = 5.92 INCHES

MAXIMUM ROTATION = 3.09 DEGREES

REACTIONS

RJ = .284E+02 LB/IN

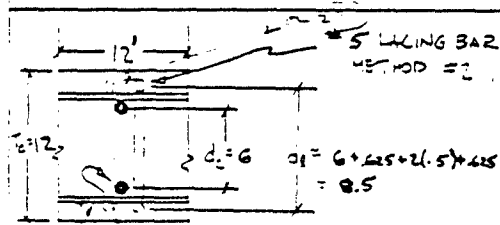
END REACTION = .154E+02 KIP/FT

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COMPUTATION SHEET

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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: LACING / LONG RAY / DIVIDING WALL
BY: S. K. K. DATE: 2/25/79 82 CHECKED BY: DATE CHECKED: 19

LACING DESIGN -



4#10 steel
TEMP. STEEL
5#12 steel
FLEXURAL STEEL

FIXED-FIXED SUPPORTS

$$L = 12 \text{ FT} = 144 \text{ in}$$

$$M = \frac{w L^2}{12}$$

$$M = \frac{.31 (90,000) 6}{12} = 11,100 \text{ in-lb}$$

$$f_u = \frac{16 M}{L^2} = \frac{16 (11,100)}{(144)^2} = 87 \text{ psi}$$

FLEX. BAR, $A_s > .0025 b d_c = .0025 (12) (6) = .18 \text{ in}^2$
TEMP. BAR, $A_s > .0015 b d_c = .0015 (12) (6) = .108 \text{ in}^2$
 $A_s = 4 \text{ BAR} = .20 \text{ in}^2 > 1/2 \checkmark$

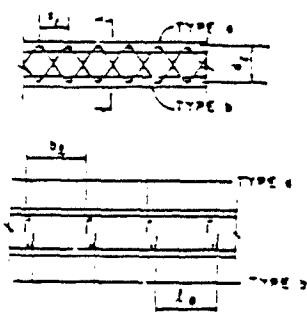
$$v_u = \frac{f_u (L/2 - d_c)}{d_c} = \frac{87 (144/2 - 6)}{6} = 60.3 \text{ psi}$$

$$p = A_s / d_c b = \frac{.31}{(6)(12)} = .0043$$

$$f_c = \phi (.85 \sqrt{f_c'} + 2500 p) = .85 (.85 \sqrt{4000} + 2500 (.0043)) = 111 \text{ psi}$$

$$f_{c \text{ max}} = 2.28 \phi \sqrt{f_c'} = 2.28 (.85) \sqrt{4000} = 123 \text{ psi}$$

$$v_c > v_u \text{ use } v_c \quad f_u' = 111 = 111 \text{ psi}$$



METHOD #2

$$s/d_1 = \frac{12}{8.5} = 1.4, \quad \frac{7 d_0}{d_1} = \frac{7 (6.5)}{8.5} = .51$$

FROM FIG. 6-19 OF TMS, $\alpha = 40^\circ$

$$A_T = \frac{v_u' s_1 b_1}{.85 f_y (\sin \alpha + \cos \alpha)} = \frac{(111) (12) (12)}{.85 (75,000) (\sin 40^\circ + \cos 40^\circ)}$$

$$A_T = .18 \text{ in}^2$$

USE 5 BAR

SHEET NO. 11 OF 32

DOUBLE STICKED WALL A

$T_c = 24"$
 $F_{LX} = 7 @ 2$
 $F_{HY} = 6 @ 12$
 $L_{HN} = 5 \text{ in. } 2$

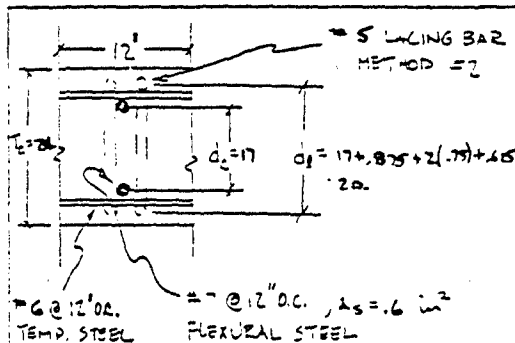
REQUIRED STEEL AREA = .0303 SQ IN/IN * .4115 SQ IN/FT

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COMPUTATION SHEET

SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CER
SUBJECT: LACING/DOUBLE STACKED WALL A
BY: DKETCHUM DATE: 23 SEP 83 CHECKED BY: DATE CHECKED: 19

LACING DESIGN



FIXED-FIXED SUPPORTS

$$L = 20 \text{ FT} = 240 \text{ in}$$

$$M = \frac{1}{6} F_u L_c$$

$$M = \frac{.6 (72000) 17}{12} = 61 \times 10^4 \frac{\text{in-lb}}{\text{in}}$$

$$r_u = \frac{16 M}{L^2} = \frac{16 (61 \times 10^4)}{(240)^2} = 17 \text{ psi}$$

$$\text{FLEX. BAR, } A_s > .0025 b d_c (.0025 (12) (17)) = .51 \text{ in}^2$$

$$\text{TEMP. BAR, } A_s > .0015 b d_c (.0015 (12) (17)) = .37 \text{ in}^2$$

$$A_s = 6 \text{ BAR} = .44 \text{ in}^2 > \frac{1}{4}$$

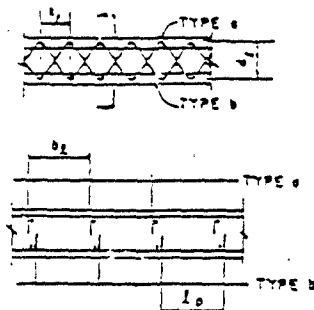
$$v_u = \frac{r_u \left(\frac{L}{2} - d_c \right)}{d_c} = \frac{17 \left(\frac{240}{2} - 17 \right)}{17} = 103 \text{ psi}$$

$$p = A_s / d_c b = \frac{.6}{(17)(12)} = .0029$$

$$v_c = \phi (1.9 \sqrt{f'_c} + 2500 p) = .85 (1.9 \sqrt{4000} + 2500 (.0029)) = 108 \text{ psi}$$

$$v_{c \text{ max}} = 2.28 \phi \sqrt{f'_c} = 2.28 (.85) \sqrt{4000} = 123 \text{ psi}$$

$$v < v_c \text{ USE } v_c \therefore v_u' = 108 = 108 \text{ psi}$$



METHOD #2

$$s_l / d_c = \frac{12}{20} = .6, \quad \frac{7 D_s}{d_l} = \frac{7 (.625)}{20} = .22$$

FROM FIG. 6-19 OF TMS, $\alpha = 67^\circ$

$$A_T = \frac{v_u' s_l b_l}{.85 f_y (\sin \alpha + \cos \alpha)} = \frac{108 (12) (12)}{.85 (60000) (\sin 67 + \cos 67)}$$

$$A_T = .23 \text{ in}^2$$

USE 5 BAR

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
13 OF 32

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: Reinforcing Bar Requirements
BY: D. J. H. DATE: 2-2-92 CHECKED BY: DATE CHECKED: 19

DOUBLE STACKED WALL B

SUMMARY -

TOT 24"
FLEX - #8 @ 12
T.M.P. - #6 @ 12
L.W.I. - #7 MESH 2

COMPUTER OUTPUT -

FIXED - FIXED SUPPORT CONDITIONS

LOAD PARAMETERS F1 = .5549E+06 F2 = .2638E+04
T2 = .3300E+03 T4 = .1000E+02

BEAM PARAMETERS L = .75.00 THICK = 24.00
D = 3.50 D' = 3.50
ROTATION = 1.00 T.N.A.T.L. = .396E-02

FC' = 4000. FDY = 72000.

IE = .1152E+04 IC = .1230E+03 IA = .5375E+03
XEL = .1006E+07 KELP = .2012E+06
L1 = .1070E+05 L2 = .1427E+05

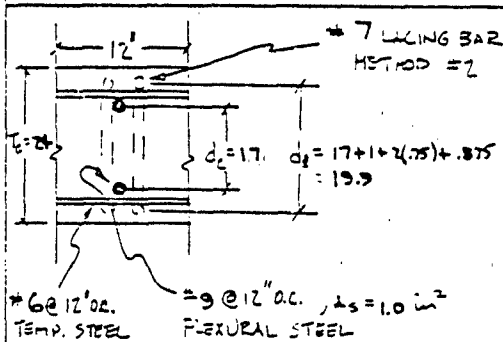
REQUIRED STEEL AREA = .0699 SQ IN/IN = .8393 SQ IN/FT

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
14 OF 37

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: LACING/DOWBLE STACKED/WALL B
BY: D. KETTER DATE: 12 SEP 79 BY: CHECKED BY: DATE CHECKED: 19

LLONG DESIGN



FIXED-FIXED SUPPORTS

$L = 8 \text{ FT} = 96 \text{ in}$

$M = \frac{w L^2}{12}$

$M = \frac{1.0 (72000) 17}{12} = 1.0 \times 10^5 \frac{\text{in-lb}}{\text{in}}$

$r_u = \frac{16 M}{L^2} = \frac{16 (1.0 \times 10^5)}{(96)^2} = 1.77 \text{ psi}$

FLEX. BAR, $A_s > .0025 b d_c = .0025 (12)(17) = .51 \text{ in}^2$
TEMP. BAR, $A_s > .0015 b d_c = .0015 (12)(17) = .38 \text{ in}^2$
 $A_s = 6 \text{ BAR} = .44 \text{ in}^2 > \frac{1}{4}$

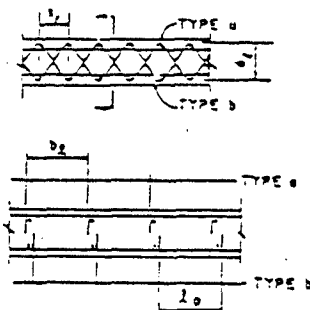
$v_u = \frac{r_u (\frac{L}{2} - d_c)}{d_c} = \frac{1.77 (\frac{96}{2} - 17)}{17} = 323 \text{ psi}$

$p = A_s / d_c b = \frac{1.0}{(17)(12)} = .005$

$v_c = \phi (1.9 \sqrt{f'_c} + 2500 p) = .85 (1.9 \sqrt{4000} + 2500 (.005)) = 113 \text{ psi}$

$v_{c \text{ max}} = 2.28 \phi \sqrt{f'_c} = 2.28 (.85) \sqrt{4000} = 123 \text{ psi}$

$v_u > 2 v_c$ USE $v_u - v_c$ $\therefore v_c = 323 - 113 = 210 \text{ psi}$



METHOD #2

$x/d_c = \frac{12}{19.9} = .60$, $\frac{7 D_s}{d_c} = \frac{7(.875)}{19.9} = .31$

FROM FIG. 6-19 OF TMS, $\lambda = 70^\circ$

$A_v = \frac{v_u' S_t b_f}{.85 f_y (\sin \lambda + \cos \lambda)} = \frac{(210)(12)(12)}{.85 (60,000) (\sin 70^\circ + \cos 70^\circ)}$

$A_v = .46 \text{ in}^2$

USE 7 BAR

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COMPUTATION SHEET

SHEET NO.
15 OF 32

PROJECT NO.: 02-7012 SPONSOR: CERL
SUBJECT: REINFORCING FOR REPAIRS
BY: KATHELYN DATE: 9/23/19 82 CHECKED BY: DATE CHECKED: 19

DOUBLE STACKED WALL C

SUMMARY -

$T_c = 24"$
FLX = #9@12
TEMP = #6@12
LACING = #5@12

COMPUTER OUTPUT -

FIXED - FIXED SUPPORT CONDITIONS

LOAD PARAMETERS F1 = .1459E+05 F2 = .6048E+04
T2 = .6800E-02 T4 = .1000E+02

BEAM PARAMETERS L = 215.00 THICK = 24.00
D = 3.50 D* = 3.50
ROTATION = 1.00 TNATL = .200E-01

FC* = 4000. FDT = 72000.

IE = .1152E+04 IC = .1261E+03 IA = .5391E+03
KEL = .8855E+05 KELP = .1771E+05
LI = .4904E+04 L2 = .5539E+04

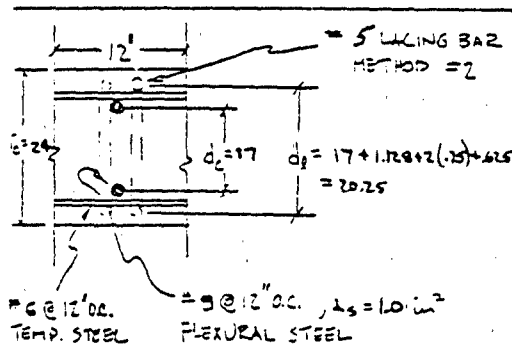
REQUIRED STEEL AREA = .0721 SQ IN/IN = .3654 SQ IN/FT

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
16 OF 32

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: LACING / DOUBLE STACKED / WALK
BY: DS DATE: 24 SEP 82 CHECKED BY: DATE CHECKED: 19

LACING DESIGN



FIXED-FIXED SUPPORTS

$$L = 18 \text{ FT} = 216 \text{ in}$$

$$M = \frac{w L^2}{8}$$

$$M = \frac{1.0 (72000) 17}{8} = 1.5 \times 10^5 \frac{\text{in-lb}}{\text{in}}$$

$$f_u = \frac{16 M}{L^2} = \frac{16 (1.5 \times 10^5)}{(216)^2} = 35 \text{ psi}$$

FLEX. BAR, $A_s = .0025 d_c = .0025 (12)(17) = .51 \text{ in}^2$
TEMP. BAR, $A_s = .0015 d_c = .0015 (12)(17) = .37 \text{ in}^2$
 $A_T = .6 \text{ BAR} = .44 \text{ in}^2 > \frac{1}{2}$

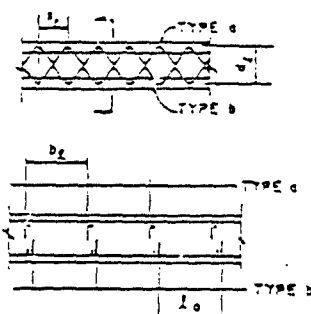
$$v_u = \frac{f_u \left(\frac{L}{2} - d_c \right)}{d_c} = \frac{35 \left(\frac{216}{2} - 17 \right)}{17} = 187 \text{ psi}$$

$$p = A_s / d_c b = \frac{1.0}{(17)(12)} = .005$$

$$v_c = \phi (1.9 \sqrt{f'_c} + 2500 p) = .85 (1.9 \sqrt{4000} + 2500 (.005)) = 112 \text{ psi}$$

$$v_{c, \text{max}} = 2.28 \phi \sqrt{f'_c} = 2.28 (.85) \sqrt{4000} = 123 \text{ psi}$$

$$v_c < v_u < 2 v_c \text{ use } v_c \therefore v_c = 112 = 112 \text{ psi}$$



METHOD #2

$$s/d_e = \frac{12}{20.25} = .59, \quad \frac{7 d_e}{d_e} = \frac{7(.25)}{20.25} = .216$$

FROM FIG. 6-19 OF TMS, $\alpha = 66^\circ$

$$A_T = \frac{v_u' s_e b_1}{.85 f_y (\sin \alpha + \cos \alpha)} = \frac{(112)(12)(12)}{.85 (60000) (\sin 66 + \cos 66)}$$

$$A_T = .24 \text{ in}^2$$

USE #5 BAR

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
17 OF 32

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: REINFORCING BAR REQUIREMENTS
BY: D. Korman DATE: SEP 19 87 CHECKED BY: DATE CHECKED: 19

DOUBLE STACKED DIVISION WALL

SUMMARY -

MAX. ROTATION = 6.44°
TL = 12"
FLEX - #5 @ 12"
TEMP - #4 @ 12"
LACING - #7 @ 12"

COMPLETE OUTPUT -

STRUCTURAL ELEMENT : DIV WALL

LOADS ON 1" SECTION :

PEAK SHOCK FORCE = .5549E+05 LB
PEAK QUASI-STATIC FORCE = ***** LB
SHOCK FORCE DURATION = .3300E-03 SEC
QUASI-STATIC FORCE DURATION = .1000E+01 SEC

MATERIAL STRENGTHS :

FC' = 4000. PSI
FDY = 90000. PSI

CROSS-SECTION :

TOTAL CONCRETE THICKNESS = 12.0 INCHES
COVER (TOP AND BOTTOM) = 2.00 INCHES
EFFECTIVE STEEL MOMENT ARM = 7.00 INCHES
STEEL SIZE AND SPACING = #5 AT 12 INCHES
STEEL AREA (EACH FACE) = .31 SQ IN/FT
EFFECTIVE MOMENT OF INERTIA I = .9113E+03 IN⁴/FT
PLASTIC MOMENT CAPACITY = .1966E+03 (IP-IN/FT)
ELEMENT LENGTH = 144.00 INCHES
NATURAL PERIOD = .1323E-01 SECONDS

SUPPORT CONDITIONS : FIXED - FIXED

RESPONSE

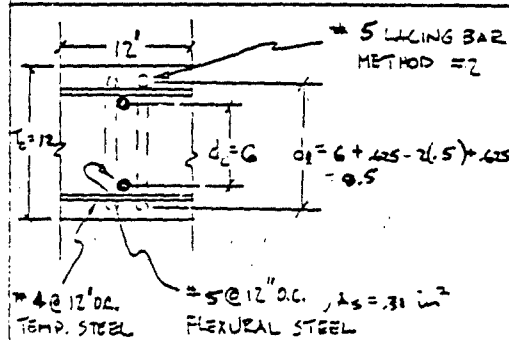
MAXIMUM DEFLECTION = 9.13 INCHES
MAXIMUM ROTATION = 6.44 DEGREES

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
18 OF 22

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: LACING/DOUBLE STACKED/DIVIDING WALL
BY: D. KETNUN DATE: 24 SEP 82 CHECKED BY: DATE CHECKED: 18

LACING DESIGN -



FIXED-FIXED SUPPORTS

$$L = 8 \text{ FT} = 96 \text{ in}$$

$$M = \frac{w L^2}{12}$$

$$M = \frac{.31 (20000) 6}{12} = 1.4 \text{ in-lb}$$

$$r_u = \frac{16 M}{L^2} = \frac{16 (1.4 \times 10^6)}{(96)^2} = .18 \text{ in}^2$$

$$\text{FLEX. BAR, } A_s = .0025 \text{ in}^2 \text{ (12) (6)} = .18 \text{ in}^2$$

$$\text{TEMP. BAR, } A_s = .0015 \text{ in}^2 \text{ (12) (6)} = .13 \text{ in}^2$$

$$A_T = .31 \text{ in}^2 = .25 \text{ in}^2 + \frac{1}{4} \text{ in}^2$$

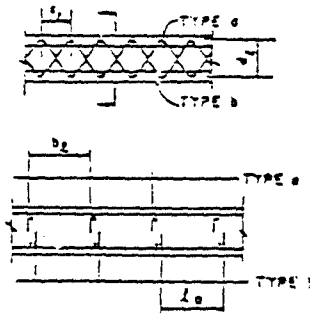
$$v_u = \frac{r_u (\frac{L}{2} - d_c)}{d_c} = \frac{24 (\frac{96}{2} - 6)}{6} = 168 \text{ psi}$$

$$p = A_s / d_c b = \frac{.31}{(2)(12)} = .0043$$

$$v_c = \phi (1.9 \sqrt{f'_c} + 2500 p) = .85 (1.9 \sqrt{4000} + 2500 (.0043)) = 111 \text{ psi}$$

$$v_{c \text{ max}} = 2.28 \phi \sqrt{f'_c} = 2.28 (.85) \sqrt{4000} = 123 \text{ psi}$$

$$v_u < v_u < 2 v_c \quad \text{USE } v_c = 111 = 111 \text{ psi}$$



METHOD #2

$$x/d_c = \frac{12}{8.5} = 1.4, \quad \frac{7 d_o}{d_c} = \frac{7 (6.25)}{8.5} = 5.1$$

FROM FIG. 6-19 OF TMS, $\alpha = 40^\circ$

$$A_T = \frac{v_u S_L b_L}{.85 f_y (\sin \alpha + \cos \alpha)} = \frac{(111)(12)(12)}{.85 (75000) (\sin 40^\circ + \cos 40^\circ)}$$

$$A_T = .18 \text{ in}^2$$

USE 5 BAR

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COMPUTATION SHEET

SHEET NO.
19 OF 37

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: ~~INTERMITTENT BURST DETECTION~~
BY: ~~DKH/THH~~ DATE: 24 SEP 19 92 CHECKED BY: DATE CHECKED: 19

PITS WALL A

SUMMARY -

$T_c = 24"$

FLEX = #10 @ 12

TEMP = #6 @ 12

LACING = #5 @ 12

COMPUTER OUTPUT -

FIXED - FIXED SUPPORT CONDITIONS

LOAD PARAMETERS F1 = .3600E+06 F2 = .4080E+04
T2 = .5800E-03 T4 = .1000E+02

BEAM PARAMETERS L = 240.00 THICK = 24.00
D = 3.50 D* = 3.50
ROTATION = 1.00 INATL = .244E-01

FC* = 4000. FDY = 72000.

IE = .1152E+04 IC = .1544E+03 IA = .5532E+03
KEL = .6599E+05 KELP = .1320E+05
L1 = .5641E+04 L2 = .7522E+04

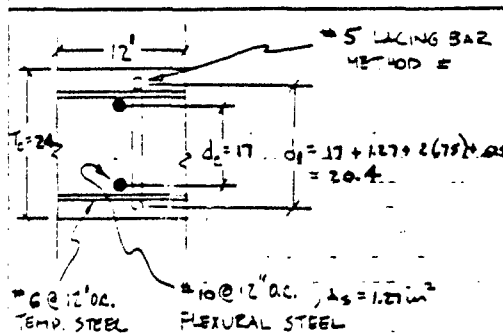
REQUIRED STEEL AREA = .0922 50 IN/IN = 1.1061 50 IN/FT

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COMPUTATION SHEET

SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CEEL
SUBJECT: LASUS / PITS / WALL A
BY: D. K. M. H. M. DATE: 23 SEP 82 CHECKED BY: _____ DATE CHECKED: 19

LACING DESIGN



FIXED-FIXED SUPPORTS

$L = 20 \text{ FT} = 240 \text{ in}$

$\gamma = \frac{A_s E_s d_c}{b}$

$M = \frac{1.27 (7.5 \times 10^5) 17}{2} = 1.32 \times 10^6 \text{ in-lb}$

$r_u = \frac{16 M}{L^2} = \frac{16 (1.32 \times 10^6)}{(240)^2} = 36 \text{ psi}$

FLEX. BAR, $A_s > 0.015 b d_c \times 0.025 (12)(17) = 0.061 \text{ in}^2$
TEMP. BAR, $A_s > 0.015 b d_c \times 0.015 (12)(17) = 0.046 \text{ in}^2$
 $A_s = 6 \text{ BAR} = 44 \text{ in}^2 > \frac{1}{2}$

$r_u = \frac{r_u (L^2 - d_c)}{d_c} = \frac{36 (240^2 - 17)}{17} = 218 \text{ psi}$

$p = A_s / d_c b = \frac{1.27}{(17)(12)} = 0.0062$

$\sigma_c = \phi (1.9 \sqrt{f'_c} + 2500 p) = 0.85 (1.9 \sqrt{4000} + 2500 (0.0062)) = 115 \text{ psi}$

$\sigma_{c \text{ max}} = 2.28 \phi \sqrt{f'_c} = 2.28 (0.85) \sqrt{4000} = 123 \text{ psi}$

$\sigma_c < \sigma_{c \text{ max}} \text{ Use } \sigma_c \therefore r_u = 115 = 115 \text{ psi}$

$\frac{d_c}{d_t} = \frac{17}{20.4} = 0.83 \quad \frac{r_u}{d_t} = \frac{7(1)}{20.4} = 0.34$

$\alpha = 18.5^\circ \text{ or } 73^\circ, \alpha = 73^\circ$

$A_T = \frac{r_u S_1 b_1}{0.85 f_y (\sin \alpha + \cos \alpha)} = \frac{(115)(12)(12)}{0.85 (60,000) (\sin 73^\circ + \cos 73^\circ)}$

$A_T = 0.26 \text{ in}^2$

USE 5 BAR

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
21 OF 31

PROJECT NO: 02-7092 SPONSOR: CERL
SUBJECT: REINFORCING BAR REQUIREMENTS
BY: DATE 21 MAY 19 82 CHECKED BY: DATE CHECKED: 19

PITS Wall B

SUMMARY -

$T_c = 24"$
FLEX - #10 @ 12
TEMP - #6 @ 12
LACING - #5 MESH

COMPUTER OUTPUT -

FIXED - FIXED SUPPORT CONDITIONS

LOAD PARAMETERS $F_1 = .7680E+06$ $F_2 = .4090E+04$
 $T_2 = .3600E-03$ $T_4 = .1000E+02$

BEAM PARAMETERS $L = 240.00$ THICK = 24.00
 $D = 3.50$ $D' = 3.50$
ROTATION = 1.00 $TNATL = .243E-01$

$FC' = 4000.$ $FDY = 72000.$

$IE = .1152E+04$ $IC = .1701E+03$ $IL = .5511E+03$
 $KEL = .6578E+05$ $KE_P = .1336E+05$
 $L1 = .6351E+04$ $L2 = .9468E+04$

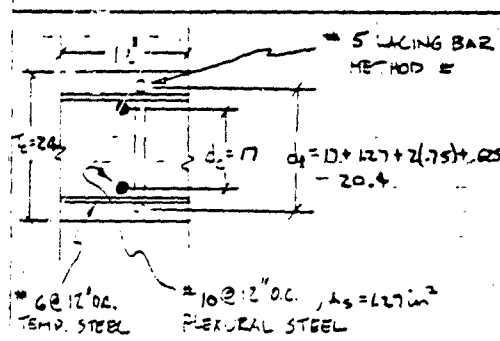
REQUIRED STEEL AREA = .1039 SQ IN/IN = 1.2454 SQ IN/FT

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COMPUTATION SHEET

SHEET NO.
22 OF 32

PROJECT NO.: 07-7027 SPONSOR: CEC
SUBJECT: LACING / FITS / WALL B
BY: DKETUNA DATE: 24 SEP 82 CHECKED BY: DATE CHECKED: 19

LACING DESIGN



FIXED-FIXED SUPPORTS

$L=20 \text{ FT}=240 \text{ in}$

$M = \frac{A_s F_y d_c}{b}$

$M = \frac{127 (72,000) 17}{12} = 1.3 \times 10^5 \frac{\text{in-lb}}{\text{in}}$

$r_u = \frac{16 M}{L^2} = \frac{16 (1.3 \times 10^5)}{(240)^2} = 1 \text{ psi}$

FLEX. BAR, $A_s > .0025 b d_c .0025 (17) (17) = .51 \text{ in}^2$
TEMP. BAR, $A_s > .0015 b d_c .0015 (17) (17) = .38 \text{ in}^2$
 $A_s = 6 \text{ BAR} = 44 \text{ in}^2 > \frac{1}{2} \checkmark$

$r_u = \frac{r_u (\frac{L}{d_c} - 17)}{d_c} = \frac{36 (\frac{240}{17} - 17)}{17} = 218 \text{ psi}$

$p = A_s / d_c = \frac{127}{(17)(17)} = .0062$

$r_c = \phi (1.9 \sqrt{f_c} + 2500 p) = .85 (1.9 \sqrt{4000} + 2500 (.0062)) = 115 \text{ psi}$

$r_{c \text{ max}} = 2.28 \phi \sqrt{f_c} = 2.28 (.85) \sqrt{4000} = 123 \text{ psi}$

$V_c < V_u < 2V_c$ USE V_c $r_c = 115 = 115 \text{ psi}$

$\frac{V_u}{b d_t} = \frac{12}{20.4} = .59$, $\frac{V_u}{b d_t} = \frac{7(62)}{20.4} = .23$

FROM FIG. 6-19 OF TMS, $\alpha = 70^\circ$

$A_T = \frac{r_u' S_t b_t}{.85 F_y (\sin \alpha + \cos \alpha)} = \frac{(115) (12) (12)}{.85 (60,000) (\sin 70 + \cos 70)}$

$A_T = .25 \text{ in}^2$

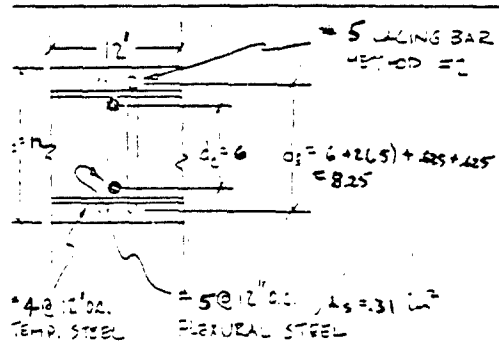
USE * 5 BAR

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
23 OF 32

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: LASING / PITS / DIVIDING WALL
BY: D. KETANM DATE: 24 SEP 19 82 CHECKED BY: DATE CHECKED: 19

LONG DESIGN



FIXED-FIXED SUPPORTS

$L = 5 \text{ FT} = 60 \text{ in}$
 $M = \frac{w L^2}{6}$
 $M = \frac{.31 (36000) 6}{12} = 1810 \text{ in-lb}$
 $f_u = \frac{16 M}{L^2} = \frac{16 (1810)}{(60)^2} = 62 \text{ psi}$

FLX. BAR, $A_s = .0025 (60) (12) (6) = .108 \text{ in}^2$
TEMP. BAR, $A_s = .0015 (60) (12) (6) = .065 \text{ in}^2$
 $A_s = .173 \text{ in}^2 > 4 \text{ BAR} = .20 \text{ in}^2 > 1/2 \text{ in}^2$

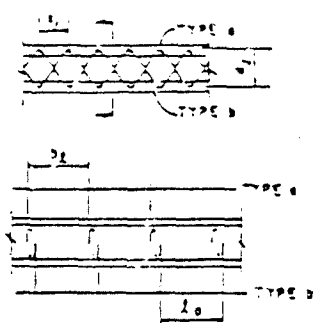
$v_u = \frac{f_u (L - d_c)}{d_c} = \frac{62 (60 - 6)}{6} = 248 \text{ psi}$

$p = \frac{A_s}{d_c b} = \frac{.31}{(6)(12)} = .0043$

$f_c = \phi (1.9 \sqrt{f_c'} + 2500 p) = .85 (1.9 \sqrt{4000} + 2500 (.0043)) = 111 \text{ psi}$

$f_c \text{ max} = 2.29 \phi \sqrt{f_c'} = 2.29 (.85) \sqrt{4000} = 123 \text{ psi}$

$v_u > 2v_c$ USE $v_u - v_c$ $f_c' = 248 - 111 = 137 \text{ psi}$



$\frac{f_u}{f_c'} = \frac{12}{248} = 1.45$, $\frac{f_u}{f_c'} = \frac{7(60)}{9.25} = .53$

FROM FIG. 6-19 OF TMS, $\lambda = 40^\circ$

$A_T = \frac{v_u' S_L b_L}{.85 f_y (\sin \lambda + \cos \lambda)} = \frac{(137)(12)(12)}{.85 (7500) (\sin 40 + \cos 40)}$

$A_T = .22 \text{ in}^2$
USE 5 BAR

METHOD #2

SHEET NO.

24 OF 32

MAINTENANCE BAY WALL

Summary-

$T_c = 24''$
 $F_{ux} = 2^{*}9 @ 12$
 $T_{inf} = 6 @ 12$
 $L_{acing} = 7 \text{ mm} @ 2$

Computer Output -

FIXED - FIXED SUPPORT CONDITIONS

```
LOAD PARAMETERS      F1 = .6643E+06      F2 = .8880E+04
                    T2 = .3000E-03      T4 = .1000E+02
```

```

BEAM PARAMETERS  L = 240.00      THICK = 24.00
                  D = 3.50        D* = 3.50
                  ROTATION = 1.00  TNALE = .239E-01

```

FC* = 4000. FDY = 72000.

```
IE = .1152E+04      IC = .2178E+03      IA = .5849E+03
XEL = .6917E+05      XELP = .1394E+05
L1 = .8659E+04      L2 = .1155E+05
```

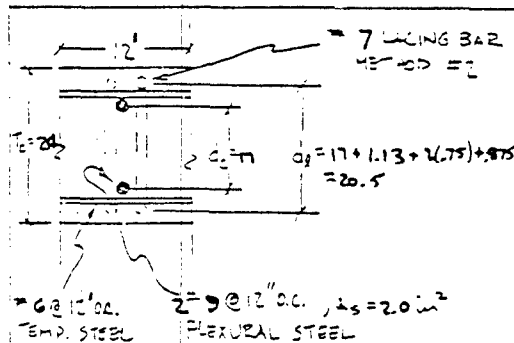
REQUIRED STEEL AREA = .1415 SQ IN/IN = 1.6979 SQ IN/FT

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
25 OF 32

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: LACING / MAINTENANCE BAY / WALL
BY: SKR DATE: 23 SEP 82 CHECKED BY: _____ DATE CHECKED: _____ 19

LONG DESIGN



FIXED-FIXED SUPPORTS

$L = 20 \text{ FT} = 240 \text{ in}$

$M = \frac{w L^2}{12}$

$M = \frac{2.0 (72000) 17}{12} = 20,105 \frac{\text{in-lb}}{\text{in}}$

$f_u = \frac{16 M}{L^2} = \frac{16 (20,105)}{(240)^2} = .57 \text{ psi}$

FLEX. BAR, $A_s = .0025 \text{ od}_c (12)(17) = .51 \text{ in}^2$
TEMP. BAR, $A_s = .0015 \text{ od}_c (12)(17) = .38 \text{ in}^2$
 $A_s = 6 \text{ BAR} = .44 \text{ in}^2 > \frac{1}{2}$

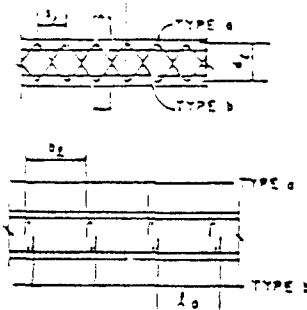
$v_u = \frac{f_u (L - d_c)}{d_c} = \frac{(57)(17 - 17)}{17} = 343 \text{ psi}$

$p = A_s / d_c = \frac{2.0}{(17)(17)} = .010$

$v_c = \phi (1.9 \sqrt{f'_c} + 2500 p) = .85 (1.9 \sqrt{4000} + 2500 (.010)) = 123 \text{ psi}$

$v_{c \text{ max}} = 2.28 \phi \sqrt{f'_c} = 2.28 (.85) \sqrt{4000} = 123 \text{ psi}$

$v_u > 225 \text{ Use } v_{c \text{ max}} \therefore v_c = 343 - 123 = 220 \text{ psi}$



METHOD #2

$\frac{v_u}{d_c} = \frac{12}{20.5} = .55, \frac{7(1)}{d_c} = \frac{7(1)}{20.5} = .32$

FROM FIG. 6-19 OF TMS, $\alpha = 75^\circ$

$A_v = \frac{v_u' S_1 b_1}{.85 f_y (\sin \alpha + \cos \alpha)} = \frac{(220)(12)(12)}{.85 (60000) (.525 + .85)} = .51 \text{ in}^2$

$A_v = .51 \text{ in}^2$
USE #7 BAR

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
26 OF 32

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: REINFORCING BAR REQUIREMENTS
BY: D. KENNEDY DATE: 24 SEP 19 92 CHECKED BY: DATE CHECKED: 19

MAZE WALL A

SUMMARY -

$T_c = 24"$
FLUX - #7 @ 12
TEMP - #6 @ 12
LACING - #5 METHOD 2

COMPUTER OUTPUT -

FIXED - FIXED SUPPORT CONDITIONS

LOAD PARAMETERS $F1 = .1757E+06$ $F2 = .1824E+04$
 $T2 = .5100E-03$ $T4 = .1000E+02$

BEAM PARAMETERS $L = 240.00$ THICK = 24.00
 $D = 3.50$ $D' = 3.50$
ROTATION = 1.00 $TNATL = .253E-01$

$FC' = 4000.$ $FDY = 72000.$

$IE = .1152E+04$ $IC = .5212E+02$ $IA = .5071E+03$
 $KE = .5132E+05$ $KEPL = .1226E+05$
 $LI = .1840E+04$ $L2 = .2453E+04$

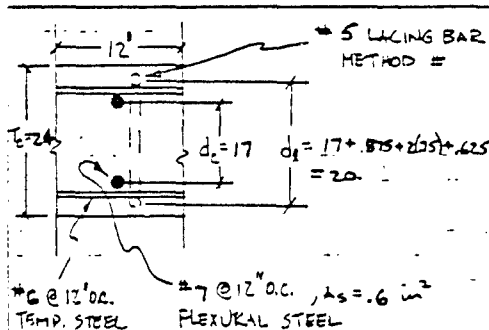
REQUIRED STEEL AREA = .0301 SQ IN/IN = .3507 SQ IN/FT

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
27 OF 32

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: LACING / MATE / WALL A
BY: D. KETUM DATE: 13 SEP 92 CHECKED BY: DATE CHECKED: 19

LACING DESIGN



FIXED-FIXED SUPPORTS

L = 20 FT = 240 in

M = $\frac{A_s f_y d_e}{b}$

M = $\frac{.6 (72,000) 17}{12} = 4.3 \times 10^4 \frac{\text{in-lb}}{\text{in}}$

r_u = $\frac{16 M}{L^2} = \frac{16 (4.3 \times 10^4)}{(240)^2} = 12 \text{ psi}$

FLEX. BAR, A_s > .0025 b d_e = .0025 (12) (17) = .6 in² ✓
TEMP. BAR, A_t > .0015 b d_e = .0015 (12) (17) = .3 in² ✓
A_t = .6 BAR = .44 in² > .3 ✓

r_u = $\frac{r_u (\frac{L}{d_e} - d_e)}{d_e} = \frac{12 (\frac{240}{17} - 17)}{17} = 73 \text{ psi}$

p = $\frac{A_s}{d_e b} = \frac{.6}{(12)(17)} = .0029$

v_c = $\phi (1.9 \sqrt{f'_c} + 2500 p) = .85 (1.9 \sqrt{4000} + 2500 (.0029)) = 108.4 \text{ psi}$

v_{c max} = $2.28 \phi \sqrt{f'_c} = 2.28 (.85) \sqrt{4000} = 123 \text{ psi}$

v_c > v_u use v_c ∴ v_c = 108 = 108 psi

s₁/d_t = $\frac{12}{20} = .6$, $\frac{7D_0}{d_t} = \frac{7(.625)}{20} = .22$

FROM FIG. 6-19 OF TMS, α = 67°

A_T = $\frac{r_u' s_1 b_1}{.85 f_y (\sin \alpha + \cos \alpha)} = \frac{(108)(12)(12)}{.85 (.0025) (\sin 67 + \cos 67)}$

A_T = .23 in²

use #5 BAR

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: REINFORCING BAR REQUIREMENTS
BY: DKH DATE: 25 SEP 19 82 CHECKED BY: _____ DATE CHECKED: 19

MAZE WALL B

SUMMARY -

$T_c = 24"$
FLEX = #10 @ 12
TENS = #6 @ 12
LACING = #5 MESH 2

COMPUTER OUTPUT -

FIXED - FIXED SUPPORT CONDITIONS

LOAD PARAMETERS F1 = .5209E+06 F2 = .4550E+04
T2 = .4009E+03 T4 = .1000E+02

BEAM PARAMETERS L = 240.00 THICK = 24.00
D = 3.50 D* = 3.50
ROTATION = 1.00 TNAFL = .245E-01

FC* = 4000. FDY = 72000.

IE = .1152E+04 IC = .1471E+03 IA = .5495E+03
KE = .6561E+05 KEPL = .1312E+05
L1 = .5316E+04 L2 = .7088E+04

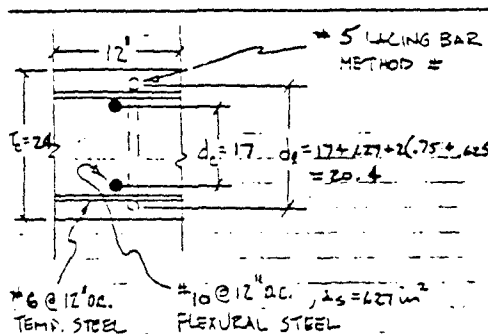
REQUIRED STEEL AREA = .0869 SQ IN/IN = 1.0423 SQ IN/FT

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: LACING / MATE / WALL B
BY: D. KOTCHUM DATE: 23 SEP 19 82 CHECKED BY: DATE CHECKED: 19

LACING DESIGN



FIXED-FIXED SUPPORTS

$$L = 20 \text{ FT} = 240 \text{ IN}$$

$$M = \frac{1}{12} F_c d_c$$

$$M = \frac{1.27 (12,000) 17}{12} = 1.3 \times 10^5 \frac{\text{in-lb}}{\text{in}}$$

$$r_u = \frac{16 M}{L^2} = \frac{16 (1.3 \times 10^5)}{(240)^2} = 36 \text{ psi}$$

$$\text{FLEX. BAR, } A_s = 0.0025 b d_c = 0.0025 (12) (17) = 5.1 \text{ in}^2$$

$$\text{TEMP. BAR, } A_s = 0.0018 b d_c = 0.0018 (12) (17) = 3.6 \text{ in}^2$$

$$A_T = 6 \text{ BAR} = .44 \text{ in}^2 \checkmark$$

$$r_u = \frac{r_u \left(\frac{L}{d_c} - 1 \right)}{d_c} = \frac{36 \left(\frac{240}{17} - 1 \right)}{17} = 218 \text{ psi}$$

$$p = A_s / d_c b = \frac{1.27}{(17)(12)} = 0.0062$$

$$r_c = \phi (1.9 \sqrt{f_c} + 2500 p) = .85 (1.9 \sqrt{4000} + 2500 (0.0062)) = 115 \text{ psi}$$

$$r_{c \text{ max}} = 2.28 \phi \sqrt{f_c} = 2.28 (.85) \sqrt{4000} = 123 \text{ psi}$$

$$V_u > V_c \quad \text{use } V_c \quad \therefore r_c = 115 = 115 \text{ psi}$$

$$x/d_1 = \frac{12}{20.4} = .59, \quad \frac{7 D_9}{d_1} = \frac{7(1.625)}{20.4} = .21$$

$$\text{FROM FIG. 6-19 OF TMS, } \alpha = 67^\circ$$

$$A_T = \frac{r_u' S_1 b_1}{.85 f_y (\sin \alpha + \cos \alpha)} = \frac{(115)(12)(12)}{.85 (60,000) (\sin 67 + \cos 67)}$$

$$A_T = .25 \text{ in}^2$$

$$\text{use } 5 \text{ BAR}$$

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.

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PROJECT NO.: 02-7092

SPONSOR: CERL

SUBJECT: LACING / MAZE / DIVIDING WALL

BY: D. K. R. H. M.

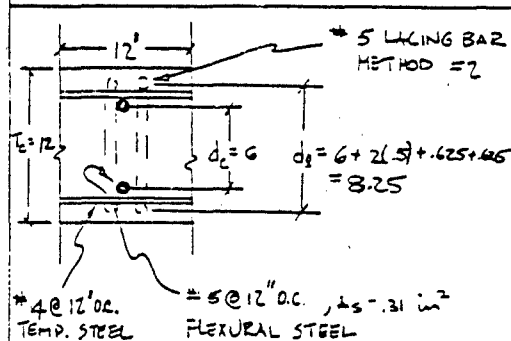
DATE: 2/25/79

CHECKED BY:

DATE CHECKED:

19

LACING DESIGN -



FIXED-FIXED SUPPORTS

$$L = 5 \text{ FT} = 60 \text{ in}$$

$$M = \frac{A_s F_y d_e}{b}$$

$$M = \frac{.31 (36000) 6}{12} = 1.4 \times 10^4 \frac{\text{in-lb}}{\text{in}}$$

$$r_u = \frac{16 M}{L^2} = \frac{16 (1.4 \times 10^4)}{(60)^2} = 62 \text{ psi}$$

$$\text{FLEX. BAR, } A_s > .0025 b d_e = .0025 (12) (6) = .18 \text{ in}^2$$

$$\text{TEMP. BAR, } A_s > .0015 b d_e = .0015 (12) (6) = .13 \text{ in}^2$$

$$A_T = 4 \text{ BAR} = .60 \text{ in}^2 > \frac{1}{4} \checkmark$$

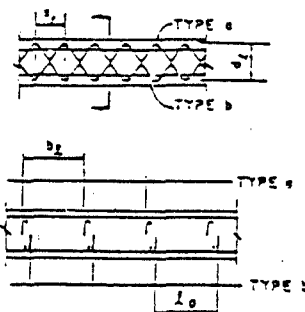
$$v_u = \frac{r_u \left(\frac{L}{d_e} - d_e \right)}{d_e} = \frac{62 \left(\frac{60}{6} - 6 \right)}{6} = 248 \text{ psi}$$

$$p = A_s / d_e b = \frac{.31}{(6)(12)} = .0043$$

$$v_c = \phi (1.9 \sqrt{f'_c} + 2500 p) = .85 (1.9 \sqrt{4000} + 2500 (.0043)) = 111 \text{ psi}$$

$$v_{c \text{ max}} = 2.29 \phi \sqrt{f'_c} = 2.29 (.85) \sqrt{4000} = 123 \text{ psi}$$

$$V_u > 2V_c \text{ Use } V_u - V_c \therefore \tau_c = 248 - 111 = 137 \text{ psi}$$



METHOD #2

$$s/d_l = \frac{12}{8.25} = 1.45, \frac{700 - (.625)}{d_l} = .53$$

FROM FIG. 6-19 OF TMS, $\alpha = 40^\circ$

$$A_T = \frac{v_u' s_l b_l}{.85 f_y (\sin \alpha + \cos \alpha)} = \frac{(137)(12)(6)}{.85 (7500) (\sin 40 + \cos 40)}$$

$$A_T = .22 \text{ in}^2$$

USE #5 BAR

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
31 OF 32

PROJECT NO.: 02-7092 SPONSOR: CSE
SUBJECT: REINFORCING BAR REQUIREMENTS
BY: D. K. K. DATE: 20 SEPT 19 92 CHECKED BY: DATE CHECKED: 19

DOUBLE STACKED CONCRETE/FLOOR
SUMMARY -

$T_c = 36"$

FLXING BAR = (2) #9 @ 12

TEMP. BAR = #7 @ 12

LACING BAR = #5 METHOD 2

COMPUTER OUTPUT -

MAINT ROOF

FIXED EDGES

PRESSURE - TIME VALUES

TIME	PRESS
0.	.23800E+03
.27000E-02	.37000E+02
.20000E+01	.37000E+02

SLAB PARAMETERS A = 540.00 B = 660.00
THICK = 36.00 TOP COVER = 4.00 BOTTOM COVER = 4.00
MAX DEFL = 4.70 TNATL = .628E-01

FCP = 4000. FDY = 72000.

IE = .2838E+04 IC = .6484E+03 IA = .2268E+04
EL = .1994E+08 KELPL = .5997E+07
L1 = .7493E+07 L2 = .1547E+08
MUMENT = .2838E+06

REQUIRED STEEL AREA = .1408 SQ IN/IN = 1.6894 SQ IN/FT

STEEL AREA IN EACH FACE FOR DOUBLY REINFORCED SECTION

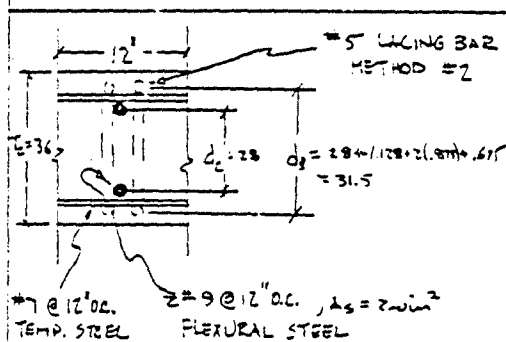
BAR SIZE & SPACING	AREA (SQ IN/FT) = (SQ IN/IN)
#10 @ 9"	1.59 .1408
#9 @ 7"	1.71 .1425

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
32 OF 32

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: LACING / DOUBLE STACKED / CEILING - 5 BAR
BY: D. K. M. DATE: 30 SEPT 82 CHECKED BY: DATE CHECKED: 19

LONG DESIGN



FIXED-FIXED SUPPORTS

$$L = 45 \text{ FT} = 540 \text{ in}$$

$$M = \frac{w L^2 d_c}{b}$$

$$M = \frac{2.0 (72000) 28}{12} = 341,000 \text{ in-lb}$$

$$r_u = \frac{16 M}{L^2} = \frac{16 (341,000)}{(540)^2} = 0.18 \text{ psi}$$

$$\text{FLEX. BAR, } A_s = 0.0025 \text{ } d_c = 0.0025 (12) (28) = 0.84 \text{ in}^2$$

$$\text{TEMP. BAR, } A_s = 0.0015 \text{ } d_c = 0.0015 (12) (28) = 0.504 \text{ in}^2$$

$$A_s = 0.7 \text{ BAR } 0.60 \text{ in}^2 \times 1/4$$

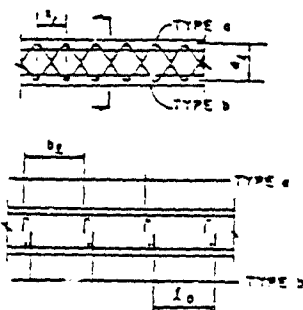
$$r_u = \frac{r_u \left(\frac{L}{d_c} - 1 \right)}{d_c} = \frac{0.18 \left(\frac{540}{28} - 1 \right)}{28} = 153 \text{ psi}$$

$$p = \frac{A_s}{d_c b} = \frac{0.84}{(28)(12)} = 0.006$$

$$r_c = \phi (1.9 \sqrt{f'_c} + 2500 p) = 0.85 (1.9 \sqrt{4000} + 2500 (0.006)) = 103.4 \text{ psi}$$

$$r_{c \text{ max}} = 2.29 \phi \sqrt{f'_c} = 2.29 (0.85) \sqrt{4000} = 123 \text{ psi}$$

$$r_u > r_c \text{ USE } r_c \therefore r_c = 103 = 103 \text{ psi}$$



METHOD #2

$$s/d_c = \frac{12}{31.5} = 0.38, \quad \frac{7 d_c}{d_l} = \frac{7 (12)}{31.5} = 1.6$$

$$\text{FROM FIG. 6-19 OF T145, } \alpha = 76^\circ$$

$$A_r = \frac{r_u' S_L b_r}{0.85 f_y (\sin \alpha + \cos \alpha)} = \frac{(103) (12) (12)}{0.85 (60,000) (\sin 76 + \cos 76)}$$

$$A_r = 0.24 \text{ in}^2$$

$$\text{USE } 5 \text{ BAR}$$

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COMPUTATION SHEET

SHEET NO.
1 OF 8

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: BLAST DOOR
BY: M. WHITNEY DATE: JAN 19 82 CHECKED BY: DATE CHECKED: 19

BLAST DOOR DESIGN SUMMARY

LOADING

LONG BAY

PIT BAY

$P_{SHOCK} = 100 \text{ psi}$

180

$t_{SHOCK} = 0.0046 \text{ sec}$

3.2×10^{-3}

$P_{QUASI} = 21 \text{ psi}$

26

FRAGMENT = 1.6 in STEEL MIN

RATE

2 in THICK

A36 STEEL

RESPONSE - ELASTIC

REBOUND PINS

14 - 1.0 in DIAMETER PINS

ANCHOR BOLTS

20 - 1.5 in DIAMETER BOLTS

$\frac{1}{2}$ in RATE LINING JAM

JAM FORMED INTO CONCRETE

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COMPUTATION SHEET

SHEET NO.
2 OF 3

PROJECT NO.: 02-7092

SPONSOR: CERL

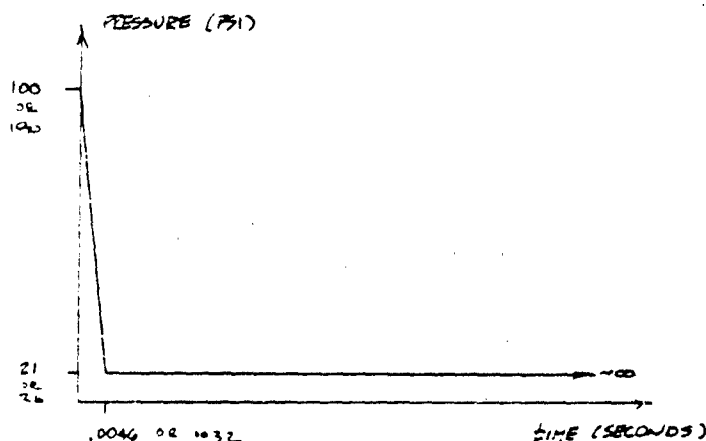
SUBJECT: BLAST DOOR DESIGN

BY: H. WHITNEY DATE: 17 AUG 19 52

CHECKED BY:

DATE CHECKED: 19

BLAST LOADING

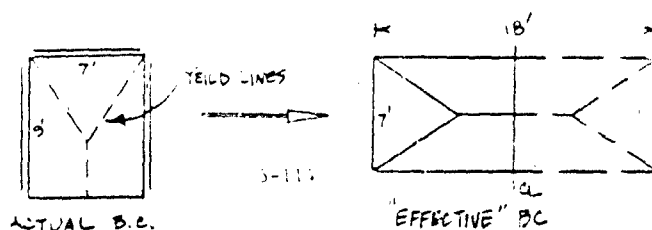


FRAGMENT LOADING

MINIMUM STEEL PLATE THICKNESS = 1.6 in

BOUNDARY CONDITION

THE DOOR WILL BE SUPPORTED ON THREE EDGES BY THE JAM. THE FLOOR EDGE WILL BE FREE. THE DOOR WILL BE ANALYZED AS A 4 SIDED SIMPLY-SUPPORTED PLATE OF TWICE THE AREA AS SHOWN. THE LOAD WILL BE APPLIED TO THE ENTIRE "EFFECTIVE" AREA.



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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: BLAS-2002
BY: J. WHITNEY DATE: JAN 19 82 CHECKED BY: DATE CHECKED: 19

THE STIFFNESS AND RESISTANCES FOR AN EQUIVALENT
1 D.O.F. SYSTEM ARE OBTAINED FROM THE SUPPRESSIVE
SHIELDS MANUAL. ⁴⁾ THESE VALUES ARE USED AS
INPUT INTO A 1 D.O.F. COMPUTER PROGRAM FOR
THE DETERMINATION OF DEFLECTION.

FOR THIS DOOR:

$$\frac{a}{b} = \frac{\text{WIDTH}}{\text{LENGTH}} = \frac{7'}{18'} = 0.39$$

THE CLOSEST VALUE IN THE SUPPRESSIVE SHIELDS MANUAL
IS $\frac{a}{b} = 0.5$, USE THIS VALUE. THE CORRESPONDING
TERMS ARE:

STRAIN	%	KLM	L_1	K
ELASTIC	.5	0.75	$\frac{1}{2}(12M_a + 9M_b)$	$201EI_a/a^2$
PLASTIC	.5	0.59	$\frac{1}{2}(12M_a + 9M_b)$	0

WHERE: $M_a = T Z_a = \frac{T a t^2}{4}$, $t = \text{PLATE THICKNESS}$
 $T = \sigma_y \times \text{DLF}$
 $DLF = 1.2$, DYNAMIC LOAD FACTOR

$$M_b = T Z_b = \frac{T b t^2}{4}$$

$E = \text{YOUNG'S MODULUS}$

$$I_a = \frac{1}{12} \frac{a t^3}{a} \quad 3-113$$

ALSO, $\text{MASS} = \rho V$, $\text{LOADS} = P a b$, $X_{cl} = L_1/K$

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COMPUTATION SHEET

SHEET NO.
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PROJECT NO.: 7-7092

SPONSOR: CERL

SUBJECT: BLAST DOOR

BY: M. WHITNEY

DATE: 7 AUG 19 82

CHECKED BY:

DATE CHECKED:

19

CONSIDER A SOLID STEEL PLATE ($a = 84$ in, $b = 216$ in)
WITH A36 STEEL, $\sigma_y = 36$ KSI

$$\bar{\sigma} = 36 \times 1.2 = 43.2 \text{ KSI}$$

THE MINIMUM PLATE THICKNESS IS FIXED BY FRAG. PENETRATION AS 1.5 in.
DETERMINE STIFFNESS AND RESISTANCE VALUES FOR
BOTH A 1.6 in AND 2 in PLATE:

	t THICKNESS (in)	
	1.6	2.0
$M_a = \frac{\bar{\sigma} a^3}{4}$	$(43,200)(84)(1.6)^2 / 4 = 2.3 \times 10^6$ lb-in	$(43,200)(84)(2)^2 / 4 = 3.6 \times 10^6$ lb-in
$M_b = \frac{\bar{\sigma} b^3}{4}$	$(43,200)(216)(1.6)^2 / 4 = 6.0 \times 10^6$ lb-in	$(43,200)(216)(2)^2 / 4 = 9.2 \times 10^6$ lb-in
$L_1 = \frac{1}{a} [12 M_a + 9 M_b]$	$\frac{1}{84} [12(2.3 \times 10^6) + 9(6.0 \times 10^6)] = 9.7 \times 10^5$ lb	$\frac{1}{84} [12(3.6 \times 10^6) + 9(9.2 \times 10^6)] = 1.5 \times 10^6$ lb
$I_a = \frac{1}{12} \frac{a^3}{a}$	$\frac{1}{12} (1.6)^3 = .34 \frac{\text{in}^4}{\text{in}}$	$\frac{1}{12} (2)^3 = .67 \frac{\text{in}^4}{\text{in}}$
$K = K_{EL} = \frac{201 E I_a}{a^2}$ (ELASTIC)	$\frac{201 (30 \times 10^6) (.34)}{(84)^2} = 2.9 \times 10^5 \frac{\text{lb}}{\text{in}}$	$\frac{201 (30 \times 10^6) (.67)}{(84)^2} = 5.7 \times 10^5 \frac{\text{lb}}{\text{in}}$
$WSS = \rho V$	$(1.6)(84)(216)(7.4 \times 10^{-4}) = 21 \frac{\text{lb}}{\text{in}^2}$	$2(84)(216)(7.4 \times 10^{-4}) = 27 \frac{\text{lb}}{\text{in}^2}$
$X_{FL} = L_1 / K_{EL}$	$9.7 \times 10^5 / 2.9 \times 10^5 = 3.3 \text{ in}$	$1.5 \times 10^6 / 5.7 \times 10^5 = 2.6 \text{ in}$
$F_1 = P_1 a b$ 120000 PIT	$120(84)(216) = 1.8 \times 10^6 \text{ lb}$ OR $P_2(84)(216) = 3.3 \text{ EG}$	$1.8 \times 10^6 \text{ lb}$ OR 3.3 EG
$F_2 = P_2 a b$ 120000 PIT	$21(84)(216) = 3.8 \times 10^5 \text{ lb}$ OR $2.7(84)(216) = 4.7 \text{ ES}$	$3.8 \times 10^5 \text{ lb}$ OR 4.7 ES

THE RESULTS ARE AS FOLLOWS, COMPUTER OUTPUT ATTACHED.

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COMPUTATION SHEET

SHEET NO.
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PROJECT NO.: 22-7092 SPONSOR: CEZL
SUBJECT: SW-57-2002
BY: V. J. H. N. B. DATE: 17 AUG 19 82 CHECKED BY: _____ DATE CHECKED: _____ 19 ____

THE DESIGNS FOR THE TWO BAYS ARE SO SIMILAR, ONLY THE 120 IN DOORS ARE DISCUSSED FURTHER. THE AT AND OTHER BAY DOORS WILL HAVE THE SAME DESIGN.

\bar{P} THICKNESS	MAX. DEFL. (X)	ELASTIC DEFL. (X _{EL})	STRAIN RATIO X / X _{EL}	MAX ROTATION ARCTAN ($\frac{2X}{a}$)
1.6	3.5	3.3	1.1	4.8°
2.0	3	2.6	0.73	2.6°

TYPICALLY BLAST RESISTANT DOORS ARE ALLOWED SOME PLASTIC DEFORMATION SUCH AS SUGGESTED IN "DESIGN OF STEEL STRUCTURES TO RESIST THE EFFECTS OF HE EXPLOSIONS" BY HEALEY ET. AL (REF. 2). FOR PLATE ELEMENTS REF. 2 SUGGESTS

$$\text{REUSABLE DESIGN: } \theta_{\text{MAX}} = 2^\circ \text{ OR } \frac{X}{X_{\text{EL}}} = 5$$

$$\text{NON-REUSABLE DESIGN: } \theta_{\text{MAX}} = 4^\circ \text{ OR } \frac{X}{X_{\text{EL}}} = 10$$

BOTH THE 1.6 IN AND 2.0 IN PLATES MEET THE X/X_{EL} REUSABLE CRITERIA. FOR THE VALUE OF $a = 84$ IN, THE ROTATIONS OF 4.8° AND 2.6° ARE NOT EXCESSIVE. EITHER PLATE THICKNESS WOULD PROVIDE THE PROTECTION DESIRED, HOWEVER FOR LITTLE EXTRA COST, THE 2 IN PLATE WILL PROVIDE ELASTIC RESPONSE ($X/X_{\text{EL}} < 1$). IN ADDITION,

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COMPUTATION SHEET

SHEET NO.
2 OF 8

PROJECT NO.: 02-7092

SPONSOR: CERL

SUBJECT: Blast Door

BY: M. WHITNEY DATE: 17 April 82

CHECKED BY:

DATE CHECKED: 19

ADDITIONAL SAFETY FOR FRAGMENT PROTECTION
IS PROVIDED.

REBOUND

BECAUSE TOTAL CONTAINMENT IS REQUIRED, THE DOOR
MUST WITHSTAND REBOUND. THIS WILL BE
PROVIDED BY SHEAR PINS WHICH SLIDE INTO AND
OUT OF THE JAM AS THE DOOR IS OPERATED. THE
SHEAR PINS MUST WITHSTAND THE ENTIRE
REBOUND SHEAR. FROM TMS-1300⁽³⁾ SHEAR
IS DETERMINED BY CONSIDERING L_1 , THE ULTIMATE
RESISTANCE OF THE PLATE APPLIED AS A STATIC
LOAD. AS SHOWN EARLIER IN THE COMPUTER OUTPUT,
THE DOOR DEVELOPES LESS THAN L_1 IN REBOUND.
THIS VALUE IS:

$$L_R = 3.2 \times 10^5 \text{ lbs}, \text{ REBOUND RESISTANCE, MAX.}$$

THIS IS APPLIED AS A STATIC LOAD

$$V_R = L_R = 3.2 \times 10^5 \text{ lbs}$$

USE STEEL SHEAR PINS WITH $\sigma_y = 60,000 \text{ psi}$.

$$\tau_f = 0.55 \sigma_y = 0.55 (60,000) = 33,000 \text{ psi}$$

D.L.F. FOR SHEAR = 1.0, $\tau = \tau_y$

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: BLAST DOOR
BY: M. WHITNEY DATE: JUNE 19 87 CHECKED BY: DATE CHECKED: 19

AREA OF SHEAR STEEL REQUIRED FOR REBOUND

$$A_{SR} = \frac{V_{MT}}{\tau} = \frac{3.2 \times 10^5 \text{ lb}}{33,000 \text{ psi}} = 9.7 \text{ in}^2$$

THIS WILL BE PROVIDED BY 14 - 1" DIAMETER
BOLTS

$$14 \left[\frac{\pi D^2}{4} \right] = 14 (3.14) \frac{(1.0)^2}{4} = 11.0 \text{ in}^2 > 9.7 \text{ in}^2 \text{ reqd.}$$

THE SHEAR PINS CAN BE ARRANGED AS 5 PER
SIDE 4 AT TOP, AND 0 AT BOTTOM. BY THIS
PLACEMENT, THE 3 SIDED SIMPLY-SUPPORTED RESPONSE
IS ALSO DEVELOPED IN REBOUND.

DOOR JAM

THE DOOR JAM IS REQUIRED TO WITHSTAND THE SUPPORT
SHEARS AS THE DOOR DEFLECTS OUTWARD. THE MAXIMUM
RESISTANCE DEVELOPED BY THE DOOR IS PROVIDED
IN THE COMPUTER OUTPUT DISCUSSED EARLIER AS,

$$L_{max} = 1.1 \times 10^6 \text{ lb}$$

AGAIN, THIS IS EQUATED TO THE SHEAR LOAD

$$V_{MAX} = L_{MAX} = 1.1 \times 10^6 \text{ lb}$$

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
8 OF 8

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: BLAST DOOR DESIGN
BY: M. WHITNEY DATE: 17 AUG 19 8 - CHECKED BY: DATE CHECKED: 19

AREA OF STEEL REQUIRED FOR JAM:

$$A_{s,j} = \frac{V_{TOT}}{\tau} = \frac{1.1 \times 10^6}{33000} = 33 \text{ in}^2$$

THIS AREA IS PROVIDED BY 20 - 1.5 in dia BOLTS

$$19 \left[\pi \frac{(1.5)^2}{4} \right] = 34 \text{ in}^2 > 33 \text{ Req'd}$$

THE SHEAR BOLTS CAN BE PLACED AS
12 ON EACH SIDE AND 10 ACROSS
THE TOP. A JAM WILL BE FORMED INTO
THE CONCRETE ON THE THREE SIDES OF THE
DOOR TO PROVIDE A LIP FOR THE DOOR. IN
ADDITION A $\frac{1}{2}$ in PLATE TO WHICH THE ANKOR
BOLTS WILL BE ATTACHED WILL LINE THE
JAM LIP.

BLAST DOOR , PIT BAY , 2 IN PL

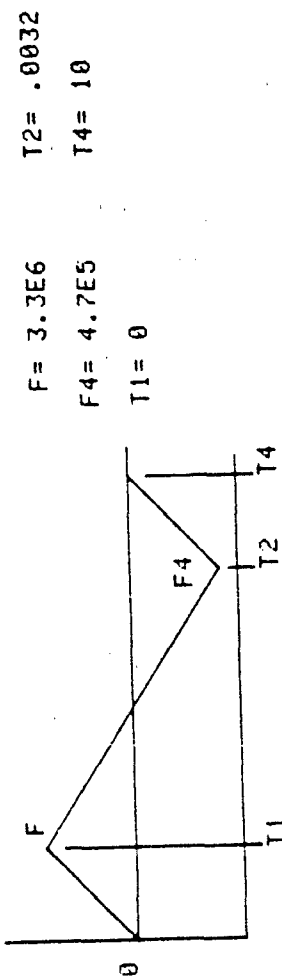
ENTER THE MASS. $M = 27$
 ENTER THE MASS-LOADED FACTORS. $K9(1) = .75$ $K9(2) = .59$ $K9(3) = .59$

ENTER THE ELASTIC SPRING CONSTANT. $K = 5.7 \times 10^6$

ENTER THE ELAS-PLAS. SPRING CONSTANT. $K5 = 0$

ENTER THE YIELD LOAD FOR THE SPRING. $L1 = 1.5E6$ $L2 = 1.5E6$

ENTER FRACTION OF CRITICAL VISCOUS DAMPING. $B = 0$



ENTER INITIAL DISPLACEMENT $(D0)$ AND VELOCITY $(V0)$.

$D0 = 0$ $V0 = 0$

THE PERIOD OF ELAS VIB., $T7$, IS 0.0374502701016 SEC

ENTER THE INTEGRATION TIME STEP, ABOUT $T7/10$. $T3 = .0016$

ENTER THE NUMBER OF RESPONSE CYCLES OVER WHICH THE NUMERICAL
 INTEGRATION IS TO BE PERFORMED. $N1 = 1.5$
 WANT COPY? (1 FOR YES, 2 FOR NO) 1

TIME	DISP	FORCE	RESISTANCE
0	0	3300000	0
0.0016	0.160113848011	1885000	91264.8933663
0.0032	0.541190936704	470000	308478.833921
0.0048	0.985179623411	469924.775928	561552.385344
0.0064	1.41750518157	469849.551857	807977.953497
0.008	1.80757399997	469774.327785	1030317.17998
0.0096	2.12780245913	469699.103713	1212847.40171
0.0112	2.35556682561	469623.879641	1342673.0906
0.0128	2.47480091349	469548.65557	1410636.52069
0.0144	2.47712868952	469473.431498	1411963.35302
0.016	2.36245197919	469398.207426	1346597.62814
0.0176	2.13895204082	469322.983355	1219202.66327
0.0192	1.82250527658	469247.759283	1038828.00765
0.0208	1.43555480576	469172.535211	818266.239282
0.0224	1.00551810216	469097.31114	573145.318231
0.024	0.562843674107	469022.087068	320820.894241
0.0256	0.130854526398	468946.862996	79147.080047
0.0272	-0.236468858359	468871.638924	-134787.249265
0.0288	-0.536609101039	468796.414853	-305867.187592
0.0304	-0.74038607793	468721.190781	-422020.064342
0.032	-0.833452185205	468645.966709	-475067.745567
0.0336	-0.809303158084	468570.742638	-461302.800108
0.0352	-0.669733227739	468495.518566	-381747.939811
0.0368	-0.424702773865	468420.294494	-242080.581103
0.0384	-0.0916282164981	468345.070423	-52228.0834039
0.04	0.305855228935	468269.846351	174337.480493
0.0416	0.739570042467	468194.622279	421554.924206
0.0432	1.17879322086	468119.398207	671912.13589
0.0448	1.59243192408	468044.174136	907686.196723
0.0464	1.95122387966	467968.950064	1112197.61141
0.048	2.22980682591	467893.725992	1270989.89077
0.0496	2.40851064266	467818.501921	1372851.06632
0.0512	2.47474553656	467743.277849	1410604.95584

0.0520
0.0544
0.056

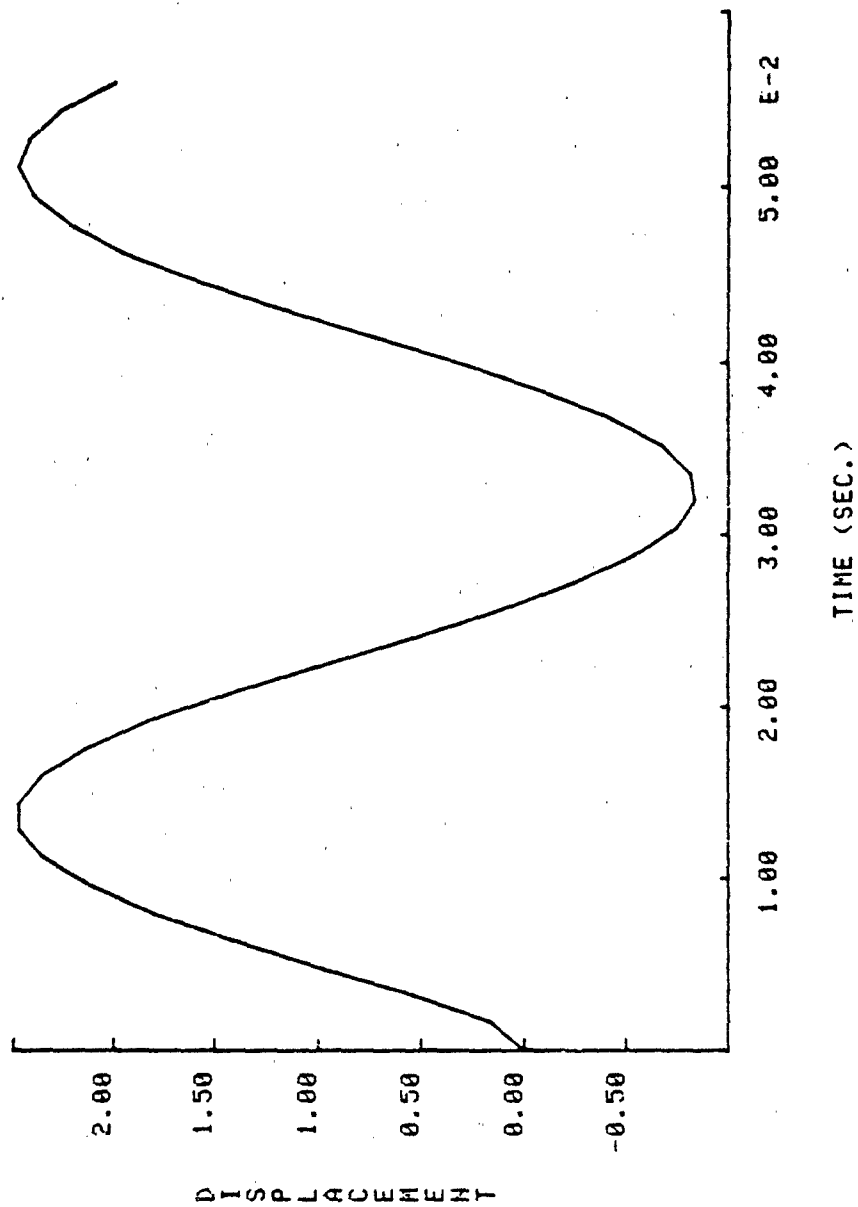
WAIT COPY? (1 FOR YES, 2 FOR NO)1

2.42388831339
2.25960434962
1.99358284643

467668.053777
467592.829706
467517.605634

1381616.33863
1287974.47928
1136342.22247

BLAST DOOR , PIT BAY , 2 IN PL



BLAST DOOR , PIT BAY , 1.6 IN PL

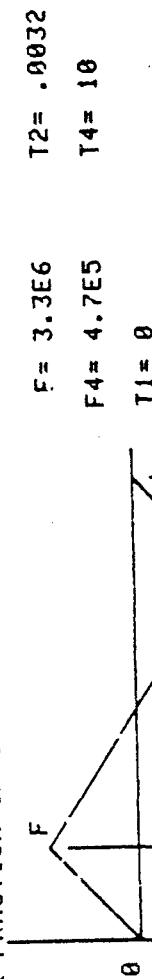
ENTER THE MASS. $M=21$
 ENTER THE MASS-LOADED FACTORS. $K9(1)=.75$ $K9(2)=.59$ $K9(3)=.59$

ENTER THE ELASTIC SPRING CONSTANT. $K=2.9E5$

ENTER THE ELAS-PLAS. SPRING CONSTANT. $K5=0$

ENTER THE YIELD LOAD FOR THE SPRING. $L1=9.7E5$ $L2=9.7E5$

ENTER FRACTION OF CRITICAL VISCOUS DAMPING. $B=0$



ENTER INITIAL DISPLACEMENT (D0) AND VELOCITY (V0).
 $D0=0$ $V0=0$

THE PERIOD OF ELAS VIB., $T7$, IS 0.0463042813716 SEC

ENTER THE INTEGRATION TIME STEP, ABOUT $T7/10$. $T3=.0016$

ENTER THE NUMBER OF RESPONSE CYCLES OVER WHICH THE NUMERICAL
 INTEGRATION IS TO BE PERFORMED. $N1=1.5$
 WANT COPY? (1 FOR YES, 2 FOR NO) 1

TIME	DISP	F.L.R.C.E	RESISTANCE
0	0	3300000	0
0.0016	0.207531672065	1985000	60184.1851308
0.0032	0.706760322951	470000	204960.493656
0.0048	1.30456232406	469924.775928	378323.073978
0.0064	1.91691168353	469849.551857	555904.388225
0.008	2.51526840502	469774.327785	729427.837456
0.0096	3.07175221033	469699.103713	890802.140996
0.0112	3.5630505729	469623.879641	970000
0.0128	3.97623591462	469548.65557	970000
0.0144	4.30807805368	469473.431498	970000
0.016	4.5585647632	469398.207426	970000
0.0176	4.72768381628	469322.983355	970000
0.0192	4.81542298602	469247.759283	970000
0.0208	4.82177004552	469172.535211	970000
0.0224	4.74755721635	469097.31114	970000
0.024	4.59644282228	469022.007068	948489.879542
0.0256	4.37535709036	468946.862396	904655.105261
0.0272	4.0946457644	468871.658924	840540.243004
0.0288	3.76738914848	468796.414853	759133.958476
0.0304	3.40883204215	468721.190781	664229.53986
0.032	3.03567302053	468645.965709	560247.979023
0.0336	2.66528619354	468570.742638	452031.862755
0.0352	2.31491171401	468495.518566	344619.682927
0.0368	2.00085275059	468420.294494	243011.083863
0.0384	1.73771632933	468345.070423	151933.984471
0.04	1.5377333942	468269.846351	75624.422306
0.0416	1.41018973686	468194.622279	17629.371188
0.0432	1.36099427272	468119.398207	-19358.2895102
0.0448	1.39240472538	468044.174136	-33624.9741099
0.0464	1.50292352304	467968.950064	-24515.9399402
0.048	1.68736850137	467893.725992	7534.50848068
0.0496	1.93711541604	467818.501921	61023.552197
0.0512	2.24050057551	467743.277849	133450.157453
			221431.853697

0.0528
0.0544
0.0556
0.0576
0.0592
0.0608
0.0624
0.064
0.0656
0.0672
0.0688

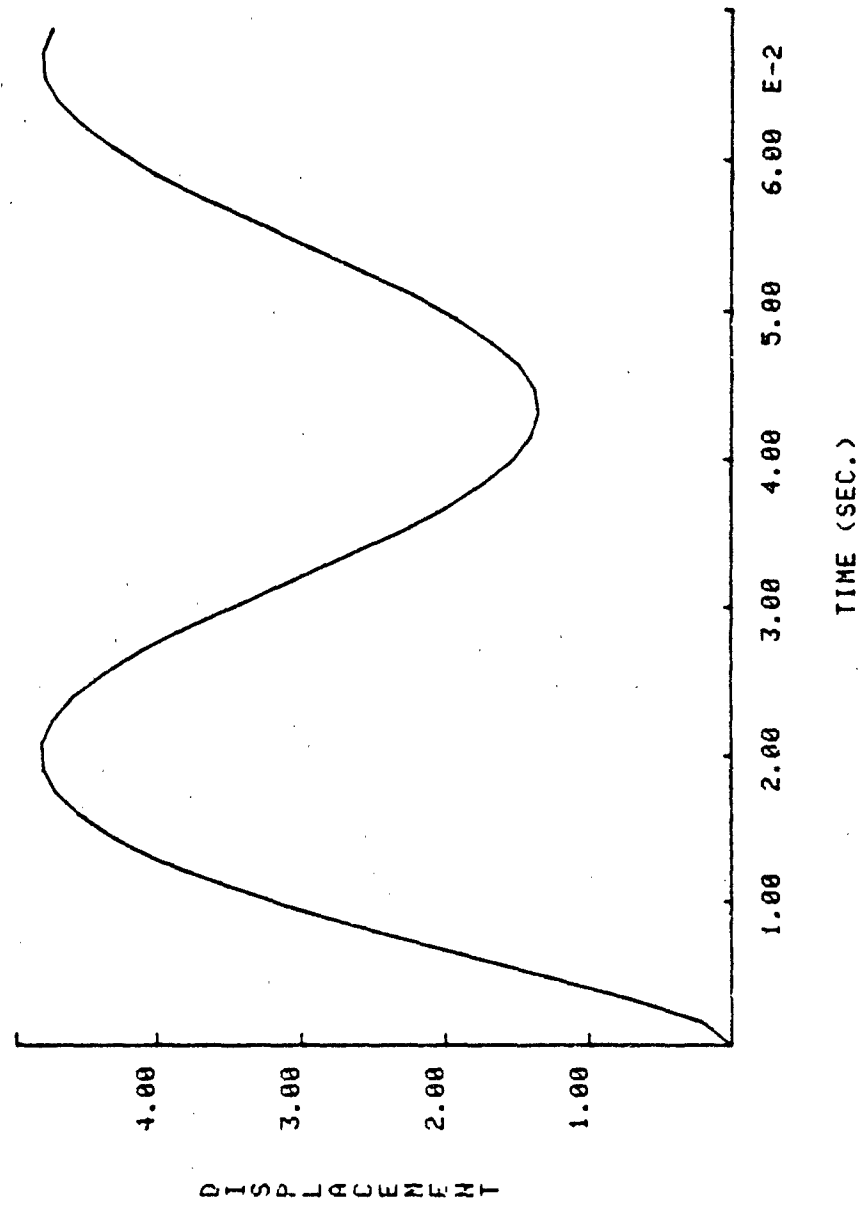
2.58336500205
2.94971466403
3.32246600821
3.68424202418
4.01818177314
4.30872564368
4.5423397593
4.70814586753
4.75842729311
4.80898752561
4.73934471663

467668.053777
467592.829706
467517.605634
467442.381562
467367.15749
467291.933419
467216.709347
467141.485275
467066.261204
466991.037132
466915.81306

320862.537394
427103.939602
535201.82918
640116.873814
736959.401011
821217.122596
888965.216998
937048.988385
963230.601802
966293.069228
946096.654622

WANT COPY? (1 FOR YES, 2 FOR NO)1

BLAST DOOR , PIT BAY , 1.6 IN PL



BLAST DOOR, 120X40 BAY, 1.6 IN PL

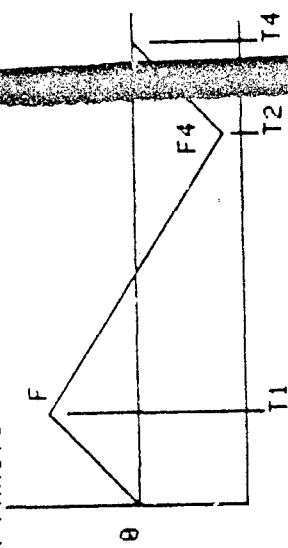
ENTER THE MASS. $M = 11$.
 ENTER THE MASS-LOADED FACTORS. $K9(1) = .75$ $K9(2) = .59$ $K9(3) = .59$

ENTER THE ELASTIC SPRING CONSTANT. $K = 2.9E5$

ENTER THE ELAS-PLAS. SPRING CONSTANT. $K5 = 0$

ENTER THE YIELD LOAD FOR THE SPRING. $L1 = 9.7E5$ $L2 = 9.7E5$

ENTER FRACTION OF CRITICAL VISCOUS DAMPING. $B = 0$



$F = 1.8E6$ $T2 = .0046$

$F_4 = 3.8E5$ $T_4 = 10$

$T1 = 0$

ENTER INITIAL DISPLACEMENT $(D0)$ AND VELOCITY $(V0)$.
 $D0 = 0$ $V0 = 0$

THE PERIOD OF ELAS VIB., $T7$, IS 0.0463042813716 SEC

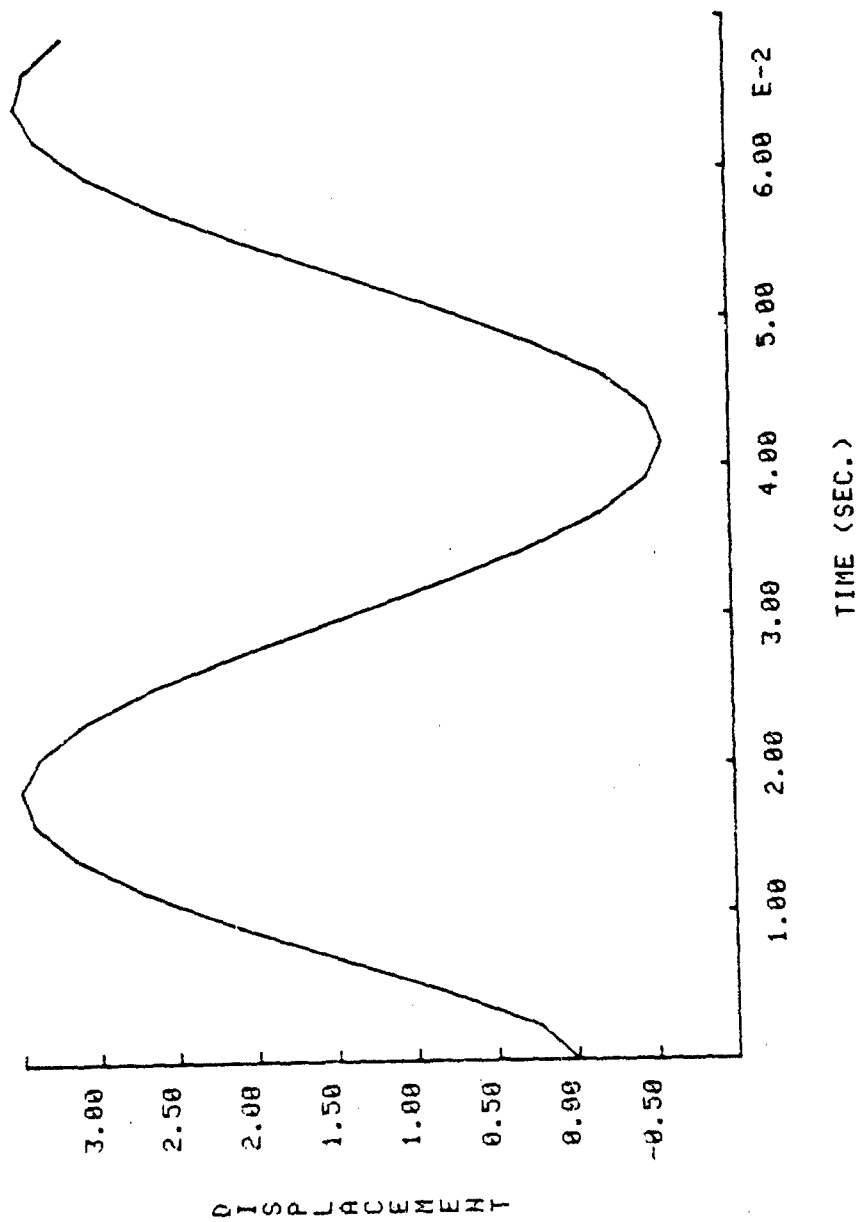
ENTER THE INTEGRATION TIME STEP ABOUT $T7/10$. $T3 = .0023$

ENTER THE NUMBER OF RESPONSE CYCLES OVER WHICH THE NUMERICAL INTEGRATION IS TO BE PERFORMED. $N = 1.5$
 WANT COPY? (1 FOR YES, 2 FOR NO)

TIME	DISP	FORCE	RESISTANCE
0	0	1800000	0
0.0023	0.235307356916	1090000	68239.1335057
0.0046	0.802257344649	380000	232654.629948
0.0069	1.47352585617	379912	427322.49829
0.0092	2.12846197399	379025	617253.972456
0.0115	2.70481696304	379737	784396.919282
0.0138	3.14788357964	379650	912886.238094
0.0161	3.41757443228	379562	970000
0.0184	3.49374910437	379475	970000
0.0207	3.3745571596	379387	935434.336016
0.023	3.07329300067	379300	848067.729924
0.0253	2.6187734311	379213	716257.054753
0.0276	2.05434514549	379125	552572.851924
0.0299	1.43374820402	379038	372599.738899
0.0322	0.81599671745	378950	193451.907792
0.0345	0.259764403	378863	32144.4366016
0.0368	-0.182190853311	378775	-96022.5877286
0.0391	-0.468033633046	378688	-178916.993852
0.0414	-0.570813707602	378600	-208723.215473
0.0437	-0.481009501598	378513	-182679.995732
0.046	-0.207412271821	378426	-103336.799097
0.0483	0.223731055195	378338	21694.7657381
0.0506	0.771232864577	378251	180470.290459
0.0529	1.38209668543	378163	357852.798506
0.0552	2.0049054652	378076	536955.018223
0.0575	2.56528799562	377988	700746.27846
0.0598	3.02364953873	377901	833671.125963
0.0621	3.33211403986	377813	923125.831291
0.0644	3.46151681979	377726	960652.637471
0.0667	3.39974425988	377639	942738.595096
0.069	3.15286475408	377551	871143.538415

WANT COPY? (1 FOR YES, 2 FOR NO) 1

BLAST DOOR , 120X40 BAY , 1.6 IN PL



BLAST DOOR , 120X40 BAY , 2 in ψ

ENTER THE MASS. $M=27$

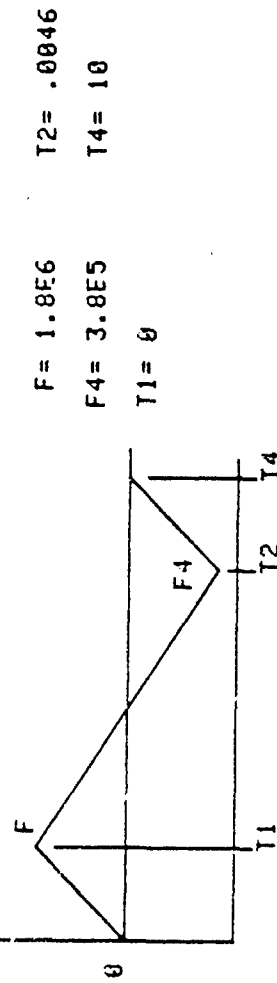
ENTER THE MASS-LOADED FACTORS. $K9(1)=.75$ $K9(2)=.59$ $K9(3)=.59$

ENTER THE ELASTIC SPRING CONSTANT. $K=5.7E5$

ENTER THE ELAS-PLAS. SPRING CONSTANT. $K5=0$

ENTER THE YIELD LOAD FOR THE SPRING. $L1=1.5E6$ $L2=1.5E6$

ENTER FRACTION OF CRITICAL VISCOUS DAMPING. $B=.6$



ENTER INITIAL DISPLACEMENT $(D0)$ AND VELOCITY $(V0)$.

$D0=0$ $V0=0$

THE PERIOD OF ELAS VIB., $T7$, IS 0.0374502701016 SEC

ENTER THE INTEGRATION TIME STEP, ABOUT $T7/10$. $T3=.0023$

ENTER THE NUMBER OF RESPONSE CYCLES OVER WHICH THE NUMERICAL INTEGRATION IS TO BE PERFORMED. $N1=1.5$

WANT COPY? (1 FOR YES, 2 FOR NO) 1

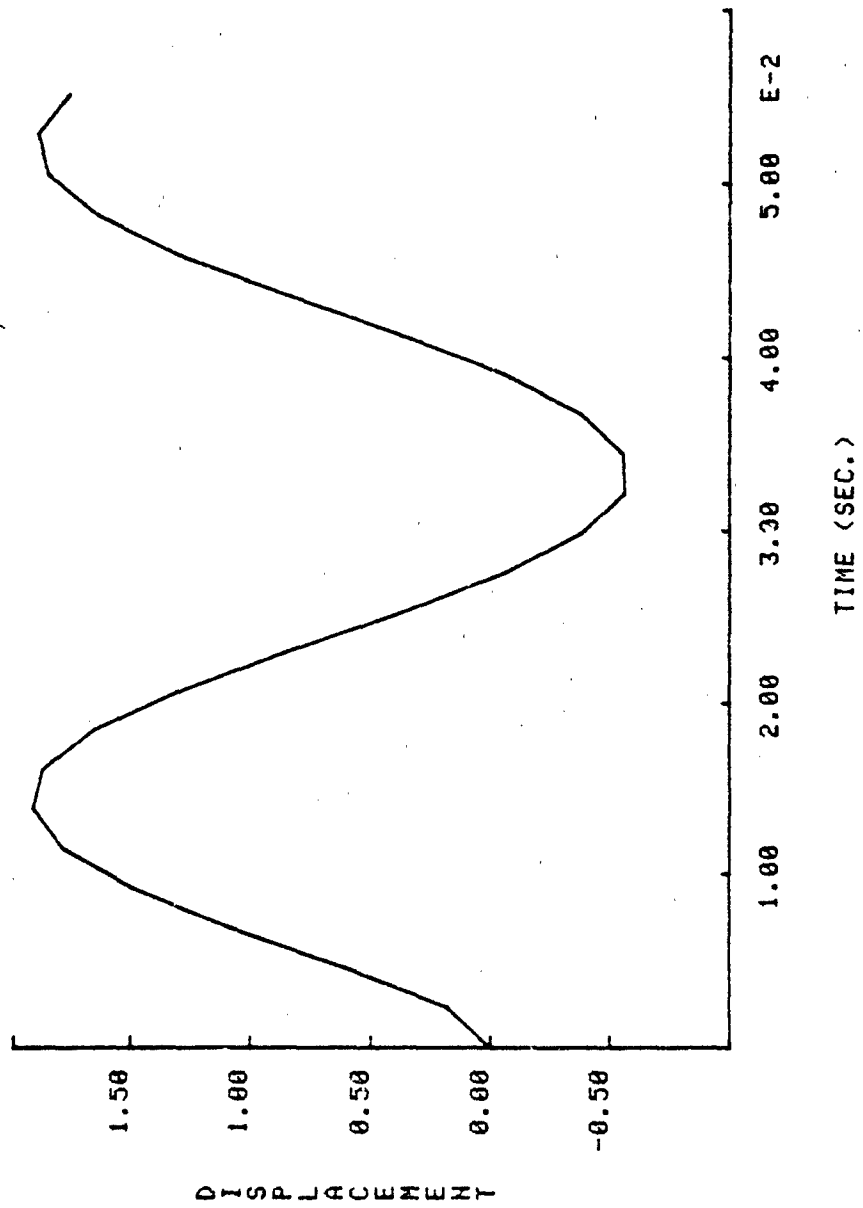
EAST DOOR, 120X40 BAT, 2.0 in

TIME	DISP (INCHES)	FORCE	RESISTANCE
0	0	1800000	0
0.0023	0.179989746502	1800000	102594.155506
0.0046	0.604370691174	3800000	344491.293969
0.0069	1.07941739464	379912.559778	615267.914942
0.0092	1.49395286781	379825.119555	851553.134652
0.0115	1.78869476315	379737.679333	1019556.015
0.0138	1.9216987187	379650.23911	1095368.26966
0.0161	1.87432347515	379562.798887	1068364.38084
0.0184	1.65383871298	379475.350665	942680.066398
0.0207	1.29231105332	379387.918442	736617.300391
0.023	0.841938241405	379300.47822	479904.797601
0.0253	0.367510250985	379213.037997	209480.843061
0.0276	-0.0629139467127	379125.597775	-35860.9496262
0.0299	-0.387772419804	379038.157552	-221030.279288
0.0322	-0.560807826016	378950.71733	-319660.460829
0.0345	-0.557656476935	378863.277107	-317864.191853
0.0368	-0.379280452502	378775.836885	-216189.057926
0.0391	-0.0517594852663	378688.396562	-29502.9066018
0.0414	0.377517460934	378600.95644	215184.952732
0.0437	0.84670496184	378513.516217	482621.828249
0.046	1.28840739812	378426.075995	734392.21693
0.0483	1.6393552161	378338.635772	934433.043179
0.0506	1.84946870885	378251.19555	1054197.16405
0.0529	1.88899135931	378163.755327	1076725.0748
0.0552	1.75270897991	378076.315105	999044.118549

MAX.

WANT COPY? (1 FOR YES, 2 FOR NO) 1

BLAST DOOR , 120X40 BAY , 2 in it



SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
1 OF 18

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: ROOF DESIGN
BY: S. MOORE DATE: SEPT 19 82 CHECKED BY: SYSTEMS DATE CHECKED: SEPT 19 82

ROOF SLABS FOR THE CONCEPTUAL STRUCTURES HAVE BEEN ANALYZED WITH THE COMPUTER CODES DESCRIBED IN THE DISCUSSION OF METHODS USED AND IN APPENDIX 6. THE CODES WERE ABLE TO CALCULATE REQUIRED REBAR SIZES FOR VARYING THICKNESSES OF CONCRETE SLABS. LOADINGS ON THE SLABS WERE FROM THE TAT IMPACT CRITERIA.

WHAT FOLLOWS IS THE TABULATED RESULTS OF THESE COMPUTER RUNS. THE SLAB SIZE, DEPTH UNDER GROUND, AND THICKNESS ARE NOTED IN EACH TABLE. FOR EVERY SET OF CONDITIONS, A REQUIRED AREA OF STEEL REINFORCEMENT IS LISTED ALONG WITH A SELECTED BAR SIZE AND SPACING THAT WILL PROVIDE THAT REINFORCEMENT. A SET OF CALCULATIONS OF STIRRUP SIZE FOR A TYPICAL ROOF AND A SUMMARY ARE ALSO INCLUDED.

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
2 OF 10

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: ROOF DESIGNS - 16' BURIAL - 30" MIN THICK
BY: B. MORRIS DATE: 24 SEP 82 CHECKED BY: DK DATE CHECKED: 24 SEP 82

SEE "ROOF DESIGNS", 15 SEP 82 FOR LOADINGS

25' 1-WAY $\theta = 2^\circ$ 3' COVER

THICK (IN)	$A_s (R)$ (IN ² /FT)	BARS	$A_s (P)$ (IN ² /FT)	TMS-1300 MINIMUMS	
				MAIN	OTHER
24	3.86	—	—	—	—
30	2.83	2-#11@12	3.12	#9@12	#6@12
36	2.20	2-#10@12	2.54		
42	1.78	2-#9@12	2.00		
48	1.50	#11@12	1.56		

54 x 37 (648" x 444") $\Delta = 7.8"$

THICK (IN)	$A_s (R)$ LONG	BARS	$A_s (P)$ LONG	$A_s (R)$ SHORT $= A_s (L) \frac{S_1}{S_2} \frac{I-1}{I-2}$	BARS	$A_s (P)$ SHORT
24	2.18	2-#10@12	2.54	2.83		
30	1.41	#11@12	1.56	1.89	2-#9@12	2.00
36	1.06	#10@12	1.27	1.44	#11@12	1.56
42	.84	#9@12	1.00	1.16	#10@12	1.27
48	.68	#8@12	.78	.95	#9@12	1.00

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
3 OF 18

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: ROOF DESIGNS - 17' BURIAL - 2' MIN THICKNESS
BY: E. MORRIS DATE: 23 SEP 19 82 CHECKED BY: D. K. H. DATE CHECKED: 24 SEP 19 82

SEE SHEETS FOR 15, 20, 23 SEP FOR LOADS, Δ 's
65' x 55' (780" x 660")

THICK (IN)	$A_s(R)$ LONG	BARS	$A_s(P)$ LONG	$A_s(R)$ SHORT $= A_s(L) \frac{T-8}{T-6}$	BARS	$A_s(P)$ SHORT
24	3.55	—	—	—	—	—
30	1.94	2-#9@12	2.00	2.10	2-#10@12	2.54
36	1.37	#11@12	1.56	1.51	#11@12	1.56

71' x 46'

THICK (IN)	$A_s(R)$ LONG	BARS	$A_s(P)$ LONG	$A_s(R)$ SHORT $= A_s(L) \frac{T-8}{T-6}$	BARS	$A_s(P)$ SHORT
24	2.65	2-#11@12	3.12	3.63	—	—
30	1.63	2-#9@12	2.00	2.31	2-#10@12	2.54
36	1.13	#10@12	1.27	1.63	2-#9@12	2.00

71 x 56

THICK (IN)	$A_s(R)$ LONG	BARS	$A_s(P)$ LONG	$A_s(R)$ SHORT $= A_s(L) \frac{T-8}{T-6}$	BARS	$A_s(P)$ SHORT
24	4.00	—	—	—	—	—
30	2.04	2-#9@12	2.00	2.37	2-#10@12	2.54
36	1.36	#11@12	1.56	1.61	2-#9@12	2.00

25' L-WAY

THICK (IN)	$A_s(R)$	BARS	$A_s(P)$
24	3.14	2-#11@12	3.12
30	2.28	2-#10@12	2.54
36	1.78	2-#9@12	2.00

5-137

SOUTH WEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
4 OF 18

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: ROOF DESIGN
BY: B. MORRIS DATE: 23 SEP 82 CHECKED BY: D. K. THOMAS DATE CHECKED: 24 SEP 82

SEE "ROOF DESIGN'S" , 14 & 20 SEP 82

SURFACE STRUCTURES

CONCEPT 1 - 73'x60' (876"x720") PRESS = $P = \frac{7.11}{73 \times 60} = .175 P_0$

PRESS	TIME	$\theta = 2^\circ$	FDT = 72,000	$f_c' = 4000$
0	0			
6.2	.028	$\Delta = 12.6"$		
6.2	.236			
32.6	.32	4" COVER LONG WAY, 3" SHORT		
32.6	.377			
6.2	.408	TMS-1300 MIN REINF.		
3.4	.49			
0	.524			

THICK (IN)	$A_s(R)$ LONG	BARS	$A_s(P)$ LONG	$A_s(R)$ SHORT = $A_s(L)$	BARS	$A_s(P)$ SHORT	$A_s(MIN)$ TMS-1300
48	1.21	#10@12	1.27	1.47	#10@10	1.52	1.32/#9@9

19' BURIAL

25' (3.12") 1-WAY PRESS = $\frac{10(.350) + 25(.302 P_0)}{12.5} = .345 P_0$

PRESS	TIME	$\theta = 2^\circ$	3' COVER
0	0		
12.1	.028		
12.1	.236		
63.2	.32		
63.2	.377		
12.1	.408		
6.6	.49		
0	.524		

THICK (IN)	$A_s(R)$ (IN/FT)	BARS	$A_s(P)$ (IN/FT)	MIN STEEL MAIN	TMS-1300 OTHER
42	1.44	#10@10	1.52	1.14 #9@10	.50 #6@10

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
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PROJECT NO: 02-7092 SPONSOR: CERL
SUBJECT: ROOF DESIGNS
BY: R. MORRIS DATE: 23 SEP 19 82 CHECKED BY: D. MORRIS DATE CHECKED: 24 SEP 19 82

24' BURIAL

62'x54' (744" x 648")

AREA = 3348 FT²

R INT	Δ AREA	\bar{P}	FORCE	
0-5	157	.255 P _o =	40.0 P _o	
5-10	471	.22 P _o =	103.6 P _o	PRESS = 483.9 P _o
10-15	786	.175 P _o =	137.6 P _o	
15-20	1099	.125 P _o =	137.4	PRESS = .145 P _o
20-25	3348 - 2513 = 835	.075 P _o =	65.3 P _o	
			483.9 P _o	

PRESS	TIME
0	0
5.1	.028
5.1	.236
26.5	.332
26.5	.377
5.1	.408
2.8	.49
0	.524

$$\Delta = \frac{1}{2} (648) \tan 2^\circ = 11.3"$$

THICK (IN)	A _s (R) LONG	BARS	A _s (P) LONG	A _s (R) SHORT = A _s (L) $\frac{7-8}{5-7}$	BARS	A _s (P) SHORT	A _s (MIN)
30	1.57	#11@12	1.56	1.65	#10@9	1.69	.74/#8@12
36	1.17	#9@10	1.20	1.25	#10@12	1.27	.94/#8@12
42	.88	#8@10	.94	.95	#8@10	.94	1.14/#9@10

26' (312") 1-WAY - SEE PAGE 4 OF 8, 15 SEP 82 FOR PRESS

THICK (IN)	A _s (R) (IN ² /FT)	BARS	A _s (P) (IN ² /FT)	MINIMUM MAIN	STEEL TMS-1300 OTHER
24	2.36	#11@8	2.34	.54	#7@12 .29 #5@12
30	1.70	#10@9	1.69	.72	#7@10 .36 #5@10
36	1.32	#9@9	1.33	.96	#9@12 .43 #6@12
42	1.07	#8@9	1.05	1.14	#9@10 .50 #6@10

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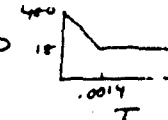
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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
6 OF 18

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: ROOF DESIGN
BY: B. MORRIS DATE: 23 SEP 19 82 CHECKED BY: Diagram DATE CHECKED: 20 SEP 19 82

71' x 46' (852" x 552")

INTERNAL BLAST P



$\theta = 1^\circ$ $\Delta = 4.8"$

THICK (IN)	$A_s(R)$ LONG	BARS	$A_s(P)$ LONG	$A_s(R)$ SHORT $= A_s(L) \frac{7}{8} \frac{I}{L}$	BARS	$A_s(P)$ SHORT	$A_s(MIN)$
42	.72	*7@10	.72	1.05	*8@9	1.05	1.14/*8@10
48	.60	*7@12	.60	.88	*8@10	.94	1.33/*9@9
36	.93	*8@10	.94	1.35	*9@9	1.33	1.96/*9@12
30	1.36	*9@9	1.33	1.88	*11@10	1.87	1.78/*11@12

24' BURIAL

40' 1-WAY

SEE 15 SEP FOR PRESS

4" COVER

THICK (IN)	$A_s(R)$ (IN ² /FT)	BARS	$A_s(P)$ (IN ² /FT)
30	4.18	2-*11@9	4.16
36	3.13	2-*11@12	3.12

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COMPUTATION SHEET

SHEET NO
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PROJECT NO: 02-7092 SPONSOR: CERL
SUBJECT: MINIMUM REBAR PER TM 5-1300
BY: E. MORRIS DATE: 20 SEP 19 52 CHECKED BY: DATE CHECKED: 19

Table 6-1. Minimum Area of Flexural Reinforcement

Pressure design range	Reinforcement	Two-way elements	One-way elements
Intermediate and low	Non-Fiber	$A_s = 0.0025bd$ $A_s = 0.0018bd$	$A_s = 0.0025bd$ $A_s = A_s'$ $= 0.0020bd$
High	Non-Fiber	$A_s = A_s'$ $= 0.0025bd$ $A_s = A_s'$ $= 0.0018bd$	$A_s = A_s'$ $= 0.0025bd$ $A_s = A_s'$ $= 0.0018bd$

* Not less than A_s/A_s' used in same direction

USE INTERMEDIATE +
LOW PRESSURE RANGE

MINIMUM SPACING = 9"

1-WAY ELEMENTS, 3" COVER

THICKNESS T_c (IN)	$d = T_c - 3$ (IN)	MAIN		OTHER	
		$A_s = A_s'$ (IN ² /FT)	BARS	$A_s = A_s'$ (IN ² /FT)	BARS
24	21	.63	#7@12	.29	#5@12
30	27	.81	#7@9	.36	#5@10
36	33	.99	#9@12	.43	#6@12
42	39	1.17	#9@10	.50	#6@10
60	57	1.71	#10@9	.72	#7@10
72	69	2.07	#11@9	.86	#8@10

2-WAY ELEMENTS, MAIN REINFORCEMENT ONLY

THICKNESS T_c (IN)	3" COVER			4" COVER		
	$d = T_c - 3$ (IN)	$A_s = A_s'$ (IN ² /FT)	BARS	$d = T_c - 4$ (IN)	$A_s = A_s'$ (IN ² /FT)	BARS
30	27	.81	#7@9	26	.78	#7@9
36	33	.99	#9@12	32	.96	#9@12
42	39	1.17	#9@10	38	1.14	#9@10
48	45	1.35	#9@9	44	1.32	#9@9
60	57	1.71	#10@9	56	1.68	#10@9

5-1.1

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SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: ROOF DESIGNS - 747 OBLIQUE IMPACT
BY: B. MORRIS DATE: 20 SEP 1982 CHECKED BY: DATE CHECKED: 19

SEE "ROOF DESIGNS", 14 SECS, FOR LOADINGS AND OTHER CASES.

SURFACE STRUCTURES

MECH/ELEC ROOM, 73'x71' (876" x 852")

$$PRESS = P_o \frac{751}{73 \times 71} = .151 P_o$$

PRESS	TIME
0	0
5.3	.028
5.3	.236
27.6	.32
27.6	.377
5.3	.408
2.9	.49
0	.524

$$FDY = 72,000 \quad A_s = 4000$$

$$\theta = 2^\circ \quad \Delta = 14.9"$$

4" COVER LONG WAY, 3"
COVER SHORT WAY

TMS-1300 MIN. REINF

THICK (IN)	A _s (R) LONG	BARS LONG	A _s (P) LONG	A _s (R) SHORT = A _s (L) $\frac{27}{37}$ (T-F)	A _s (P) SHORT	A _s (L MIN) TMS-1300
4.8	1.55	#11 @ 12	1.56	1.53 #11 @ 12	1.56	1.32

19' BURIAL - 71'x56'

SLAB AREA = 3976 FT²

R INTERVAL	ΔA	x	P _{AVG}	=	FORCE
0-5	157	1	.356 P _o	=	55.9 P _o
5-10	471		.302 P _o	=	142.2 P _o
10-15	786		.211 P _o	=	165.8 P _o
15-20	1099		.126 P _o	=	138.5 P _o
20-25	1463		.070 P _o	=	102.4 P _o
					604.8 P _o

PRESS	TIME
0	0
5.3	.028
5.3	.236
27.6	.32
27.6	.377
5.3	.408
2.9	.49
0	.524

$$AVG PRESS = \frac{604.8 P_o}{3976}$$

$$= 0.152 P_o$$

$$(852 \times 672")$$

$$\theta = 1.2^\circ \quad \Delta = 11.7"$$

SOUTHWEST RESEARCH INSTITUTE
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SHEET NO.
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PROJECT NO. 02-7092 SPONSOR: CERL
SUBJECT: ROOF DESIGNS-747 OBLIQUE IMPACT
BY: P. MORRIS DATE: 20 SEP 82 CHECKED BY: DATE CHECKED: 19

THICK (IN)	A _s (R) LONG	BARS LONG	A _s (P) LONG	A _s (R) SHORT = A _s (L) $\frac{J}{T-F}$	A _s (P) SHORT	A _s (MIN) TMS-1300
42	1.07	*4@10	1.2	1.28 *9@9	1.33	1.14

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SHEET NO.
10 OF 18

PROJECT NO.: 02-7092

SPONSOR: CERL

SUBJECT: ROOF DESIGN

BY: B. MORRIS

DATE: 14 SEP 19 82

CHECKED BY:

DATE CHECKED: 19

747 OBLIQUE IMPACT - SURFACE

TIME	PRESS.	FORCE (LB) $\times 112,454 \text{ IN}^2$	PRESSURE	
			63' x 40'	71' x 44'
0	0	0	0	0
.028	35	3.937×10^6	10.8	8.8
.236	35	3.937×10^6	10.8	8.0
.320	183	2.058×10^7	56.7	45.7
.377	183	2.058×10^7	56.7	45.7
.408	35	3.937×10^6	10.8	8.8
.49	19	2.137×10^6	5.9	4.8
.524	0	0	0	0

FDY = 72,000 $f'_c = 4000$ $\theta = 2^\circ$ 3" COVER - 1 WAY +
2-WAY SHORT
4" COVER - 2-WAY LONG TYPICAL BAR SPACING

63' x 40' (756" x 480" x 48" DEEP)

$\Delta = 8.4"$

THICK	$A_s(R)$ LONG SIDE	BARS	$A_s(P)$ LONG SIDE	$A_s(R)$ SHORT SIDE $= A_s(L) \frac{L}{T} \frac{T-8}{T-6}$	$A_s(P)$ SHORT
48	1.08	#9 @ 10	1.20	1.80 #11 @ 10	1.87

71' x 44' (852" x 528" x 48" DEEP)

$\Delta = 9.2"$

THICK	$A_s(R)$ LONG	BARS	$A_s(P)$ LONG	$A_s(R)$ SHORT $= A_s(L) \frac{L}{T} \frac{T-8}{T-6}$	$A_s(P)$ SHORT
48	1.02	#9 @ 12	1.00	1.54 #11 @ 12	1.56

30' 1-WAY FIXED-FIXED

THICK	$A_s(R)$ (IN ² /FT)	BARS	$A_s(P)$ (IN ² /FT)	TEMP. STEEL, EACH FACE *
12"	3.38	2-#10 @ 9"	3.38	#8 @ 10"

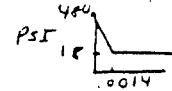
$$* A_s + A_s' = .002 b T_c = .002 (12)(72) = 1.73 \text{ IN}^2/\text{FT}$$

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SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: ROOF DESIGNS
BY: E. MORRIS DATE: 14 SEP 19 82 CHECKED BY: DATE CHECKED: 19

71' x 46' (852" x 552" x 48" DEEP)



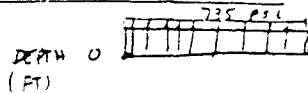
$f_{dy} = 72,000$ $f_c' = 4000$ $\theta = 1^\circ$ $\Delta = 4.8"$
4' COVER LONG SIDE, 3" COVER SHORT SIDE

71' x 46'

THICK	$A_s(R)$ LONG	BARS	$A_s(P)$ LONG	$A_s(R)$ SHORT $= A_{s(L)} \frac{T}{T-L}$	$A_s(P)$ SHORT
48	.60	*7@12	.60	.88 *8@10	.94

EARTH COVERED CASES

EARTH PRESSURES



5	.66	.63	.74	.73	.29	.27
10	.514	.456	.294	.125	.44	.29
15	.353	.323	.228	.125	.66	.33
20	.250	.215	.187	.118	.66	.37
25	.176	.149	.132	.95	.66	.44
	→ R					
0	5	10	15	20	25	

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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: ROOF DESIGNS
BY: B. MORRIS DATE: 5 SEP 80 CHECKED BY: _____ DATE CHECKED: _____ 19__

AT 24' BURIAL, THE SOIL PRESSURES ARE

<u>R(LAT)</u>	<u>P</u>	<u>P₀ = SURFACE PRESSURE</u>
0	.26 P ₀	
5	.25 P ₀	
10	.19 P ₀	
15	.16 P ₀	
20	.09 P ₀	
25	.06	

THESE PRESSURES WERE CALCULATED FOR A 22' DIAMETER CIRCULAR LOAD. THE OBLIQUE IMPACT LOADED AREA IS A 44'x22' OVAL, SO THE ABOVE PRESSURES COULD APPLY OVER RADII OF 2R, ESPECIALLY FOR A 1-WAY SLAB.

$$\text{AVG. PRESS ON 40' SLAB} = \frac{10(.255 P_0) + 10(.22 P_0)}{20} = .238 P_0$$

<u>PRESS (PSI)</u>	<u>TIME (SEC)</u>	<u>FORCE ON 480" x 1"</u>
0	0	0
8.3	.028	3984
8.3	.236	3984
43.6	.32	20,928
43.6	.377	20,928
8.3	.408	3984
4.5	.49	2160
0	.524	0

40' 1-WAY SLAB FIXED-FIXED, 2", 3" COVERS

<u>THICK (IN)</u>	<u>A_s (R) (IN²/FT)</u>	<u>BARs</u>	<u>A_s (P) (IN²/FT)</u>	<u>TEMP STEEL</u>	<u>TEMP STEEL +</u>
36	2.91	*11 @ 6	3.12	*5 @ 9 (.41)	*6 @ 12 (.71)
42	2.36	*11 @ 8	2.34	*6 @ 12 (.45)	*6 @ 10 (.53)
48	1.92	*11 @ 9	2.08	*6 @ 10 (.53)	*7 @ 12 (.60)
54	1.65	*10 @ 9	1.69	*6 @ 9 (.59)	*7 @ 10 (.72)

$$*A_s = \frac{1}{2} (.0018) (\text{THICK}) (12) + \text{TMS-1300 } A_s \text{ WRITTEN } A_s, A_s = .0026 T_c$$

$$\text{AVG PRESS ON 26' SLAB} = \frac{10(.255 P_0) + 3(.27 P_0)}{13} = .247 P_0$$

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SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: ROOF DESIGN
BY: B. MORRIS DATE: 15 SEP 82 CHECKED BY: DATE CHECKED: 19

PRESS (PSI)	TIME (SEC)	FORCE ON 3/2"
0	0	0
3.6	.028	2683
8.6	.236	2683
45.2	.32	14,102
45.2	.377	14,102
8.6	.408	2683
4.7	.49	1466
0	.524	0

26' 1-WAY SLAB, FIXED, 2' 3' COVER

THICK (IN)	A _s (R) (IN ² /FT)	BARS	A _s (P) (IN ² /FT)	TEMP STEEL *	TEMP STEEL +
42	1.07	#8@9"	1.05	#5@8 (.46)	#6@10 (.53)

$A_s = \frac{1}{2} (1.0018) (THICK) (IN)$
+ TMS-1300

AT 16' BURIAL

R (FT)	PRESS	ELLIPSE AREA = πR^2
0	.452 P _o	0
5	.416 P _o	157
10	.298 P _o	628
15	.168 P _o	1414
20	.090 P _o	2513
25	.045 P _o	

FOR A 25' 1-WAY SPAN, THE AVG PRESS COULD BE

$$P = \frac{10(.434 P_o) + 25(.357)}{12.5} = .419 P_o$$

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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: ROOF DESIGNS
BY: B. MORRIS DATE: 15 SEP 19 82 CHECKED BY: DATE CHECKED: 19

PRESS	FORCE ON 300"	TIME
0	0	0
14.7	4410	.028
14.7	4410	.236
76.7	23010	.32
76.7	23010	.377
14.7	4410	.408
8.0	2400	.49
0	0	.524

25' 1-WAY SLAB 1-WAY, FIXED, 3" COVER

THICK (IN)	A _s (R) (IN ² /FT)	BAR	A _s (P) (IN ² /FT)	TEMP* STEEL	TEMP STEEL +
60	1.13	#4@10	1.20	*7@10(.72)	*9@9(1.33)

FOR A 54X37 SLAB, AREA = 1998 FT²

R INTERVAL	Δ AREA	FORCE = Δ AREA × P̄
0-5'	157 #	157 × .434 P _o = 68.1 P _o
5-10	471 #	471 × .357 P _o = 168.1 P _o
10-15	786 #	786 × .233 P _o = 183.1 P _o
15-20	1998 - 1414 = 584	584 × .124 P _o = 75.3 P _o
		494.6 P _o

$$AVG PRESS = \frac{494.6 P_o}{1998} = .248 P_o$$

PRESSURE	TIME
0	0
8.7	.028
8.7	.236
45.4	.32
45.4	.377
8.7	.408
4.7	.49
0	.524

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
15 OF 18

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: ROOF DESIGNS
BY: B. MORRIS DATE: 15 SEP 82 CHECKED BY: DATE CHECKED: 19

54' x 37' (648' x 444') Δ = 7.8 4" Ls. COVER

THICK (IN)	A _s (R) LONG	BARS	A _s (P) LONG	A _s (R) SHORT = A _s (L) $\frac{54}{37} \times \frac{7-8}{7}$	A _s (P) SHORT
60	.50	#6@10	.53	.74 #8@12	.78

TEMP STEEL = $\frac{1}{2}(6.0018)(60)(12) = .65 \text{ IN/FT} \quad \#7@10 (.72)$
= 1.30 PER TMS-1300 - #9@9

19' BURIAL DEPTH

R(FT)	PRESS	AREA = 2πR'(FT')
0	.368 P ₀	0
5	.344 P ₀	157
10	.261 P ₀	628
15	.162 P ₀	1414
20	.090 P ₀	2513
25	.049 P ₀	3927

FOR A 65' x 55' SLAB; AREA = 3757

R INTERVAL	Δ AREA	Δ AREA * P _{AVG} = FORCE
0-5	157	157 x .356 P ₀ = 55.9 P ₀
5-10	471	471 x .302 P ₀ = 142.2 P ₀
10-15	786	786 x .211 P ₀ = 165.8 P ₀
15-20	1099	1099 x .126 P ₀ = 138.5 P ₀
20-25	1244	1244 x .070 P ₀ = 87.1 P ₀
		589.5 P ₀

AVG PRESS = $\frac{589.5 P_0}{3757} = .157 P_0$

PRESS	TIME
0	0
5.5	.028
5.5	.236
28.7	.32
28.7	.377
5.5	.408
3.0	.49
0	.524 5-149

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
16 OF 18

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: ROOF DESIGN
BY: B. MORRIS DATE: 16 SEP 19 82 CHECKED BY: DATE CHECKED: 19

65' x 55' (780" x 660") $\Delta = 11.5"$ 4" COVER LONG, 3" COVER SHORT							
THICK (IN)	$A_s(R)$ LONG	BARS	$A_s(P)$ LONG	$A_s(R)$ - SHORT $= A_s(L) \frac{T-L}{T}$	$A_s(P)$ SHORT	TEMP ACI	TEMP TMS-1300
42	1.07	#8 @ 9	1.05	1.17	#9 @ 10	1.20	#6 @ 10 (1.2)

FOR A 71' x 46' SLAB, AREA = 3266'

R	INT	Δ	AREA	$\Delta A \times P =$	FORCE
0-5		157		157 x .356 P_o	= 55.9 P_o
5-10		471		471 x .302 P_o	= 142.2 P_o
10-15		786		786 x .211 P_o	= 165.8 P_o
15-20		1099		1099 x .126 P_o	= 138.5 P_o
20-25		3266-2513	= 753	753 x .070 P_o	= 52.7 P_o
					555.1 P_o

$$AVG PRESS = \frac{555.1 P_o}{3266} = .170 P_o$$

PRESS	TIME
0	0
6.0	.028
6.0	.236
31.1	.32
31.1	.377
6.0	.408
3.2	.49
0	.524

71' x 46' (852" x 552") $\Delta = 9.6"$

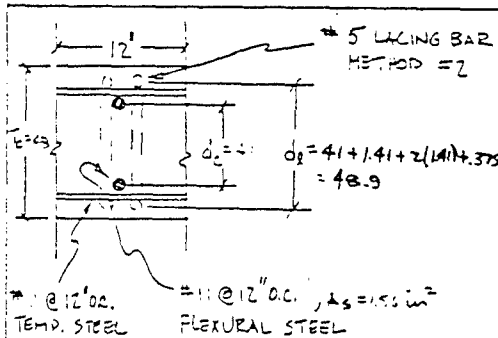
THICK (IN)	$A_s(R)$ LONG	BARS	$A_s(P)$ LONG	$A_s(R)$ SHORT $= A_s(L) \frac{T-L}{T}$	$A_s(P)$ SHORT	TEMP ACI	TEMP TMS-1300
42	.88	#8 @ 10	.94	1.37	#9 @ 9	1.33	#6 @ 10 (1.33) #9 @ 12 (1.2)

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
17 OF 18

PROJECT NO.: 02-7022 SPONSOR: CER
SUBJECT: STIRRED CALCULATIONS
BY: D. KATHAM DATE: 27 SEP 19 92 CHECKED BY: DATE CHECKED: 19

BEAM DESIGN



FIXED-FIXED SUPPORTS

$L = 40 \text{ FT} = 480 \text{ IN}$

$M = \frac{w L^2}{12}$

$M = \frac{1.56 (72000) 41}{12} = 38710 \text{ in-lb}$

$r_u = \frac{M}{L^2} = \frac{16 (38710)}{(480)^2} = 27 \text{ psi}$

FLEX BAR, $A_s > .0025 b d_c$, $.0025 (12) (41) = .123 \text{ in}^2$
TEMP. BAR, $A_s > .0015 b d_c$, $.0015 (12) (41) = .091 \text{ in}^2$
 $A_s = 1.56 \text{ in}^2 > \frac{1}{2}$

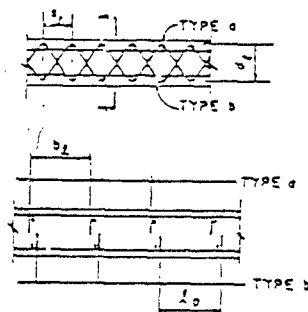
$r_u = \frac{16 (r_u' - d_c)}{d_c} = \frac{27 (\frac{480}{12} - 41)}{41} = 129 \text{ psi}$

$p = A_s / d_c b = \frac{1.56}{(41)(12)} = .0032$

$r_c = \phi (1.9 \sqrt{f'_c} + 2500 p) = .85 (1.9 \sqrt{4000} + 2500 (.0032)) = 109 \text{ psi}$

$r_{c \text{ max}} = 2.28 \phi \sqrt{f'_c} = 2.28 (.85) \sqrt{4000} = 123 \text{ psi}$

$V_u > V_c$ USE $V_u - V_c$ $\therefore r_u' = 20 = 20 \text{ psi}$



METHOD #2

$s/d_t = \frac{12}{48.9} = \frac{7(625)}{48.9}$

FROM FIG. 6-19 OF TMS, $\alpha = 45^\circ$

$A_v = \frac{r_u' s b_f}{.85 f_y (\sin \alpha + \cos \alpha)} = \frac{(109)(12)(12)}{.85 (60000) (\sin 45 + \cos 45)}$

$A_v = .05 \text{ in}^2$

USE #5 BAR

$s = \frac{A_v}{.0015} = \frac{.05}{.0015} = 33.3 \text{ in}$

APPENDIX 6

ROOF SYSTEM COMPARISON

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: ROOF/747 IMPACT CALCULATIONS
BY: B. MORRIS DATE: 27 SEP 82 CHECKED BY: DEK DATE CHECKED: 2 OCT 82

APPENDIX 6 SUMMARY

TOPIC	PAGE
INTRODUCTION	6-1
COMPUTER CODE DESCRIPTION	6-2
CALCULATIONS	
40' 1-WAY ROOF	6-4
I-BEAM ROOF	6-5
ARCHES	6-6

INTRODUCTION

THIS APPENDIX CONTAINS THE CALCULATIONS USED TO EVALUATE THE STRUCTURAL REQUIREMENTS OF DIFFERENT ROOF TYPES SUBJECTED TO AN OBLIQUE SURFACE IMPACT OF A 747. A 40 FOOT ONE-WAY ROOF, AN I-BEAM ROOF, AND AN ARCH ROOF, EACH OF REINFORCED CONCRETE, ARE CONSIDERED. THE ANALYSES ALLOW A 6° ANGLE OF ROTATION AND ARE PERFORMED WITH COMPUTER CODES.

EACH ANALYSIS CONTAINS SUMMARY TABLES OF THE REQUIRED AREAS OF STEEL REINFORCING FOR VARIOUS THICKNESSES FROM A PRESSURE HISTORY OF THE IMPACT (SEE FIGURE 4.1) AND SOME BASIC DESIGN PARAMETERS. ALSO INCLUDED IN THE APPENDIX ARE A BRIEF DESCRIPTION OF THE ARCH CALCULATION METHOD, AN APPROXIMATION OF THE NATURAL FREQUENCY ACQUIRED BY THE ROOF, AND A FIGURE TAKEN FROM INTRODUCTION TO STRUCTURAL DYNAMICS BY SIESS USED IN THE CALCULATIONS.

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SHEET NO.
1 OF 2

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: ROOF SYSTEM COMPARISON CASE
BY: B. MORRIS DATE: 27 SEP 19 82 CHECKED BY: D. KETTER DATE CHECKED: 27 SEP 19 82

ROOF SYSTEM COMPARISON CALCULATIONS

THE ANALYSIS DETAILS WHICH FOLLOW WERE PRODUCED USING A NUMBER OF IN-HOUSE DEVELOPED COMPUTER CODES DESIGNED TO NUMERICALLY INTEGRATE THE EQUATIONS OF MOTION SUGGESTED IN BIGGS' INTRODUCTION TO STRUCTURAL DYNAMICS OR TO PERFORM OTHER MATHEMATICAL COMPUTATIONS. CODES USED ARE DESCRIBED BELOW.

'BEAMDES' - ITERATIVE DESIGN OF REINFORCED CONCRETE ONE WAY SLABS OR BEAMS SUBJECTED TO TIME-VARYING PRESSURE LOADS. IT USES BIGGS' SINGLE-DEGREE-OF-FREEDOM MODEL WITH STRUCTURAL PARAMETERS AS CALCULATED BY BIGGS OR TMS-1300.

'SLABDES' - SIMILAR TO ABOVE BUT FOR TWO-WAY REINFORCED CONCRETE SLABS.

'ARCH' - CALCULATES REACTIONS FOR FIXED-END ARCHES BASED ON ARCH SPAN, THICKNESS, AND RISE. GOVERNING EQUATIONS ARE FOUND IN FORMULAS FOR STEEL AND STONE BY ZORR AND YOUNG.

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SHEET NO.

2 OF 2

PROJECT NO. 92-792

SPONSOR. CRL

SUBJECT: ROPE SYSTEM COMPARISON CALC.

BY: P. MORRIS

DATE: 21 AUG 19 82

CHECKED BY: J. KETCHUM

DATE CHECKED: 21 SEP 19 82

'ONEDE' - SINGLE-DEGREE-OF-FREEDOM MAXIMUM DISPLACEMENT OF
AN UNDAMPED MASS-SPRING SYSTEM TO A
SPECIFIED TIME-VARYING LOAD. EQUATIONS OF
MOTION ARE FROM BILLS.

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COMPUTATION SHEET

SHEET NO.
1 OF 1

PROJECT NO.: CD-12 SPONSOR: CEAL
SUBJECT: Roof System Calc.
BY: J. K. H. W. DATE: 27 Nov 1992 CHECKED BY: _____ DATE CHECKED: 19

Roof System Calculations

EACH OF THE FOLLOWING ROOF ANALYSIS IS BROKEN INTO THREE PARTS. THE FIRST IS A PRESSURE HISTORY FOR THE OBLIQUE IMPACT OF A 747 ON A CONCRETE SLAB. THIS IMPACT IS COMPLETELY DESCRIBED IN SECTION 4.1.

THE SECOND PART IS A LISTING OF THE DESIGN PARAMETERS USED. THE THIRD PART IS A SUMMARY TABLE GENERATED FROM THE AFORESAID COMPUTER PROGRAMS. THE TABLE LISTS THE REQUIRED A_s AND THE REBAR SIZE AND SPACING NEEDED FOR THAT AREA OF STEEL.

THE REMAINING CALCULATIONS AND TABLES SHOULD BE SELF EXPLANATORY.

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COMPUTATION SHEET

SHEET NO.

1 OF 2

PROJECT NO: 02-7092 SPONSOR: CERL
SUBJECT: 747 OBLIQUE IMPACT ON 40' 1-WAY ROOF
BY: B. MORRIS DATE: 27 AUG 82 CHECKED BY: _____ DATE CHECKED: _____ 19____

PRESSURE - TIME HISTORY:

PRESSURE (PSI)	FORCE ON 1"x480" (LB)	TIME (SEC)
0	0	0
35	16,800	.028
35	16,800	.236
183	87,840	.320
183	87,840	.377
35	16,800	.408
19	9120	.49
0	0	.524

DESIGN PARAMETERS:

$\theta = 6^\circ$ $f_c = 4000$ $FD = 90,000$
FIXED-FIXED SUPPORTS 3" COVER
MINIMUM BAR SPACING - 9"

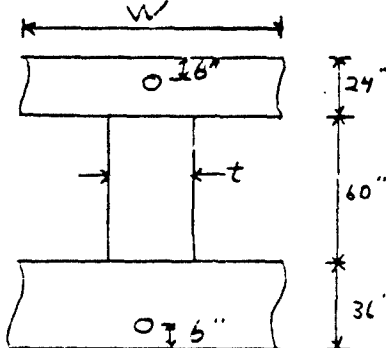
THICK (IN)	A_s (REQ'D) (IN ² /FT)	BAR S	A_s (PROV) (IN ² /FT)
48	6.93	—	—
54	5.95	—	—
60	5.13	—	—
72	4.01	2-#11@9	4.16
84	3.27	2-#10@9	3.38
96	2.72	2-#10@10	3.04
108	2.33	2-#9@10	2.40

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COMPUTATION SHEET

SHEET NO.
2 OF 2

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: I-BEAM ROOF DESIGN
BY: B. MORRIS DATE: JAN 82 CHECKED BY: DATE CHECKED: 19

CROSS-SECTION:



TIME (SEC)	LOADING P (PSI)	FORCE ON 1"x40" (LB)
0	0	0
.028	35	16800
.236	35	16800
.320	183	87840
.377	183	87840
.408	35	16800
.490	19	9120
.524	0	0

$f'_c = 4000$ $f_y = 70,000$ $\theta = 6^\circ$ FIXED-FIXED

REQUIRED STEEL (EACH FACE) = .1784 IN/IN = 2.14 IN/FT

USE #10 @ 7" $A_s = 2.17$ IN/FT
OR #11 @ 8" 2.34 IN/FT

FOR #10 @ 7, SUPPORT SHEAR = WALL REACTION = $\frac{8A_s f_y (TRAIL-12)}{12 L}$

SUPPORT SHEAR = 29,295 LB/IN

ASSUME USE #5 STIRRUPS AT 6" C-C.

NOW $V_u = V_{CR} + \frac{A_y f_y d}{s}$

$29,295 W = 1.9 \sqrt{f'_c} t (114) + \frac{(2 \times 31)(69,000)(114)}{6}$

$29,295 W = 1.37115 t + 7.07 \times 10^5$

$t = 2.14 W - 51.6$

t (IN)	W (IN)	$\frac{W}{t}$	CLEAR SPACE BETWEEN WEBS (IN)
6	26.9	4.48	21
12	29.7	2.43	18
18	32.5	1.81	15

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COMPUTATION SHEET

SHEET NO.
1 OF 2

PROJECT NO.: 02-7092

SPONSOR: CERL

SUBJECT: ARCHES

BY: B. M. M. DATE: 18 AUG 82

CHECKED BY:

DATE CHECKED: 19

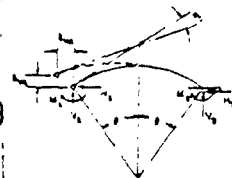
REFERENCE: FORMULAS FOR STRESS AND STRAIN, 5th EDITION
ROARK & YOUNG, MCGRAW HILL, 1975

TABLE 18 Reaction and deformation formulas for circular arches

NOTATION: W = load (pounds); w = unit load (pounds per linear inch); M_0 = applied couple (inch-pounds); θ_0 = externally created concentrated angular displacement (radian); Δ_0 = externally created concentrated lateral displacement (inches); $T - T_0$ = uniform temperature rise (degrees); T_0 and T_1 = temperatures on outside and inside, respectively (degrees); H_1 and H_2 are the horizontal end reactions at the left and right, respectively, and are positive to the left (pounds); V_1 and V_2 are the vertical end reactions at the left and right, respectively, and are positive upward (pounds); M_1 and M_2 are the reaction moments at the left and right, respectively, and are positive clockwise (inch-pounds); I is the area moment of inertia for bending in the plane of the arch.

The axial stress deformation factor is $\alpha = 1/AR^2$, where A is the cross-sectional area. The transverse shear deformation factor is $\beta = FE/IGAR^2$, where G is the shear modulus of elasticity and F is a shape factor for the cross section (see page 185). R is arch radius; $\phi = \sin \theta$; $\psi = \cos \theta$; $\epsilon = \sin \phi$; and $\zeta = \cos \phi$. γ = temperature coefficient of expansion (inches per inch per degree).

General reaction and deformation formulas for arches for loads W to W_4 , right and fixed at all 10 cases.



Deformation constants

$$\text{Horizontal deformation at } \theta = \theta_0 = \frac{R^3}{EI} \left(\theta_0 M_1 + \theta_0 M_2 + \theta_0 \frac{M_0}{R} - \Delta_0 \right) \quad (2)$$

$$\text{Vertical deformation at } \theta = \theta_0 = \frac{R^3}{EI} \left(\theta_0 M_1 + \theta_0 M_2 + \theta_0 \frac{M_0}{R} - \Delta_0 \right) \quad (3)$$

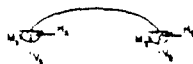
$$\text{Angular rotation at } \theta = \theta_0 = \frac{R^2}{EI} \left(\theta_0 M_1 + \theta_0 M_2 + \theta_0 \frac{M_0}{R} - \Delta_0 \right) \quad (4)$$

$$\begin{aligned} \text{where } \theta_0 &= \theta = 360^\circ - 2\alpha + \alpha \phi = \alpha \phi - 2\alpha \\ \theta_{10} &= \theta_{10} = 360^\circ - 2\alpha \\ \theta_{20} &= \theta_{20} = 360^\circ - 2\alpha \\ \theta_{30} &= \theta_{30} = 360^\circ - 2\alpha \\ \theta_{40} &= \theta_{40} = 360^\circ - 2\alpha \\ \theta_{50} &= \theta_{50} = 360^\circ - 2\alpha \\ \theta_{60} &= \theta_{60} = 360^\circ - 2\alpha \\ \theta_{70} &= \theta_{70} = 360^\circ - 2\alpha \\ \theta_{80} &= \theta_{80} = 360^\circ - 2\alpha \\ \theta_{90} &= \theta_{90} = 360^\circ - 2\alpha \end{aligned}$$

and where θ_{10} , θ_{20} , and θ_{30} are loading terms given below for several types of load

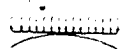
When θ desired, H_1 , V_1 , and M_1 can be obtained from equilibrium equations after obtaining H_2 , V_2 , and M_2 .

1. Left and fixed, right and fixed



where $\theta_{10} = 0$; $\theta_{20} = 0$; and $\theta_{30} = 0$. See Table 1 for general formulas for H_1 , V_1 , and M_1 .

2. Uniformly distributed load



$$\theta_{10} = -\alpha \left(\frac{1}{2} - \frac{1}{2} \phi \right) = -\frac{\alpha}{2} (1 - \phi) = -\frac{\alpha}{2} (1 - \sin \phi)$$

$$\theta_{20} = -\alpha \left(\frac{1}{2} - \frac{1}{2} \phi \right) = -\frac{\alpha}{2} (1 - \phi) = -\frac{\alpha}{2} (1 - \sin \phi)$$

$$\theta_{30} = -\alpha \left(\frac{1}{2} - \frac{1}{2} \phi \right) = -\frac{\alpha}{2} (1 - \phi) = -\frac{\alpha}{2} (1 - \sin \phi)$$

THESE EQUATIONS HAVE BEEN PROGRAMMED
IN "ARCH" WITH THE FOLLOWING RESULTS

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SHEET NO.
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PROJECT NO.: 02-7092

SPONSOR:

CERL

SUBJECT: ECHES

BY: B. MORRIS DATE: 18 AUG 82

CHECKED BY:

DATE CHECKED: 19

40' SPAN, 1" WIDE SECTION

RISE (INCHES)	THICKNESS (IN)	HA (LB)	V ₁ (LB)	MA (LB-IN)	MC (LB-IN)	HC (LB)
60	24	-396W	240W	-3084W	1956W	-396W
	36	-317W	240W	-6249W	3531W	-317W
	48	-240W	240W	-9337W	5063W	-240W
	60	-173W	240W	-12000W	6420W	-173W
120	24	-228W	240W	54W	1494W	-228W
	36	-206W	240W	-1686W	2394W	-206W
	48	-179W	240W	-3805W	3515W	-179W
	60	-150W	240W	-6097W	4703W	-150W
230	24	-135W	240W	5025W	2775W	-135W
	36	-130W	240W	4258W	3158W	-130W
	48	-123W	240W	3238W	3744W	-123W
	60	-114W	240W	2014W	4594W	-114W

W = UNIFORM VERTICAL LOAD

MC = MOMENT AT CROWN, HC = THRUST AT CROWN

VC = SHEAR AT CROWN = 0.

ASSUME MC IS POSITIVE COUNTER CLOCKWISE

$$\Sigma M_C = 0 \Rightarrow M_C - \text{RISE} \cdot H_A - M_A - \frac{\text{SPAN}}{2} V_A + W \frac{\text{SPAN}}{2} \times \frac{\text{SPAN}}{4} = 0$$

$$M_C - \text{RISE}(H_A) - M_A - 240V_A + 28,800W = 0$$

$$M_C = M_A + 240V_A + \text{RISE} H_A - 28,800W \quad + \int$$

$$H_C = H_A \quad + \text{TO THE RIGHT}$$

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COMPUTATION SHEET

SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: ARCH ROOF DESIGN
BY: E. MORRIS DATE: 19 AUG 82 CHECKED BY: DATE CHECKED: 19

USE THE 747 OBLIQUE IMPACT, UNMOUNDED

TIME	PRESSURE (PSI)	FORCE	TOTAL IMPULSE
0.	0	0	0
.028	35	$.168 \times 10^5$	235
.236	35	.168	3727
.320	183	$.878 \times 10^5$ F1	8123
.377	183	.878	13,128
.408	35	.168	14,749
.490	19	.091	15,811
.524	0	0	15966

FLAT ROOF DESIGN

THICK (IN)	A_s (IN ²)	NATL AREA	A_y	$A = \frac{A_y}{A_s}$	R_m	$\frac{R_m}{F_1}$
36	9.06	.0551	3.5"	14.6	$.68 \times 10^5$.77
48	6.00	.0420	1.48	34.5	$.63 \times 10^5$.72
60	4.38	.0361	.78	65.4	$.59 \times 10^5$.67

BY .408 SEC, THE SYSTEM HAS REACHED MAX RESPONSE.

TREAT THE LOAD AS A TRIANGLE

F



$$I = \frac{1}{2} F t_d = \frac{1}{2} (.878 \times 10^5) t_d = 14,749$$

$$t_d = .336 \text{ SEC}$$

USE FIG 2.26 IN BIGGS TO CHECK RESPONSE:

THICK	$\frac{t_d}{T}$	$\frac{R_m}{F_1}$	M	CLOSENESS
36	6.1	.77	213	OK
48	7.8	.72	235	REAL GOOD
60	9.3	.67	280	MODERATELY OK

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COMPUTATION SHEET

SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: ARCH ROOF DESIGN
BY: B. MORRIS DATE: 19 AUG 19 52 CHECKED BY: _____ DATE CHECKED: 19

CONSIDER THE 40' SPAN, 5' RISE ARCH. BIGGS
GIVES AN APPROXIMATION TO THE NAT'L PERIOD
FOR AN ARCH IN RADIAL COMPRESSION AS

$$T_L = \frac{R}{1810} = \frac{42.5}{1810} = .0235 \text{ SEC.}$$

SINCE WE DON'T HAVE TRUE RADIAL COMPRESSION,
ASSUME NAT'L PERIOD IS BETWEEN THIS
AND THE FLAT ROOF VALUE.

USE	ARCH THICK	NAT'L PERIOD	$\frac{T_L}{T}$
	36	.0393	8.5
	48	.0332	10.1
	60	.0298	11.3

ASSUME WANT SAME M 'S AS CALCULATED FOR
THE 36" AND 48" CASES

ARCH THICK	M
36	15
48	35

FROM BIGGS FIGURE 2.26, FIND
(SEE PAGE 4 OF THIS WRITE-UP)

ARCH THICK	M	$\frac{T_L}{T}$	$\frac{R_n}{F_i}$	$\frac{R_n}{F_i}$
36	15	8.5	.8	70,420
48	35	10.1	.75	65,850

COMPARE SUPPORT MOMENTS FOR FLAT AND ROOF

SECTION THICK	ARCH MOMENT	FLAT MOMENT	FLAT ARCH
36	6249W	19200W	3.08
48	9337W	19200W	2.06

W = APPLIED UNIFORM VERT. LOAD.

FOR THE FLAT, $R_n(FLAT) = \frac{16 M}{L}$, M = MOMENT CAPACITY OF
THE CROSS-SECTION

$$USE R_n(ARCH) = R_n(FLAT) \times \left(\frac{FLAT}{ARCH} \right) = \frac{16 M}{L} \left(\frac{FLAT}{ARCH} \right)$$

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SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: ARCH ROOF DESIGN
BY: B. MORRIS DATE: 19 44 82 CHECKED BY: DATE CHECKED: 10

$$\text{THUS, } M(\text{ARCH}) = M(\text{FLAT}) \frac{A_3(\text{ARCH})}{A_3(\text{FLAT})}$$

$$\text{MOMENT CAP} = A_3 \text{FDY}(\text{THICK} - 6)$$

$$\text{THUS, } A_3(\text{ARCH}) = A_3(\text{FLAT}) \left(\frac{A_3(\text{ARCH})}{A_3(\text{FLAT})} \right) \text{ FOR OTHER CONDITIONS CONSTANT}$$

THICK	$A_3(\text{FLAT}) (\text{IN}^2/\text{FT})$	$A_3(\text{ARCH}) (\text{IN}^2/\text{FT})$
36	6.80	1.95
48	4.38	2.13

CONCLUSION: WHILE SOME STEEL SAVINGS MAY BE POSSIBLE, THE COST OF FABRICATION PLUS THE NEED FOR IMPROVED WALL DESIGN TO TAKE THE ARCH THRUST WILL OFFSET ANY SAVINGS.

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SHEET NO.
4 OF 4

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: Arch Roof Design
BY: DATE: 19 CHECKED BY: DATE CHECKED: 19

78 Introduction to Structural Dynamics

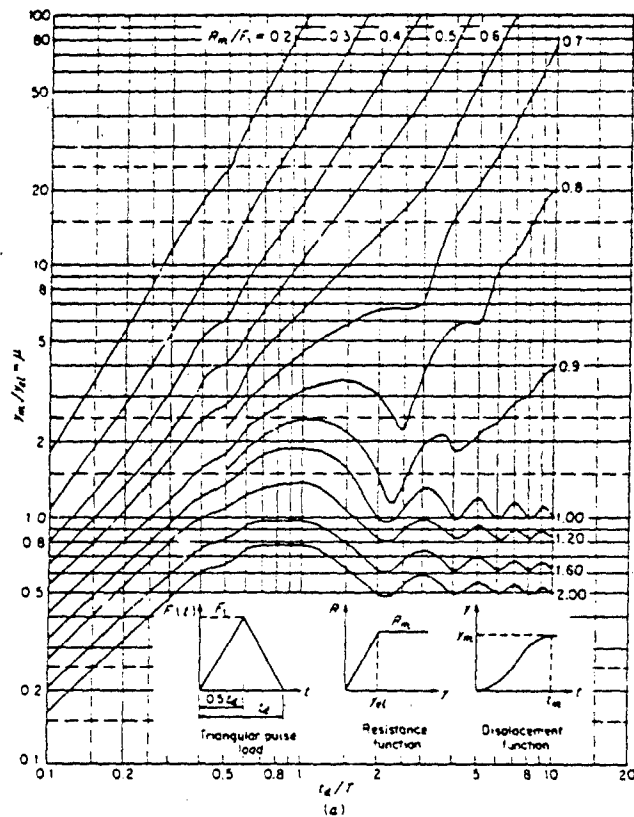


FIGURE 2.26 Maximum response of elasto-plastic one-degree systems (undamped) due to equilateral triangular load pulses.

APPENDIX 7

DIVIDING WALL COMPARISON

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COMPUTATION SHEET

SHEET NO.
OF

PROJECT NO: DA-7092 SPONSOR: CERL
SUBJECT: Appendix 7: Dividing Wall Comparisons
BY: KM DATE: 23 Sep 72 CHECKED BY: _____ DATE CHECKED: 19

Appendix 7: Dividing Wall Comparisons Calculations

This appendix contains the calculation work sheets for
blast loads,
fragment threat analysis,
spall analysis,
and design for interior dividing walls.

The blast loads were determined for a 1' x 5' wall panel with a 78.5 lb charge on the floor 3.5' from the wall.

For the fragment threat analysis, a worst case steel cube fragment weighing 0.4 lb having an impact velocity of 3000 ft/sec, was used to calculate depth of penetration into concrete, steel, spaced armor, and sand. The penetration depths calculated were used to specify wall thicknesses necessary to contain the worst case fragment threat.

The spall analysis work sheets are included to show how calculations of spalling and spall velocity were made.

Work sheets are included for design calculations for a concrete cantilever wall, a steel suppressive shield type wall, a wall consisting of two steel sections with sand in between

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SHEET NO. _____
OF _____

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: Appendix 7: Dividing Wall Comparisons
BY: TKM DATE: 22 SEP 82 CHECKED BY: _____ DATE CHECKED: _____ 19__

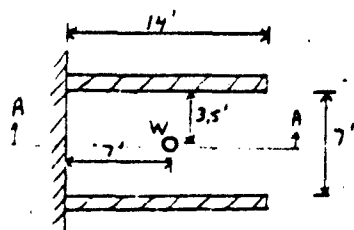
the steel sections, and a steel/concrete composite wall.
The thicknesses and construction descriptions for each
wall type are indicated in the work sheets.

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COMPUTATION SHEET

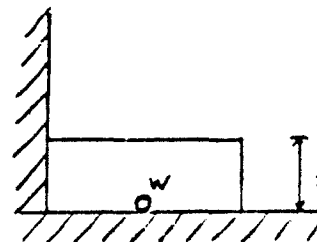
SHEET NO.
OF

PROJECT NO: 02-7092 SPONSOR: CERL
SUBJECT: BLAST LOADS ON INTERIOR DIVIDING WALLS
BY: B. MORRIS DATE: 22 JUL 19 53 CHECKED BY: DATE CHECKED: 19

7.1 Blast Loads



PLAN VIEW



SECTION A-A

CHARGE WT = 78.5^{lb} BUT DOUBLE SINCE ITS ON THE FLOOR

$$W_e = 157 \text{ LB}$$

SWRI has developed a small computer program to predict wall blast FROM PROGRAM INPUT ATTACHED AS PAGE 7-5, loading.

$$P_{max} = 15,000 \text{ PSI}$$

$$i = 1.05 \text{ PSI-SEC}$$

$$T_D = \frac{2i}{P} = .00714 \text{ SEC}$$

MULTIPLY THIS PRESSURE AND IMPULSE BY 1.75 TO ACCOUNT FOR REFLECTIONS WITHIN THE STORAGE AREA (PER DOE/TIC-11268)

$$P = 26,250 \text{ PSI}$$

$$i = 1.84 \text{ PSI-SEC}$$

$$T_D = .00014 \text{ SEC}$$

QUASI-STATIC PRESSURES WILL NOT AFFECT THE WALLS SINCE THEY WILL ACT ON BOTH SIDES.

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COMPUTATION SHEET

SHEET NO.
OF

PROJECT NO.: 62-7092 SPONSOR: CERL
SUBJECT: Living Wall II
BY: NRS DATE: 22 Jul 82 CHECKED BY: _____ DATE CHECKED: 19

1' x 5' wall panel
charge on floor

$$R = 3.5' \quad w = 157$$

$$w^{1/3} = 5.39$$

$$x/w^{1/3} = 0.649 \text{ ft/lb}^{1/3}$$

$$x_{imp} = 1.05 \text{ psi-sec}$$

$$x/w^{1/3} = 0.19 \quad \frac{\text{psi-sec}}{\text{lb}^{1/3}}$$

$$x/r = .7$$

$$\text{Pressure} = \underline{15,000 \text{ psi}}$$

1.5

INPUT PARAMETERS:

CHARGE WEIGHT =	157.00 LB
PERPENDICULAR STANDOFF =	3.50 FT
PLATE LENGTH, XTOT =	1.00 FT
PLATE WIDTH, YTOT =	5.00 FT

CHARGE POSITION IN PLANE PARALLEL TO PLATE

LENGTHWISE DISTANCE FROM PLATE CORNER, XC =	0.50 FT
WIDTHWISE DISTANCE FROM PLATE CORNER, YC =	0.00 FT

OUTPUT:

TOTAL APPLIED IMPULSE =	0.7544E+03 LB-SEC
AVERAGE SPECIFIC IMPULSE =	1.05 PSI-SEC

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
1 OF

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: FRAGMENT THREAT ANALYSIS
BY: LMV DATE: 8/3 19 82 CHECKED BY: DATE CHECKED: 19

1.2 FRAGMENT THREAT ANALYSIS:

ASSUME A WORST CASE STEEL FRAGMENT WEIGHING 0.4 lbs, IN SHAPE OF A CUBE, WITH AN LD RATIO CLOSE TO 1.0 AND AN IMPACT VELOCITY OF 3000 fps.

1.2.1 CALCULATE DEPTH OF PENETRATION INTO CONCRETE

MODIFIED NDRC EQUATION FOR DEPTH OF PENETRATION "X"

$$G(x/d) = \alpha KN \frac{W}{d} \left(\frac{V}{1000d} \right)^{1.8}$$

WHERE:

α = CONSTANT = 1.0

N = MISSILE SHAPE FACTOR = 0.72

$$K = \frac{180}{\sqrt{E}} \left(\frac{E}{29,000} \right)^{1.25} \quad E = \text{MODULUS OF ELASTICITY OF MISSILE IN KIPS/IN}^2$$

$E = 30,000 \text{ KIPS/IN}^2$
 $E_c = 4000 \text{ PSI}$

$$K = 2.97$$

W = MISSILE WEIGHT (lbs)

V = IMPACT VELOCITY (fps)

d = MISSILE DIAMETER (in)

- a) CALCULATE MISSILE DIAMETER. A CUBE STEEL FRAGMENT WEIGHING 0.4 lbs, CALCULATE THE PRESENTED AREA

$$S Vol = W$$

$$Vol = W/\rho = 0.4 \text{ lbs} / 0.283 \text{ lbs/IN}^3$$

$$VOLUME = 1.413 \text{ IN}^3$$

$$AREA = 1.259 \text{ IN}^2$$

CONVERT THE PRESENTED AREA OF THE CUBE (1.259 IN²) TO A PRESENTED AREA OF A CYLINDER AND DETERMINE THE DIAMETER "d".

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
2 OF

PROJECT NO.: 22-7092 SPONSOR: CCEL
SUBJECT: FRAGMENT THREAT ANALYSIS
BY: LMV DATE: 8/3 19 82 CHECKED BY: DATE CHECKED: 19

$$A_{cyl} = \frac{\pi d^2}{4} = 1.259 \text{ in}^2$$

$$d = 1.266 \text{ in}$$

b) SOLVE MODIFIED NDRC EQUATION FOR DEPTH OF PENETRATION "X"

$$G(x/d) = \alpha K N \frac{w}{d} \left(\frac{v}{1000 \text{ ft}} \right)^{1.8}$$

$$G(x/d) = (1.0)(0.72)(2.97) \left(\frac{0.4}{1.266} \right) \left(\frac{3000}{1266} \right)^{1.8}$$

$$G(x/d) = 3.19$$

$$G(x/d) = \frac{x}{d} - 1 \quad \text{for } \frac{x}{d} \geq 2.0$$

$$\therefore \frac{x}{d} - 1 = 3.19$$

$$\frac{x}{d} = 4.19$$

$$x = 5.30 \text{ in.}$$

c) SOLVE MODIFIED NDRC EQUATION FOR DEPTH TO PREVENT PERFORATION "C"

$$\frac{C-a}{d} = 1.32 + 1.24 \left(\frac{x}{d} \right) \quad \text{for } 3 \leq \frac{x}{d} \leq 17$$

$$a = \text{MAX. AGGREGATE SIZE} = 1.5 \text{ in}$$

$$\frac{C-1.5}{1.266} = 1.32 + 1.24 (4.19)$$

$$C = 9.75 \text{ in.}$$

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.

3 OF

PROJECT NO.: 02-7092

SPONSOR: CERL

SUBJECT: FRAGMENT THREAT ANALYSIS

BY: LMY

DATE: 2/2 19 82

CHECKED BY:

DATE CHECKED: 19

1.2.2 CALCULATE DEPTH OF PENETRATION INTO STEEL

THOR EQUATION FOR BALLISTIC PROTECTION VELOCITY (V_0)

$$V_0 = 10^{C_1} (tA)^{\alpha_1} (7000W_s)^{\beta_1} (\sec \theta)^{\delta_1}$$

Where:

t = TARGET THICKNESS IN INCHES.

A = AVERAGE IMPACT AREA OF THE FRAGMENT IN SQUARE INCHES.

W_s = WEIGHT OF THE ORIGINAL FRAGMENT IN LBS.

θ = IMPACT ANGLE (DEGREES)

$C_1, \alpha_1, \beta_1, \delta_1$ = EMPIRICAL CONSTANTS DEPENDANT ON TARGET MATERIAL

SOLVE THOR EQUATION FOR t .

$$V_0 = 3000 \text{ fps}$$

$$A = 1.259 \text{ in}^2$$

$$W_s = 0.4 \text{ lb}$$

$$\theta = 0^\circ$$

$$C_1 = 6.523 \text{ for MILD HOMOGENEOUS STEEL}$$

$$\alpha_1 = 0.906 \text{ for MILD HOMOGENEOUS STEEL}$$

$$\beta_1 = -0.963 \text{ for MILD HOMOGENEOUS STEEL}$$

$$\delta_1 = 1.276 \text{ for MILD HOMOGENEOUS STEEL}$$

$$(tA)^{\alpha_1} = V_0 / 10^{C_1} (7000W_s)^{\beta_1} (\sec \theta)^{\delta_1}$$

$$t = \left[V_0 / 10^{C_1} (7000W_s)^{\beta_1} (\sec \theta)^{\delta_1} \right]^{1/\alpha_1} / A$$

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COMPUTATION SHEET

SHEET NO.
4 OF

PROJECT NO.: 62-7092 SPONSOR: CELL
SUBJECT: FRAGMENT THREAT ANALYSIS
BY: LMV DATE: 8/3 19 82 CHECKED BY: _____ DATE CHECKED: _____ 19 _____

$$t = \left[3000 / 10^{6.523} (7000 - 0.4)^{-0.963} (\sec 0)^{1.296} \right]^{1/906} / 1.259$$

$$t = \left[3000 / (3.334 \times 10^4) (4.79 \times 10^{-4}) \right]^{1/906} / 1.259$$

$$t = (1.878)^{1.104} / 1.259$$

$$t = 1.59 \text{ inches}$$

∴ A STEEL PLATE 1.59 INCHES WILL DEFEAT A STEEL CUBE WEIGHING
0.4 lbs, AT 3000 f/s.

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COMPUTATION SHEET

SHEET NO.
5 OF

PROJECT NO.: C2-7092 SPONSOR: CEEL
SUBJECT: FRAGMENT THREAT VS SPACED ARMOR
BY: LMV DATE: 2/4 19 82 CHECKED BY: _____ DATE CHECKED: _____ 19 _____

2.3 CALCULATE DEPTH OF PENETRATION INTO SPACED ARMOR

$$V_R = V_S - 10^C (tA)^\alpha (7000 W_S)^\beta V_S^\lambda$$

MILD Homoc. STEEL

$$\begin{array}{ll} C = 6.399 & t = .8 \text{ in} \\ \alpha = .889 & A = 1.259 \text{ in}^2 \\ \beta = -.945 & W_S = 0.4 \text{ lb} \\ \lambda = .019 & V_S = 3000 \text{ fps} \end{array}$$

$$V_S^\lambda = 3000^{.019} = 1.1643$$

$$(7000 W_S)^\beta = (7000(.4))^{-.945} = .0005326$$

$$10^C = 10^{6.399} = 2.5 \times 10^6$$

$$(tA)^\alpha = (.8(1.259))^{.889} = 1.0064$$

$$V_R = V_S - 1618.8$$

$$V_R = 1381.2 \text{ fps}$$

(CALCULATE THICKNESS TO STOP FRAG. WITH IMPACT VEL = 1381 fps. ASSUME NO MASS LOSS AFTER 1st IMPACT, NORMAL IMPACT)

$$V_0 = 10^{C_1} (tA)^{\alpha_1} (7000 W_S)^{\beta_1}$$

$$tA^{\alpha_1} = V_0 / 10^{C_1} (7000 W_S)^{\beta_1}$$

$$t = \frac{1}{A} \left[V_0 / 10^{C_1} (7000 W_S)^{\beta_1} \right]^{1/\alpha_1}$$

$$\begin{array}{ll} C_1 = 6.523 & A = 1.259 \text{ in}^2 \\ \alpha_1 = 0.906 & V = 1381 \text{ fps} \\ \beta_1 = -0.963 & \end{array}$$

$$t = \frac{1}{1.259} \left[1381 / 3334260 (.0004791) \right]^{1/.906} = 1.1037524$$

$$t = .676 \text{ in.}$$

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COMPUTATION SHEET

SHEET NO.
6 OF

PROJECT NO.: 02-7092 SPONSOR: CELL
SUBJECT: FRAGMENT PENETRATION INTO STEEL
BY: LMV DATE: 8/4 19 82 CHECKED BY: _____ DATE CHECKED: _____ 19 _____

FRAGMENT PENETRATION INTO STEEL - PANTEX MANUAL - 6.4.1.1

ASSUME FRAGMENT IS A SPHERE

$$\frac{4}{3} \pi r^3 \rho = W_f$$

$$\pi r^3 = 0.4 / .283 \left(\frac{3}{4} \right)$$

$$r = 0.695 \text{ in.} = a$$

$$\begin{aligned} V_{50} &= 3000 \text{ fps} \\ \rho_p &= .283 \text{ lb/in}^3 \\ \rho_t &= .283 \text{ lb/in}^3 \\ \sigma_t &= 36 \times 10^3 \text{ psi} \end{aligned}$$

$$\frac{V_{50} \rho_p}{\sqrt{\sigma_t \rho_t}} = 8.4$$

$$\text{From Fig. 6.21, } \frac{h}{a} = 2.1, a = 0.695 \text{ in.} \\ h = 1.46 \text{ in.}$$

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
7 OF

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: Fragment Threat Analysis
BY: TKM DATE: 8/3 19 82 CHECKED BY: DATE CHECKED: 19

7.2.4 CALCULATE DEPTH OF PENETRATION INTO SAND

Figure 1 from TM5-1300 is used.

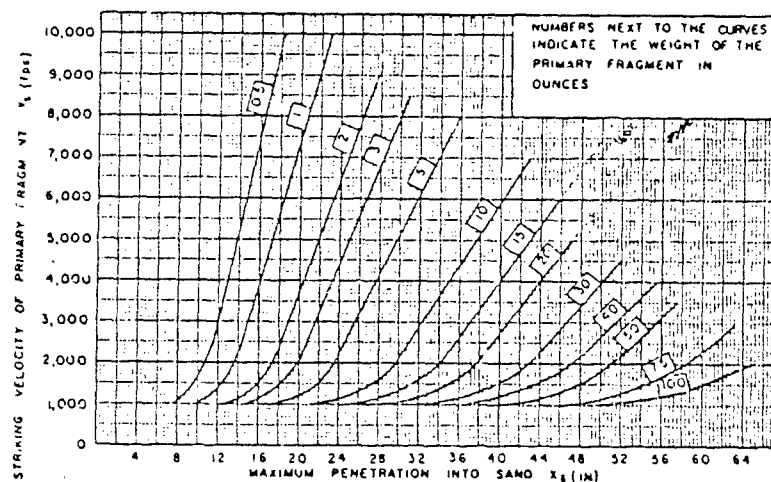


Figure 1

Steel Fragment weight = 0.4 lb = 6.4 oz
Striking velocity = 3000 ft/sec

Interpolating from the Figure for a weight of 6.4 oz
yields a maximum penetration into sand of
26.28 inches.

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SHEET NO.
OF

PROJECT NO: 22-7092 SPONSOR: CSH
SUBJECT: Calculation of Spall for Dividing Wall in 120A40 Bay
BY: TKM DATE: 1 SEP 1982 CHECKED BY: DATE CHECKED: 19

7.3 SPALL ANALYSIS

7.3.1 DETERMINE WHETHER SPALL WILL OCCUR

$$S_c = 4000 \text{ psi}$$

$$G_u = 300 \text{ psi}$$

$$H = \text{wall thickness} = 12'$$

$$E = 2 \times 10^6 \text{ lb/in}^2$$

$$\rho = 2.25 \times 10^{-4} \text{ lb-sec}^2/\text{in}^4$$

$$P_r = 14,175 \text{ psi}$$

$$i_r = 1.3 \text{ psi-sec.}$$

Input parameters from blast loads.

Using Figure 1 from DOE/TIC 11262, determine whether or not spalling will occur on the backface of a dividing wall which has been loaded by blast from the detonation of an explosive inside a bay.

First calculate the stress wave velocity, v , through the wall:

$$v = \sqrt{\frac{E}{\rho}} = \sqrt{\frac{2 \times 10^6 \text{ lb/in}^2}{2.25 \times 10^{-4} \text{ lb-sec}^2/\text{in}^4}} = 94280 \frac{\text{in}}{\text{sec}} = 7862 \frac{\text{ft}}{\text{sec}}$$

$$\frac{i_r v}{P_r H} = \frac{(1.3 \text{ psi-sec})(94280 \frac{\text{in}}{\text{sec}})}{(14,175 \text{ psi})(12 \text{ in})} = 0.72$$

$$\frac{P_r}{G_u} = \frac{14,175 \text{ psi}}{300 \text{ psi}} = 44$$

Using Figure 2 we determine that backface spalling will occur.

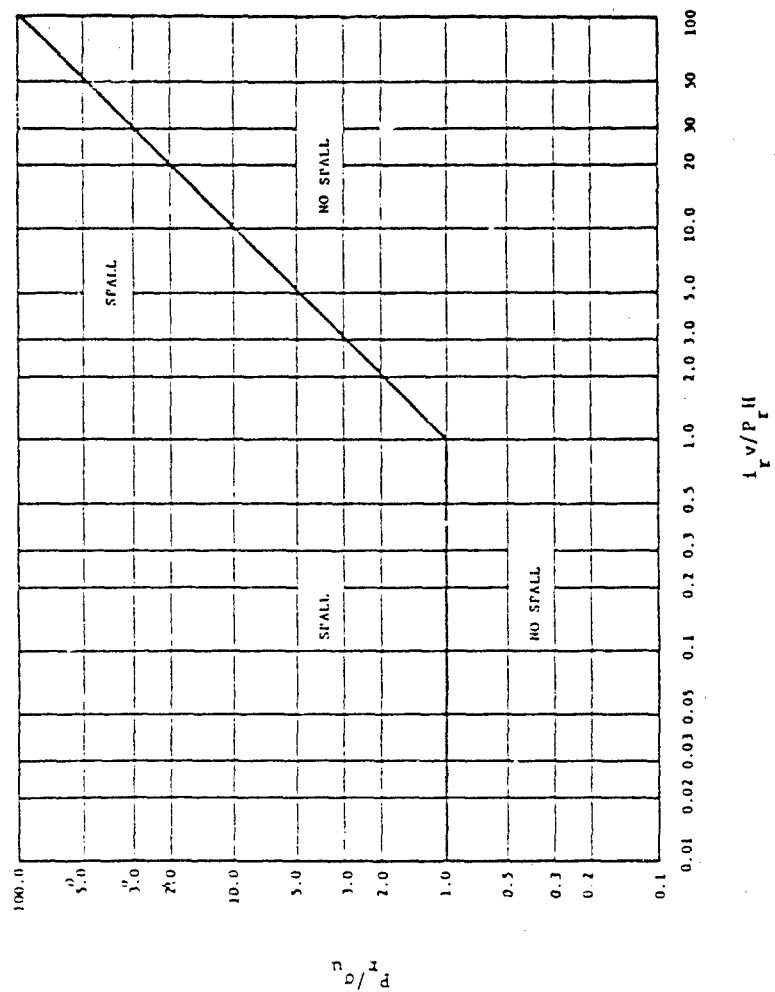


Figure 1. Spall Threshold for Blast Waves Loading Walls

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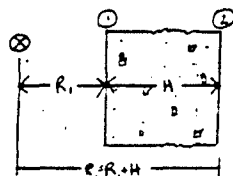
SHEET NO.
OF

PROJECT NO.: 02-7092 SPONSOR: CEPL
SUBJECT: Spall Velocity Calculation for Dividing Wall
BY: JKM DATE: 3 Sep 82 CHECKED BY: DATE CHECKED: 19

7.3.2 DETERMINE SPALL VELOCITY

R=3.5'

First, attenuate i_r linearly through the wall to obtain estimate at backface of wall, i_{r2} .



$$i_{r2} = i_r \frac{R_1}{(R_1 + H)} = (1.3 \text{ psi-sec}) \left(\frac{3.5'}{3.5 + 1} \right) = 1.01 \text{ psi-sec}$$

Now determine a spall velocity for a 3'x3'x3' chunk of spall.

$$\text{Area } A = 9 \text{ in}^2$$

$$\text{Impulse } i_{r2} = 1.01 \text{ psi-sec}$$

$$\text{Mass } m = (27 \text{ in}^3) (2.3 \times 10^{-4} \frac{\text{lb-sec}^2}{\text{in}^3}) = 6.2 \times 10^{-3} \frac{\text{lb-sec}^2}{\text{in}}$$

Velocity, then, is

$$u = \frac{i_{r2} A}{m} = \frac{(1.01 \text{ psi-sec})(9 \text{ in}^2)}{(6.2 \times 10^{-3} \frac{\text{lb-sec}^2}{\text{in}})} = 1466 \frac{\text{in}}{\text{sec}} = 122 \frac{\text{ft}}{\text{sec}} \approx 120 \frac{\text{ft}}{\text{sec}}$$

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COMPUTATION SHEET

SHEET NO.
1 OF

PROJECT NO.: 02-7092

SPONSOR: CERL

SUBJECT: DIVIDING WALL DESIGN

BY: B. MORRIS

DATE: 20 JUL 19 82

CHECKED BY:

DATE CHECKED: 19

7.4 DIVIDING WALL DESIGN CALCULATIONS

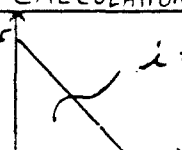
PRESSURE-TIME

26,250 PSI

$\lambda = 1.84$ PSI-SEC

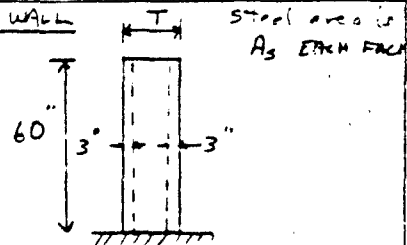
FORCE = $60 \times 26,250 = 1,575,000$

.00014 SEC



7.4.1 DESIGN FOR CONCRETE CANTILEVER WALL

$f'_c = 4000$ PSI
 $f_y = 90,000$ PSI
 $\theta = 12^\circ$



T (IN)	A _s (IN ² /FT)	FLOOR REACTION (LB/IN)	SHEAR STRESS IN WALL = $\frac{\text{REACTION}}{L}$
12	3.38	5070	845 PSI
15	1.77	3982	442
18	1.12	3360	280
21	.78	2925	195
24	.56	2520	140

FOR CANTILEVERS, (TMS-1300, TABLE 5-5) $r_u = \frac{2M_u}{L}$

(TMS-1300, TABLE 5-12) SUPPORT SHEAR $V_u = r_u L$

$\therefore V_s = \text{FLOOR REACTION} = \frac{2M_u}{L} = \frac{2}{L} (F_y)(T-6)$

$V_s = \frac{A_s F_y (T-6)}{L}$

SHEAR STRENGTH OF CONCRETE (TMS-1300) $\leq 2.28(.85) \sqrt{f'_c}$
 $\leq 2.28(.85) \sqrt{4000} = 123 \text{ PSI} = \tau_c$

NOW DETERMINE STEEL AREAS REQUIRED TO PROVIDE SUPPORT STRENGTH.

$A_y = \frac{12" (\text{TOTAL REACTION} - \tau_c (T-6))}{75,000} \text{ IN}^2/\text{FT}$

7-17 $\rightarrow 75,000 = \frac{1}{2} (f_y + f_u) = \frac{1}{2} (64,000 + 90,000)$

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SHEET NO.
2 OF

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: DIVIDING WALL DESIGN
BY: B. MORRIS DATE: 27 JUL 19 82 CHECKED BY: DATE CHECKED: 19

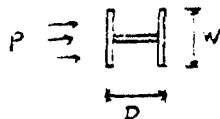
I	REACTION	A _v (IN ² /FT)
12	5070	.69
15	3982	.46
18	3360	.30
21	2925	.17
24	2520	.05

1.4.2 DESIGN FOR STEEL SUPPRESSIVE SHIELD TYPE WALL

CONSIDER AN INTERLOCKING I-BEAM SYSTEM

$$\theta = 12^\circ$$

$$FDY = 60,000 - 1.2 \times 50,000$$



$$L = 60" \quad \text{ASSUME } 30" / \text{FT } SU$$

$$WT = 150^2$$

USE A SPECIAL-PURPOSE PROGRAM DEVELOPED AT SWRI WHICH CALCULATES THE MINIMUM ELASTIC & PLASTIC SECTION MODULUS TO RESIST THE APPLIED LOADS.

W (IN)	D (IN)	REQ'D S (IN ³)	$\frac{S}{W}$	SECTION
6	6	33.98	38.74	
	9	33.98	38.74	
	12	30.91	37.52	
4	4	15.29	17.43	
	8	14.75	16.82	
	12	14.75	16.82	12x4B (16.8)
5	5	23.30	26.56	
	12	23.30	26.56	12I5 (31.8 44.1)
7	7	45.73	52.14	
	12	45.73	52.14	

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COMPUTATION SHEET

SHEET NO.
3 OF 5

PROJECT NO.: 02 7092 SPONSOR: CERL
SUBJECT: DIVIDING WALL DESIGN
BY: B. M. M. M. DATE: 22 SEP 19 82 CHECKED BY: _____ DATE CHECKED: _____ 19__

NOW VARY BEAM WEIGHT / FT WITH LESS D VARIATION

WT/FT (LB)	W (IN)	D (IN)	S (IN)	Z (IN)	POSSIBLE SECTIONS PER 1973 AISC MANUAL
40	4	4	11.55	13.16	
		12	11.01	12.55	
	5	5	17.96	20.47	
		12	17.42	19.86	S 12 x 40.8 (Z=52.1)
	6	6	25.44	29.00	
		12	25.44	29.00	S 15 x 42.9 (Z=69.3)
30	4	4	15.29	17.43	
		12	14.75	16.82	
	5	5	23.30	26.56	
		12	23.30	26.56	S 12 x 31.8 (Z=42.0)
	6	6	33.98	38.74	
		12	32.91	37.52	W 10 x 27 (Z=37.7)
20	4	4	23.30	26.56	
		12	22.23	25.34	W 12 x 22 (Z=29.3)
	5	5	35.05	39.96	
		12	35.05	39.96	W 14 x 26 (Z=40.0)
	6	6	50.01	57.01	
		12	50.01	57.01	
15	4	4	30.78	35.09	
		12	29.71	33.87	
	5	5	47.87	54.57	
		12	45.73	52.14	
	6	6	67.10	76.50	
		12	67.10	76.50	

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COMPUTATION SHEET

SHEET NO.
4 OF 5

PROJECT NO.: 02-7092

SPONSOR: CERL

SUBJECT: DIVIDING WALL DESIGN

BY: MORGAN

DATE: 22 JUL 19 82

CHECKED BY:

DATE CHECKED: 19

WT/FT (LB)	W (IN)	D (IN)	S (IN)	Z (IN)
50	4	12	9.34	9.51
	5	12	13.15	14.99
	6	12	19.49	21.08
	8	12	32.91	37.52
40	8	12	44.67	50.92
30	8	12	58.55	66.75
20	8	12	88.47	101.85
15	8	12		

W 14 x 48 (Z = 78.4)

W 10 x 45 (Z = 54.9)
W 12 x 40 (Z = 57.9)

NOW DETERMINE THE TOTAL STEEL PER FOOT OF
DIVIDING WALL FOR SOME OF THESE SECTIONS

SECTION	WT/FT x (5 FT HEIGHT)	x (12 ^{IN} FT)	x (1.5708 W)	= WT/FT OF WALL
12 IS	31.8 x 5	x $\frac{12}{5}$		= 382 LB
S 12 x 40.8	40.8 x 5	x $\frac{12}{5}$		= 490
S 15 x 42.9	42.9 x 5	x $\frac{12}{5}$		= 429
W 12 x 27	27 x 5	x $\frac{12}{5}$		= 270
W 14 x 26	26 x 5	x $\frac{12}{5}$		= 312

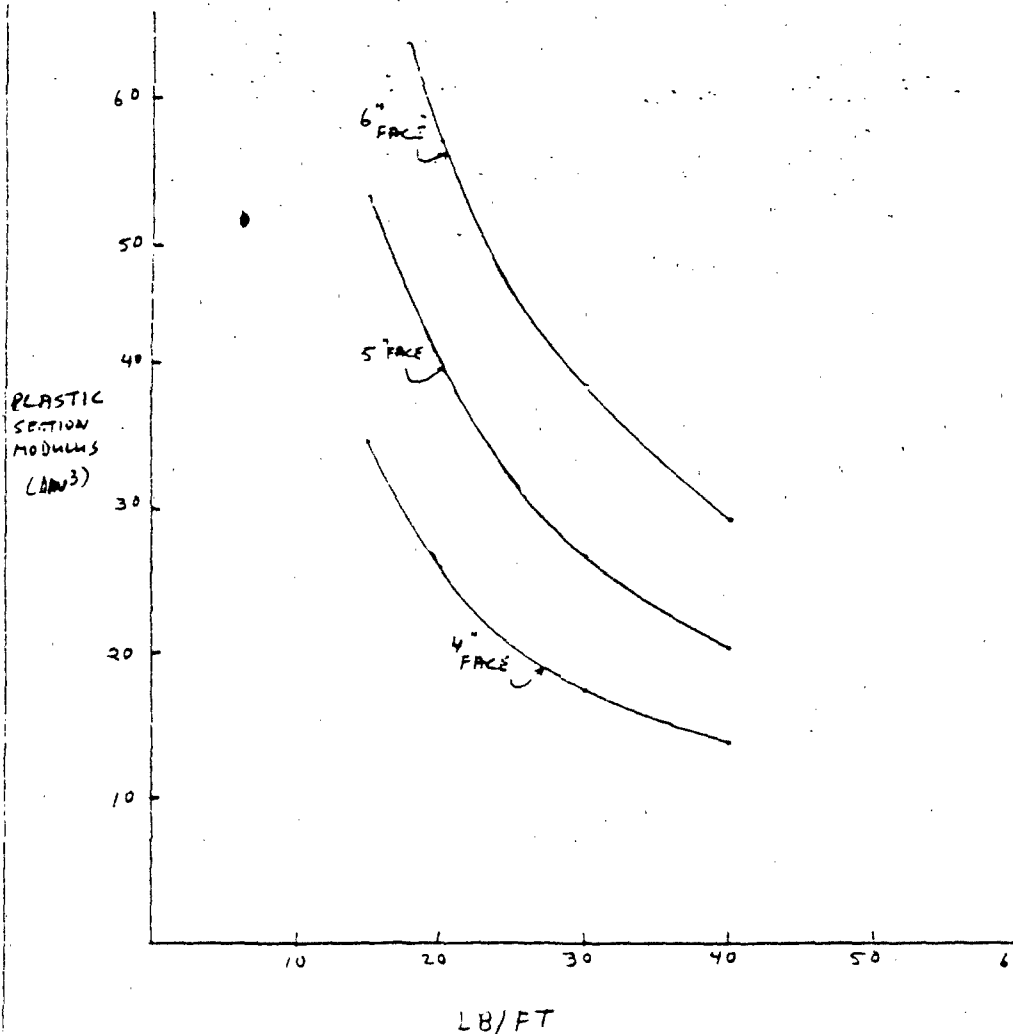
NOW DETERMINE CONCRETE WTS

THICKNESS (IN)	$\frac{I}{12} \times 5 \times 150 = \text{WT OF WALL}$
12	$1 \times 5 \times 150 = 750 \text{ LB}$
15	= 938
18	= 1125

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COMPUTATION SHEET

SHEET NO.
5 OF 5

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: DIVIDING WALL DESIGN
BY: B. MORRIS DATE: 23 JUL 19 82 CHECKED BY: DATE CHECKED: 19



STEEL WT IN 15" RC WALL
FLEXURE, EACH FACE #10 @ 8" $WT = 4.3 \text{ LB/FT} \times 5' \text{ HIGH} \times \frac{12}{8} = 32.3 \times 2 \text{ FACES} = 64.6 \text{ LB/FT}$
SUPPORT SHEAR - #5 @ 8" $WT = 1.04 \text{ LB/FT} \times 4' \times \frac{12}{8} = 6.2 \text{ LB/FT}$
LACING - #5 $WT = 1.04 \times 4 \times \frac{12}{8} = 6.2$
TOTAL 77 LB/FT

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COMPUTATION SHEET

SHEET NO.
1 OF 1

PROJECT NO.: 02-7092 SPONSOR: CERL

SUBJECT: STEEL DIVIDING WALLS

BY: R. MORRIS

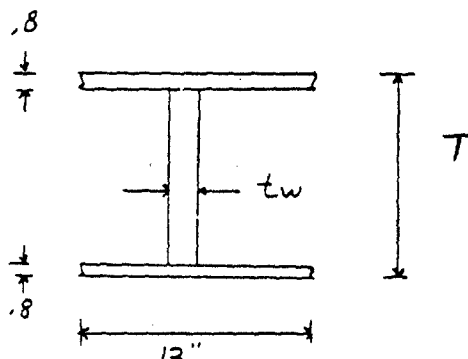
DATE: 7 AUG 19 82

CHECKED BY:

DATE CHECKED:

19

1.4.3. DESIGN FOR TWO STEEL SECTIONS WITH SAND IN BETWEEN
FROM THE WORK STARTING ON PAGE 7-10, TWO SECTIONS
OF STEEL, EACH 0.8 IN THICK, WILL STOP
THE DESIGN FRAGMENT.



DETERMINE WEIGHT OF 1 FT OF WALL (JUST
THE WEIGHT OF THE 0.8" PLATES)

$$WT = 2(0.8)(12) \times 60 \text{ INCH HEIGHT} \times .283 \frac{\text{LB}}{\text{IN}^3}$$

$$WT = 326 \text{ LB/FT OF DIVIDING WALL}$$

IF THE WEB THICKNESS IS 0.8", THE WEIGHT
OF THE ENTIRE PIECE IS 467 LB/FT.

THIS IS TOO MUCH AND A STEEL DIVIDING
WALL SEEMS IMPRACTICAL.

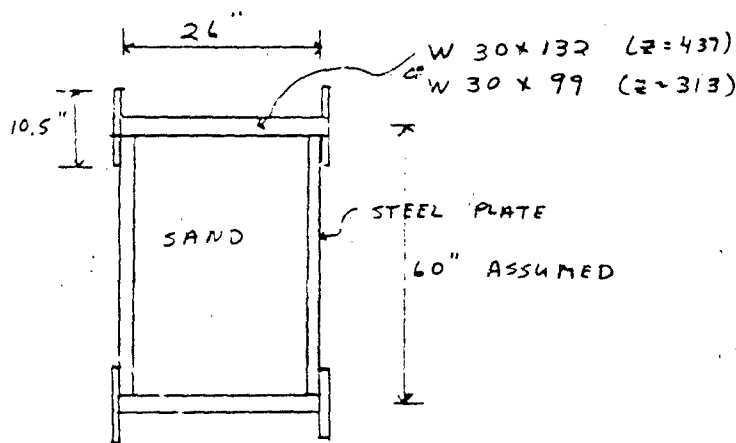
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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
1 OF 3

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: SAND-FILLED DIVIDING WALLS
BY: E. MORRIS DATE: 4 AUG 19 82 CHECKED BY: DATE CHECKED: 19

THE DESIGN FRAGMENT REQUIRES 26"-28" OF SAND TO STOP IT.

ASSUME THE WALL CROSS-SECTION IS AS FOLLOWS:



EACH W30x132 (W30x99) MUST WITHSTAND DESIGN LOADS OVER A 5' LENGTH OF WALL

WT PER VERTICAL FOOT = $(5 \times \frac{26}{12} \times 100) + 132 = 1215 \text{ LB/FT}$

DESIGN PEAK PRESS IS 26,250 PSI, .00014 SEC = T₀.

USE "STEEL" PROGRAM TO DETERMINE REQ'D Z.

FDY = 60,000 L = 60" $\theta = 12^\circ$ W = 1215 x 5
CANTILEVER = 6075 lb
TOTAL PEAK LOAD = 26,250 x 60 x 60 = 9.45 x 10⁷ LB

REQ'D Z = 93.55 < Z OF W30x99. OK

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
2 OF 3

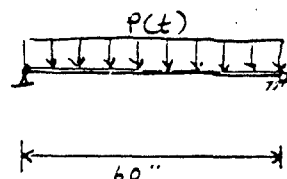
PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: SAND-FILLED DIVIDING WALLS
BY: B. MORRIS DATE: 4 AUG 1982 CHECKED BY: DATE CHECKED: 19

DETERMINE THICKNESS OF STEEL PLATES, TREATING THE 2 PLATES AS A SINGLE ONE.

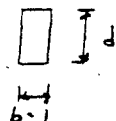
CONSIDER A SIMPLY SUPPORTED BEAM 60" LONG, 1" WIDE, WITH A WEIGHT OF

$$WT = 1" \times 60" \times 26" \times \frac{100 \text{ lb}}{1728 \text{ in}^3} = 90 \text{ LBS}$$

$$MASS = \frac{90}{5.4} = 16.7 \text{ slugs}$$



CROSS-SECTION



$$S = \frac{bd^2}{6}$$

$$I = \frac{bd^3}{12}$$

FOR ELASTIC RESPONSE, $K = \frac{384EI}{5L^3} = \frac{384(30 \times 10^6)}{5(60^3)} \left(\frac{1 \text{ in}^4}{12} \right)$

$$K = 889 d^3$$

FROM BIGG'S, $K_{LM} = .78 \Rightarrow M_c = .78 \frac{90}{36.4} = 1.82$

$$NAT'L PERIOD = T = 2\pi \sqrt{\frac{M_c}{K}} = 6.28 \sqrt{\frac{1.82}{889 d^3}}$$

$$T = .09 d^{1/2}$$

$$M_p = F_D Y * Z = 60,000 \frac{1 \times d^2}{4} = 15,000 d^2$$

MAX RESISTANCE $R_m = \frac{8M_p}{L} = \frac{8 \times 15,000 d^2}{60}$

$$R_m = 2000 d^2$$

PEAK FORCE = $F_1 = 26,250 \times 60 = 1.58 \times 10^5$

ELASTIC DEFL = $\frac{R_m}{K} = \frac{2000 d^2}{889 d^3} = \frac{2.25}{d}$

FOR 12° ROTATION, $\Delta_{max} = 30 \tan 12^\circ = 6.4"$

ROTATION = $\tan^{-1} \frac{\Delta_{max}}{30}$

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COMPUTATION SHEET

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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: SAND-FILLED DIVIDING WALLS
BY: B. MORRIS DATE: 4 AUG 19 82 CHECKED BY: _____ DATE CHECKED: _____

Use whole automated solution to elastic-plastic, one degree of freedom
system. In a program, ONE OF
USE THIS PROGRAM TO DETERMINE MAX DEFL FOR
VARIABLE "d."

$$FDY = 60,000$$

<u>d</u>	<u>$I = \frac{d^3}{12}$</u>	<u>$CS = \frac{d^2}{4}$</u>	<u>$Z = \frac{d^2}{4}$</u>	<u>Δ_{MAX}</u>	<u>ROTATION</u>
1	.083	.17	.25	1.06	2.02°
.5	.010	.042	.025	13.7	24.5°
.75	.035	.0938	.141	2.86	5.44°

CONCLUSION: CAN USE TWO $\frac{3}{8}$ " PLATES
SEPARATED BY LESS THAN 26" OF
SAND SINCE THE PLATES WILL SERVE
TO REDUCE FRAGMENT SPEED.

DETERMINE WEIGHT OF STEEL PER FOOT OF WALL:

$$WT = \frac{1}{8}(6)(99) + .75(12)(60)(1.283) = 153 \text{ LB/FT}$$

1.6" STEEL OR 26" SAND WILL STOP THE FRAGMENT
SO 1" STEEL \approx 16" SAND.

IF USE TWO $\frac{1}{2}$ " PLATES, NEED .8" STEEL \approx 3" SAND
ADDITIONAL.

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SHEET NO.
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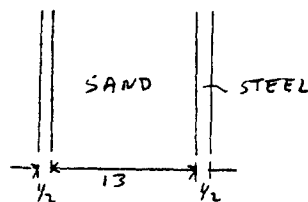
PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: REFINEMENT OF SAND DIVIDING WALL
BY: E. MORRIS DATE: 6 AUG 82 CHECKED BY: DATE CHECKED: 19

FROM PAGE 7-24, TWO $\frac{3}{8}$ " PLATES WITH 26" OF SAND
WILL WORK.

1.6" STEEL OR 26" SAND IS REQUIRED TO STOP
FRAGMENT \Rightarrow 1" STEEL \approx 16" SAND

WITH TWO $\frac{1}{2}$ " PLATES, HAVE 1" STEEL SO NEED
.8" STEEL OR 13" SAND.

LOOK AT:



FOR A 1" X 60" STRIP, $WT = 2(\frac{1}{2})(1)(60)(2.43) + 1(60)(4.4) = 177$

WT = 62 LB

FROM PAGE 7-24, $I = .083$ $S = .17$ $Z = .25$ $FDY = 69,000$

PEAK FORCE = 1.58×10^5 $T_D = .00014$ SEC

USE "ONEDP" TO DETERMINE MAX DEFL. OF THIS
SECTION:

$\Delta_{MAX} = 1.44"$
ROTATION = 2.74°

NOW SELECT A STEEL SECTION WITH 14" BETWEEN
FLANGES:

TRY W18 X 35 $I = 513$ $S = 57.9$ $Z = 66.8$
TOTAL WT = 5 WT X 35 + 60 WT X 12 = 3895 LB
7-26



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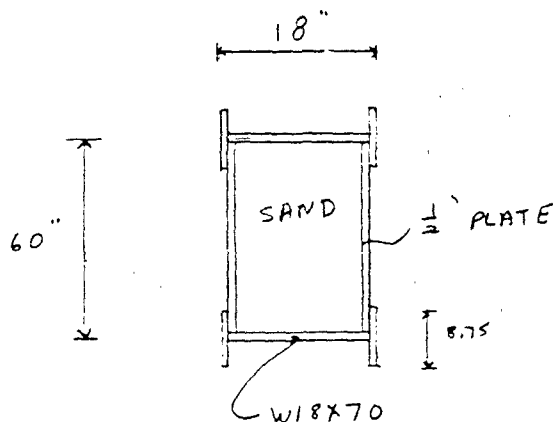
PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: REFINEMENT OF SAND DIVIDING WALL
BY: B. MORRIS DATE: AUG 19 82 CHECKED BY: DATE CHECKED: 19

USE "STEEL" PROGRAM TO DETERMINE REQUIRED
 Z

$$Z(\text{REQ'D}) = 144.7 > Z(W18 \times 35)$$

TRY W18X70 $Z = 145$ OK SINCE WT WILL
INCREASE SLIGHTLY.

FINAL SOLUTION:



STEEL WT PER FT OF WALL: $\frac{70 \times 5 + 1 \times 60 \times 60 \times 2.83}{5} = 274 \text{ LB}$

CHECK FLOOR REACTION SHEAR CAPABILITY OF W18X70:

$$M_{\text{CAP}} = F_D Y * Z = 60,000 \times 145 = 8.7 \times 10^6$$

$$\text{FLOOR REACTION (PER TMS-1300)} = V_s = \frac{2M}{L} = \frac{2(8.7 \times 10^6)}{60}$$

7-27

$$V_s = 2.9 \times 10^5 \text{ LB}$$

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COMPUTATION SHEET

SHEET NO.
3 OF 3

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: REFINEMENT OF SAND DIVIDING WALL
BY: B. MORRIS DATE: 6 AUG 19 82 CHECKED BY: DATE CHECKED: 19

FOR W18X70, WEB THICKNESS = .438"

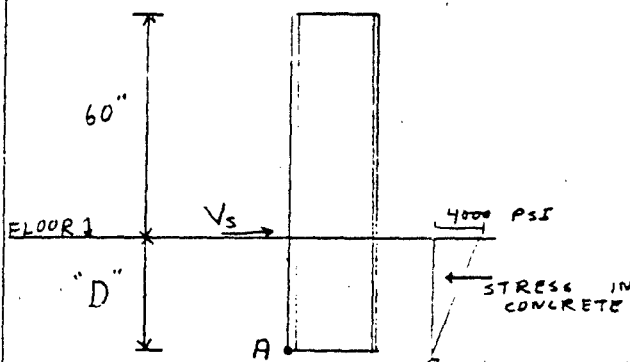
$$\text{WEB HEIGHT} = 18 - 2t_f = 18 - 2(.751) = 16.5"$$

$$\text{WEB AREA} = 7.2 \text{ IN}^2$$

$$\text{SHEAR STRESS} = \frac{V_s}{\text{WEB AREA}} = \frac{2.9 \times 10^5 \text{ LB}}{7.2 \text{ IN}^2} = 40300 \text{ PSI}$$

THIS IS LESS THAN $F_y = 50,000$ OK

LOOK AT FLOOR CONNECTION.



ASSUME THAT FLANGES OF THE W18X70 ARE REINFORCED BELOW THE FLOOR SO THE WHOLE FLANGE AREA RESISTS THE FORCE V_s .

ASSUME BEAM ROTATES ABOUT POINT "A".

$$\sum M_A = 0 \Rightarrow V_s D = \frac{1}{2} (4000) (8.75) (D) \times \frac{2}{3} D$$

$$\text{Force} \times \text{Dist} = \text{Ave STRESS} \times \text{AREA} \times \text{DIST}$$

$$2.9 \times 10^5 = 11,667 D$$

$$D = 24.9"$$

7-28

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COMPUTATION SHEET

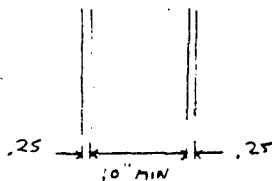
SHEET NO. 1
OF 1

PROJECT NO. 02-7092 SPONSOR: CERL
SUBJECT: STEEL-CONCRETE COMPOSITE DIVIDING WALL
BY: E. MORRIS DATE: SEP 19 82 CHECKED BY: DATE CHECKED: 19

14.4 STEEL/CONCRETE COMPOSITE WALL

$P_{max} = 26,250 \text{ PSI}$ $T_D = .0004 \text{ SEC}$
1.6" STEEL OR $1.6 \times 9 = 14.4$ " CONCRETE WILL STOP THE
FRAGMENT

USE THE FOLLOWING SECTION:



WITH $2 \times .25 \times .5$ " PLATE, NEED
ADDITIONAL 1.1" STEEL = 9.9"
CONCRETE, HENCE 10" MINIMUM

FROM PAGE 7-16, "DIVIDING WALL DESIGN,"
A 12" CONCRETE WALL WITH $A_s = .282 \text{ IN}^2/\text{IN}$ WITH
A MOMENT ARM OF $12 - 6 = 6$ " WILL SUFFICE.

USE "ONEDF2" TO CHECK MAX DEFLECTION

$$FDY = 90,000 \quad F_c' = 4000 \quad A_s = .25 \text{ IN}^2/\text{IN}$$

$$DC = 10.25" \quad \text{TOTAL THICK} = 10 + .5 \times \frac{4000}{150} \approx 12$$

$$\Delta_{MAX} = 8.38" \quad \text{ROTATION} = 7.95^\circ \quad \text{OK}$$

APPENDIX 8

WEAPON LOADOUT



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DEPARTMENT OF BALLISTICS AND EXPLOSIVES SCIENCES
COMPUTATION SHEET

SHEET NO.
1 of 31

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: REQUIREMENTS
BY: rwf DATE: 6-23-1982 CHECKED BY: DATE CHECKED: 19

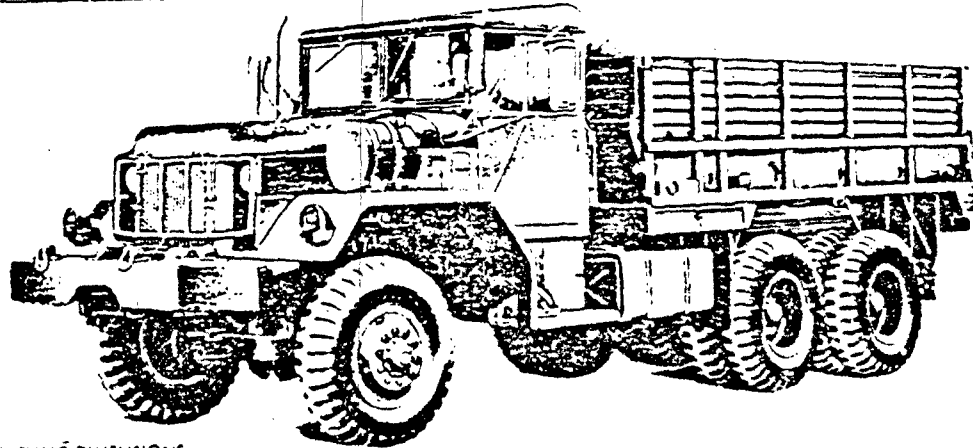
Weapon Movement and Handling (M/H)
Requirements and Constraints

- ✓ 1. M/H system must be compatible with bay sealing requirements
- ✓ 2. Must be able to move weapons without building power
- ✓ 3. M/H equipment must be on current NRC-approved list
- ✓ 4. Cannot "do" anything to round except move it (i.e. mis hazard)
- ✓ 5. Workflow (a) storage to dock to truck
(b) truck to dock to storage
(c) storage to maintenance to storage
- ✓ 6. Must provide 4,000 pound overhead crane in maintenance area
- ✓ 7. Must have ≤ 5000 lb dead load on dock at any one time
- ✓ 8. Loadout cannot compromise chemical collective protection system
- ✓ 9. Loadout 57 weapons of specified mix in ≈ 2 hours.
- ✓ 10. External weapon transport by $2\frac{1}{2}$ and 5 ton standard trucks only
- ✓ 11. Crew of two per truck
- ✓ 12. Weapons are processed in/out of facility by serial number
- ✓ 12. M/H compatible with 12 foot clear ceiling height
- ✓ 14. Weapons transported beyond facility as a group.
- ✓ 15. Load dock ≤ 500 ft²
- ✓ 16. M/H must minimize manpower requirements
- ✓ 17. M/H must minimize cost (investment and O&S)
- ✓ 18. M/H system must be able to operate in a d.bis environment following an explosion in a bay.

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
2 OF 3

PROJECT NO.: 02-7692 SPONSOR: CERL
SUBJECT: 5-TON VEHICLES
BY: guth DATE: 6-23 19 82 CHECKED BY: _____ DATE CHECKED: _____ 19

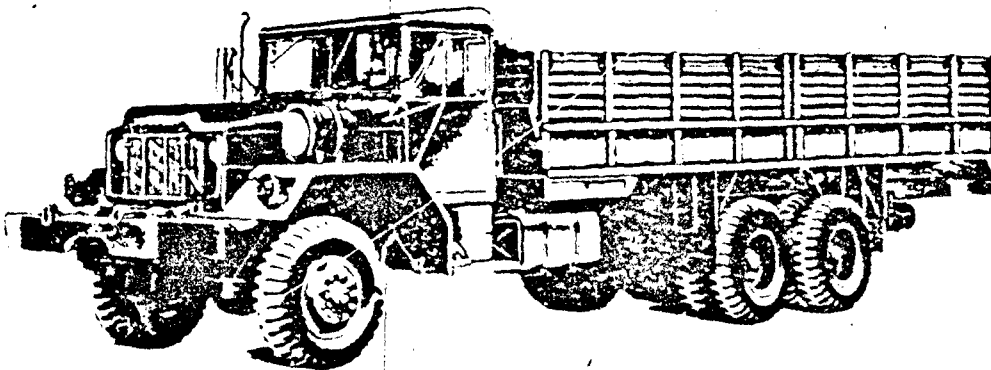


SHIPPING DIMENSIONS

OVERALL LENGTH	317 in.	SHIPPING CUBAGE	1538 cu. ft.
OVERALL HEIGHT	85-1/2 in.	SHIPPING WEIGHT	21686 lbs.
OVERALL WIDTH	96 in.	SHIPPING TONNAGE	10.84 tons

AT 26062

TRUCK, Cargo, M813, w/w - left front view



SHIPPING DIMENSIONS

OVERALL LENGTH	392-13/16 in.	SHIPPING CUBAGE	1878 cu. ft.
OVERALL HEIGHT	85 -1/2 in.	SHIPPING WEIGHT	25724 lbs.
OVERALL WIDTH	96 in.	SHIPPING TONNAGE	12.86 tons

AT 26056

TRUCK, Cargo, M814, w/w - left front view

X-3

SOUTHWEST RESEARCH INSTITUTE
DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
2 OF 31

PROJECT NO. 02-7092 SPONSOR CERL
SUBJECT: 5-TON VEHICLES
BY: nwb DATE 6-23-92 CHECKED BY DATE CHECKED: 19

5-TON VEHICLES

1-5. Tabulated Data

a. General.

Crew	2
Vehicle weight (empty) with front winch (except as noted).	
M813	21,686 lbs.
M813A1	21,686 lbs.
M814	25,724 lbs.
M815	21,790 lbs.
M816	36,000 lbs.
M817	24,323 lbs.
M818	19,260 lbs.
M819	24,480 lbs.
M820	27,340 lbs. w/o/w
M820A1	27,550 lbs. w/o/w
M820A2	29,020 lbs. w/o/w
M821	28,710 lbs.

NOTE

Deduct 714 pounds for vehicles not equipped with a front winch.

Payload (on highway):

M813, M813A1, M814, M815, and M821	20,000 lbs. <i>10T</i>
M816 (on crane)	See crane safe load plate
M817	20,000 lbs. cargo
M818	25,000 lbs. weight on fifth wheel
M819	16,000 lbs. weight on fifth wheel
M820, M820A1 and M820A2	15,000 lbs.

Payload (off highway):

M813, M813A1, M814, M815, M817, and M821	10,000 lbs. cargo <i>5T</i>
M816 (on crane)	See crane safe load plate
M818	15,000 lbs. weight on fifth wheel
M819	12,000 lbs. weight on fifth wheel
M820, M820A1, and M820A2	5,000 lbs.

NOTE

Payload limits shown are without personnel (400 lb) 15,000 and 12,000-pound loads on fifth wheel are for limited cross country operation.

b. Dimensions.

Height overall:

M813, M813A1, and M814	86 1/2 in.
M815	118 in.
M816	106 in.
M817	110 5/8 in.
M818	89 in.
M819	132 in.
M820, M820A1, and M820A2	136.37 in.
M821	120 3/4 in.

Length, overall, w/o front winch:

M813, M813A1, and M815	304 in.
M814	377 5/16 in.
M817	273 in.

M818	364 3/8 in.
M820, M820A1, and M820A2	360 in.

Length, overall, w/front winch:

M813, M813A1, and M815	317 in.
M814	392 13/16 in.
M816	356 in.
M817	388 1/2 in.
M818	280 in.
M819	359 1/4 in.
M821	372 1/2 in.

Width:

M813, M813A1, and M815	96 5/8 in.
M814, M816, and M818	97 in.
M817	97 1/4 in.
M819	97 11/16 in.
M820, M820A1, and M820A2	98 in.
M821	115 in.

Ground clearance, minimum

M813, M813A1, M814, M815, M817, and M819	10 1/2 in.
M816, and M819	11 in.
M820, M820A1, M820A2, and M821	13 in.

Turning circle (dia), w/o front winch:

M813, and M813A1	83 ft. 8 in.
M814, M820, and M820A1	93 ft. 10 in.
M817	78 ft. 10 in.
M818	76 ft. 8 in.
M820A2	94 ft. 10 in.

Turning circle (dia) w front winch:

M813, M813A1, and M815	84 ft. 8 in.
M814	95 ft. 4 in.
M816, and M819	94 ft. 10 in.
M817	80 ft. 2 in.
M818	78 ft. 6 in.
M821	95 ft. 7 in.

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COMPUTATION SHEET

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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: 5-TON VEHICLES
BY: mmh DATE: 6-23-82 CHECKED BY: _____ DATE CHECKED: _____ 19__

c. Capacities.

Cooling system 42 qta.
Crankcase: (wet refill) 23 qta.
Oil filter 2 qta.
Differentials, each 12 qta.
Transmission w/o power takeoff 13 qta.
w/ power takeoff 14 qta.
Transfer w/o power takeoff 5 qta.
w/ power takeoff 5 1/2 qta.

Winch:

Front 2.6 qta.
Rear 3.0 qta.
Midship 2.6 qta.

Fuel tank:

M813, M813A1, M814, M815,
M819, M820, M820A1, M820A2,
and M821 78 gals.
M816 133 gals.
M817, and M818 110 gals.

d. Performance.

Maximum speed 59 mph @ 2100 rpm.

Fording Depth:

With fording kit installed 78 in.
Without fording kit 30 in.
Midship winch capacity (max) 20,000 lbs.
Front winch capacity (max) 20,000 lbs.

Maximum towed load (on highway):

All models except M818, and
M819 30,000 lbs.
(pintle)
M818 56,000 lbs. (fifth
wheel)
M819 46,000 lbs. (fifth
wheel)

Maximum towed load (off highway):

All models except M818, and
M819 15,000 lbs.
(pintle)
M818 20,000 lbs. (fifth
wheel)
M819 30,000 lbs. (fifth
wheel)

e. Tire Pressures.

Highway (psi):

M813, M813A1, M814, M816,
M817, and M818 70
M817 80
M819 75
M820, M820A1, and M820A2 90
M821 55

Cross country (psi):

All models 35

Mud, sand, and snow (psi):

All models 15

f. Electrical System.

Batteries (four connected in series-parallel)

Model 6TN23
Voltage (each) 12
Ground negative

g. Cooling System.

Surge tank filler cap:

Opening pressure 7 psi

Thermostat:

Starts to open at 160°F.
Fully open 185°F.

h. Brakes.

Service brakes:

Type Air-hydraulic
Brake pedal free travel 1/4 to 1/2 in.

Hand brakes:

Location transfer rear out-
put shaft

i. Front Winch.

Capacity 20,000 lbs.

Cable length:

All models except M816, and
M819 200 ft.
M816, and M818 280 ft.

j. Rear Winch (M816).

Capacity 46,000 lbs.
Cable length 280 ft.

k. Midship Winch (M815).

Capacity 20,000 lbs.
Cable length 200 ft.

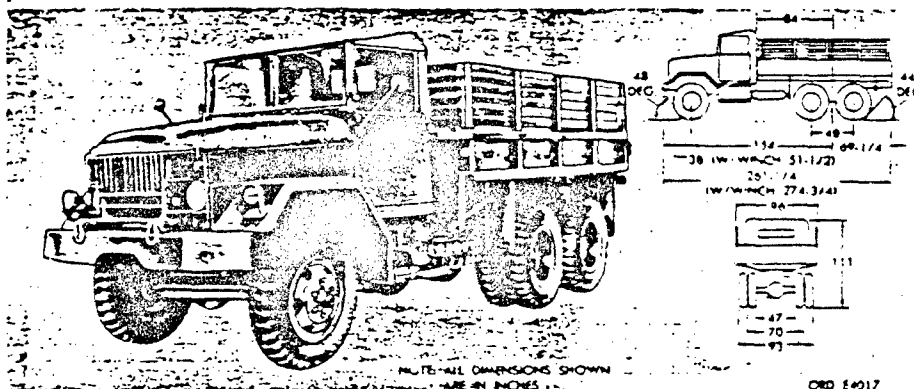
TM 9-2320-260-10
April 1970

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COMPUTATION SHEET

SHEET NO
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: 2 1/2-TON VEHICLES
BY: [signature] DATE: 6-22-1972 CHECKED BY: DATE CHECKED: 19

TRUCK, CARGO: 2 1/2-TON, 6x6, M35 AND M35A1, W/ AND W/O WINCH, W/E R&D TM 9-500



Major Item

Model	Line Item No.	Federal stock No.
M35, w/w, soft top	4-60110-19	2320-425-4444
M35, w/o, soft top	4-60110-28	2320-425-4445
M35, w/w, hard top	4-60110-31	2320-425-4446
M35A1, w/o, hard top	4-60110-37	2320-425-4447
M35A1, w/w	4-60110-32	2320-442-4424
M35A1, w/o	4-60110-29	2320-442-4425

Cross-country:

M35	3,350 lb
M35A1	3,350 lb

Rear axle gear ratio:

M35	6.72:1
M35A1	6.72:1

Axle load:

Empty	
Front	
M35	3,810 lb
M35A1	

Rear	
M35	3,335 lb
M35A1	

Loaded	
Front	
M35	6,700 lb
M35A1	

Rear	
M35	3,140 lb
M35A1	

Time

PH	10
Size	9.00 x 20

Pressure

Highway	
Front	70 psi
Rear	45 psi

Cross-country	
Rear	75 psi
Front	45 psi

Sand	15 psi
Tread, center-to-center front	57 1/2 in.

Electrical system

Number of batteries (12 volts each)	2
Voltage	24
Ground	Negative

Fuel system

Fuel system rating (gallons/hr)	30
Compression-ignition engine type	35-37
Diesel	40

Cooling system

Cooling system	
M35 (w/o booster)	22 qt
M35A1	34 qt

General

TRUCK, CARGO: 2 1/2-ton, 4 x 4, M35 and M35A1 is a vehicle used to transport general cargo or personnel. The truck has iron axles for 14 passengers whom it can carry in lieu of cargo. It is dual-tired on the rear wheels. The body is a 12-foot steel flatbed type. Sides and tailgate frame are removable.

Differences among models

The M35 cargo truck has a spark plug ignition engine. The M35A1 has a multifuel compression ignition engine.

Data plate location

Classification: M35—Standard A (OTCM 36481)
M35A1—Standard A (OTCM 37471)

CHARACTERISTICS

Crew	3
Passenger	14

Length overall	278 in.
W/Winch	182 in.
W/o Winch	194 in.

Width overall	74 in.
Height (as supplied to travel)	112 in.
Weight net	

M35	
W/o Winch	12,445 lb
W/Winch	12,580 lb

M35A1	
W/o Winch	13,443 lb
W/Winch	13,540 lb

Payload	
M35	10,350 lb
M35A1	10,000 lb

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DEPARTMENT OF ENERGETIC SYSTEMS
COMPUTATION SHEET

SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: 2 1/2-TON VEHICLES
BY: [signature] DATE: 6-22-92 CHECKED BY: DATE CHECKED: 19

Crankcase refill:
M33 9 at
M33A1 18 at
Steering gear 1 ps
Fuel tank 50 gal
Axle each 7 at
Transmission:
W/power take-off 10 1/4 pt
W/o power take-off 8 1/4 pt
Transfer 1 at
Winch clutch housing 1 at
Winch worm housing 1 1/4 pt
Brakes:
Manufacturer Bendix-Westinghouse
Type Air-hydraulic
Parking type Transfer
Engine:
M33 truck:
Manufacturer Rex Motors Inc. Model OA-331
Continental Model COA-331
Type gasoline, valve-in-head 4-cycle
Number of cylinders (in line) 6
Displacement 321 cu in.
Bore 4 1/4 in.
Stroke 4 1/4 in.
Compression ratio 6.73:1
Governed speed 3,600 rpm
Brake horsepower (w/std accessories) 146 @ 3,600 rpm
Torque (w/std accessories) 248 lb-ft @ 1,400 rpm
M33A1 truck:
Manufacturer Continental Model LDS-427-2
Type multifuel, compression-ignition
Number of cylinders (in line) 6
Displacement 427 cu in.
Bore 4.3125 in.
Stroke 4.875 in.
Compression ratio 28:1
Governed speed 2,800 rpm
Brake horsepower (w/std accessories) 140 @ 2,800 rpm
Torque (w/std accessories) 340 lb-ft @ 1,400 rpm
Transmission:
Manufacturer Spicer Mfg Co. Model 3052
M33 Spicer Mfg Co. Model 3052
M33A1 Spicer Mfg Co. Model 3052A
Type synchromesh
Speeds:
Forward 5
Reverse 1
Gear ratio:
High:
M33 1:1
M33A1 0.79:1
Low:
M33 7.58:1
M33A1 3.02:1
Reverse:
M33 7.36:1
M33A1 4.90:1
Transfer:
Manufacturer Wisconsin Axle Div. Model T-136-10
M33 Rockwell-Standard Model T-136-21
M33A1 Rockwell-Standard Model T-136-21
Speeds:
High:
M33 1:1
M33A1 1:1
Low:
M33 1.98:1
M33A1 1.89:1
Live axle type double-reduction, full-floating
Body cargo space 456 cu ft
Winch capacity 10,000 lb
Chassis M46

Ground clearance 12.5 in.
Angle of approach:
W/o wn 38 deg
W/o wn 46 deg
Angle of departure 44 deg

PERFORMANCE

Computed grade ability in lowest gear:
M33 44 percent
M33A1 60 percent
Turning radius:
W/o wn 35 1/2 ft
W/o wn 34 1/2 ft
Fuel consumption:
M33 9.3 mpg (gasoline); 9.7 mpg (CIE fuel);
M33A1 11.3 mpg (diesel)

Cruise range:
M33 350 miles
M33A1 500 miles
Allowable speed (governed):
M33 50 mph
M33A1 54 mph
Fording depth:
W/fording kit 72 in.
W/o fording kit 36 in.
Towed load:
Highway 10,000 lb
Cross-country 6,000 lb

EQUIPMENT

Basic Issue Items:
M33—See TM 9-4022, C3
M33A1—See TM 9-2320-235-10

INSTRUCTIONAL MATERIAL
STORAGE AND SHIPMENT DATA

Within Continental United States

Shipped 1 truck, uncrated W/Winch W/o Winch
M33:
Length 274 in.
Width 74 in.
Height 112 in.
Volume 1,718 cu ft
Gross weight 24,064 lb
Ship tons 42.97
Area 259 sq ft
M33A1:
Length
Width
Height
Volume
Gross weight
Ship tons

Outside Continental United States

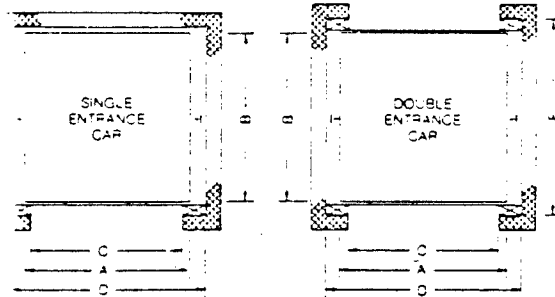
Shipped 1 truck, uncrated W/Winch W/o Winch
M33:
Height
Volume
Gross weight
Ship tons
Area
M33A1:
Length
Width
Height
Volume
Gross weight
Ship tons

References: M33—SNL G-42, TM 9-2300-223-10P, TM 9-2320-209-15P, TM 9-2320-209-10P, TM 9-4022, TM 9-1828A, TM 9-4601, TM 9-4611, TM 9-4423, TM 9-4647, TM 9-7910-210-34, LO 9-2320-209-12.
M33A1—TM 9-2320-235-10, TM 9-2320-235-11, TM 9-2320-235-10P, TM 9-2320-235-11A, TM 9-1370-235-11P, TM 9-2300-223-20P, LO 9-2320-235-10.

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SHEET NO.
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PROJECT NO. 02-7092 SPONSOR CERL
SUBJECT Down Flight Elevators



RECOMMENDED SIZES AND CAPACITIES					
DIMEN- SIONS	CAPACITY IN POUNDS				
	2500	4000 5000 6000	8000	10,000	
A	5'-4"	8'-4"	8'-4"	8'-4"	
B	7'-0"	10'-0"	12'-0"	14'-0"	
C	5'-0"	8'-0"	8'-0"	8'-0"	
D*	7'-2"	10'-2"	10'-2"	10'-2"	
E*	7'-8"	10'-8"	12'-3"	14'-8"	
F**	7'-11"	10'-11"	12'-11"	14'-11"	

DIMENSIONS SHOWN ARE FOR POWER OPERATED DOORS OF THE REGULAR TYPE WITH 8'-0" CLEAR OPENING HEIGHT. CHANGES REQUIRED FOR OTHER THAN ABOVE:

*SUBTRACT 5" IF MANUAL DOORS ARE USED

**ADD 1" IF PASS TYPE DOORS ARE USED

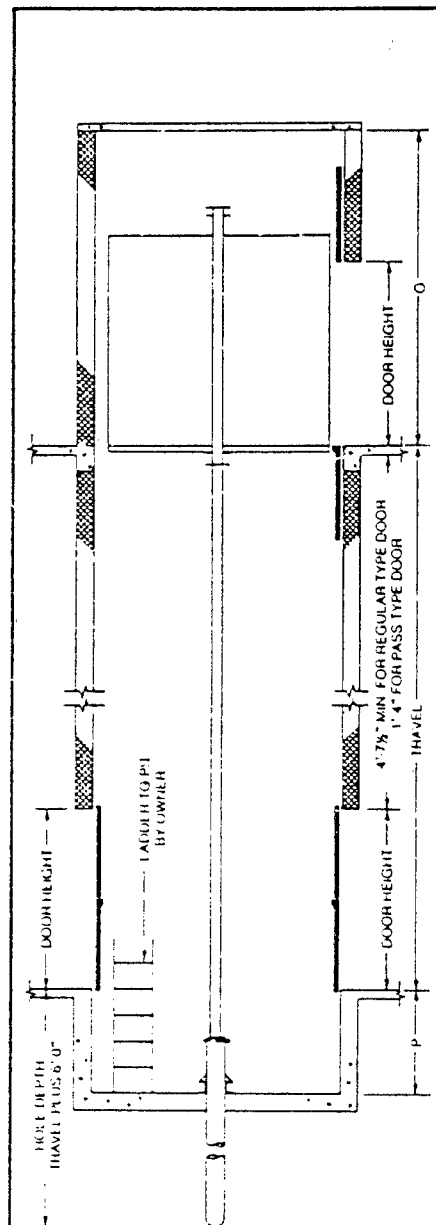
***ADD 3" IF PASS TYPE DOORS ARE USED

MINIMUM PIT AND OVERHEAD DIMENSIONS						
DIMENSIONS	SPEED (FEET PER MINUTE)					
	25	50	75	100	150	200
P	4'-6"	4'-6"	4'-6"	4'-6"	4'-6"	4'-6"
O***	15'-0"	15'-0"	15'-0"	15'-0"	15'-2"	15'-0"

DIMENSIONS SHOWN ARE FOR POWER OPERATED DOORS OF THE REGULAR TYPE WITH 8'-0" CLEAR OPENING HEIGHT. CHANGES REQUIRED FOR OTHER THAN ABOVE:

***SUBTRACT 1" IF 7'-0" CLEAR OPENING HEIGHT DOORS ARE USED

POWER UNIT (MACHINE) LOCATION: The most desirable machine room location is on the lowest floor adjacent to the elevator shaft. It may, however, be located remote from shaft if necessary. Typical size for one-car installation: 9'-3" x 7'-2" x 8'-0" high. Larger area is required when two or more power units are used or for two elevators with common machine room, etc. Closure to meet local code requirements must be provided. A sound-isolated machine room is recommended for quietest operation. Adequate heating and ventilation of machine spaces must be provided.





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PROJECT NO. 02-7092 SPONSOR CERL
SUBJECT TRUCK REQUIREMENTS
BY muft DATE 6-23-82 CHECKED BY: _____ DATE CHECKED 19

5-TON CARGO TRUCKS (6x6)

Model	Front Winch	Drop Side	Overall Length (in)	Overall Width (in)	Payload		Front Winch Capacity	Bed
					on heavy	off-highway		
M813	✓	-	317	95.6	10T	5T	10T	~35' x 5' 128 ft ²
M813A1	✓	✓	317	95.6	10T	5T	10T	~13' x 8' 128 ft ²
M814	✓	-	393	97.0	10T	5T	10T	~17' x 8' 168 ft ²

2½-TON CARGO TRUCK (6x6)

M35A1	✓	-	276	96.0	5T	2½T	5T	~12' x 8'
-------	---	---	-----	------	----	-----	----	-----------

Longest w/pn.
14' x 45' x 45'

Alpha Type 41 15 - 5-ton trucks (1500)
2 1 - 5 ton w/stack 12+3
2 - 2½ ton (8.25)
3 6 - 5 ton 200
6 - 2½ ton
1 1 - 5 ton tks
1 - 2½ w/stack (12 bottom, 3 top)

23 - big 5-ton
16 - reg 5-ton
7 - 2½ ton

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SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: EQUIPMENT REQUIREMENTS
BY: mt DATE: 9-8 19 82 CHECKED BY: _____ DATE CHECKED: _____ 19 _____

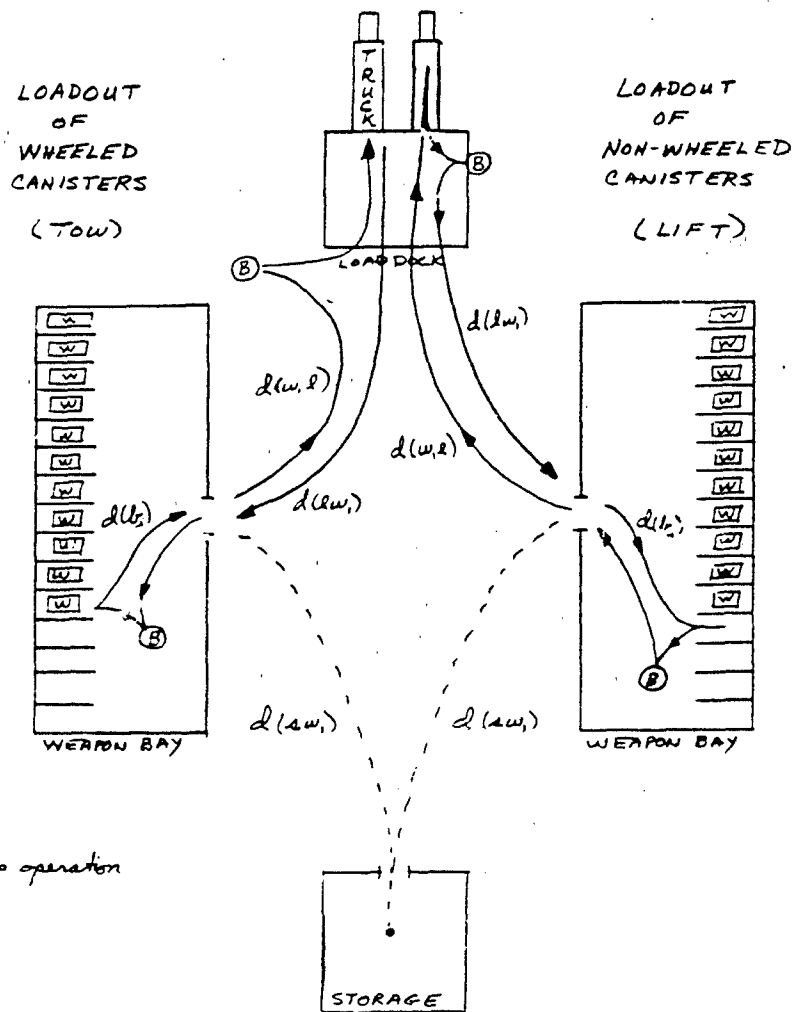
For loadout: 4 forklifts
2 trucks at each dock.
11 people: 4 Forklift drivers
1 person in each of 4 bays.
2 checkout people
1 overseer
- 2 spare forklifts
2 20ft roller tracks
1 180° rotating pad (15' diameter)

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COMPUTATION SHEET

SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: LOADOUT
BY: *mw* DATE: 7-12-82 CHECKED BY: DATE CHECKED: 19



ⓑ Backup operation

Tow wheeled canister
to dock; back it
onto the truck

8-11

Forklift the canister
to the dock; drive
onto the truck and
set in place

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SHEET NO.
17 OF 31

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: TIMING FOR LOADING
BY: mlb DATE: 7-12 19 82 CHECKED BY: DATE CHECKED: 19

ABOVE GROUND FACILITY USING FORKLIFTS

Time (sec)	Event
300	Loadmaster instructions (in control room)
120	Mount and start forklift (in storage area)
(120)	Open blast doors on weapons bay
*	Drive from storage to weapons bay - d(sw)
*	Drive from bay door to first cubicle - d(b)
20	Position forklift with tires under canister
75	Tie forklift to canister; raise canister; backup
*	Drive from first cubicle to bay door - d(b)
*	Drive from weapons bay to loading dock - d(wl)
30	Logout weapon
120	Drive weapon onto truck; position it; set it down; backup
*	Drive from loading dock to weapons bay - d(wl)
*	Drive from bay door to second cubicle - d(b)
20	Position forklift with tires under canister
75	Tie forklift to canister; raise canister; backup
*	Drive from second cubicle to bay door - d(b)
*	Drive from weapons bay to loading dock - d(wl)
30	Logout weapon
120	Drive weapon onto truck; position it; set it down; backup
:	Repeat series of steps for cubicles 3 through 15
:	

* Time computed based on distances travelled and assumed forklift rate of travel.

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SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: TIMELINE FOR LOADOUT
BY: HWK DATE: 7-12 19 82 CHECKED BY: DATE CHECKED: 19

UNDERGROUND FACILITY USING FORKLIFTS

Time (min)	Event
300	Loadmaster instructions (in control room)
120	Mount and start forklifts (in storage area)
(120)	Open blast doors on weapons bay
*	Drive from storage to weapons bay - d(w)
*	Drive from bay door to first cubicle - d(b ₁)
20	Position forklift with tires under canister
75	Tie canister to forklift; raise canister; backup
*	Drive from first cubicle to bay door - d(b ₁)
*	Drive from weapons bay to platform lift - d(wl)
30	Logout weapon
30	Drive onto lift
30	Ride lift to surface (30 ft @ 1 fpm)
120	Drive weapon onto truck; position it; set it down; backup
15	Drive onto platform lift
30	Ride lift to bottom (30 ft @ 1 fpm)
*	Drive from lift to weapons bay - d(wl)
*	Drive from bay door to second cubicle - d(b ₂)
20	Position forklift with tires under canister
75	Tie canister to forklift; raise canister; backup
*	Drive from second cubicle to bay door - d(b ₂)
*	Drive from weapons bay to platform lift - d(wl)
30	Logout weapon
30	Drive onto lift
30	Ride lift to surface (30 ft @ 1 fpm)
120	Drive weapon onto truck; position it; set it down; backup
...	Repeat series of steps for cubicles 3 through 15.

* Time computed based on distance travelled and assumed forklift rate of travel.



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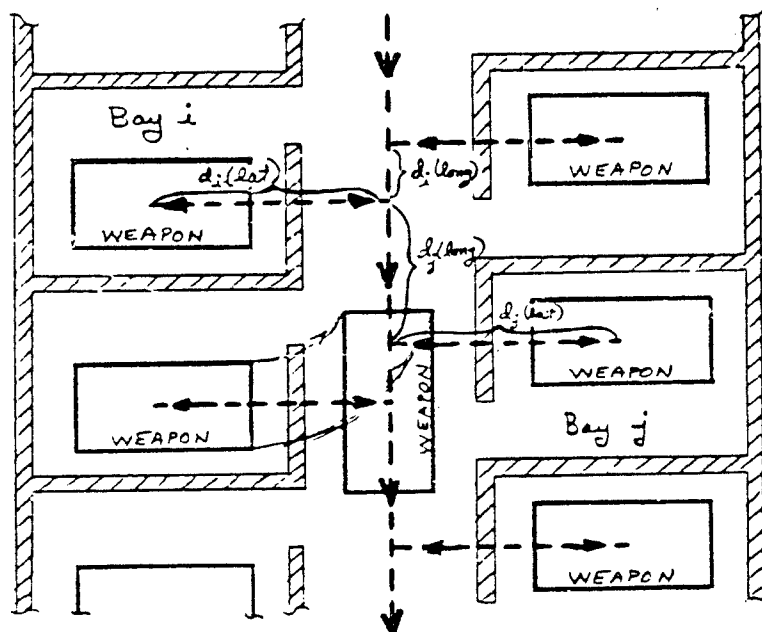
SHEET NO.

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PROJECT NO. 02-7092 SPONSOR CERL

SUBJECT CRANE WEASE FOR LOADOUT

BY mtb DATE 7-14 1982 CHECKED BY: DATE CHECKED: 19



* $d_1(\text{long})$ rate 1.5 fpe

* $d_1(\text{lat})$ rate 1.5 fpe

30 sec Fine-tune positioning and hookup weapon

25 sec Lift weapon to 6 ft height (rate 0.25 fpe)

* $d_2(\text{lat})$

30 sec Position weapon longitudinally and lower to floor

15 sec Unhook weapon

— Forklift moves weapon from this point

* Time computed based on distance travelled and on assumed crane travel rates.

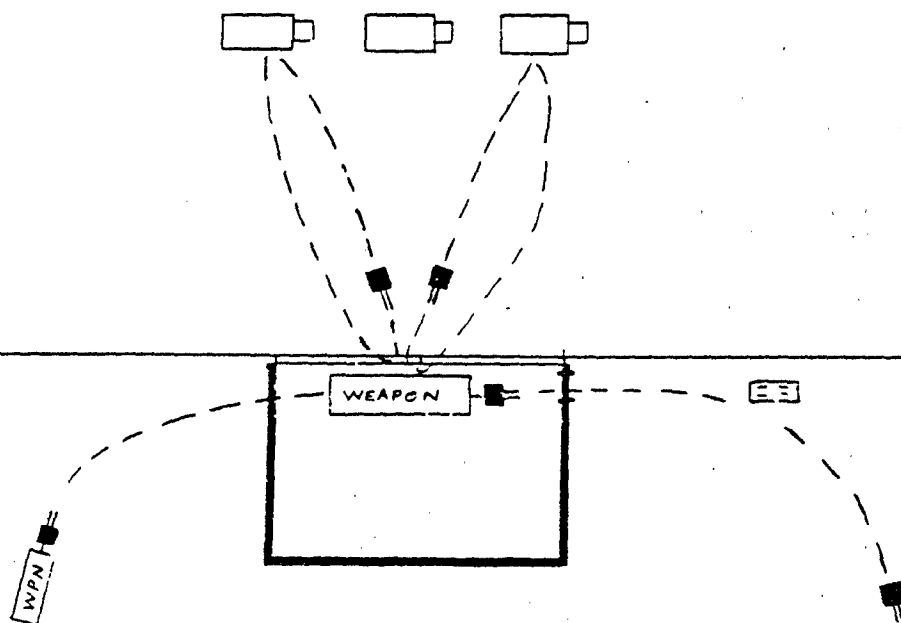


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SHEET NO.
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PROJECT NO. 02-7092 SPONSOR CERL
SUBJECT LOADOUT ON DOLLIES
BY WPN DATE 7-8-1982 CHECKED BY _____ DATE CHECKED: _____ 19__

LOADOUT OF WEAPONS ON DOLLIES



Must be able to receive/ship weapons following accident
(1) at loading dock
(2) at maintenance bay

Dropside trucks

2 Load docks

Forklifts & Dollies

Store on Dollies

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SHEET NO.
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PROJECT NO.: 02-7092

SPONSOR: CERL

SUBJECT: LOADOUT ON DOLLIES

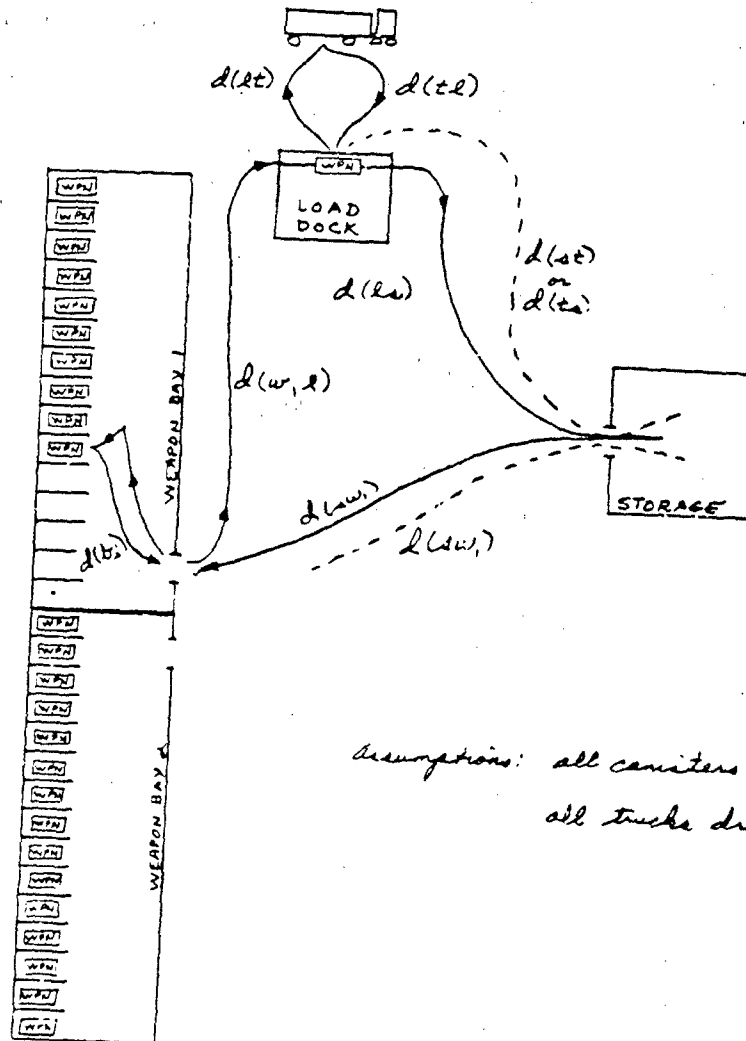
BY: mwb

DATE: 7-8 1982

CHECKED BY:

DATE CHECKED: 19

LOADOUT SCENARIO - WEAPONS ON DOLLIES



assumptions: all canisters on dollies
all trucks dropside

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SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: LOADOUT ON DOLLIES
BY: mtb DATE: 7-12-82 CHECKED BY: _____ DATE CHECKED: _____ 19

LOADOUT TIMELINE - WEAPONS ON DOLLIES

seconds	
300	Loadmaster instructions (in control room)
120	Mount and start forklifts (in storage area)
(120)	Open blast doors on weapons Bay
	d(lw, ₁)
	d(l, ₁)
10	position forklift for towing
30	dismount; hookup weapon; mount
	d(l, ₁)
	d(lw, ₁)
30	logout weapon
120	Back weapon onto truck
20	dismount; unhook weapon; mount
	d(lw, ₁)
	d(l, ₁)
10	position forklift for towing
* 30	dismount; hookup weapon; mount
	d(l, ₁)
	d(lw, ₁)
30	logout weapon
120	Back weapon onto truck
20	dismount; unhook weapon; mount
	∴ complete all 15 bags in W ₁
	d(lw, ₂)
	d(l, ₂)
20	position forklift with tines under canister.
75	tie canister to forklift; pick up canister; hookup.
	d(l, ₂)
	d(lw, ₂)
30	logout weapon
60	Drive weapon onto truck; position it; set it down; hookup
	∴ complete all 15 bags in W ₂

W₃ and W₄ are emptied at the same time as the above, using the second loading dock.

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SHEET NO.
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PROJECT NO.: C2-7092 SPONSOR: CERL
SUBJECT: LOADOUT TIMELINE CALCULATIONS
BY: mtt DATE: 9-8 19 82 CHECKED BY: _____ DATE CHECKED: _____ 19

LOADOUT CALCULATIONS

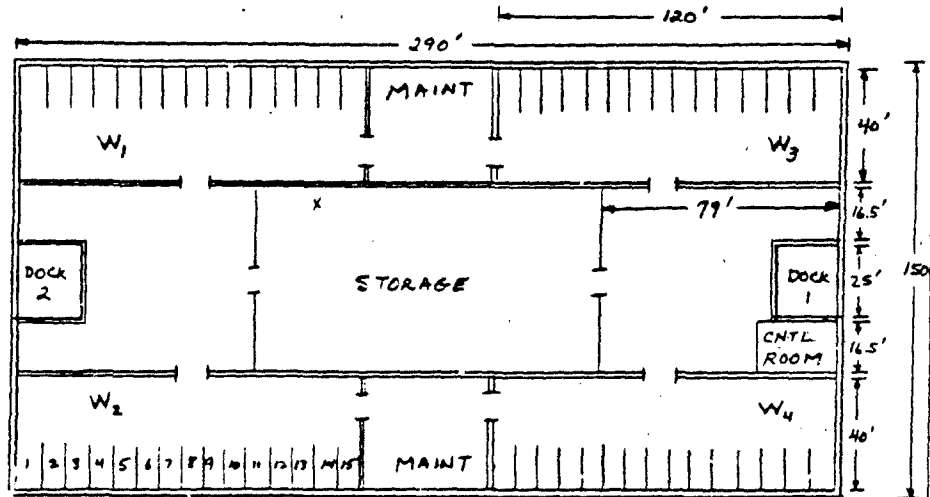
EXAMPLE 1	RECTANGULAR FACILITY	LONG BAYS (DOORS IN MIDDLE)
EXAMPLE 2	SQUARE FACILITY	LONG BAYS (DOORS ON ENDS)
EXAMPLE 3	"L"-SHAPED FACILITY	LONG BAYS
EXAMPLE 4	SEMICIRCULAR FACILITY	LONG BAYS
EXAMPLE 5	SQUARE FACILITY	PIT BAYS
EXAMPLE 6	UNDERGROUND FACILITY (LIFTS)	LONG BAYS

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SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: EXAMPLE LOADOUT 1
BY: [signature] DATE: 7-13-1982 CHECKED BY: DATE CHECKED: 19

Scale $\rightarrow 10'$



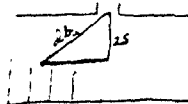
Dimensions in feet

$dl_1 = 61$	$d(w_1, l) = 70$	$d(lw_1) = 70$
$dl_2 = 54$	$d(w_2, l) = 70$	$d(lw_2) = 70$
$dl_3 = 47$	$d(w_3, l) = 70$	$d(lw_3) = 70$
$dl_4 = 41$	$d(w_4, l) = 70$	$d(lw_4) = 70$
$dl_5 = 35$		
$dl_6 = 30$	$d(lw_1) = 70$	
$dl_7 = 26$	$d(lw_2) = 70$	
$dl_8 = 25$	$d(lw_3) = 70$	
$dl_9 = 25$	$d(lw_4) = 70$	
$dl_{10} = 30$		
$dl_{11} = 35$		
$dl_{12} = 41$		
$dl_{13} = 47$		
$dl_{14} = 54$		
$dl_{15} = 61$		

Assume: W_1 and W_3 have wheeled containers
 W_2 and W_4 have non-wheeled containers
 W_1 and W_2 use dock 2
 W_3 and W_4 use dock 1

No loadout time penalty for any days

Forklift travel rate: loaded = 2 mph = 3 fpm
unloaded = 3 mph = 4.5 fpm



* 57.5-35-11.1 for 15-35
6.75 for 75-35

SHEET NO. 19 OF 31

SPONSOR: CEKL

BY: msk. DATE: 7-13-92

CHECKED BY:

DATE CHECKED: _____ 19____

[illegible]

$\Sigma 4073$

$\Sigma 3698$

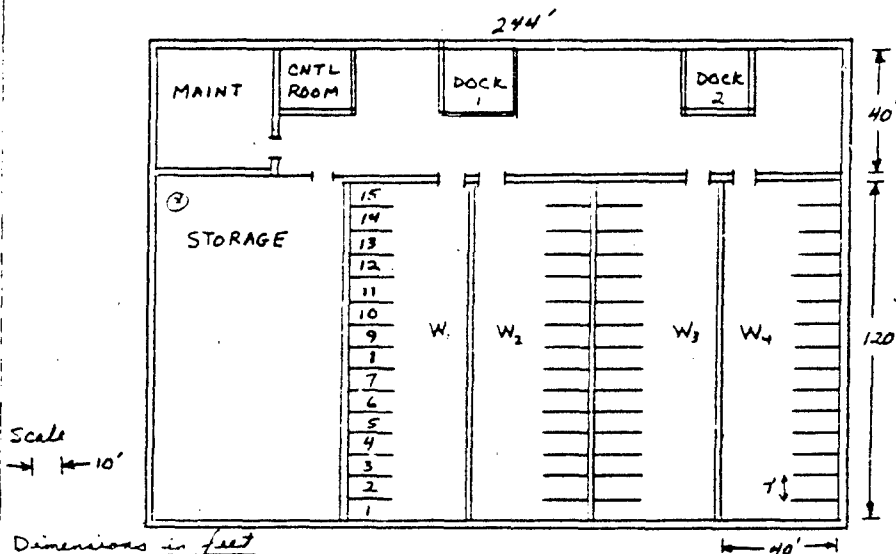
Σ 7771
8-20 *Piedmont* 420

$$\Sigma = 8191 \text{ sec} = 2.28 \text{ hrs}$$

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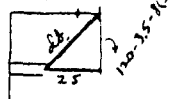
SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: EXAMPLE LOADOUT 2
BY: MCB DATE: 7-12-82 CHECKED BY: DATE CHECKED: 19



Dimensions in feet

$db_{15} = 25$	$d(w_1, l) = 40$	$d(lw_1) = 110$
$db_{14} = 28$	$d(w_2, l) = 40$	$d(lw_2) = 120$
$db_{13} = 32$	$d(w_3, l) = 40$	$d(lw_3) = 190$
$db_{12} = 38$	$d(w_4, l) = 40$	$d(lw_4) = 210$
$db_{11} = 44$		
$db_{10} = 51$	$d(lw_1) = 40$	
$db_9 = 58$	$d(lw_2) = 40$	
$db_8 = 65$	$d(lw_3) = 40$	
$db_7 = 73$	$d(lw_4) = 40$	
$db_6 = 80$		
$db_5 = 88$		
$db_4 = 96$		
$db_3 = 104$		
$db_2 = 112$		
$db_1 = 119$		



Removal starts with bay 1 and progresses to bay 15
Assume: W_1 and W_3 have wheeled canisters
 W_2 and W_4 have non-wheeled canisters
 W_1 and W_2 use dock 1
 W_3 and W_4 use dock 2

Time penalty for removal from b_{10} and b_{15} in W_1 and W_3 is 90 seconds

Forklift travel rate: loaded = 2 mph = 3 fpe
unloaded = 3 mph = 4.5 fpe

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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: EXAMPLE 2
BY: gmk DATE: 7-13 19 82 CHECKED BY: _____ DATE CHECKED: _____ 19 _____

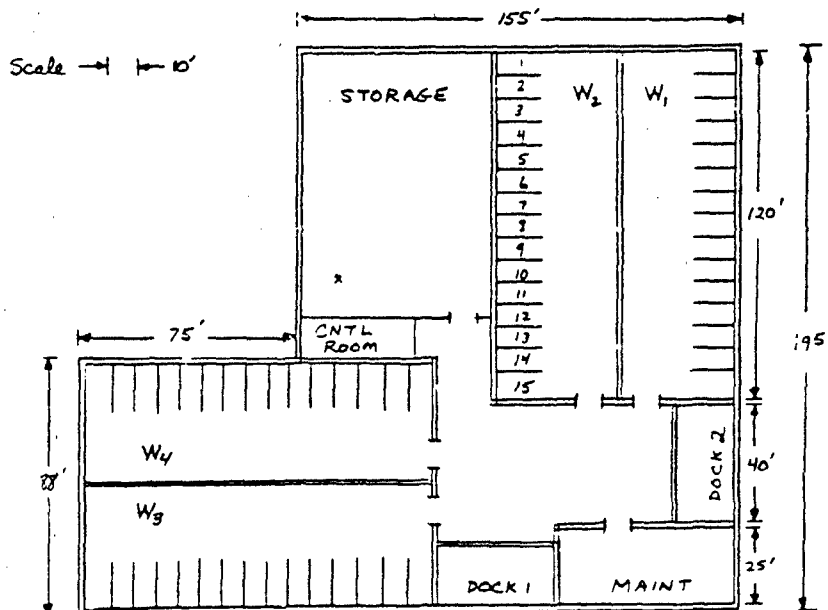
[illegible]

$$\left. \begin{array}{l} \Sigma 7913 \\ 8-22 \text{ leadtime } 420 \end{array} \right\} \Sigma = 8330 \text{ sec} = 2.31 \text{ hrs.}$$

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COMPUTATION SHEET

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PROJECT NO.: 12-7092 SPONSOR: CERL
SUBJECT: EXAMPLE LOADOUT 3
BY: mtb DATE: 7-13-82 CHECKED BY: _____ DATE CHECKED: _____ 19__



Dimensions in feet

$dt_1 = 119$	$d(w_1, l) = 45$	$d(lw_1) = 155$
$dt_2 = 112$	$d(w_2, l) = 65$	$d(lw_2) = 145$
$dt_3 = 104$	$d(w_3, l) = 45$	$d(lw_3) = 135$
$dt_4 = 76$	$d(w_4, l) = 65$	$d(lw_4) = 115$
$dt_5 = 88$		
$dt_6 = 80$	$d(lw_1) = 45$	
$dt_7 = 73$	$d(lw_2) = 65$	
$dt_8 = 65$	$d(lw_3) = 45$	
$dt_9 = 58$	$d(lw_4) = 65$	
$dt_{10} = 51$		
$dt_{11} = 44$		
$dt_{12} = 38$		
$dt_{13} = 32$		
$dt_{14} = 28$		
$dt_{15} = 25$		

Assume: W_1 and W_3 have wheeled canisters
 W_2 and W_4 have non-wheeled canisters
 W_1 and W_2 use dock 2
 W_3 and W_4 use dock 1

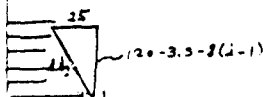
Time penalty for removal from dt_{14} and dt_{15} in W_1 and W_3 is 90 seconds.

Forklift travel rate:

loaded = 2 mph = 3 fpm

unloaded = 3 mph = 4.5 fpm

Removal starts with bay 1 and progresses to bay 15.



SHEET NO. 23 OF 31

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: EXAMPLE 3
BY: nmh DATE: 7-12 19 82 CHECKED BY: _____ DATE CHECKED: _____

[illegible]

$\Sigma 4112$

Σ 3879

8-24

Σ 799.

leadtime

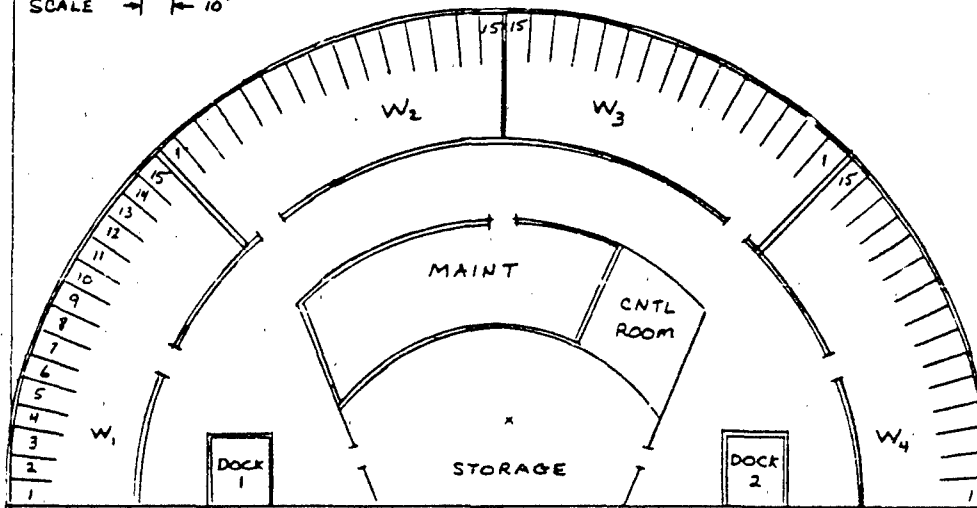
$$\Sigma = 8411 \text{ sec} = 2.34 \text{ hrs}$$

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SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: EXAMPLE LOADOUT 4
BY: mlt DATE: 7-13-1982 CHECKED BY: DATE CHECKED: 19

SCALE 1" = 10'



For W_1, W_4

$db_1 = 61$
 $db_2 = 54$
 $db_3 = 47$
 $db_4 = 41$
 $db_5 = 35$
 $db_6 = 30$
 $db_7 = 26$
 $db_8 = 25$
 $db_9 = 26$
 $db_{10} = 30$
 $db_{11} = 35$
 $db_{12} = 41$
 $db_{13} = 47$
 $db_{14} = 54$
 $db_{15} = 61$

(ft)
height
altitude

For W_2, W_3

$= 30$
 $= 30$
 $= 30$
 $= 33$
 $= 38$
 $= 42$
 $= 49$
 $= 57$
 $= 61$
 $= 67$
 $= 73$
 $= 80$
 $= 86$
 $= 93$
 $= 99$

(measured)

$d(w_1, 2) = 60 = d(2w_1)$
 $d(w_2, 2) = 100 = d(2w_2)$
 $d(w_3, 2) = 100 = d(2w_3)$
 $d(w_4, 2) = 60 = d(2w_4)$

$d(1w_1) = 140$
 $d(1w_4) = 140$

Assume: W_1 and W_4 have wheeled canisters
 W_2 and W_3 have non-wheeled canisters
 W_1 and W_2 use dock 1
 W_3 and W_4 use dock 2
No loadout time penalty for any bays
Forklift travel rates:
loaded = 2 mph = 3 fpm
unloaded = 3 mph = 4.5 fpm
Removal starts with bay 1 and
progresses to bay 15.

SHEET NO. :
25 OF 31

SPONSOR: CERL

SUBJECT: EXAMPLE 4

BY: myb

DATE: 7-14-79 82

CHECKED BY:

DATE CHECKED: _____ 19____

[illegible]

Σ 4509

Σ 4.90

5-2:

Σ 3079
line 420

$$T = 851^{\circ} \text{ sec} = 2.37 \text{ hrs}$$

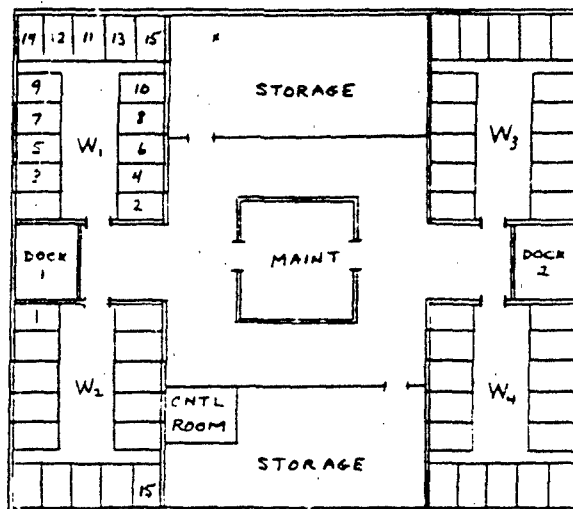


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PROJECT NO. 02-7092 SPONSOR: CERL
SUBJECT: EXAMPLE LOADOUT 5
BY: nwt DATE: 7-14-82 CHECKED BY: DATE CHECKED: 19

Scale → 1" = 10'



$db_1 = 6$
 $db_2 = 6$
 $db_3 = 16$
 $db_4 = 16$
 $db_5 = 26$
 $db_6 = 26$
 $db_7 = 36$
 $db_8 = 36$
 $db_9 = 46$
 $db_{10} = 46$
 $db_{11} = 46$
 $db_{12} = 46$
 $db_{13} = 46$
 $db_{14} = 46$
 $db_{15} = 46$

$lw_1 = w_1 l = 35$
 $lw_2 = w_2 l = 35$
 $lw_3 = lw_4 = 110$

Assumptions: W_1 and W_3 have wheeled canisters
 W_2 and W_4 have non-wheeled canisters
 W_1 and W_2 use dock 1
 W_3 and W_4 use dock 2
 No loadout time penalties for any bays

Forklift travel rates: loaded = 2 mph = 3 fpm
 unloaded = 3 mph = 4.5 fpm
 Crane travel rates: longitudinal = 1.5 fpm
 lateral = 1.5 fpm
 vertical = 0.25 fpm

Removal starts with bay 1 and progresses to bay 15

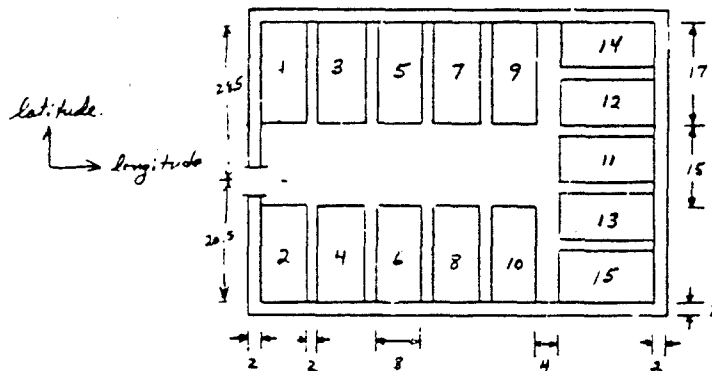


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PROJECT NO. 02-7092 SPONSOR CERL
SUBJECT EXAMPLE 5
BY mwb DATE 7-14-1982 CHECKED BY: DATE CHECKED 19

WEAPON BAY
PIT DESIGN



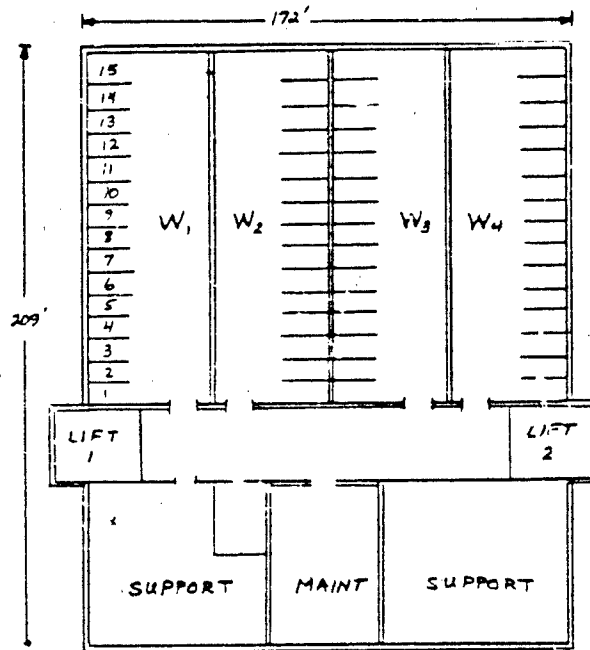
$d(\text{lat})$ bay 1 = 20
2 = 12
3 = 20
4 = 12
5 = 20
6 = 12
7 = 20
8 = 12
9 = 20
10 = 12
11 = 4
12 = 14
13 = 6
14 = 24
15 = 16

$d(\text{long})$ bay 1 = 4
2 = 0
3 = 10
4 = 0
5 = 10
6 = 0
7 = 10
8 = 0
9 = 10
10 = 0
11 = 17
12 = 17
13 = 17
14 = 17
15 = 17

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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: EXAMPLE LOADOUT 6
BY: [signature] DATE: 9-8-1982 CHECKED BY: DATE CHECKED: 19



Dimensions in feet:

$db_1 = 25$
 $db_2 = 28$
 $db_3 = 32$
 $db_4 = 38$
 $db_5 = 44$
 $db_6 = 51$
 $db_7 = 58$
 $db_8 = 65$
 $db_9 = 73$
 $db_{10} = 80$
 $db_{11} = 88$
 $db_{12} = 96$
 $db_{13} = 104$
 $db_{14} = 112$
 $db_{15} = 119$

$d(w, l) = 25' = d(lw_1)$
 $d(lw_1, l) = 45' = d(lw_2)$
 $d(lw_2, l) = 60'$

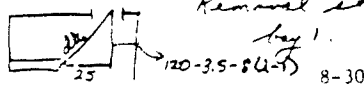
Assume: W_1 and W_2 use lift 1
 W_3 and W_4 use lift 2

Time penalty for removal from bays 14, 15 is 30 seconds.

Forklift travel rate - loaded = 2 mph = 3 fps
empty = 3 mph = 4.5 fps.

Lift rate = 1 fps for 30 ft.

Removal starts with bay 15 and progresses to



SHEET NO.
30 OF 31

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: EXAMPLE 6
BY: wf DATE: 9-8-1982 CHECKED BY: _____ DATE CHECKED: _____ 19____

[illegible]
$$\frac{\sum}{8-31} \left. \begin{array}{l} \sum 1284 \\ \text{bedtime} \quad 420 \end{array} \right\} \sum = 1284 \text{ sec} = 3.53 \text{ hr}$$

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COMPUTATION SHEET

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PROJECT NO.: C2-7092 SPONSOR: CERL
SUBJECT: 6 CONCEPTS - LOADOUT TIME
BY: mw DATE: 9-15 1982 CHECKED BY: _____ DATE CHECKED: _____ 19____

LOADOUT TIMES FOR 6 CONCEPTS SELECTED

ASSUME EACH BAY LOADING OUT SIMULTANEOUSLY
SO LOADOUT TIME = TIME FOR SLOWEST BAY.

CONCEPT 1 - Use times from Example 1, bay no. 2.

$$\begin{aligned} \text{Time} &= 3698 \text{ sec} \\ &+ 420 \text{ sec lead time} \\ &\underline{4118} \\ &\sim 1.14 \text{ hrs} \end{aligned}$$

CONCEPT 2 - Use times from Example 1, bay no. 1
CONCEPT 5

$$\begin{aligned} \text{Time} &= 4073 \text{ sec} \\ &+ 420 \text{ lead time} \\ &\underline{4493 \text{ sec}} \\ &\sim 1.25 \text{ hrs.} \end{aligned}$$

CONCEPT 3 - Use times from Example 6, bay no. 1

$$\begin{aligned} \text{Time} &= 6037 \text{ sec} \\ &- 100 \text{ sec (shorter legs)} \\ &+ 420 \text{ sec (lead time)} \\ &\underline{6357 \text{ sec}} \\ &\sim 1.77 \text{ hrs.} \end{aligned}$$

CONCEPT 4 - Use times from Example 6, bay no. 2
CONCEPT 6

$$\begin{aligned} \text{Time} &= 6247 \text{ sec} \\ &+ 420 \text{ sec lead time} \\ &\underline{6667 \text{ sec}} \\ &\sim 1.85 \text{ hrs.} \end{aligned}$$

APPENDIX 9

SURVIVABILITY

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SHEET NO. _____
OF _____

PROJECT NO.: 02-7092-001 SPONSOR: CERL
SUBJECT: Free-Flow for 747 Crash
BY: Philip T. Hall DATE: 26 Sept 19 82 CHECKED BY: _____ DATE CHECKED: _____ 19 _____

CONTENTS -

<u>Topic</u>	<u>Analysis Method</u>	<u>Page</u>
Large Aircraft Crash	Replica Modeling Specific Momentum Extrapolation Soil Pressure Attenuation	9-2
General Purpose Bomb	Wall Damage - 500 pound Bomb Penetration	9-18
Large H.E. Charge	Pentex Manual	9-32

LARGE AIRCRAFT CRASH



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COMPUTATION SHEET

SHEET NO.
1 OF 7

PROJECT NO. C2-7092-001 SPONSOR C E R L
SUBJECT Force-Time for 747 Crash
BY W.E. Eaker DATE 2/24 19 92 CHECKED BY Philip T. Hall DATE CHECKED: 6/25 19 82

Analysts and designers of nuclear power plants have considered the effects of aircraft crash on those plants, and have developed ways of estimating the force-time histories for crashes of a spectrum of aircraft into rigid walls. These estimates are reported in a number of papers in the SMIRT (Structural Mechanics in Reactor Technology) Conferences.

One paper in particular is useful for our purpose of estimating dynamic loads for a 747 crash.¹ The authors compare the force-time histories for crash of a number of different aircraft, with several curves being given for the 707 aircraft, but none for the 747. Fig 1 is taken from Ref. 1, and shows four force-time crash curves for the 707 aircraft.

¹ G. Larsson & P. Lundsäker, "On the Response of a Reactor Building to Aircraft Crash", Transactions of the 4th International Conference on Structural Mechanics in Reactor Technology, 1977, San Francisco, CA, Paper J9/A.



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SHEET NO.
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PROJECT NO.: 02-7092-001 SPONSOR: C E R L
SUBJECT: Force-Time for 747 Crash
BY: W. E. Baker DATE: 6/29/82 CHECKED BY: JIN DATE CHECKED: 6/05/82

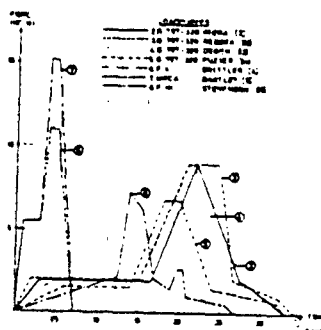


Fig. 1. Force-Time Curves for Aircraft Crashes (Ref. 1)

The 707 aircraft is very similar in configuration to the 747. Both have rather long cylindrical fuselages, swept low wings and horizontal stabilizers with the same sweep angles, swept vertical stabilizer, and four engines mounted on pylons below the wings. To predict the 747 force-time history in a crash, we assume that it is essentially a replica of the 707, to a larger scale, and use similitude relations to scale an appropriate curve in Fig. 1.

Some general characteristics of the two aircraft appear in Table 1, from Ref. 1.



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PROJECT NO. 22-7092-001 SPONSOR CERL
SUBJECT Force-Time for 747 Crash
BY W. E. Baker DATE 6/24/82 CHECKED BY: PTN DATE CHECKED: 6/25/82

Table 1. Aircraft Characteristics

Aircraft	Wing Span, m	Fus. Length, m	Fus. Dia., m	Take-off Wt, kg	Ldg. Wt., kg
B 707	44.4	49.9	3	150,000	112,000
B 747	59.6	68.4	6.5	372,000	286,000

In "replica" scale modeling of structural crushing under impact loads, Ref. 2 shows that the "model" and "prototype" structures should be exact geometric scale models of each other, and employ all of the same materials in the same locations. This is, of course, not exactly true when comparing a 707 to a 747. But, neglecting for the moment deviations from exact replication, we can use the inferences of the replica modeling law to scale the force-time history of the crashing 707 to a 747. If λ is the geometric scale factor obtained by scaling any comparable lengths for the two aircraft, then the law states that time scale by

2. W. E. Baker, P. S. Westine & F. T. Dodge, "Similarity Methods in Engineering Dynamics", Hayden Books, New Rochelle, N.Y., 1973.



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PROJECT NO.: 02-7092-001 SPONSOR: CERL
SUBJECT: Force-Time for 747 Crash
BY: Y.E. Baker DATE: 6/25/82 CHECKED BY: P.W. DATE CHECKED: 6/25/82

the same scale factor λ as do lengths, velocities are unchanged, weights or masses scale as λ^3 , and forces scale as λ^2 (See Ref. 2)

What should λ be, when scaling up from a 707 to a 747? From Table 1, we can get several values by making length ratios and by taking the cube roots of mass ratios. These factors are given in Table 2.

Table 2. Geometric Scale Factors, 747/707

Parameter	Symbol	Value
Wing span ratio	λ_s	1.34
Fuselage length ratio	λ_L	1.54
Fuselage diameter ratio	λ_D	1.70
(Take-off mass ratio) ^{1/3}	λ_{TO}	1.35
(Landing mass ratio) ^{1/3}	λ_{LG}	1.37

To scale the force-time history of a 707 to a 747, it seems appropriate to scale the time by λ_L , because the duration



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PROJECT NO. 02-7092-001

SPONSOR: CERL

SUBJECT: Force-Time for 747 Crash

BY: W.F. Baker

DATE 2/25/82

CHECKED BY

ATN

DATE CHECKED

8/25/82

of the crash force should be proportional to fuselage length. But, to scale the force amplitudes, a scale factor related to mass seems more appropriate. So, we assume that $\lambda_t = \lambda_a = 1.54$, and take an average between the two values based on mass scaling for $\lambda_F = \left(\frac{\lambda_{t0} + \lambda_{tG}}{2} \right)^2 = 1.85$.

Figure 1 shows four curves for force-time histories for a 707 crash. We scale the one labeled ③ as a conservative upper limit. Table 3 gives F, t coordinates from this curve, and the values scaled up to a 747, using the scale factors for force and time listed above.

Table 3
Force-Time Coordinates for 707 and 747 Aircraft

707		747	
$F, 10^7 N$	$t, s.$	$F, 10^7 N$	$t, s.$
0	0	0	0
1.9	0.018	3.5	0.028
1.9	0.153	3.5	0.236
3.3	0.208	18.3	0.320
3.0	0.245	17.1	0.377
1.9	0.265	11.1	0.408
1.0	0.318	1.9	0.490
0	0.340	0	0.524



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SHEET NO.

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PROJECT NO. 22-7002-001 SPONSOR CERL

SUBJECT Force-Time for 747 Crash

BY W.E. Baker DATE 2/25 19 82 CHECKED BY: ATN DATE CHECKED 6/35 19 82

The force-time history for the 747 from Table 3 is plotted in Fig 1. We can use this plot as a conservative (upper limit) forcing function for a 747 crash, because it is based on head-on crash into a rigid wall. Earth cover or oblique impact can only attenuate the loads.

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PROJECT NO 22-7292-001 SPONSOR CERL
SUBJECT Force-Time for 747 Crash
BY W.F. Baker DATE 6/25 19 92 CHECKED BY FIN DATE CHECKED 6/25 19 92

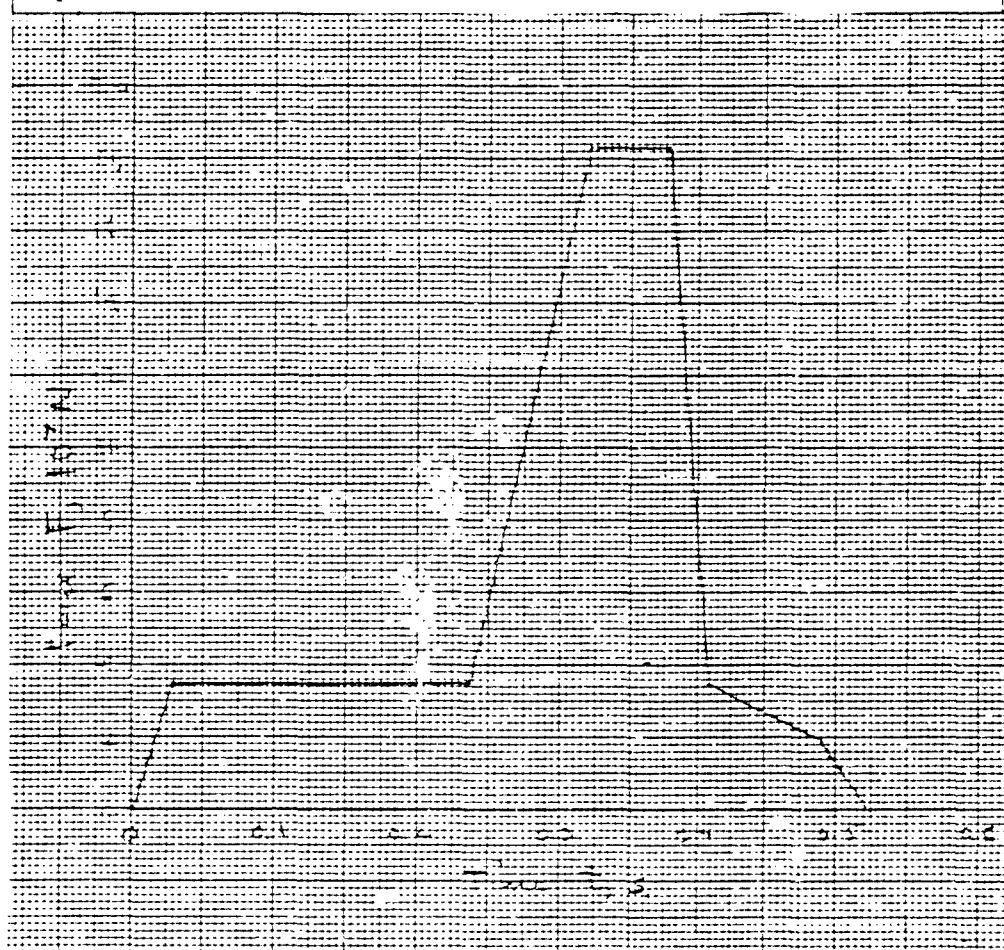


Fig 1.
Force-Time History for Crash of a
747 Aircraft

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SHEET NO.

1 OF 2

PROJECT NO. 22-7092-001 SPONSOR CEPL
SUBJECT Time-Time 787 crash
BY Paul T. H. L. DATE 6-1-82 CHECKED BY: _____ DATE CHECKED: _____

Dr. Baker used replica modeling techniques to develop a force-time history for a 787 crash into a rigid wall based upon similar histories for 707 aircraft. Time durations and peak forces can be determined in an alternate manner as a check. Important aircraft parameters are shown in Table 1.^①

TABLE 1. Aircraft Parameters

Aircraft Type	Wing Span (m)	Length (m)	Wing Area (m ²)	Max Take-off Weight (kg)
B-720	39.0	39.2	4.0	106,100
B707-320	44.4	44.4	3.8	151,300
B747-200	59.6	68.4	6.8	351,000
B747-200C	59.6	68.4	6.8	372,000
F4	11.7	17.8	2.0	20,865
MRCR	8.6-13.9	14.1	1.8	18,145
F111	16.3-21.9	20.8	1.9	41,500

Reference 2 details force-time curves for several aircraft with impact data given in Table 2.

1. Graham, J. The mobile aircraft 1978-79, 4th Edition, John Wiley & Sons.
2. Jackson, G. and Landberg, P., "On the Response of a Reinforced Building and its Surrounding Soil," Proceedings of the 4th International Conference on Structural Mechanics in Reactor Technology, Vol. 5 (6), Trondheim, Norway, 15-19 August 77, 9-11.

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PROJECT NO. 12-7692-501 SPONSOR: CERL
SUBJECT: 767-320 for 767 crash
BY: Chapman DATE: 6/28/92 CHECKED BY: _____ DATE CHECKED: 19

TABLE 2

IMPACT DATA
(from Reference 2)

Aircraft Type	Aircraft Mass 1,000 kg	Impact Speed (m/s)	Impact Area m ²	Peak Load 10 ³ N	Momentum M·v 10 ⁶ Ns	Impulse ∫ F(t)dt 10 ⁶ Ns
B720	72.4	103	18-36	7.1	7.46	7.37
B707-320	97.6	103	18-36	8.9	10.1	9.24
B707-320	--	103	28	8.8	--	11.4
B707-320	90.0	103	10-40	8.8	9.27	9.12
B707-320	100.0	83	28	6.8	8.81	7.26
F4	20.0	215	7	10.8	4.30	4.31
MRCA	25.0	215	7	15.1	5.38	5.35
F111	41.5	89	--	7.3	3.69	4.91

Peak force during the aircraft crash should vary according to the ^{specific} impact momentum (momentum per unit area, M·v/A). Using the impact data and aircraft characteristics giving specific momentum the peak force are shown in Figure 1. For 767 aircraft impacting at speeds between 100 and 120 m/sec, peak forces of 12. to 17. x 10³ N can be expected, depending upon aircraft weight. Location of the load will vary with the length and

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PROJECT NO.: 02-7092-001 SPONSOR: CERL
SUBJECT: Force-Time for 747 Crash
BY: Philip T. Reid DATE: 6/6/52 19 52 CHECKED BY: DATE CHECKED: 19

stiffness of the aircraft fuselage. Figure 2 plots fuselage length versus load duration for the aircraft data given. The upper curve is expected to be more representative of the 747 aircraft because the 707, similar in stiffness to the 747, is the major influence on the curve development. Using the upper curve, the load duration for the 747 aircraft is approximately .49 seconds. Thus, load-time histories for 747 aircraft crashing onto rigid walls can expect peak forces and load durations near the following:

Peak Force - $12-17 \times 10^3$ N.

Load Duration - $\approx .49$ seconds.

These values are very near those developed by Dr. Baker. I recommend using the force-time curve developed using replica modeling.

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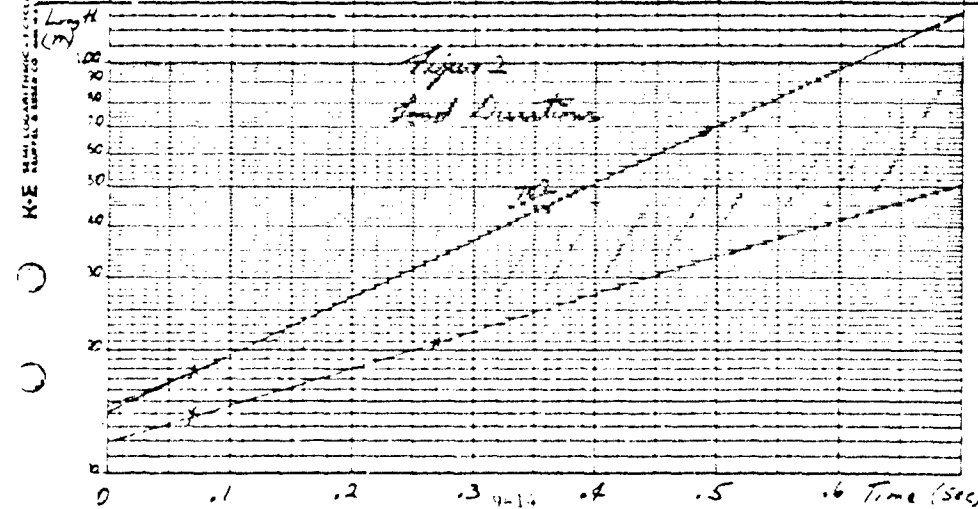
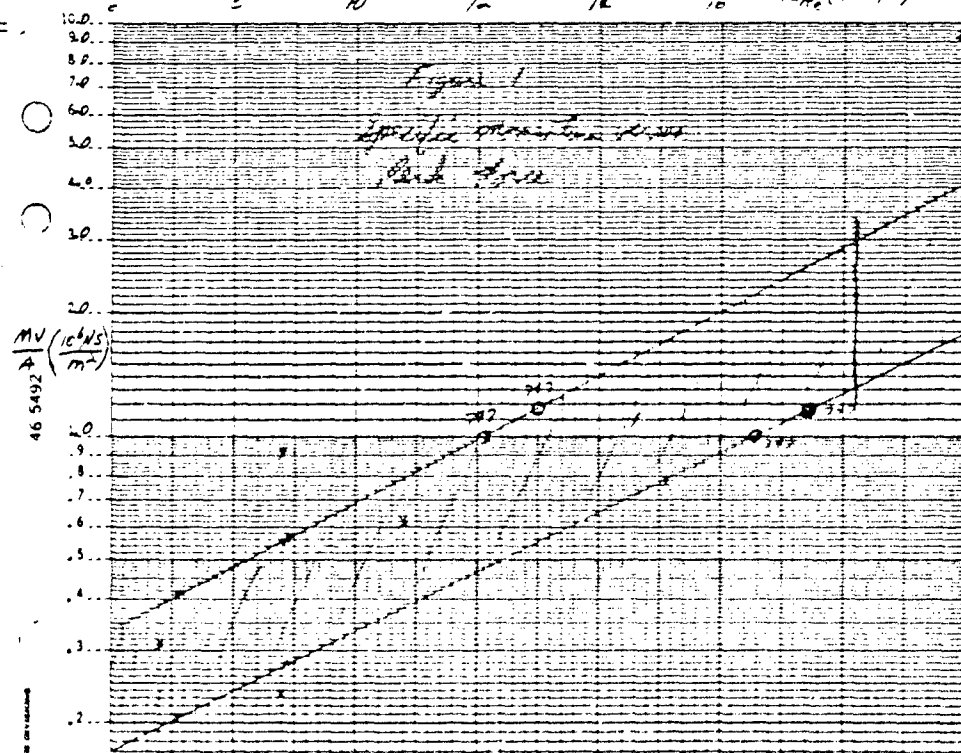
SHEET NO.
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PROJECT NO.: 02-7092

SPONSOR: CERL

SUBJECT:

1000 $F_{TTC}(10^7 N)$ 4/L

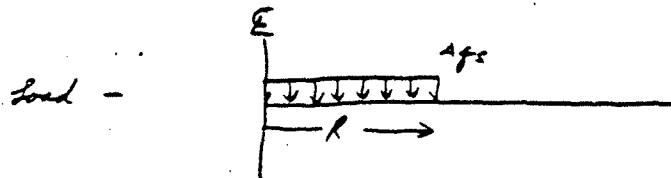


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SHEET NO.
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PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: Soil Pressure Attenuation
BY: William F. Hall DATE: 2 Aug 19 92 CHECKED BY: DATE CHECKED: 19

A Boussinesq solution for loading at the surface of an elastic half-space was used to calculate the attenuation of the aircraft impact stresses through soil. Although a static solution, for the relatively long load durations it should represent the stress attenuation due to geometric dispersion.



$$R \text{ for a } 747 = 34 \text{ m} = 11.12 \text{ ft.}$$

$$\text{Peak Force} = 18.3 \times 10^7 \text{ N} = 4.114 \times 10^7 \text{ lb}$$

$$Ags = \frac{F}{\pi R} = \frac{4.114 \times 10^7}{\pi (11.12)^2 / 4} = 735 \text{ psi}$$

The chart shown on page 2 was used to determine stress coefficients at 5 ft. increments vertically and horizontally. For example the stress coefficient at 25' depth, 10' horizontally, from the centerline is 0.18. Stresses calculated are shown on page 3 ($.18 \times 735 = 132$ for the example) and illustrated graphically on page 4.

$$\frac{15.2 \times 10^6}{4.114 \times 10^7} = 16$$

$$\pi (11.2)^2 144 = 135 \text{ psi}$$

Ch. 8 Stresses within a Soil Mass

2/4

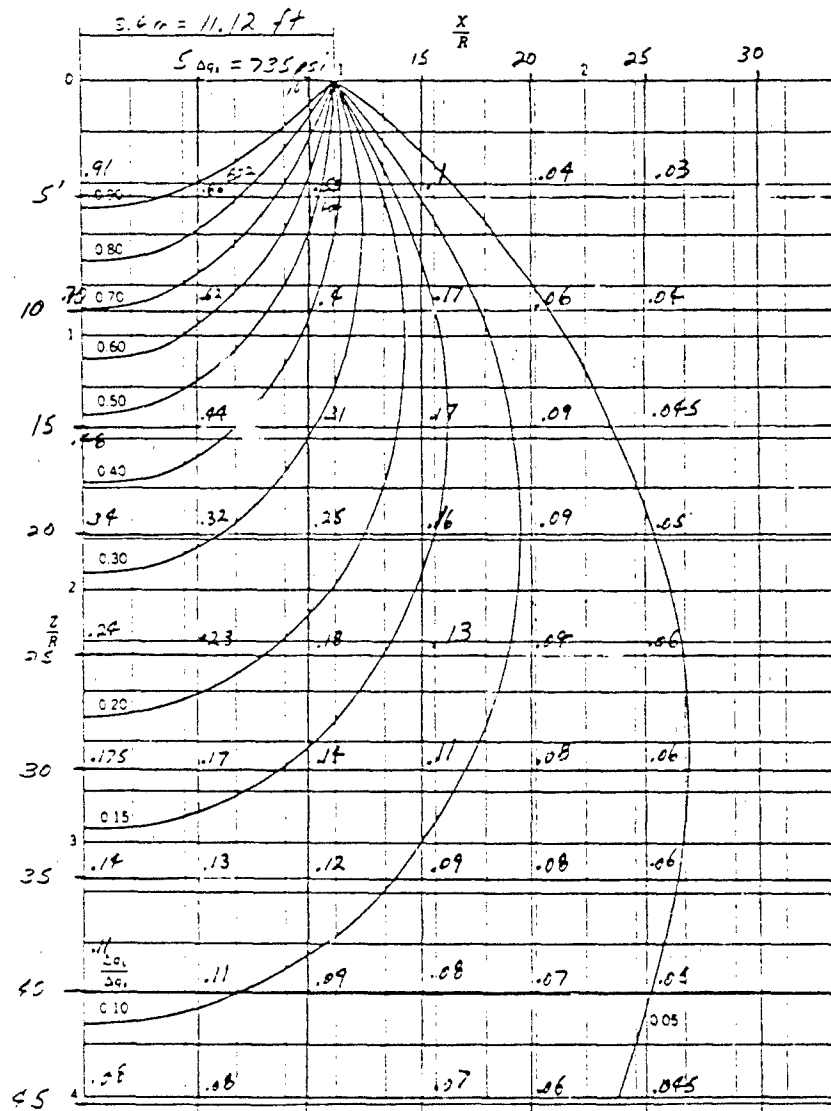


Fig 8.4 Vertical stresses induced by uniform load on circular area.

It is a very tedious matter to obtain the elastic solution for a given loading and set of boundary conditions. In this book, we are concerned not with how to obtain solutions but rather with how to use these solutions. This section presents several solutions in graphical form.

Uniform load over a circular area. Figures 8.4 and 8.5 give the stresses caused by a uniformly distributed normal stress Δq , acting over a circular area of radius R on the

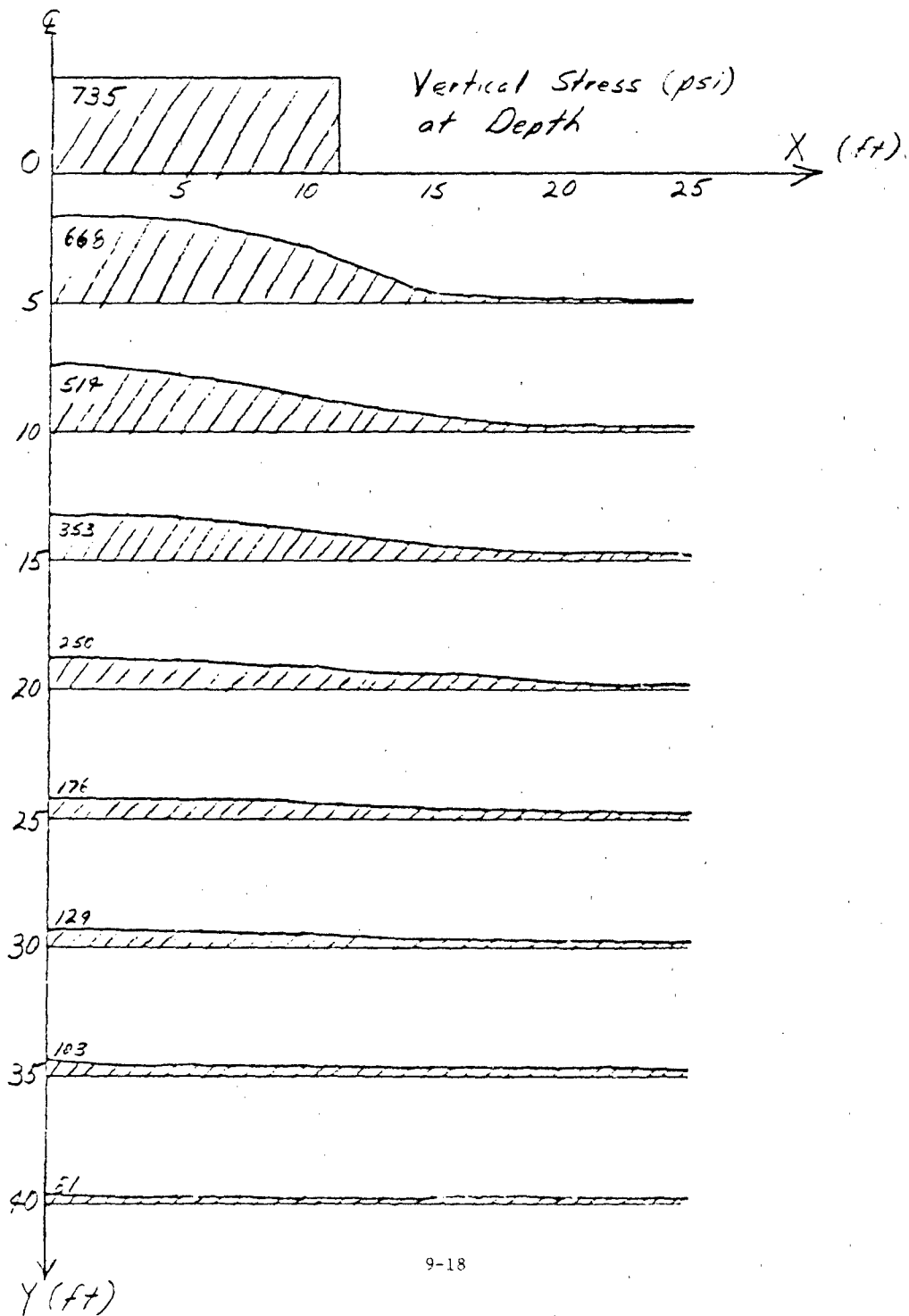
surface of an elastic half-space.¹ These stresses must be added to the initial geostatic stresses. Figure 8.4 gives

¹ In general, the stresses computed from the theory of elasticity are functions of Poisson's ratio μ . This quantity will be defined in Chapter 12. However, vertical stresses resulting from normal stresses applied to the surface are always independent of μ , and stresses caused by a strip load are also independent of μ . Thus of the charts presented in this chapter only those in Fig. 8.5 depend upon μ , and are for $\mu = 0.45$.

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PROJECT NO.: 02-7092 SPONSOR: CESL
SUBJECT: Soil Pressure Attenuation
BY: Wally T. Hawk DATE: 9 Aug 1982 CHECKED BY: _____ DATE CHECKED: _____ 19____

	9-16					
5	668	632	404	73	29	22
10	514	456	294	125	44	29
15	353	323	228	125	66	33
20	250	235	183	118	66	37
25	176	169	132	95	66	44
30	129	125	103	81	59	44
35	103	95	88	66	59	44
40	81	81	66	59	51	37
45	59	59	56	51	44	33



GENERAL PURPOSE BOMB

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SHEET NO.
1 OF 7

PROJECT NO.: 02-7892-001 SPONSOR: CERL
SUBJECT: Well Damage - 500 lb Bomb
BY: Phillip T. Smith DATE: 2/29 19 52 CHECKED BY: DATE CHECKED: 19

Worst case damage from a 500 pound bomb can be expected when the weapon strikes in contact with the wall. Breaching radius for a contact explosion can be determined from the formula:

$$R = \left[\frac{W}{K C} \right]^{\frac{1}{3}}$$

where: R - breaching radius in feet

W - weight of TNT in pounds (≈ 200)

C - tamping factor (2.3 - air, 1.25 soil)

K - material factor (1.5 for approximately 5' thickness reinforced concrete)

$$R \text{ for untamped explosion in air} = \left[\frac{200}{(1.5)(2.3)} \right]^{\frac{1}{3}} = 5.6'$$

$$R \text{ for explosion in earth} = \left[\frac{200}{(1.5)(1.25)} \right]^{\frac{1}{3}} = 6.8'$$

A summary of concrete thicknesses required for typical placements and explosive weights is given in Figure 1.

Pl. + E. Gumpf et al.

"Protection from Nonnuclear Weapons," AFWL-TR-70-127, February 1971,

Third Printing, May 1981.



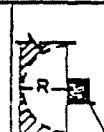



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SHEET NO.
2 OF 7

PROJECT NO.: 02-7092-001 SPONSOR: CERL
SUBJECT: Wall Damage 500 lb Bomb
BY: [Signature] DATE: 6/25/82 CHECKED BY: _____ DATE CHECKED: 19

BREACHING CHARGE WEIGHT REQUIRED TO PERFORATE REINFORCED CON-
CRETE SLABS

Figure 1.

THICK- NESS OF CONCRETE IN FEET	METHODS OF PLACEMENT						
	TNT (lb)						
							
2	16	28	15	8	8	16	1
2 1/2	31	55	28	16	16	31	2
3	41	67	38	21	21	41	4
3 1/2	59	107	55	33	33	59	6
4	88	159	81	49	49	88	8
4 1/2	126	226	116	63	63	126	11
5	157	282	144	79	79	157	16
5 1/2	208	375	192	104	104	208	20
6	270	486	249	135	135	270	21
6 1/2	344	618	316	172	172	344	26
7	369	664	340	185	185	369	33
7 1/2	454	817	418	227	227	454	40
8	551	991	507	276	276	551	49

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SHEET NO.
3 OF 7

PROJECT NO.: 02-7092-001 SPONSOR: CERL
SUBJECT: Wall Damage - 500 lb Bomb
BY: Philip T. Galt DATE: 6/22 19 82 CHECKED BY: DATE CHECKED: 19

Damage producing efficiency drops off drastically as distance between the wall and weapon increases. Data collected during and following World War II for damage to reinforced concrete wall panels was summarized and can be used to determine wall thickness required to resist breaching.² Damage relationships developed for detonations in air and earth are given in Figures 2 and 3 respectively. Using these relationships, thickness required to resist damage from the 500 pound bomb at various standoffs were determined and are shown in Figure 4.

"Effects of Impact and Explosion," Summary Technical Report of Division 2, NDPC, Volume I, 1946. 9-22

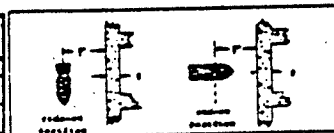
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COMPUTATION SHEET

SHEET NO.
4 OF 7

PROJECT NO.: 02-7092 SPONSOR: CORL
SUBJECT: Wall Damage - 500 lb Bomb
BY: Philip Reed DATE: 6/29/82 CHECKED BY: DATE CHECKED: 19

DAMAGE CRITERIA ADAPTED:

DESCRIPTION OF DAMAGE	TYPE OF DAMAGE	Approx. Spall Depth (in.)
SLIGHT	Slight Cracking and Spalling	0.1
Moderate	Light Puncturing and Cracking with Possible Spall Sealing	0.5
HEAVY	Heavy Puncturing, Shattering, or Possible Perforation	1.5
BLEEDING	Perforation with Extensive Scaling. Spall May Be Sealed or Replaced.	—



For contact and very near contact shots, maximum position of a bomb will cause less damage than side-on. Damage given corresponds to side-on position, the bomb detonating 10 ft from a point opposite center of wall. Position approximately offside usually results in localized damage.

The table gives max. distance at which an air-blasted reinforced concrete wall or wall panel that will experience various degrees of damage due to explosion of TNT-filled bombs in air nearby or in side-on contact with the wall.

These data derive from tests, both model and full scale, on rectangular panels with face dimensions from 3 to 15 times the thickness. Charges used in tests ranged from about 10 to 1700 lb. Test panels were supported along all four edges, and results showed no appreciable difference between freely supported and fixed edges.

Tests involved various degrees of reinforcing and different explosives, but all data were reduced to a basis of 25 wt % of volume and bomb-filled TNT. Degree of damage and R_{0.1} of central deflection to each correlate fairly well for slight, moderate and heavy damage.

Tabulated data refer to about 25 reinforcing steel by volume. For 15 steel, initially thickness values by 0.9 for bleeding and by 0.7 for other degrees of damage.

Projections from nearby bomb detonations usually cause surface scars and a few perforations; this type of damage is not included in the present analysis.

The graph gives damage curves in terms of scale variation r/r_0 and r/r_0 , where r is wall thickness (ft), r_0 is distance from wall to bomb (ft) and r_0 is weight of charge (lb).

Fig. 2. Explosions in Air

DAMAGE CRITERIA ADAPTED:

DESCRIPTION OF DAMAGE	TYPE OF DAMAGE	Approx. Spall Depth (in.)	R _{0.1} (ft)
SLIGHT	Light Cracks	Light Cracks	0.02
Moderate	Medium Cracks	Light Spalling	0.1
HEAVY	Heavy Cracks and Spalling	Severe Spalling	0.2
BLEEDING	Cracked	Cracked	0.5

The table gives values of the maximum distance at which various degrees of damage will result from detonation of bombs in earth. The figures given in the table apply to lightly reinforced rectangular concrete wall panels supported along either two opposite edges or along all four edges. The bomb is assumed to be approximately opposite the center of the face of the wall; any other location from this position will generally result in less damage. Under the same conditions the freely and heavily supported walls will be damaged differently as described by the Damage Criteria above, but the ratio of central deflection to span is a good measure of damage for both types.

This information is based on tests of walls whose face dimensions are in the ratio of about 2:1, and whose span to thickness ratio is between 10:1 and 15:1. The test walls were reinforced with steel bars, about one percent by volume. Damage and central deflection were measured for bombs detonated at various distances. Charges weights used in these tests ranged from 1/8 to 1000 lb.

The graph gives the ratio of central deflection to span for various wall thicknesses and charge distances in terms of the scale variation r/r_0 and r/r_0 .

Tabulated values based on only heavier data are given in tables. Similarly, the corresponding parts of the curves on the graph are dotted.

SOURCE:

Tests by Ministry of Home Security and Sand Research Laboratory (British) and by Committee on Fortification Design (U.S.A.).

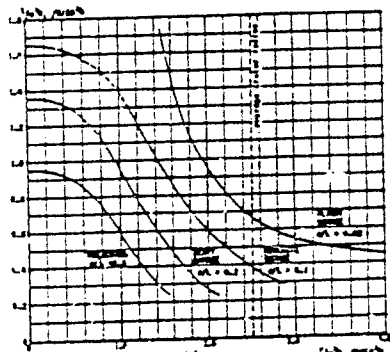
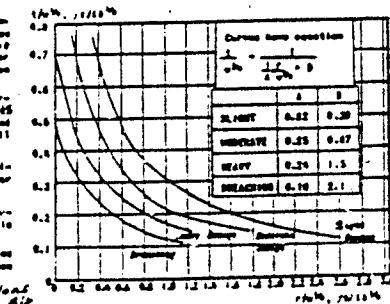


Fig. 3. Explosions in Earth

Calculations of breaching radius for underground explosions
using W.E.S. TR-5L-80-1

5/7

$$\frac{L}{W^2} = \frac{20'}{(6'')^2} = 3.42$$

$\frac{L}{W^2}$	$\frac{L}{W^2}$	$R(f)$	$t(f)$	Free Curve
.8	.168	.98	6.6	
.6	.28	1.63	3.5	
.4	.56	3.22	2.34	
.3	.66	5.6	1.75	
.2	1.92	11.22	1.12	

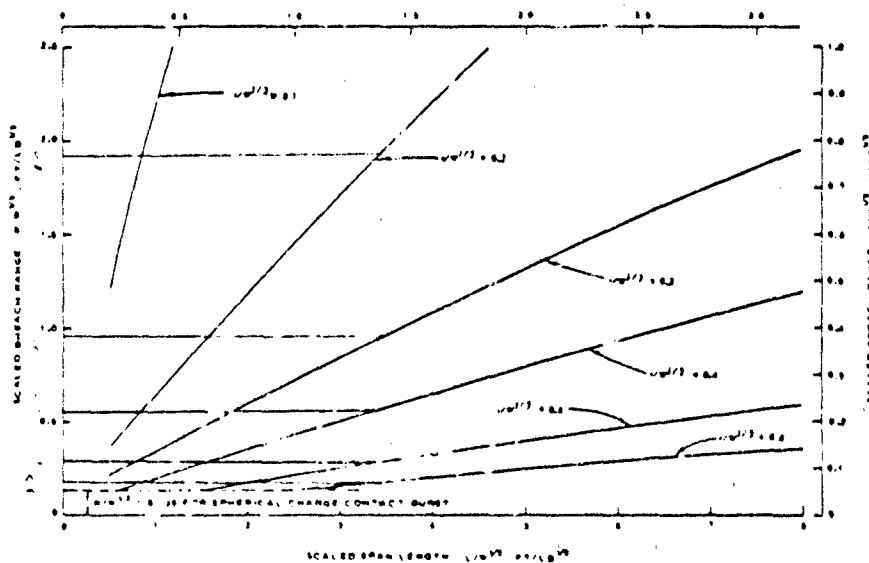


Figure 4.14 Scaled breach range-scaled span length for various scaled wall thicknesses, prototype box structures.

Ref.

Kerr, SA & G.E. Alberts, 'Response of Solid Hardened Core Structures to the Effects of Localized Explosions', TR-5L-80-1, I.S. Army Engineer Waterways Experiment Station, Vicksburg, Miss. 1980.

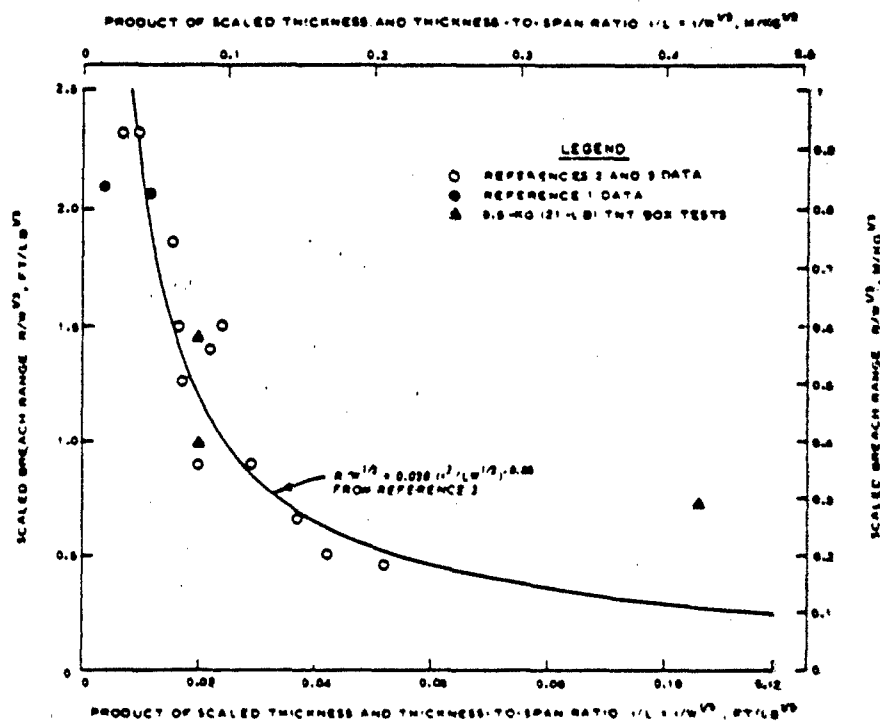


Figure 4.13 Scaled breach range-product of scaled thickness and thickness-to-span ratio for box structures.

$$\frac{L}{W^2} = 0.038 \left(\frac{t}{LW^2} \right)^{-0.18}$$

$$t = 1.6573 \left(\frac{L}{W^2} \right)^{-0.5682}$$

Ref N.F.S. TP SL 20-1

Ref 3 Anderson, H. P., and J. W. Kiers, "Report of Guided Missile Tests
77
in Underground Explosions," ~~DAFATL~~ TP-77-115, Sept 1977.

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SHEET NO.
2 OF 2

PROJECT NO. 62-7092-001 SPONSOR: CEPL
SUBJECT: Walt Damage - see 16 BOMB
BY: Phil T. Shaw DATE: 6/28/82 CHECKED BY: _____ DATE CHECKED: _____ 19__

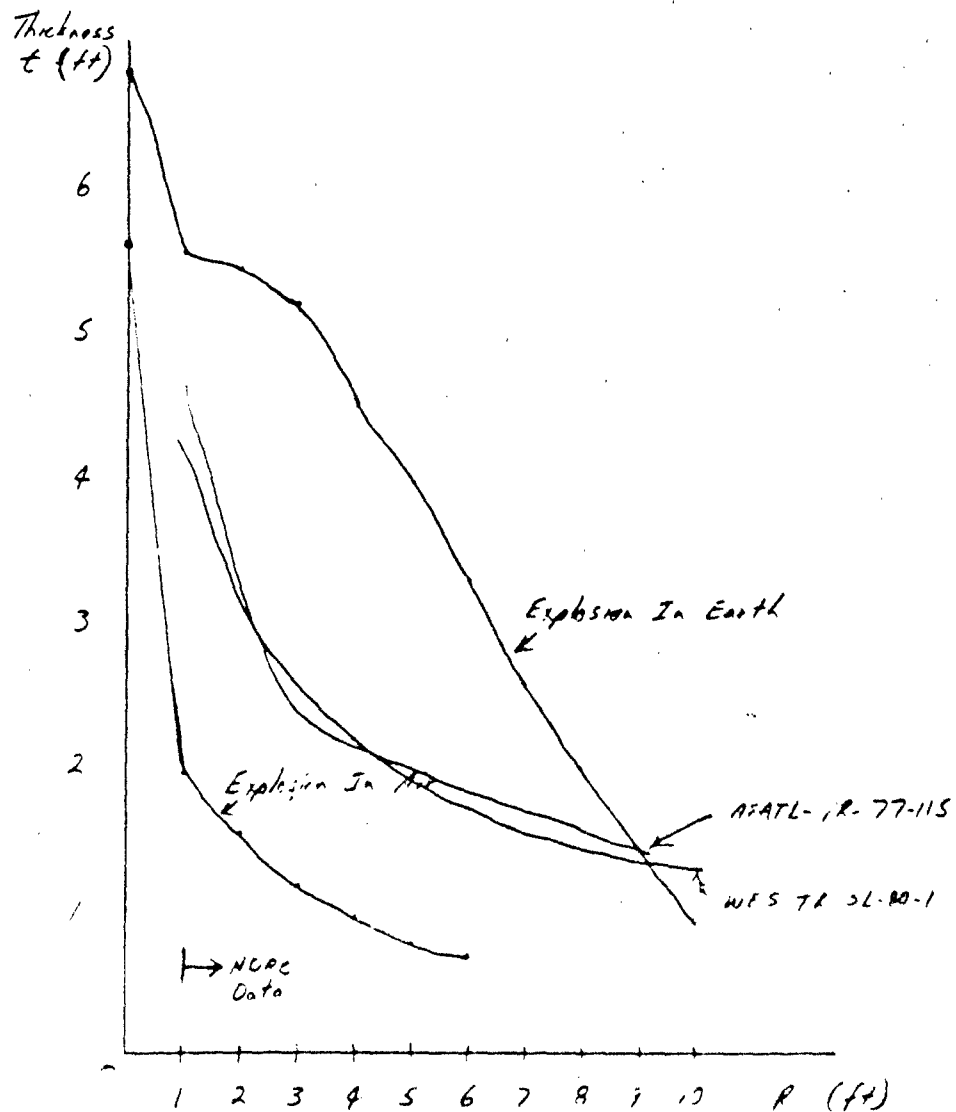


Figure 4. Thickness versus Standoff

9-26

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SHEET NO.

1 OF 6

PROJECT NO.: 02-7092

SPONSOR: CGRI

SUBJECT: PENETRATION BY 500 lb Bomb (Concrete & Soil)

BY: Philip T. Clark DATE: 6/29/82

CHECKED BY:

DATE CHECKED: 19

NDRC Formula ^①

$$X = K N W \left(\frac{V}{1000d} \right)^{1.8} + d \quad \text{for } \frac{X}{d} > 2.0$$

W - missile weight = 500 pounds

d - missile diameter = 14 in.

V - impact velocity = 900 feet per second

f_c' - compressive strength = 5000 psi

N - shape factor = 1.00 except bullet nose

K = $\frac{180}{f_c'}$ - penetrability factor = $\frac{180}{5000} = 2.546$

$$X = 2.546 (1.00) (500) \left(\frac{900}{1000(14)} \right)^{1.8} + 14 = 23 \text{ in.} \approx 2'$$

Weapon Data for penetration into reinforced concrete and soil are given in Figures 1 and 2 respectively. ^②

Concrete thicknesses greater than approximately 29" are expected to resist penetration by the 500 pound bomb. Soil thickness of 15 feet can also resist penetration. Concrete thicknesses can be decreased when combined with soil overburden, or additional layers of concrete protection (barrier slabs).

^① Baker, D.C., "Comment of Empirical Concrete Impact Formula," ASCE Journal of the Structural Division, May 1980.
^② Table of Impact and Explosion, Training Technical Report of Division 2, NSWC, Volume I, 1986.

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SHEET NO.
2 OF 6

PROJECT NO.: 02-7042-001

SPONSOR: CEIL

SUBJECT: Concrete Perforation (500 lb Bomb)

BY: Philip T. Reid

DATE: 21 April 1982

CHECKED BY:

DATE CHECKED: 19

Perforation

$$NDRC - \frac{x}{d} = \frac{23}{14} = 1.64$$

$$\frac{t_p}{d} = 1.32 + 1.24 \frac{x}{d}$$

$$t_p = 14 \{ 1.32 + 1.24 (1.64) \} = 46.8"$$

~~Perforation~~

CFA - EDF

$$\begin{aligned} t_p &= .765 (f'_c)^{-\frac{3}{8}} \left(\frac{W}{d} \right)^{\frac{1}{2}} V^{\frac{3}{2}} \\ &= .765 (5000)^{-\frac{3}{8}} \left(\frac{500}{14} \right)^{\frac{1}{2}} (900)^{\frac{3}{2}} \\ &= .765 (.041) (5.976) (164) = 30.8" \end{aligned}$$

The above formulas were used to determine the thickness of concrete required to resist perforation by a 500 lb bomb. The formulas apply to air backed concrete, soil backed concrete will require less thickness.

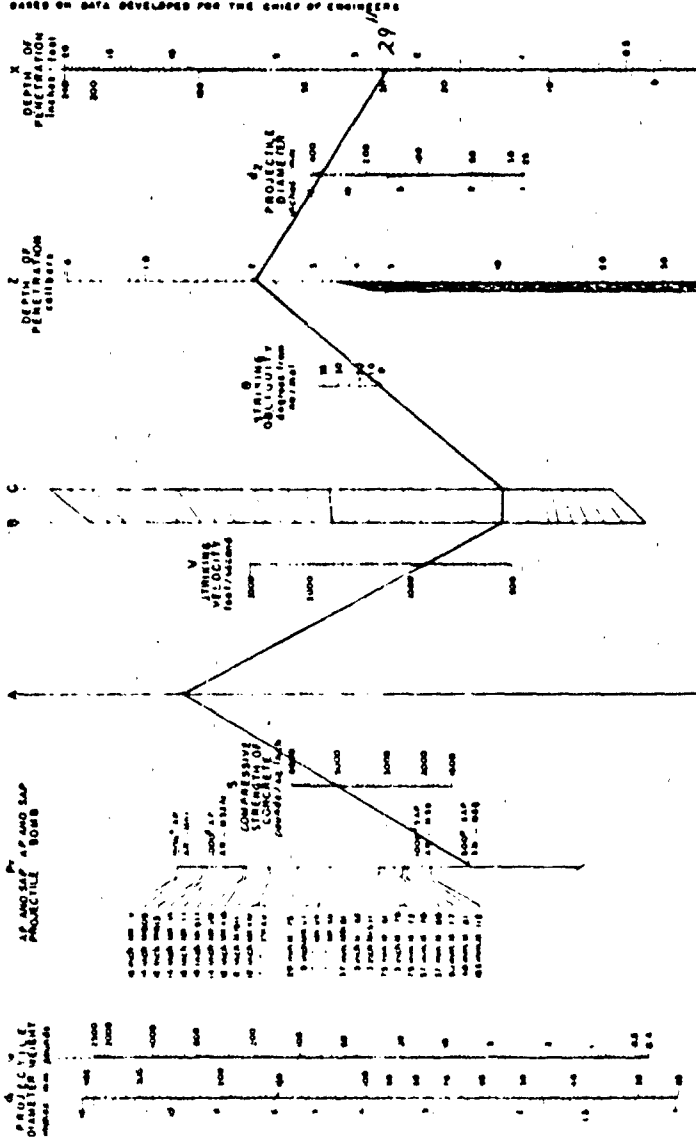
Reference
Sliter, G. E., "Assessment of Empirical Concrete Impact Formulas," ASCE Journal of the Structural Division, May 1980.

Figure 1.

WEAPON DATA

PENETRATION OF REINFORCED CONCRETE
BY AP PROJECTILES AND AP AND SAP BOMBS

BASED ON DATA DEVELOPED FOR THE CHIEF OF ENGINEERS



DIRECTIONS: Determine the diameter and weight of that part of the projectile which penetrates the slab. (The total weight minus the weight of the unsplashed part will usually be correct.) Draw a line through the proper points on the d , w , and v scales at the left of the chart to the P scale. (If the projectile is a type listed on this scale, the above may be omitted.) Continue from the point through the S scale to the A scale. From the A scale draw a line through the V scale to the B scale and follow the guide lines to the C scale. From the C scale draw a line through the D scale to the E scale. (If the point falls within the shaded area the projectile will probably stick in the target.) From the E scale draw a line through the F scale and read the depth of penetration from the X scale. For oblique fire this depth is the maximum penetration normal to the target face. If the projectile or bomb causes scaling (see 2 B-1 sheet) the penetration will be somewhat more than shown here. The penetration for oblique fire at velocities above 2000 ft/sec may be somewhat more than shown here.

CAUTION: For use with 100 pound projectiles, 1000 ft/sec, consult with velocity of 1000 ft/sec at an obliquity of 20°. The projectile will penetrate to a depth of 15 inches at 1000 ft/sec. The foregoing are probably not valid for targets.

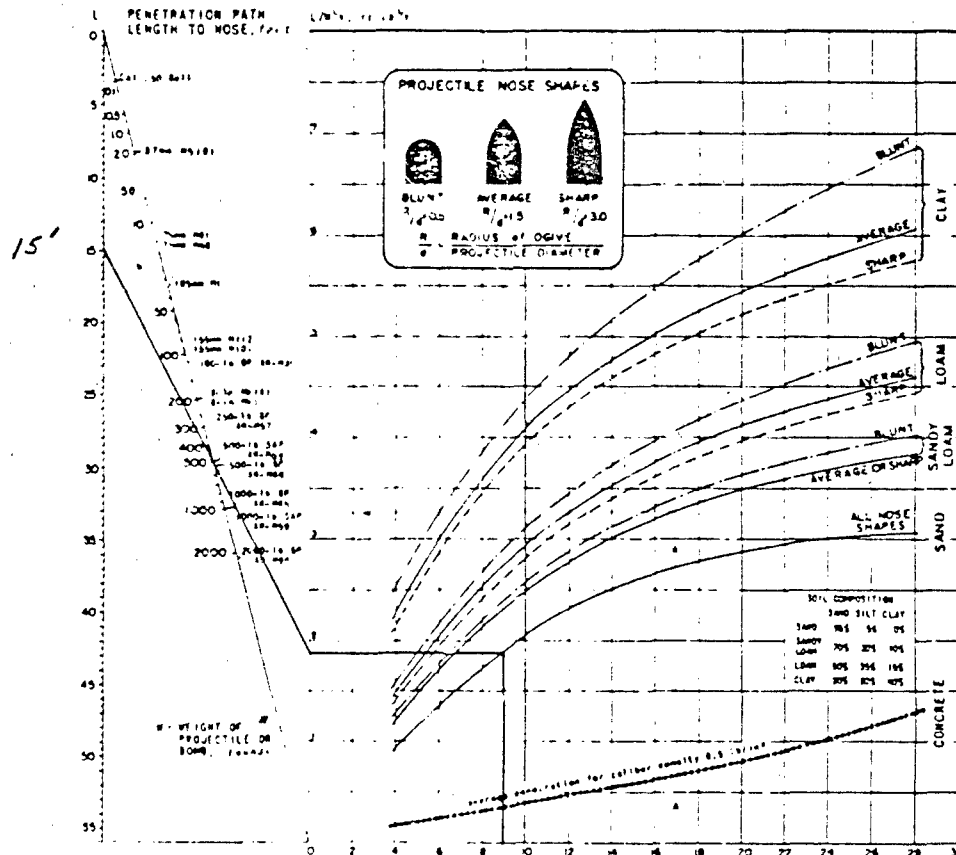
June 1943

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WEAPON DATA

Figure 2.

PENETRATION OF BOMBS AND PROJECTILES INTO SOIL



The graph and nomogram give the relation between striking velocity and penetration path length measured to the nose for projectiles or bombs of various weights penetrating into several soils. Curves marked "average" and "sharp" are for projectiles of different nose shapes as sketched. Where no appreciable effect of nose shape on penetration has been observed only a single curve is drawn. The dependence of penetration path length on projectile weight, as given by the nomogram, agrees with observations for projectiles or bombs having caliber densities from 0.15 to 0.65 lb/in.³. Most bombs and antitank projectiles have caliber density values (weight of projectile in pounds divided by the cube of the diameter in inches) within the above range.

Trajectories in soils are usually straight for two-thirds or more of the path length, but curve near the end of the path (see center). For this reason range distance from the surface is usually 10% to 20% less than the penetration path given here.

Curves given are for average soil types. Penetrations into rich elastic clay are approximately 20% greater than those observed in clay. The dotted curve at the bottom of the graph gives average penetration into good quality reinforced concrete, and is added here for rough comparison.

[EXAMPLE] The dotted line shows that a projectile of average nose shape and weight of 50 lb striking sandy loam soil with a velocity of 1700 ft/sec will have a path length of approximately 12.5 ft, measured to the nose. Because of the curvature of the underground trajectory, the actual penetration from the surface will be somewhat less.

SOURCE: British and American tests with bombs and large caliber projectiles at velocities below 1700 ft/sec. Small caliber tests for the Corps of Engineers, U.S.A. extending over entire velocity range. The curves agree with measurements to ±20%.

* Revision of 2A2 dated September 1945

PTM No. 173 February 1945

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SHEET NO.
5 OF 6

PROJECT NO.: 02-7092-001 SPONSOR: CERL
SUBJECT: Penetration of Soil by 500 lb Bomb
BY: Philip T. Shaw DATE: 21 Feb 1962 CHECKED BY: DATE CHECKED: 19

Ref. Young, C. Wayne, "Depth Prediction For Earth-Penetrating Projectiles," Journal of the Soil Mechanics and Foundations Division, ASCE, May 1969.

Young's formula was used as another method to determine penetration of the 500 lb bomb into soil.

$$Z = 0.0031 S N \left(\frac{W}{A} \right)^{\frac{1}{2}} (V-100)$$

Z - penetration depth (ft)

S - Soil Constant = 7

N - Nax Constant = 1.22

W - Weight = 500 lb

A - area = 153.9 in²

V - impact velocity = 900 ft/sec

$$Z = 38 \text{ ft.}$$

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COMPUTATION SHEET

SHEET NO.
62 OF 6

PROJECT NO.: 02-7092-001 SPONSOR: CEPL
SUBJECT: Penetration of Rock Bldg Overlay
BY: Philip T. Rock DATE: 4/29/80 CHECKED BY: DATE CHECKED: 19

Liverington Initiation Formula ^① (Rock)

$$KE = \frac{1}{2} W V^2$$

$$\begin{aligned} W &= 500 \text{ lb} \\ V &= 900 \text{ fps} \\ G &= 32.2 \end{aligned}$$

$$= \frac{1}{2(32.2)} 500 (900)^2 = 6.2885 \times 10^6 \text{ ft-lbs}$$

$$N_p = \frac{1 - m_1}{2.454 m_1} \cdot \frac{N}{d} \text{ GNP}$$

$$\text{GNP (from design chart)} = 1.6$$

Fig 203

$$m_1 = .25$$

$$I = 1.2 \text{ feet}$$

$$N = (\text{from Fig 199}) = 1.0 ?$$

$$NR = \frac{1 - .25}{2.484(.25)} \cdot \frac{1.0}{1.2} \cdot 1.6 = \underline{\underline{1.61'}}$$

Research has shown that projectiles encountering at least two layers of rock rubble at least the ^②caliber of the projectile have a high probability of being deflected. The Liverington formula showed 1.61 feet of granite can stop the threatening 500 pound bomb. It is reasonable that two layers of 14 inch diameter granite boulders can accomplish the same purpose.

^① Liverington, C.W. and Smith, E.L. "Bomb Penetration Project," Technical Report of Minnie Research Foundation Inc., Golden Colo. 15 June 1951.
^② David G. Rothberg for "Inertial Weapons," USWES, Jan 1980.

LARGE HIGH EXPLOSIVE CHARGE

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COMPUTATION SHEET

SHEET NO.
1 OF 1

PROJECT NO.: 02-7092 SPONSOR: CERL
SUBJECT: 300,000 LB HE @ 100 M
BY: E MORRIS DATE: 19 CHECKED BY: DATE CHECKED: 19

ASSUME THAT THE 300,000 * CHARGE IS ON THE
GROUND SO THE EFFECTIVE CHARGE WEIGHT
IS $2 \times 300,000 = 600,000$ LB.

DISTANCE = 100 M = 328 FEET

SCALED DISTANCE = $\frac{R}{W_e^{1/3}} = \frac{328}{(600,000^{1/3})} = 3.9 \text{ FT/LB}^{1/3}$

PER DOE/TIC - 11268,

$$P_{\text{SIDE-ON}} = 42 \text{ PSI}$$

$$\lambda_{\text{SIDE-ON}} = .01250 W_e^{1/3} = .0125 (94.3) = 1.05 \text{ PSI-SEC}$$

$$T_{S-O} = \frac{2\lambda}{P} = \frac{2(1.05)}{42} = .05 \text{ SEC}$$

$$P_{\text{REFL}} = 190 \text{ PSI}$$

$$\lambda_{\text{REFL}} = .039 W_e^{1/3} = 3.3 \text{ PSI-SEC}$$

$$T_{\text{REFL}} = \frac{2\lambda}{P} = \frac{2(3.3)}{190} = .035 \text{ SEC}$$

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Ammunition storage concepts for use in flat terrain / by M. G. Whitney... (et al.)
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1. Ammunition storage. I. Southwest Research Institute. II. Whitney,
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